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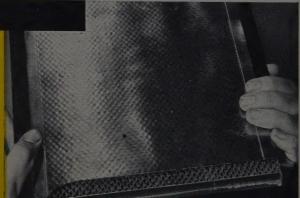
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- Production and manufacturing for missile and aircraft systems.

For further information, write to: Vice President, Defense Products Marketing, Crosley Division, Avco Manufacturing Corporation, Cincinnati 25, Ohio.





Circle No. 1 Reader Service In Product Re-Section.

Coming Next Month... A NEW Format

The February issue of SPACE/AERONAUTICS will come to you in a new format. The outside dimensions will be cut to 8\%x11\% inches—commonly referred to as "standard size"—and on the inside you'll find a more efficient package for the busy Technical Management reader.

Like many others that you've probably noticed during the past 18 months, these improvements are part of a <u>continuing</u> effort to provide you with the most <u>useful</u> technical information service possible—packaged for easier reading.

New typography, layout and styling were suggested by a professional magazine designer—based on the latest publishing techniques. But even more important, many of the improvements are the result of comments and suggestions by you, our readers.

For instance, over 2,500 of you recently took the time to fill out a 7-page questionnaire about your reading habits and use of SPACE/AERONAUTICS. And almost every one who did so also took the trouble to tell us how we could make the publication more useful and easier to read.

For example, we learned of the trend toward increased filing of technical articles. Better than 8 out of 10 of our readers are making use of SPACE/AERONAUTICS material for this purpose. Many of you commented on the difficulty of filing oversize pages or clipping articles that ran on in single columns for many pages. Many of you asked for some mechanical aid to help file articles for future reference.

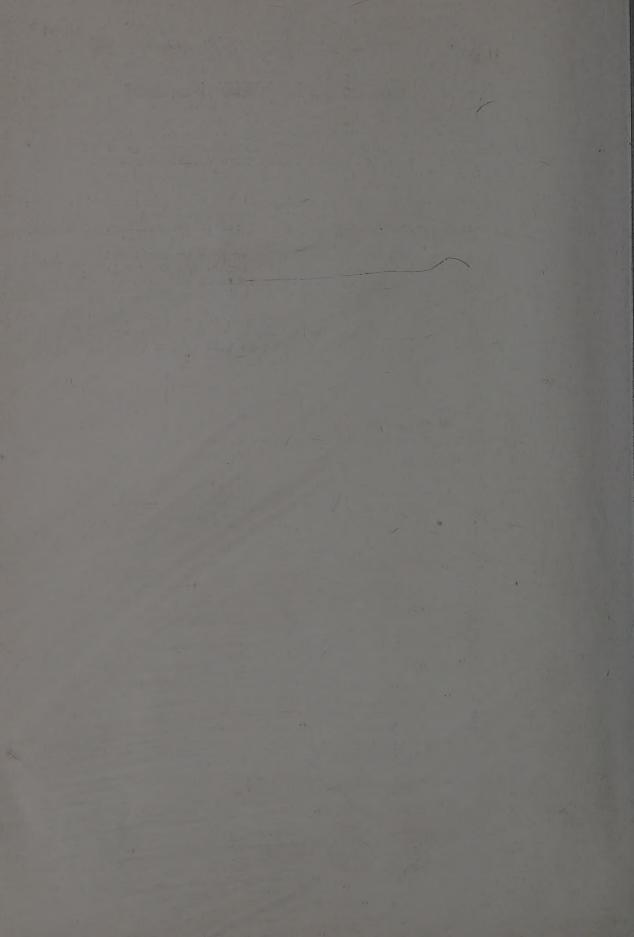
Starting in February there will be two major sections of solid editorial—each section will be uninterrupted by advertising. In addition, there'll be a new feature, "Technical Article Abstracts," covering all articles in the issue. All you have to do is cut them out, mount on 3x5 cards and you'll have a permanent reference to all issues of SPACE/AERONAUTICS. To aid you further in your need to "clip and file," some features like "Design Progress" will be printed on perforated pages—making them easy to remove and keep in a design notebook.

Most of you (about 9 out of 10) told us advertising was of definite interest. The majority use the ads to get answers to technical problems. So, even with solid editorial sections, we are making sure that advertising will not be bunched up in other parts of the magazine. You will also find a special reference in every issue which classifies all advertisements by product or service. This will appear immediately before the regular index to advertisers and the "Reader Service Card"—giving you a complete reference to all advertisements in any issue, as well as a direct channel to the advertiser.

SPACE/AERONAUTICS offers you the industry's most complete and authoritative technical information service—in a greatly improved package to make it more useful.

Bill Maass

Puhlisher



Our new format

The next issue of SPACE/-AERONAUTICS will be welcomed by those who clip and file technical articles. Their number, apparently, is legion.

One of the most insistent requests has been for some way to make it easy for you to file material in the magazine. We believe the new format and size will meet some of your filing requirements.

There will be two major sections of solid editorial "uncluttered" by ads. We have also added a new filing aid—"Technical Article Abstracts." These will cover all articles in each issue. You will be able to mount each abstract on a 3x5 card for a permanent reference to all issues of SPACE/AERO-NAUTICS.

Moreover, some special features—such as "Design Progress"—will be printed on perforated pages, making it easier to remove them for filing in a design notebook.

Since advertising is of particular interest to those wanting specific technical information, there will be a special reference section in every issue which classifies all ads by product or service.

The two major sections, Space/Aero Engineering and Space/Aero Electronics, have been designed to give you the best possible way of getting the information you want. It further streamlines our "systems and components" arrangement of editorial subjects.

Electronics, on the other hand, because it continues to be of such heavy, specialized interest, is expanded to cover even more material than before

Randolph Hawthome

Does your weapon system need fuel "on the double"?



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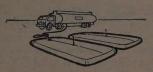
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lanuary 1959

REFEARCH DESIGN DEVELOPMENT

JANUARY 1959

VOLUME 31 NUMBER 1

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The editorial content of Space/Aeronautics is regularly examined for readability by Robert Gunning Assoc., counselors in clear writing. These consultants meet periodically with the editors and discuss comparative readability ratings.



Cover Story

This month's cover features the first published photo of the three-gyro stable platform that will provide flight data for the North American X-15. The platform, shown within its roll and pitch gimbals, is the heart of an extremely precise inertial system, specially developed for the first U.S. manned spacecraft by Sperry Gyroscope Co., Great Neck. N.Y.

The entire system consists of the stabilized platform, a computer, and a control unit. The first model was delivered to the Air Force last month.

The platform has a volume of roughly 0.446 cu ft and weighs about 27 lb. It contains all its own power supplies, amplifiers, and even a heat exchanger. It is the X-15's basic sensor for critical attitude, speed, distance, and altitude data.

The entire system is designed to direct the pilot from the moment of launch throughout the hypersonic acceleration phase and to provide re-entry control data.

See Astronautics Intelligence (p. 175) for details on Atlas orbit shot.

Randolph Hawthorne ...

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B. F. Goodrich Fabric Tread Dimple Tire proved "far superior" in F-106 tests

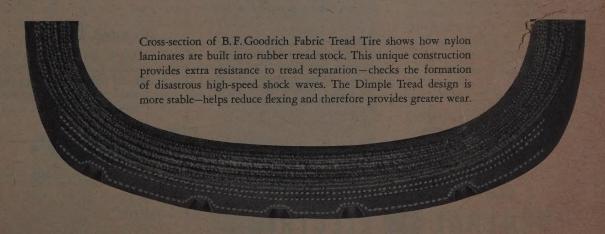
Ordinary high-performance tires used on the F-106 built by Convair, a division of General Dynamics Corporation, just couldn't take the punishment. The stress of highspeed takeoffs and landings literally tore them to pieces.

Then B.F.Goodrich submitted its revolutionary new Fabric Tread Dimple Tire for testing. Even under severe operating conditions, including repeated RTO situations, the tire proved that it could out-perform and out-wear any jet tire known today. As a result, the B.F. Goodrich Fabric Tread Dimple Tire is standard equipment on the F-106 Delta Dart, world's fastest and highest flying all-weather interceptor.

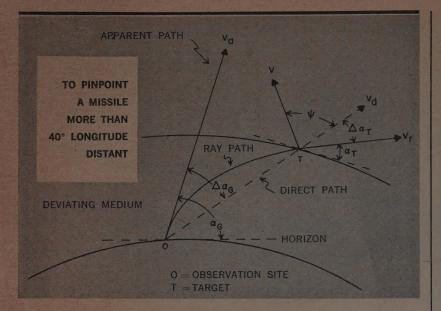
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Missile Detection Systems Section
HEAVY MILITARY ELECTRONICS DEPT.



Court Street, Syracuse, New York

In this issue

look for . . .

Nuclear flight in Russia

There can be little doubt the Reds are ahead of us in developing an airborne atomic reactor. How did they do it? A partial answer is given in the article on page 132, the first in a series of detailed engineering reports on Soviet aircraft reactor progress.

Electronics Outlook

As usual, this year's special report on space/aero electronics is introduced by a detailed analysis of electronics' share of the space/aero industry. Turn to this informative article on page 22 for a review of the trends and trouble spots of the electronics market.

State of the electronic art

Three of the most important areas of space/aero electronics—inertial guidance, radar, and infrared—are covered in detail in special "state of the art" reports (beginning on page 24). Here you'll find expert assessments of the latest development and research trends.

Magnetohydrodynamics

As we learn to combine the sciences of electrodynamics and hydrodynamics in magnetohydrodynamics, the article on page 28 points out, vast new possibilities are opening up in fields as diverse as microwave amplification and space propulsion.

Reader Service

Want to get more information on products or services advertised or mentioned editorially in this issue? Then use our handy Reader-Service Card opposite page 167. It's simple—just circle a number, drop the card into a mailbox, and we'll forward your inquiry.

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calendar

January 12-14 — Fifth National Symposium on Reliability and Quality Control in Electronics, Bellevue-Stratford Hotel, Phila., Pa.

January 13-14—Symposium on Cathode Ray Tube Recording, Systems Development Corp., Engineers Club, Dayton, Ohio.

January 21-23—South West Electronic Exhibit, Arizona State Fairgrounds, Phoenix, Ariz.

January 26-29 — Tenth Annual Plant and Engineering Show, Public Auditorium, Cleveland, Ohio.

January 26-29 — 27th Annual Meeting, Institute of the Aeronautical Sciences, Sheraton-Astor Hotel, New York, N. Y. Honors Night Dinner, Jan. 27.

January 27-29 — Fifth Annual Radar Symposium (classified), Rockham Bldg., Univ. of Michigan, Ann Arbor, Michigan.

January 27-30 — 15th Annual Technical Conference, Society of Plastics Engineers, Hotel Commodore, New York, N. Y.

January 28-29 — Fifth Annual Midwest Welding Conference, Armour Research Foundation, Illinois Institute of Technology, Chicago, III.

February 3-5—14th Annual Technical and Management Conference, Reinforced Plastics Division, Society of the Plastics Industry, Inc., Edgewater Beach Hotel, Chicago, III.

February 12-13 — 1959 Solid State Circuits Conference, Institute of Radio Engineers' Professional Group on Circuit Theory, American Institute of Electrical Engineers' Committee on Electronics and Univ. of Pennsylvania, Philadelphia.

rebruary 26-March 1—1959 Engineering Exposition, Balboa Park, San Diego, Calif.. Address Inquiries to: 422 Land Title Bldg., San Diego 1, Calif.

March 3-5—1958 Western Joint Computer Conference, sponsored by IRE, American Institute of Electrical Engineers and Assn. for Computing Machinery, Fairmont Hotel, San Francisco, Calif.

March 5-6 — Flight Propulsion Meeting (classified), IAS, Hotel Carter, Cleveland, Ohio.



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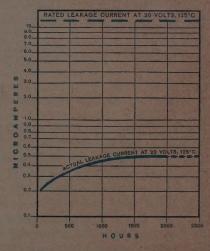
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ANALYSIS

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Circle No. 279 on Reader-Service Card

editorial

Needed: An Overall Program

VALTER LIPPMANN once said that we prefer a disagreeable fact still in the future to a disagreeable remedy in the present. The truth of this threads through the whole fabric of our national affairs.

It shows in the way we face almost any national problem you want to name today. Take just three: Inflation, defense budgets, labor unions.

The danger, the waste, and the evils of each are well known, especially by those in a spot to do something about them. The day is fast coming when either we shall master these problems, or they will master us. But the will to do something about them now is sadly lacking.

Union racketeering, looting of union funds, and lust for power have been spread over the front pages of the press for a year. But Republicans and Democrats stopped even a timid little curb on union activities from being passed. Prospects for a law this year are even less.

THE WASTE in defense spending, in civil government spending programs, and that caused by union monopolies, are major causes of inflation. But who cares enough to do anything about them? Not the Congress. Not the Administration. And, apparently, not the people.

Waste in defense spending stems from lack of a logical, overall strategic plan. If such exists, it does not coordinate a diverse effort to meet a diverse threat. As a result, parts of our efforts negate the effects of other parts, some duplicate others, and some exist merely because they are Congressional "pets." The Soviet threat, however, is a coordinated attack at once military, economic and political.

Despite "reorganization" of the Defense Department, we still have overlapping service roles and missions far beyond what is needed for effective defense. Our weapon system programs are based on these traditional, if modified, roles and missions, instead of on targets and weapons missions.

How else can we justify three parallel IRBM programs-Army, Navy and Air Force? Two parallel ICBM programs, with another coming up? And now, two parallel space flight development and operating agencies, one called "civil" and the other "military"?

NFLATION repeatedly has been called our greatest danger. But efforts to reduce the federal budget have been confined to across-the-board slashing of defense funds. Good programs have been cut or dropped along with the bad. Each service chief is forced to choose which project to drop or cut in favor of one he believes needed for his service role and mission. The "local" service view prevails over national strategic defense requirements. An overall strategic defense plan would select the weapons, according to the enemy situation and capabilities, and the cognizant service.

Much grim talk points up the dangers of an inflation "that will curl your hair." Yet we pay \$500,000 a day to store wheat which cost us \$3 billion in subsidy. We don't want this wheat. It is more than we can use. We can't even give it away.

Why, then, do we throw away billions of dollars? Because Democrats and Republicans want the "farm" vote. The money to buy this vote comes out of all our pockets, raising the cost of living.

"The only freedom man can ever have is the freedom to discipline himself," Bernard M. Baruch points out in a recent book. This is the freedom we must use to prevent the discipline of an implacable enemy being thrust upon us.

technical management intelligence

Upsurge due in missiles and civil planes

SPACE/AERONAUTICS industry's prospects for '59 look like this:

Continuing decline in military aircraft production will be balanced by continuing upsurge in missiles, a sales volume roughly comparable to '58, a steady level of employment (about 760,000), and an upturn in commercial aircraft sales.

Sales went up
in 1958 despite decline in
aircraft production

SALES IN 1958 came to about \$11.8 billion (equal to 1957), even though the number of military aircraft dropped from 5500 to 4000 and the number of commercial transports (almost all piston engine types) fell from 322 to 225. Military missile purchases, which climbed 40 per cent over '57, and inflation made up for the difference.

In '58, total sales of the top 12 airframe manufacturers amounted to \$7 billion, as against \$6.9 billion in '57. Commercial sales over the same period dropped from \$1.6 billion to \$1.5 billion. Earnings for the 12 firms declined from 2.4 to two per cent of sales.

NASA plans for taking over ABMA ruled out by "compromise" agreement on Jet Propulsion Labs NASA lost an important round in the fight to take over the Army Ballistic Missile Agency at Huntsville, Ala. But NASA's campaign is not yet over.

The "compromise" under which NASA got Jet Propulsion Laboratory (2300 people) at Pasadena, Calif., and the Army retained ABMA (3800 personnel), gave NASA the right to use some ABMA manpower and facilities for civil space programs.

DR. GLENNAN, NASA'S ADMINISTRATOR, indicated the "compromise" was "not without some shortcomings." DOD and NASA will make a joint report to President Eisenhower on the results of the "compromise" within the next year. NASA is certain to point out that there is room for improvement—in other words, that NASA still wants ABMA.

NASA has its eye on research facilities of all three services

DESPITE NASA'S PARTIAL LOSS in its battle with the Army, the civil space group is getting ready for more fights with the services. It would like to take over USAF's School of Aviation Medicine, Randolph AFB, Texas; Navy's Naval Ordnance Test Station, China Lake, Calif.; and some of the Army's Signal Corps activities at Fort Monmouth, N. J.

Naturally, none of the services will be particularly enthusiastic about losing scientists and facilities to NASA. But it is almost equally certain that Glennan & Co. will end up with more manpower and facilities than they have now.

Takeover at JPL to come on July 1; Huntsville will assist NASA on some space programs NASA'S AGREEMENTS with the Army on ABMA and JPL provide:

- JPL will continue work for the Army throughout '59 on the Sergeant ballistic missile, special intelligence studies, aerodynamic testing and research, and communications research. NASA will start reorienting JPL's activities on July 1. Army is transferring \$4 million to NASA to cover JPL activities during the January-June period.
- Army Ordnance Missile Command at Huntsville (including ABMA) will "assist in the development of broad requirements and objectives in space programs" and "in the determination of specific projects and specific methods (including hardware development) by which NASA may accomplish its overall objective." NASA will provide the funds, but the Army will be fully responsible for these assignments.

Congress will push for speedup on an airborne atomic reactor

CONGRESSIONAL ARMED SERVICES COMMITTEES will begin pushing for a speedup in the Aircraft Nuclear Propulsion (ANP) program as soon as Defense Secretary McElroy and other DOD witnesses start testifying on the defense budget for fiscal '60.

The ANP program, currently budgeted at \$150 million a year, now involves "parametric studies" of nuclear airframes and extensive work on nuclear power-plants.

In the second second

ANP WORK was cut back last vear (from \$250 million annually) on the advice of Dr. James Killian, the President's chief scientific advisor. Even now, despite Pentagon acknowledgments that the Russians "may have a slight lead over the U.S. in this area," there is no DOD policy to change the ANP program "at this time."

This attitude differs sharply from that of high USAF officials. Maj Gen. D. J. Keirn, for example, believes the best type of airborne alert system is the manned, nuclear-powered aircraft.

Nuclear plane delayed by misconceptions, says Keirn

GENERAL KEIRN, who is AF's Assistant Deputy Chief of Staff for Development in charge of nuclear systems and also chief of AEC's aircraft reactors branch, says the ANP project has been delayed by misconceptions and by the desire to be safe instead of sorry.

One of those misconceptions is that chemically powered planes can fly faster and higher than nuclear-powered turbojets. Says Keirn, that's not true of the nuclear engines using chemically powered afterburners.

The USAF's A-plane boss believes the U.S. is now entering "the experimental flight development phase looking toward prototypes," but indicates that DOD has adopted the "fly before you buy" attitude on the A-plane.

ALL this means is that it will be many years before the U.S. has a nuclear-powered bomber or patrol aircraft in service.

Just how much Congression-

al pressure to speed up the ANP program will accomplish is uncertain. Some lawmakers have been fighting to get the program speeded for several vears—without much success. But Russian progress may force DOD to change its attitude.

ARMY fired the first important salvo in a campaign to meet the threat of limited wars. If General Maxwell D. Taylor's recommendations are even partially accepted by DOD and/or Congress, it will mean additional orders for the space/aeronautics industry.

The Army's Chief of Staff concedes that "we now have a significant capability to cope with limited wars." But "we can and should take steps to improve it." As Taylor sees it, the services should:

- modernize present equipment.
- improve the strategic mobility of limited war forces,
- pre-plan the use of airand sealift.
- · expand joint planning and training.

Army needs: New missiles, light atom weapons, more tactical mobility

TAYLOR cites the need for second generation missiles. light atomic weapons, improved conventional weapons. "and many new types of air and ground vehicles to advance tactical mobility." Also required are improved lightweight signal communications based on new radio equip-

"As the Army lightens its loads," Taylor explains, "we hope that our sister services will modernize the sea- and airlift upon which we depend for our strategic mobility." Taylor's bid for allocation of a specified amount of air and sealift for use in a limited war would almost certainly entail the procurement of additional cargo aircraft by USAF and Navv.

Federal R&D spending reported on steady increase

NATIONAL Science Foundation study shows federal expenditures for R&D rose 13 per cent last year, will climb another nine per cent in fiscal '59. NSF says \$3.9 billion will be spent on R&D in this fiscal year. (Not included in this is \$800/900 million that DOD spends for R&D support out of production funds.) About 90 per cent of this total, NSF estimates, is being used for actual R&D: the remainder represents an increase in facilities.

DEVELOPMENT, which historically accounts for the bulk of federal R&D spending, will take \$2.243 billion this fiscal vear. Basic research will account for \$281 million in 1959.

Largest Government spender is the Defense Department. Its R&D expenditures will rise from \$2.085 billion in '57 to nearly \$2.4 billion in '59.

AIA'S plan to fight for outright repeal of the Renegotiation Act has been toned down. AIA now sees liberalizing amendments as more likely possibilities than complete elimination of the law.

House Ways and Means Committee is expected to hold full hearings on the law, scheduled to die on June 30. But the continuing flow of defense orders makes it politically inexpedient for Congress to let the renegotiation idea die out.

AIA member companies have been assessed up to \$10 million in excess profits by the Renegotiation Board. As a result, the industry would welcome the demise of the whole renegotiation concept. AIA maintains that there is little incentive in incentive-type contracts if another Government agency tries to take away the added earnings after negotiations with the military services have been approved.

AIA's position in future testimony before the Ways and Means Committee is still being firmed up. But the association is certain to seek at least:

- permission to appeal Renegotiation Board decisions to U. S. Tax Court.
- that the Board reveal the considerations it takes into account in making a ruling of excess earnings,
- that the Board consider the earnings of other industries in its decisions.

Privately built plants for production of solids?

EXPANDING use of solid propellant rocket engines will require additional engine plants. DOD is planning to have these financed by private industry where possible.

Dudley C. Sharp, Assistant AF Secretary for Materiel, who also reflects the views of the Army and Navy on this subject, believes the old industrial mobilization concept of standby war plants is not suited to our present idea of a force in

being. DOD's policy is aimed at obtaining "an industrial base which is adjusted to firm weapon program needs, and is mum degree . '. ."

ARPA'S program involves a two-pronged approach to detection, solid propellants, and military space missions. Dr. Herbert F. York, ARPA's chief scientist, describes the projected work this way:

• Ballistic Missile Defense —One approach is based on the present state of the art. Among other items, it covers the Nike-Zeus, its radar, computer, and rocket, and the USAF's Ballistic Missile Early Warning System. Later work involves better defenses against second-generation ICBMs, including advanced radar, power tubes, and reference comput-

• Re - Entry Research -ARPA plans to find out what interactions take place when a ballistic missile nose cone reenters the atmosphere, especially in the way of ionization and infrared and possible op-

• Satellite detection—One important projected study involves the observation of in-

• Solid propellants—ARPA is primarily interested in the fuels themselves. Other work will involve components that can withstand the temperatures resulting from the new

• Space program — Work will involve larger boosters (e.g., the forthcoming 1.5 million pound thrust cluster), high energy liquid propellant stages to be used on top of ICBMs, and space vehicles themselves, especially the prob-

more on next page



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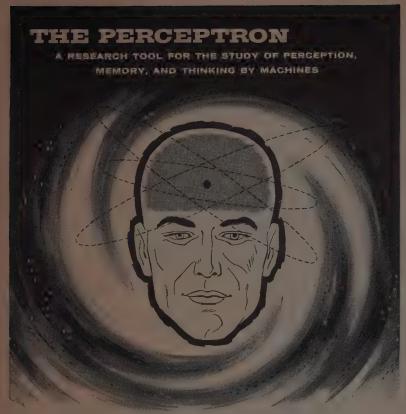
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MANAGEMENT INTELLIGENCE

lems of their orientation, stabilization, and power supplies.

NAVY and Air Force each canceled a major weapon system because the vehicle arrived too late for military requirements.

Martin got the word to stop production of the P6M Sea-Master flying boat after 14 have been built. Six development models are already flying.

BELL AIRCRAFT was told by AF that the GAM-73 Rascal is out. The cancellation does not affect any of the airto-surface missiles already contracted for. But orders for spares and other logistic support equipment are being cut back.

Technical advances, fund shortage got P6M the axe

P6M HAS four Allison J71 turbojets in the development models, will become supersonic when production aircraft are fitted with P&WA J75s. Navy's decision to give up on SeaMaster resulted from money shortages for fiscal '60 as well as technological advances since the plane was first ordered more than five years ago.

RASCAL, which has been fired from B-47s and B-52s, is to be replaced by AA's GAM-77 Hound Dog. To be carried by the B-52Gs now coming off the line at Boeing-Wichita, this missile has superior performance.

AS WE go to press, still another cancellation is announced: Fairchild was told by USAF to drop work on its Goose decoy missile originally slated for SAC. Status of Goose's J83 jet engine, also a Fairchild design, remains undecided.



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Technical Personnel Dept. 15





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SPACE/AERONAUTICS











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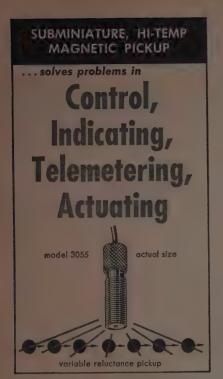
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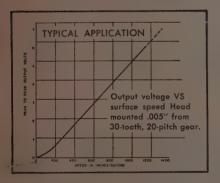
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washington briefing

by A. N. Wecksler. Washington Bureau Chief

NASA contractors face patent trouble

THE TELEVISION OF THE THE TELEVISION AND ASSOCIATED REPORT OF THE PROPERTY OF

NASA (National Aeronautics Space Administration) will follow the same contracting procedures as the armed services, with one exception — patent rights on inventions will become the property of the government.

This practice affects both companies and individual scientists and engineers where the discovery has some significance in space technology. Under Department of Defense procedures, where a contractor invents, he makes no charge to the government for the use of his invention, but does retain ownership in the patent.

Since R&D contracts can be notoriously lean as far as profit is concerned, one of the incentives for the company to engage in R&D is the yield that can possibly be obtained from a commercial use of the invention. In many instances, the company gives the individuals who develop the invention an interest in the patent.

Under the "Space Act" that set up NASA, all patent rights belong to the government. If there is an NASA contract, neither the company nor the inventor has any ownership rights—even if the discovery is made on the inventor's own time, in his own laboratory, and he uses his own materials.

The Act also says any company or individual working under an NASA contract must promptly furnish full technical information on any discovery made in course of the contract.

The Act even goes a step further and says no patent can be issued by the Patent Office for any invention that appears to the Commissioner of Patents to have "significant utility in the conduct of aeronautical and space activities" unless the applicant makes written oath on the full facts on how the invention was developed.

If NASA does not intervene within 30 days, the patent can be issued. But NASA can come back over a period of five years and request title to the patent, if it believes that the facts presented by the inventor were false.

There is a precedent for this kind of regulation. Under the Atomic Energy Act, the government retains patent ownership, and in fact has taken patent title to a vast amount of technical data.

But there are two essential differences between inventions in nuclear fields and in space:

- (1) Nuclear technology was basically a new art—developed at great cost by the government, involving materials and techniques requiring continued government control. In contrast space is an integral part of the 50-year-old technology of flight. This technology was developed under a certain set of rules. The Space Act now switches the rules in the middle of the game.
- (2) Nuclear technology can be segregated. It is separate and different from other technologies. In contrast, invention in space involves the environment, conditions, and hardware that are basic to the art of flying.

It is quite plausible that a company could be developing a unit for a non-astronautic military vehicle and at the same time be working on an adaptation of this unit for an NASA space project.

Under the military contract, the company would get the patent ownership if some commercial adaptation were possible. At the same time, it would be forfeiting these rights under the NASA contract.

There is little need to show that the Space Act provision dealing with inventions is impractical. There are few who will defend it. The American Bar Association, the American Patent Law Association, the Aircraft Industries Association—all join in questioning the wisdom and practicality of the patent ban.

NASA spokesmen show no enthusiasm for the provision. In fact, they point out that they did not advocate such a regulation.

Yet these patent provisions are now law. An effort will be made in the new Congress to get the law amended.

In the meantime, NASA is proceeding to establish criteria under which the government might forego ownership of space patents. NASA must show it is serving the "interests of the United States" before it can waive government ownership.

NASA's patent criteria will be published in the Federal Register and circulated among the industry. After a while, they will be adopted as standard operating practice—unless, of course, Congress moves in and changes the law, making everybody a bit happier.

industry viewpoint

by Robert M. Loebelson, Industry Editor

Military and industry must cooperate on reliability

A LL over the space/aeronautics industry, voices are being raised that call for more reliability in today's and tomorrow's military equipment. One of these voices belongs to Maj. Gen. Albert Boyd, now retired and vice president in charge of Westinghouse Electric's Defense Products Division, and once one of USAF's top test pilots.

Says Boyd: "We're moving into an age in which reliability and quality control are far more important than they were in the past. To meet the threat, our equipment must function properly. The need for reliability is just as evident in Mach 2 planes as it is in missiles."

Boyd, who spent a large part of his military career in R&D believes there are several ways of approaching the reliability problem.

"In the first place, industry's quality control people, who now report to the production or engineering manager, should be placed on a higher level and report directly to the top man in their echelon," Boyd explains.

'These quality control specialists should participate in design planning and the design approach, from the start of the project to the time

when the system is in service.

"Another way to insure reliability is to rotate the quality control people with others from engineering, production, and maintenance. In that way, each group gets to know the others' prob-

The Westinghouse v-p believes that environmental testing is as important as quality control.

"Whenever a program is compressed, testing suffers most," Boyd declares. "This always costs time and money. In the long run, it would be far less costly to do the job properly in the first place. What we really want to accomplish is to put the 'fix' in the weapon system before it is delivered and save millions by eliminating modi-

Boyd thinks both the services and industry can



MAJ. GEN. ALBERT BOYD (USAF, ret.), vice president, Defense Products Div., Westinghouse Electric Corp.

contribute to greater reliability. "The military can and should provide greater financial incentives for meeting quality specifications," he says. "And there should be corresponding penalties for not meeting those specifications.

'Military programers should also lay down requirements that are realistic and within the state of the art," Boyd adds. All too often, he points out, the equipment is too complex for the job it is designed to do. "The industry often permits its engineers to devise weapon systems that are unnecessarily complex—systems that meet unnecessarily complicated specifications."

Military contractors should refuse to accept requirements that can't be met, Boyd declares. "For financial reasons, the industry often refuses to tell the military the job can't be done. It's up to the industry to control the situation and insist on qualified people to write requirements."

Boyd insists that electronic equipment is now less reliable than other systems in aircraft and missiles. "We've accepted and lived with unsatisfactory electronics in the past," he states. "But we should not continue to accept this."—End



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EASTERN INDUSTRIES, INC.

aircraft · missile electronics intelligence

Transistor has fused aluminum-silicon junction

FIFTIETH OF MILLIMICROSECOND is what Sperry Semiconductor Div. engineers estimate to be the switching speed of their new silicon junction avalanche transistor. They also report no measurable delays or recovery times.

Working element of the switch is a fused junction of aluminum and silicon. Diameter of the junction layer is 0.002 in., with a thickness of $4x10^{-6}$ in.

Slight voltage rise spreads electron avalanche

ATOMIC MECHANISM in the junction layer, says Sperry, blocks current flow until triggered by a bit of computer information in the form of a small voltage pulse. Slight increase in voltage starts an electron avalanche that almost instantly spreads throughout the layer, carrying the current across the junction.

Sperry sees a wide range of uses for its device in missile and space vehicle systems.

Mesa transistor called most significant advance in commercial transistors since diffusion

MESA GERMANIUM TRANSISTORS are in production at Texas Instruments. Calling the mesa the most significant advance in commercial semiconductor technology since the discovery of the diffusion technique, TI claims it will provide the highest frequency response, switching speeds, and power capabilities now attainable. Switch speeds are in the millimicrosecond range. Alpha cutoffs of 250 mc are typical of the mesa, says TI.

The original mesa transistor was developed and produced by Bell Labs. Motorola was the first to produce mesas on a large scale. Mesas figure prominently in the new Nike-Hercules electronics, are a natural for other missile systems. (Raytheon also has a mesa type in development.)

Arinc started
Kahn SSB tests with special
2.5-kw transmitter

KAHN SINGLE SIDEBAND tests finally got underway at Arinc's leased transmitter at Centereach, N. Y. A 2.5-kw Wilcox 96A AM transmitter was adapted to the Kahn system so that peak envelope power on single sideband is 10 kw. The adapter is so installed that the transmitter can work either full carrier AM or full carrier SSB.

Arinc reports that, with the Kahn technique, undesired sideband and third-order distortion is down some 30 db.

The transmitter will be used by Pan Am and Eastern airliners in the Caribbean area. So far no operating reports have come in. Tests are to check the improvement—if any—the Kahn system gives over conventional methods.

Kahn system appears
to be the only one
for a smooth transition to
single sideband

ARINC IS PARTICULARLY INTERESTED in the Kahn system, mainly because it might offer a quick and inexpensive way of converting its many ground comm stations to SSB while letting them work conventional double sideband AM during the transition period. To date the Kahn system appears to be the only one that will do this without intolerable distortion.

Arinc, however, does not feel the Kahn system (a full-carrier SSB type) will buy enough improvement over conventional systems to justify its use for airborne SSB.

ON THE STRENGTH of developments so far, it looks as though U. S. airlines will go to suppressed-carrier SSB but will demand AFC in the receiver and a floating carrier in the transmitter. These latter capabilities would be needed when higher airliner speeds (with resulting Doppler shift) and/or data link become available.

Operation on

HF communications channels

foreseen for AGACS

AGACS (Air Ground Air Communications System), being developed for Federal Aviation Agency by RCA, may also be built to work on HF comm channels. Airlines have been prodding the agency for this particular feature.

Speaking for the airlines, Arinc has also been asking that AGACS modulation (on both VHF and HF links) be compatible with existing airline equipment. It points out this "will shorten the implementation time cycle by at least five and possibly by as much as 7-10 years. . . ."

more on next page

BECAUSE of incompatability, Arinc is against the carrier shift method originally proposed as the modulation for AGACS. It would prefer, for instance, some sort of tone modulation even though it might be several decibels inferior in performance.

AIR FORCE disclosed a high rate of recovery of recorded scientific data from some of its Thor and Atlas shots carrying the operational nose cone. Mounted in the GE-developed nose cone (which fits both missiles) is a data recovery capsule that is ejected some time before the re-entering missile impacts (see S/A, "Capsule Collects Data During Re-Entry," Dec. '58, p. 150).

Since re-entry testing is usually done over water, the 18in.-diameter plastic sphere is strong enough to withstand impact and light enough to float. If, for any reason, the telemetered flight data are interrupted, the canned data fill in

the gaps.

According to GE, Ampex makes the tape recorder, Allied Chemical supplies the Du Pont urethane foam, and Standard Plastics molds the sphere's parts.

Infrared adjunct to fire control for **Century fighters**

CENTURY series fighters will be revamped by USAF to add at least two more years of operational life. One new wrinkle will be an infrared adjunct to fire control system.

Convair F-102 and F-106 IR contracts are expected to be let soon. The systems in these airplanes will be much more sophisticated than the simple IR viewer used in Lockheed's F-104.

NAA F-108 fire control system, in design at Hughes Aircraft, will have integrated radar-IR sensors. NAA B-70 and A3J will use IR detectors for defense.

ATLAS guidance for first 2-3 operational ICBM squadrons will be the radio command system made by GE's Heavy Military Electronics Dept. This is the guidance used on the first Atlas to travel the missile's full 6000-mile-plus range.

Electrostatic and nuclear spin types among advanced gyros under study

ELECTROSTATIC gyro is being studied by Minneapolis-Honeywell under Navy contract. General Precision Labs is working on a nuclear spin gyro under a \$83,000 WADC study contract.

Arma, Autonetics, GE, Gulton Industries, Maxson, Motorola, and Northwestern U. are among others exploring "blue sky" methods of determining direction and measuring acceleration. In many cases, the work is financed with private funds. So far a breakthrough — if one is to come—seems years away.

FLAME attenuation of S-band signals has been largely responsible for opening up of C-band for missile tracking and ground guidance systems. RCA's FPS-16 instrumentation radar, widely used for tracking, operates on C.

AIRBORNE gyro compass is in the works at both Arma and Laboratory for Electronics.

Both projects are companyfinanced.

DETECTION over 200,000 miles is the aim of an Avion scheme announced at recent American Astronautical Society meeting in Washington, D. C. The system uses eighthmagnitude stars as background, scans a wide field of view with either IR or optical

If an object such as a missile or satellite appears in the field of view, at least one of the pinpoints of starlight would be blanked out. Successive images would be stored and compared electronically for dif-

Avion says its system could be installed in a satellite to watch for enemy ballistic mis-

IM-99B Super Bomarc will have triple-threat guidanceprobably command radio via Sage for launch and early flight phase, inertial for midcourse, and infrared homer for terminal phase. Minneapolis-Honeywell reportedly will make the inertial system.

BOMARC I uses Lear threegyro stable platform. The gyros are one-degree-of-freedom, floated types. Whole package costs around \$26,000.

Similar table is made by Lear for Nike-Zeus.

MINIATURE high resolution pulse radar developed by Army's Diamond Fuze Labs operates on X-band, with pulse width of 10 millisec and rep rate up to 150 kc. Peak power is 150 W. Range discrimination is 10 ft. The set has no "blind" range, say its designers, and has a statistical accuracy of about a foot.

The modulator uses a block-

ing oscillator and ferrite-cored transformer to pulse a L3028A miniature X-band magnetron. The receiver has a 400-mc, four-tube IF strip with 50-mc bandwidth and 50-db gain.

Ascop analyzers and marker use concepts of statistical telemetry

STATISTICAL telemetry, long talked about in theory, has finally been put into hardware. Associated Science Corp. of Princeton (Ascop) says this month it will deliver production versions of three units that embody the statistical data handling concept.

THE concept is based on the fact that most of the high frequency data telemetered from missiles and aircraft is random in nature. It can be analyzed at the source, and the meaningful information can be sent down as low frequency data. Obviously one of the fruits of such a system is bandwidth conservation.

On the minus side, many missile manufacturers object to the additional complexity in the airborne package that, they say, the statistical concept involves.

ASCOP'S three airborne units are all solid state devices:

- an airborne spectrum analyzer, weighing less than seven ounces and occupying less than 25 cu in.,
- an amplitude probability analyzer, weighing less than two ounces and occupying six cubic inches,
- a time of occurrence marker, weighing three ounces and occupying three cubic inches.

The probability analyzer

ives a plot of vibration amplide vs probability that a given implitude will appear in the andom signal. The time markmarks the exact time at thich a transient occurs.

ABLE between Canada and cotland has been suggested by a ternational Civil Aviation or air traffic control and weather communications on world's busiest interational air route—which also lways has had communications troubles.

ICAO had previously lanned to set up a forward catter link. It dropped the lea when none of the scatter takers would guarantee relible communications to the detect that ICAO wanted.

F IT goes through, the cable fill be 2500 miles long, spaning the Atlantic via Greenand and Iceland. It would be completed in '61.

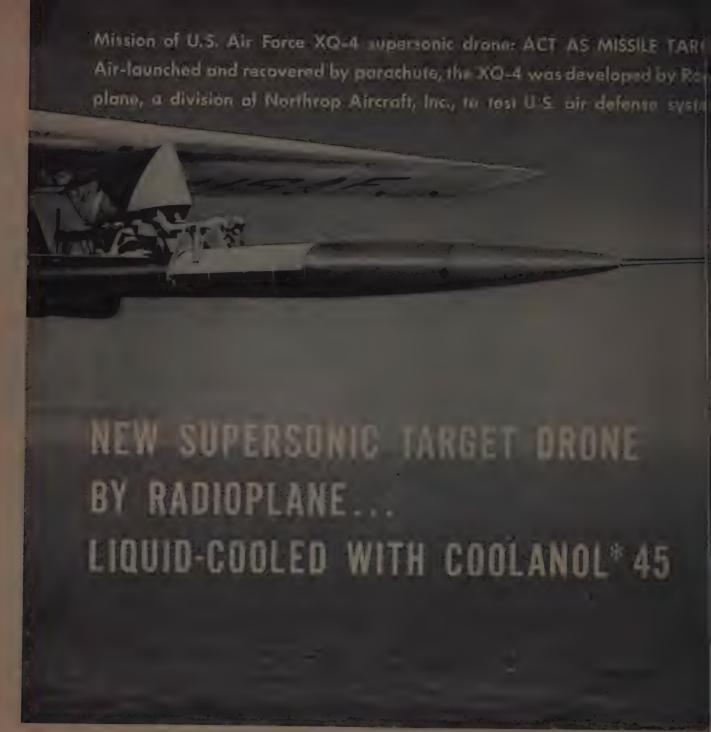
EDERAL Aviation Authory will absorb CAA's Indianpolis (Ind.) Technical Develpment Center in its own R&D
bureau by the middle of this
ear. The Bureau will be
taffed by the technical people
f the former Airways Modrnization Board. Its main failities are at the former Naval
hir station in Atlantic City,
I. J.

VEW type of sub detection ystem will be used in some 50 Lockheed Neptune Navy atrol planes. Half of Lockheed's \$36 million contract will be spent for electronics to letect subs "at greater ranges han ever before."

It is believed the system will be built around the APS-00 airborne search radar in the works at Texas Instrunents.

more on page 21





*Coolanol: Monsanto Trades

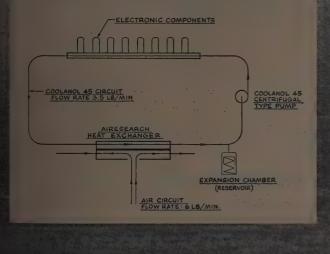
AiResearch uses Coolanol 45 for increas

Liquid cooling with Coolanol 45 keeps temperatures within critical operating limits to assure reliable function of electronic equipment. Coolanol 45 is a coolant-dielectric liquid...operable from -65°F. to 400°F. Pumped around tubes

and other jacketed components, it sorbs heat...carries it from sensitive a to be dissipated in the heat exchan

Coolanol 45 is versatile . . . operates a hydraulic fluid as well as a coolant one fluid "package" you can satisfy b

Coolanol 45 dissipates heat from electronic components of the XQ-4 through this liquid-cooling system (above) designed by The Garrett Corporation, AiResearch Manufacturing Division. System utilizes liquid cooling with Coolanol 45, forced air and cold plate as shown in the diagram.



AiResearch uses Coolanol 45 in all these missile cooling components to assure precise temperature control of electronic equipment ... a key factor in missile reliability.

MINIATURE CENTRIFUGAL FUMP Paristymanon Data

Operating temperatures: 65 degrees to 275 degrees F. Fluid flow (Cooland 145): 0.5 GPM at 10 psig. and 160 degrees F. Motor: 10,000 rpm, 3-phase, 400-cycle ac (1½ inches) Net weight: Less than 1 lb.



UNIT No. 172010 Performance Data

Heat rejection:
2 kw at 15,000 ft, under the following conditions:
Ambient air: 50 degrees C. Fluid flow (Coolanoi 45): 2 GPM Fluid temp. in: 117 degrees C. (Max. system temp.)
Envelope dimensions:
8.2 x 6.8 x 4.12 in.
Wet weight: 3.81 lbs.



UNIT No. 172030

Heat rejection:
10 kw at 5000 ft. under following conditions:
Ambient air: 53 degrees C. Fluid flow (Coolanol 45): 5 GPM Fluid temp. in: 135 degrees C. (Max. system temp.)
Envelope dimensions:
11 x 9.06 x 6.34 in.
Wet weight: 9.48 lbs.



UNIT No. 172040 Performance Data

Heat rejection:
2 kw at 60,000 ft. under following conditions:
Ambient air: 22 degrees C. Fluid flow (Coolano! 45): 1 GPM Fluid temp. in: 147 degrees C. (Max. system temp.)
Envelope dimensions:
7.5 x 8.5 x 6.6 in.



UNIT No. 171600

Heat rejection: 175 watts
Fluid flow (Coolanol 45):
1.5 GPM
Fluid temp. in: 50 degrees C
(Max. system temp.)
63 degrees C.
132 degrees C.
132 degrees C.
132 degrees C.



iability in missile cooling systems

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ELECTRONICS INTELLIGENCE

IMPROVED Sidewinder guidance unit is being worked on by Motorola's Military Electronic Div. under a \$1,086,000 BuOrd contract.

Pan Am places first airline Doppler order with Canadian Marconi

SIX Doppler radars were bought by Pan Am from Canadian Marconi. This is the first purchase of a Doppler by an airline.

PAA bought the FM-CW sets for its Boeing 707-120 jet-liners. The set doesn't meet Aric Characteristic 540-C-M and U.S. manufacturers don't expect to have sets that do until 3-6 months from now.

WHETHER this order offers a clue to PAA's or the other airlines' plans for the future is questionable. C-M in any case got the jump on U.S. manufacturers.

Competition for the Doppler radar sensor market—put at 1700 units for domestic and international airlines combined—is expected to be fierce.

SPERRY system built around a traveling-wave tube makes small test drones look like giant attacking bombers to ground radar systems. New "echo enhancer" was used successfully in recent flight tests at Cape Canaveral that involved the Sage air defense network.

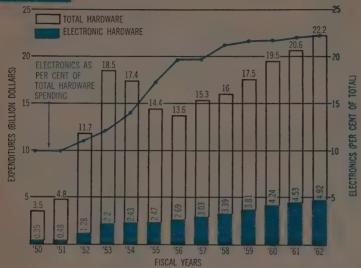
The TWT package detects a ground radar pulse and sends back a greatly intensified signal. It has worked well at Mach 2 and altitudes over 50,000 ft. It can be widely used with either surveillance, air intercept, or tracking radars. It also enables ATC ground controllers to pick up and identify aircraft with greater certainty and at greater distances.

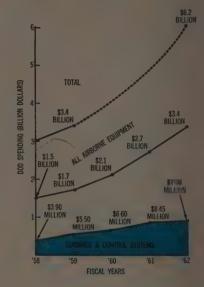
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Special Report

Space/Aero Electronics





ELECTRONICS' share of Defense Dept. hardware spending (left) has gradually increased from 10 per cent of total in fiscal 1950, should hit 22.2 per cent in fiscal 1962. Despite this rise, space/aero electronics remains a buyer's market, with specially fierce competition for prime contracts. Research and development spending (not shown) is hard to determine accurately, since funds for it are drawn from

DOD's production and procurement accounts as well as fom R&D accounts. Right: Guidance and control system costs in relation to other DOD missile expenditures. Spending on airborne equipment is expected to rise once the emphasis is switched from R&D to production, initial outlays are made on GSE, etc. All figures are estimates, except for fiscal 1958 and 1959 guidance and control system costs.

Electronics outlook

is bright in changing market

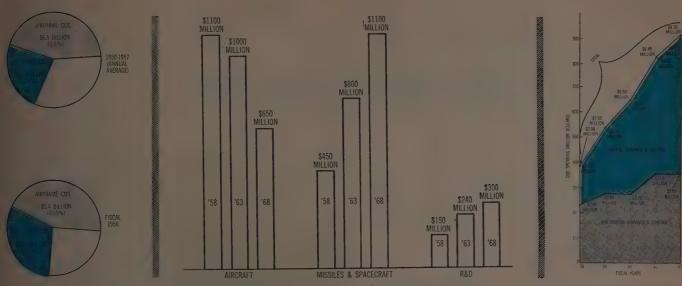
- AIRCRAFT: Military market shrinking, civil market expanding
- MISSILES: Electronics share to increase on all types except air-to-air
- SPACECRAFT: Little immediate effect on total market but potential is vast

by James Holahan, Electronics Editor

THE military will spend more on electronics in 1959 than in any past year, yet electronics will be a buyer's market. There will be no dearth of top industry talent competing for the some \$3.3 billion to be spent by the Defense Dept. on aircraft, missile, and spacecraft electronics.

Competition will be particularly fierce for prime system contracts. Under the weapons

system approach, the practice has been to award the prime to a firm with either a strong airframe or a strong electronics background and with systems management capabilities. Now, as the industry moves more rapidly from manned to unmanned combat vehicles, the airframe background is becoming less of a factor. The military has been stressing electronic skills combined with



PRIME contract awards (left) to 45 electronics and 28 airframe companies (including suppliers). Electronics category includes firms making propulsion units as well (such as GE and Westinghouse). Pie chart at top covers the period from July 1, 1950, to Dec. 31, 1957. Center: Military spending for airborne electronics (estimated for 1963 and 1968 in 1958 dollars). Figures exclude all ground equipment (even

command guidance radars, ground portions of data links, etc.), assume cold war situation continues as at present and DOD budget increases gradually. Marketing analysts at major electronics company who computed this forecast believe it's accurate within ± 10 per cent. Right: Trend to inertial guidance for missiles is shown by the results of a poll of missile and guidance system manufacturers.

systems engineering capabilities.

Except for certain special skill areas, the difference between airframe and electronic companies in technical competence for missile-making is shrinking more and more. Among major firms, both types have hired and otherwise acquired the other's skill.

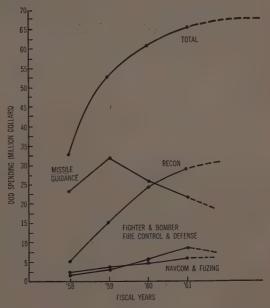
All this adds up to the fact that the military has more system primes to choose from. Nevertheless, observers don't expect any drastic change in the near future in the practice of giving electronic companies only the primes on the smaller birds.

Team bidding, started only a year ago by medium-sized firms determined not to be shut out of large contracts, is having a profound effect on the industry. "Government-sponsored trusts" of large and small

firms alike are being set up to compete for major contracts and subcontracts for which a variety of skills and facilities and large amounts of money are needed.

One disadvantage of the team system is that certain coalitions may be formed that may make it harder for the fellow who was not asked to join to sell his "better mouse-trap" or technical skill to the military. Some electronic companies also fear the team system might provide channels through which much of their proprietary information might leak to competitors.

The military—especially AF—today doesn't have the inhouse competence for the proper technical evaluation of large projects. Therefore the military is putting more stress on the past experience and present competence of a bid-



MILITARY infrared market, industry survey shows, should climb to some \$70 million a year by '61 and then level off. Hopes are high for recon and surveillance applications, especially at very high altitudes and in space. IR guidance should peak in '59 and then fall off—because of shrinking role of manned interceptors with their air-to-air missiles, many of which are IR-guided. Civil market for IR, the survey showed, should amount to about \$5 million annually for the next 10 years.

State of the art:

Inertial guidance

- Electrostatic suspension
- Gas-lubricated and nuclear spin bearings
- Applied cryogenics

by W. G. Wing, Head, Engineering Dept.-Inertial Guidance, Sperry Gyroscope Co.*

THE progress of inertial guidance (IG) is paced by the development of the components, mainly gyros and accelerometers. The engineering effort in this area is paying off in a wide variety of applica-

Sperry Gyroscope Co., Great Neck,

tions—Sperry, for instance, is working on IG hardware over a range from the B-58's primary bombing navigation system to the miniature X-15 platform and the ultra long time Sins system.

Gyros for inertial systems must be of high quality in all but the crudest applications. Long missions demand very low drift rates—tough to get even under the best conditions. Ballistic missile guidance requires only moderate drift accuracy, but in a troublesome acceleration and vibration environment. What is needed, above all, for ballistic missiles is good isoelasticity (uniform rigidity in various directions).

Suspension techniques characterize the different types of inertial gyros. The commonest is liquid flotation (with some supplemental means of establishing axes of rotation). It gives extremely low friction levels and excellent resistance to shock and vibration. The density of the moving element is limited to that of the fluid, which keeps down angular momentum and structural rigidity. Also, you have to be especially careful in assembling floated units to avoid dirt and gas bubbles.

Hydrostatic bearings using a compressed gas have been applied for some time. With careful design and fabrication, these can give very low uncertainty torques. Generally, gas bearings don't restrict the density of the moving element,

hence give high angular momentum with structural rigidity. The main drawbacks of this suspension are:

• high tolerances on parts,

• the need for a continuous gas supply,

• the need for absolute cleanliness to keep foreign particles out of the small gaps.

Ball bearings for less critical uses

The old standby of the control gyro field, the ball bearing suspension, is virtually unknown in inertial guidance. One recent design, Sperry's Rotorace* shows promise for the less critical areas (see AvAge, "Non-Floated Table' Nears Inertial Accuracy," June '58, p. 128).

Both single- and two-degree-of-freedom gyros are used. The single-degree-of-freedom gyro needs extreme mass-balance accuracy about only one axis and good isoelasticity in only one plane. On the minus side, it allows significant disturbances in the orientation of the spin vector under angular vibration. This leads to drift error. Moreover, you need

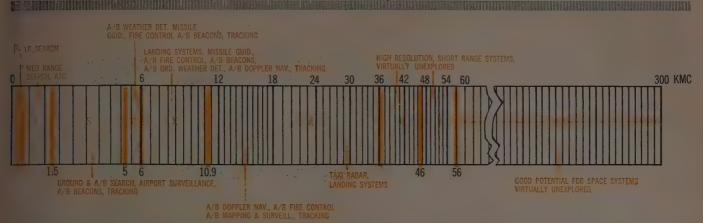
*Registered trademark.

more on page 82

ACCELERATION DATA ACCELEROMETER COMPUTER SPACED-FIXED POSITION ACCELERATION DATA ACCELEROMETER PRECESSED SYSTEM GUIDANCE ACCELERATION DATA **ACCELEROMETER** COMMAND ACCELEROMETER ROTATION (MULTI-AXIS) **ERROR** SERVO SYSTEM **GYRO** (3 AXES) **GYRO**

-

SPACE-FIXED AND PRECESSED gyro guidance systems. Space-fixed method of getting directional reference is well suited to ballistic missile needs. Limiting factor is gyros' susceptibility to time-varying direction of gravity, which leads to appreciable drift rates. Principle of precession is to rotate the entire reference element in a manner corresponding to the earth's. Gyro torquing accuracies are critical here.



P-band to MM region. Far end of spectrum is virtually unexplored,

SPECTRUM for aircraft, missile, and spacecraft radar extends from has great potential. As interference problems among "friendly" radars are becoming acute, at least some semi-rigid control is expected.

State of the art: Radar

- High power microwave amps and components
- Statistical design concepts
- Low noise UHF and microwave components

by James Holahan, Electronics Editor

PERFORMANCE gains in post-Korea radars have come at the expense of an inordinate amount of R&D and engineering. Now, thanks mainly to three developmental "dents" (if not break-throughs), there should be a bigger performance payoff on future efforts. These "dents" are:

• availability of high power microwave amplifiers and associated components,

- development of statistical design concepts,
- availability of low noise UHF and microwave com-

The basic task of any radar is to detect its target and to discriminate against other targets and natural and manmade decoys. Both detection and discrimination depend very largely on the amount of returned energy.

ment for a workable signal therefore is that the ratio of received energy to noise per unit bandwidth (E/N_o) be sufficient for the signal to be resolved in range, azimuth, or

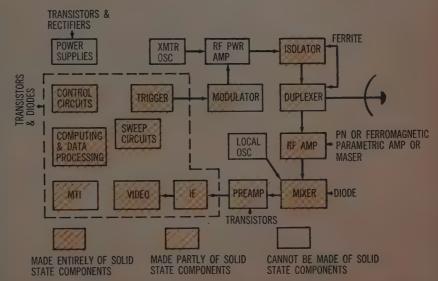
Inevitably, any return car-

ries noise. The basic require-

waveform (depending on the particular system).

Obviously, the most direct way of insuring a workable E/N_o is to put more energy on the target. This means higher radiated energy per de-

more on page 94

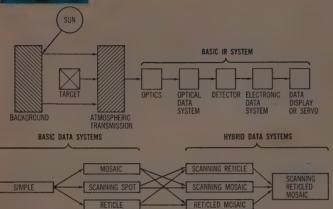


"SOLID STATE" radar progress. Use of transistors, diodes, and ferrites in future radars will mean smaller, lighter, cooler, more efficient, and more reliable equipment. In several radars, a few of the "building blocks" have been transistorized. Within a few years, we can expect to see many radars "solidified" at least to the extent shown here.

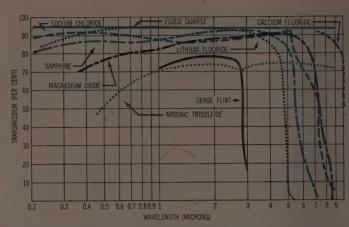


Special Report

Space/Aero Electronics



BASIC infrared design (left): Block diagram of typical IR system (top) applies equally to search, track, and dual-mode use. Main differences between these applications lie in the data unit. Bottom:



Eight possible versions of the data unit. Right: Optical transmission characteristics of some typical infrared materials. The most sensitive detectors must be operated well below ambient temperatures.

State of the art: Infrared

- Rapid-scan detectors for better recon
- Use of thermopause region of spectrum
- Needed: Adequate radiation data

by Raymond H. McFee, Director of Research Avionics Div., Aerojet-General Corp.*

THE recent success of infrared-guided missiles in the Quemoy are a conclusively proved infrared (IR) design has reached operational capability. For IR equipment, this capability hinges on two factors: (1) adequate supply of special components and (2)

*Aerojet-General Corp., Azusa, Calif.

effective detection and information processing techniques.

Generally, IR system design has two major requirements:

• The system must be able to determine angular location and other characteristics of targets of interest accurately and reliably — even though other sources of IR radiation,

not of interest to the system, may be present.

• The data received by the system must be in such a form that they can be acted on (either instantaneously and automatically or after interpretation).

IR devices use optical, electronic, and photoelectric methods to get their information. Since they operate chiefly in the wavelength region of the electromagnetic spectrum between 1 and 15 microns (10⁻⁴ to 1.5x10⁻³ cm), angular resolution limits due to diffraction effects are relatively minor.

By the same token, the mechanical tolerances of the energy-gathering devices are correspondingly more severe than in longer wavelength equipment. However, since optical techniques can be used in this spectral region, the necessary precision is easily obtained.

Trend points to use of 3-5-micron region

The trend in current IR system design is primarily toward use of the thermopause region of the spectrum (3-5 microns). Some consideration is also being given to the thermal region (8-14 microns). The interest in these regions is the result of the increasing availability of long-wavelength detectors and optical materials.

more on page 88

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FIGURE 1: Detached shock waves can be obtained with magnetohydrodynamics, STL studies show. Probe containing magnetohydrodynamic device was inserted into very high speed gas flow. Before and after

shots show that, after the device is turned on, the magnetic field pushes the gas off the probe's critical areas by changing it's density. This might be used to prevent overheating of re-entry surfaces.

Magnetohydrodynamics

opens up new electronic vistas

The new science of magnetohydrodynamics, or the interaction of electric effects and fluid flow, is paving the way for important advances in both basic electronics and overall vehicle design. Applications already under study include magnetic fields as the propulsive force for space craft and microwave amplification.

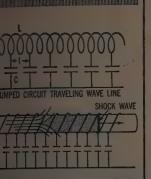
by Irwin Stambler, Engineering Editor

MAGNETOHYDRODY-NAMICS (MHD) is the combining form of the previously separate disciplines of electrodynamics and hydrodynamics. The former has to do with the complete study of electricity, magnetism, and their combinations, from propagation phenomena to MHD. Another way of defining electrodynamics, Dr. Milton Clauser, of Space Technology Labs (STL), told SPACE/AERONAUTICS, is that it deals with any phenomena described by Max-

well's equations. Hydrodynamics, of course, deals with all kinds of fluid motions.

If the fluid becomes a conductor, the two disciplines are united. MHD is thus defined as the study of the interaction of magnetic fields with electrically conducting fluids.

"Magnetoaerodynamics" is sometimes used interchangeably with MHD. However, Dr. Clauser points out, the former stresses with cases for which compression and changes in density are important, which



URE 2: Electromagnetically en shock tubes can provide s of propelling gases at eds of ever Mach 100. One p, suggested by Dr. Milton ser of STL, uses lumped meter electric circuits, which traveling magnetic fields tom) for driving shock waves. e speed is I/\sqrt{LC} , where L nductance per section and C citance per section.

lies gas phenomena. MHD more general term coverboth incompressible and pressible fluids.

here are electrically contive liquids, Dr. Clauser lains, such as mercury and water, but it hasn't been sible to do much that is unal with them. The situation lifferent with gases. In air 2500 deg K and over, example the thermal moof the molecules is such collisions occur in which molecules dissociate into ms. At about 4000 deg K, trons begin to be knocked The gas becomes ionized a good conductor.

Conductivity rises rapidly temperature. Dr. Clauser es that, at the re-entry eds of ballistic missiles, the becomes as good a contor as carbon. Once a gas this property, it can be shed" by a magnetic field. more on next page



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MAGNETOHYDRODYNAMICS

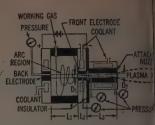


FIGURE 3: Plasma generator us in MHD research by Gianni Plasmadyne.

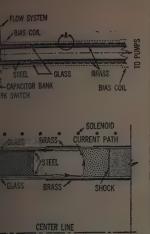


PLASMATRONS Giannini Plasmadyne are alread finding commercial and test use This unit is used for high temper ature materials studies and f flame-spraying of ceramic coa

One of the major areas current MHD work is the d velopment of magnetic powe plants for spacecraft. This based on the acceleration a stream of ionized gases, plasma, by electric means.

One approach, taken such firms as Giannini Plasm dyne, is referred to as therma plasma propulsion. It us electric arc methods involving Joule heating with and wit out electrodes-to provide h gases with specific impulses b tween 300 and 3000 second A typical plasma generat consists of a cylindrical char ber with a cylindrical ele trode at the back and an a nular one at the front (Fig. 3

In operation, a power su ply establishes a potential b tween the front and back ele trodes and maintains an a between them. The working gas passes through the arc at out the orifice, being heat



ELERATION of fully ionized tma to propel spacecraft has n suggested by such firms as o, which has designed this gnetic test accelerator. View shows detail of accelerator's akdown region.

lionized in the process. Bese the gas has a high exit ed, it imposes a great deal thrust on the chamber.

Another approach, called gneto-plasma propulsion, kes use of strong pulsed ctromagnetic fields to ionize accelerate gases. Specific pulses from 3000 seconds up can be obtained. A great my organizations are workin this area, including co, GE, Giannini Plasmate, and STL.

A typical design of this type kes use of a long tube to ich are attached a series of densers (Fig. 2). The setis similar to that of a travelwave tube. A solenoid is installed around the tube. the propellant starts to er the tube through a valve, ctrodes are fired that proe enough electric discharge ionize a small amount of propellant. The magnetic rent from the solenoid inces a current in the gas. is current repels the current the coil, and, since the coil 't move, the gas does.

The condensers are fired

more on next page



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one after the other to help push the gas out the tube. In the process, a shock wave develops that ionizes the rest of the gas. Continued "squeezing" ionizes the gas ahead of the tube to make it a good conductor. Thus when a current is induced in the gas, the whole slug of gas is moved out of the tube at very high speeds.

These MHD powerplants have low thrust-weight ratios and so wouldn't be good for such functions as takeoff from earth for which high thrust is needed. For space flight, though, where small accelerations over a long period are needed, they are very promising.

Re-entry drag brakes are being studied

The fact that gas around a re-entry body is heated and ionized to the point w ere it becomes a good conductor pinpoints another possible use for MHD. In this regime, magnetic fields can be used to control the flow around a body and also to influence the heat transfer rates.

In detached shock studies, STL has inserted a probe containing a 40-turn solenoid into very hot and ionized gas (Fig. 1). When the coil is activated, a voltage is generated in the gas that produces a current flow. The interaction between this current and that of the magnetic field displaces the shock from the nose of the probe body. The magnetic lines of force are bent back along the body.

One of the basic limitations in MHD is magnet materials. The stress imposed on coils by a magnetic flux of 5000 gauss is about 15 psi. Magnets have been built that provide about a million gauss for short periods—but the result is a tensile stress in the coils of 600,000 psi, and practically no material is available to take this. Copper, for instance, is good to 250,000 gauss, and beryllium copper yields at 750,000 gauss.

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Conclusion: With proper attention compounding, silicone rubber still pr vides better extreme temperature flex bility than any other elastomer. Thoug disappointing at 800°F, the same cor pounds tested will withstand 500°F for month or more, 700°F for several hour They will also remain flexible in the -65°F to 130°F range and provide shelf life of many years.

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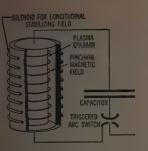








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NCH EFFECT is important in a g n e t ohydrodynamic work. nching of magnetic field comesses hot plasma and keeps away from chamber walls. One the devices used for pinching this linear discharge tube.

One group working on this oblem is that of Dr. Ralph aniek, at Giannini Plasmane. To extend the range of nventional types of magnets, told Space/Aeronautics. s group is studying designs ing tungsten alloys especially eated for high gauss work. hese have high conductivity d tensile strength. Some deces are operated in liquid trogen so that heating effects ke longer to destroy the coil. aniek's group has designed a imber of test devices; in one se 2x106 gauss, or 3x106 i was obtained in explosive

However, researchers have ome up with an alternate ethod—"force-free" coils. If ou have two parallel, infinite eets, Dr. Waniek states, and separate current in each eet, lines of force produced these currents don't intract if they are at 90 deg to each other. They do, though, roduce a net magnetic field a certain direction.

Force-free geometry, Dr. Vaniek notes, is complicated at not insoluble. Researchers are already developed successful test configurations. ike the high gauss designs, is approach can provide elds acting as "pistons" in occlerating gas to high speeds.

more on next page



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MAGNETOHYDRODYNAMICS



REACTION chamber of fusi propulsion system suggested Dr. Clauser and Dr. Weibel STL. This system uses the pin effect to contain plasma the reaches temperatures on the der of 4x108 deg K.

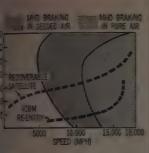
Looking further ahead, r searchers believe that ionize gas can be made to attain force-free geometry, so that magnet can eventually made of gas instead of a soli

Although originally studie because of its astronautic possibilities, MHD also show promise for a wide range other uses in fields ranging from metallurgy to electronic For instance, the plasma jet first developed for research have already been applied cut metals and for flame-sprecoating. Some researchers has suggested possible uses microwave amplification.

Other studies are aimed developing MHD generator and motors, to replace convetional power generation equiment, such as the steam to bine and the electric generator. In a conventional motor, finstance, inducing current copper conductors results magnetic forces that rotate to armature. In an MHD unthe copper is replaced by conducting fluid.

One obvious advantage that moving parts are elim nated, which overcomes surproblems as bearing performance, fatigue stress, etc. Muchigher speeds should becompossible. Another interesting aspect is that it wouldn't necessary to retain the circular geometry of convention units.

Still another application



AKING of very high speed icles by magnetohydrodynamis being studied by Avco.



GNETOHYDRODYNAMIC methos are used by Giannini Plasdyne in this hyperthermal drunnel. Hot jets of ionized les. or plasmas, simulate that very high Mach numbers d stagnation temperatures of 15.000 deg K.

HD is in controlling fusion actions. In fact, this may ove to be the most important of all MHD uses.

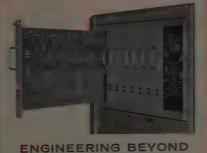
The fusion precess is based the fact that the collision two light elements—partically the isotopes of hydron—can result in their fusion to a heavier element with a lease of energy. As Dr. auser states, the reaction tes don't become significant till the relative velocity is ghe enough to overcome the bullomb repulsion of the targed particles and bring e nuclei together. This returns a speed of 10° m/sec, hich corresponds to 100,000 deg K, or 10 kev.

Obviously, at such temperares no reasonable volume of e plasma can be in touch with ny cold matter, or it would lench itself. The solution is keep the plasma from touchg the chamber walls by mag-

more on next page

MISSILE IMPACT PREDICTION





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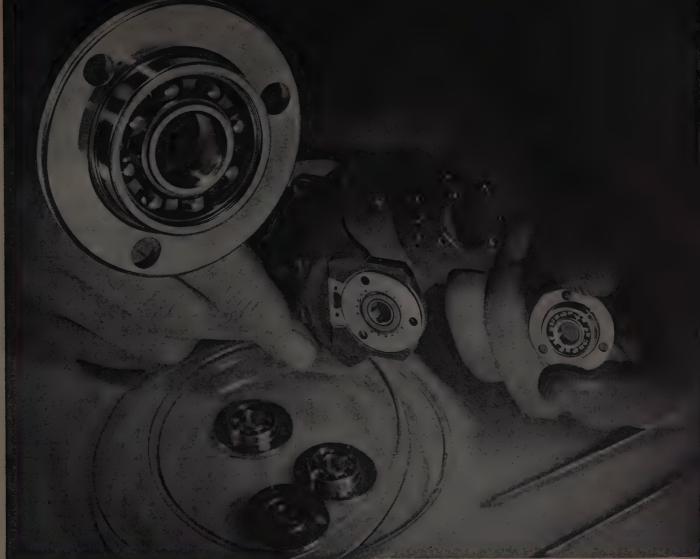
CONDUCTIVITY of air rises rapidly with increasing Mach number after dissociation begins (about Mach 12), STL graph shows.

netic or electrical force There are several ways confining the plasma for th fusion process. Perhaps th best-known is the "pinch" et fect—a confining magneti field surrounds the plasma an acts as a magnetic wall. ST is working on ways of squeez ing gas so rapidly that th molecules are literally swer up to speeds of 10° m/se The aim is to sweep the mole cules in and hold them "cook" in the center of th device while the fusion read tion is going on.

To do this, Dr. Clause states, it's necessary to apply magnetic field 10 times as ra pidly as has ever been done s far. Nevertheless, he believe an airborne fusion powerplar will be developed some day,

Another way to contain plasma is to use a standing wave RF field such as can b produced in a cavity. The ol ject is to drive both ions an electrons in a cavity towar an electric node by feeding the cavity with a waveguid The RF waves beating back wards and forwards on th plasma hold it in the center an the RF currents in the surface of the plasma heat it. The he column of plasma thus ol tained acts like a coaxial cor ductor at cutoff.—End

*M. U. Clauser & E. S. Weibel, "R dintion Pressure Confinement, the Shor Pinch and Fersibility of Fusion Propusion:" STL Report.



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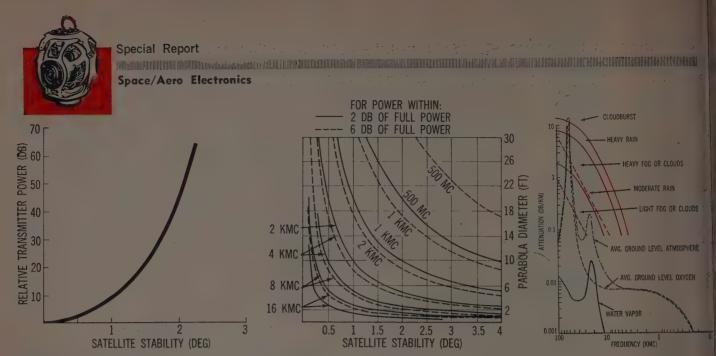


FIGURE 1: Satellite stability vs transmitter power and parabola size (left and center) and atmospheric attenuation by gases and precipitation. Attenuation curves are for water vapor of one g/m³, oxygen at 76 cm pressure at 20 deg C, and atmosphere of oxygen as above

and water vapor of 7.5 g/m³ (average for summer day in temperate zone); moderate rain of four mm/hr, heavy rain of 25 mm/hr, cloudburst of 50 mm/hr, fog or clouds of 0.32 g/m³ (about 400 ft visibility), and heavy fog or cloud of 2.3 g/m³ (about 100 ft visibility).

Reliability and stability most critical in

satellite communications

What are the most important problems facing satellite communications designers and how can they be solved? The answer to this question is given in this condensed version of a chapter from a detailed study of satellite reconnaissance written by a group of 10 scientists and engineers at Du Mont Labs.*

OF ALL the factors involved in the design of satellite communications systems, equipment reliability probably is the most critical. Exceptionally

"Space Reconnaissance Laboratory, Research & Develoyment Div., Allen B. Du Mont Laboratories, Inc., 750 Bloomfield Ave., Clifton, N.J. high reliability is essential for the substantial operating life required for unmanned, nonrecoverable communications satellites.

Probably next in importance is satellite stability. Because of the tremendous distances

spanned by high information rate satellite communications, high gain, narrow beam antennas should be used. But such antennas can be used only if a certain degree of satellite stability is insured.

Other factors that must be considered include: (1) S/N ratio vs bandwidth; (2) modulation characteristics; (3) frequency; (4) antenna directivity and gain; (5) power vs equipment size; (6) propagation in atmosphere, ionosphere, and space; (7) Doppler shift; (8) distances; (9) fixed vs movable antennas; (10) solar and cosmic noise;

(11) command link functions; (12) jamming; (13) data link considerations; (14) integration into overall vehicle.

For satellite - ground communication, the signal must pass through the atmosphere. This sounds worse than it really is—the distances involved usually are fairly short. If the signal follows a path normal to the earth, for instance, it has to go through merely 12 miles of atmosphere.

However, even if we assume that the signal will travel only through 50 miles or less of atmosphere, atmospheric absorption still does limit our choice



IGURE 2: Four-satellite comnunication system.

f frequency. Uncondensed ases in the atmosphere limit ne frequency to about 20 kmc -they may attenuate this freuency by as much as 0.2 b/km on an average summer ay in the temperate zone Fig. 1). Things get worse hen it rains even moderately: ow frequencies of 20 kmc are ttenuated by about 0.4 b/km. Taking precipitation nto account and assuming that ropagation through heavy ain is impractical, we thereore get a new upper limit of sable frequencies of around

Fading also must be condered. Standard practice calls or a fade margin of roughly db/mile for 99.9 reliability. This indicates a margin of e-25 db for satellite-ground ommunication to insure high ropagation reliability.

The Doppler shift that must be expected is a function of the communications system's operating frequency and of atellite elevation angle and period—for instance, at an orbit period of 24 hours or more in the plane of the earth's equator, it becomes zero. How important Doppler effects are depends on the type of modulation and the system bandwidth.

Generally, satellites will more on next page

VIGILANTE

The Navy's new all-weather attack weapon system packs a precision punch...
for limited war, or all-out conflict

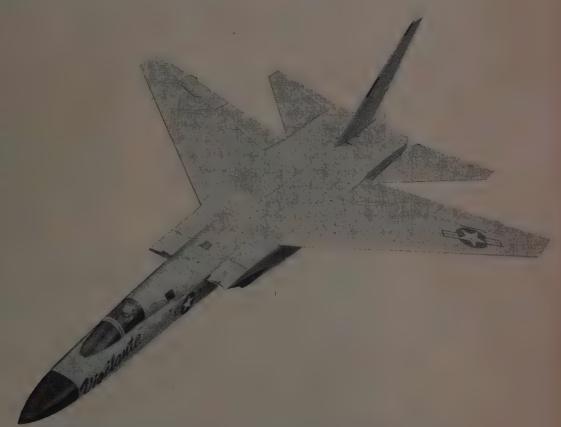
When North American's A3J Vigilante joins the fleet, the Navy will have its first supersonic carrier-based attack weapon system.

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rier decks and short runways ashore.

The A3J is a true weapon system. All electronic systems and auto-flight controls are integrated, and were designed to enable the A3J to carry out all-weather, all-attitude weapon delivery. Its precision bombing-navigation system—outstanding in tactical efficiency—is the result of a coordinated effort by North American's Columbus and Autonetics Divisions.

Most important—the A3J has men: a pilot and a bombardier-navigator. For only men can respond to the unexpected. Only men can make decisions and report results. Only men can think. That is why, now and in the future, we must have manned weapon systems like the A3J Vigilante to keep our defense in balance.



Also from Columbus—a new concept in Navy basic training

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THE COLUMBUS DIVISION OF NORTH AMERICAN AVIATION, INC.



Noise Figure vs Frequency

(mc)	Noise	Figure	(db)
		6	
		7÷8	
		7-8	
		12	
		16	
			7-8 7-8 12

have to be tracked. In special cases, however, it may be better to use suitably oriented fixed antennas with switching systems at RF or IF frequencies. For a multi-satellite system, it may prove worth while to combine these two methods. The tracking problem is not very serious—the tracking rate ranges from 18 to 60 deg/min, for instance, for a circular orbit at 300 miles above the earth.

The propagation characteristics of the ground-satellite link are quite similar to those of the satellite-ground channel. Of course, the transmitting equipment limitations now are much less stringent than on the satellite. The ground transmitter could have considerably greater power, so that a lower gain receiving antenna could be used on the satellite.

This naturally applies only to a single-satellite system. In a multi-satellite arrangement, command signals probably would have to be relayed through one or more of the satellites.

Stabilization commands are likely to be sent to each satellite of a multi-vehicle system as it passes over the main tracking station. As they are so critically important, it may prove best to have a separate link for them, with a simple antenna and receiver in each satellite. To provide for failure of this link, the general command link might have an alternate channel for stabilization information. (Generally, a reliability analysis may well show that extensive standby

why

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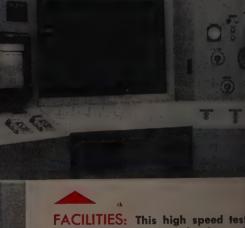
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provisions are needed for a multi-satellite system.)

Figure 2 shows a possible configuration for a four-satellite system. All inter-satellite functions could be handled by a complement of two antennas on each vehicle. Two legs of the inter-satellite link would be used for data transmission and two more for command signals.

If necessary, one additional antenna on each satellite would be enough for stabilization, command, and data transmission. However, as we have seen, in the interests of reliability a fourth antenna may prove desirable. This antenna—or at least a separate feed for the existing antenna—may be necessary for another reason, too: Very probably, stabilization and data recording will take place concurrently.

Stability indirectly affects gain

The reason why satellit stability is so important is that t is directly related to antenna beamwidth and size, which in turn affect the overall system gain. For instance, Figure 1 shows that, for a 10-ft dish and a frequency of two-kmc and assuming minimum transmitter power is desirable, the satellite must be stable within one degree to keep the system gain within two decibels of the signal that would be available with perfectly aligned antennas. A stability of 1.75 deg would be needed to keep the gain within six decibels of full

At first glance, Figure 1 appears to show that the highest stability and frequency would be best. However, when all the aspects of the problem are considered — including propagation, mechanical, cost, and state-of-the-art factors—lower frequencies and stability will often prove desirable.

In cases for which the required satellite stability be-

more on next page

comes mechanically or economically impractical, an alternative solution is conceivable: A rather wide beam antenna could be used at the transmitting end of the system and a very high gain, narrow beam antenna at the receiving end. The latter would then home in on the transmitted signal by means of a sophisticated tracking mechanism.

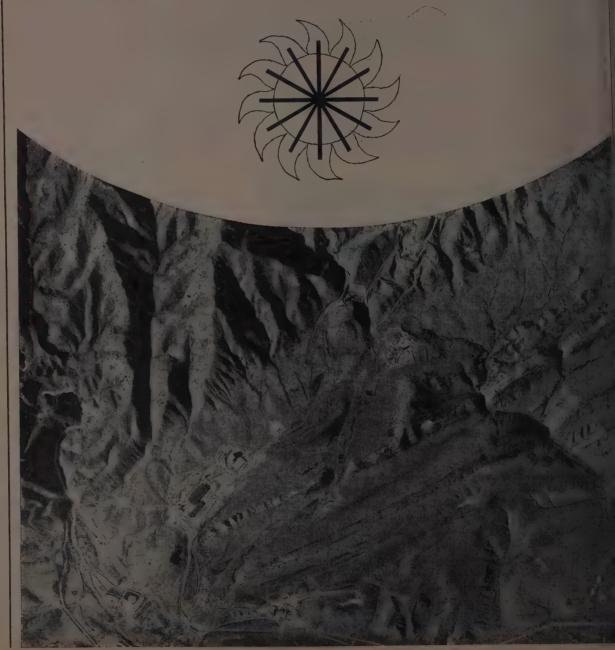
This solution could be applied, for example, in a system using 18-ft dishes at two kmc for which a required satellite stability of 0.5 deg (needed to keep the link within two decibels of full power) could not be obtained. If the stability value had to be increased to 1.5 deg, an additional 21 db of transmitter power would be needed, and this probably would be undesirable.

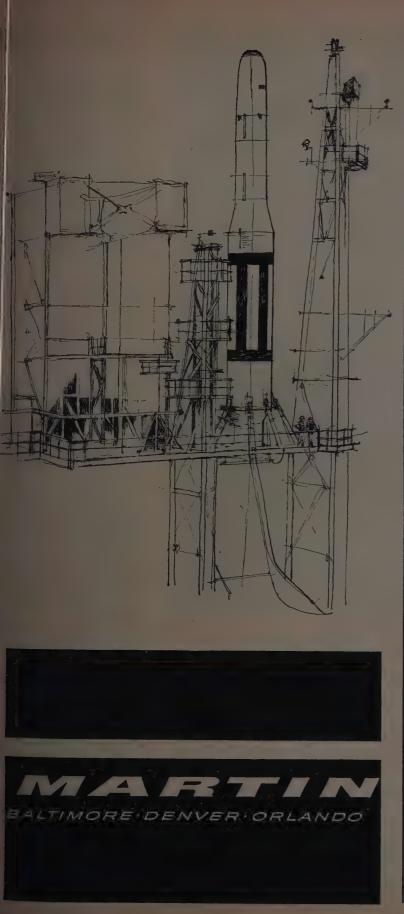
Power increase feasible for smaller antenna

On the other hand, an increase of six decibels in transmitter power in a system with a 6.5-ft transmitting antenna with one-decibel power points at ±1.5 deg and a 25-ft tracked receiving antenna would provide the necessary gain at two kmc. (Other compromises might involve a frequency change as well.)

The problem of transmitter power vs satellite stability is further illustrated in Figure 1. Let's assume that, for a communication link between two satellites, we decide that propagation considerations require 18-ft reflectors to provide the necessary voltage at a two-kmc receiver. This, of course, calls for a certain transmitter power. Figure 1 then shows how much additional power is needed to overcome the satellite instability.

We can see that, if the satellites cannot be stabilized to ±2 deg, we must consider lower frequencies or smaller dishes as well as higher transThere is nothing else like this under the sun. It is the Martin-Denver facilibirthplace of the Air Force TITAN. It is also this country's most advance and fully integrated big-missile development center. Here, our most formiable weapon systems of tomorrow are being designed, built and tested from the smallest component to the total system—within a single 7,000 ac complex. Every top military and scientific expert who has seen Martin-Denver from within, considers it one of our most valuable national resource.





SATELLITE COMMUNICATIONS

mitter power. However, it is quite impossible in this case to get enough transmitter power to overcome the system loss due to satellite instability. When power or dish size is limited, the highest usable frequency consistent with satellite stability would be used.

The minimum detectable signal, which will overcome the noise in the receiver output, depends largely on three factors: (1) receiver noise figure, (2) bandwidth, and (3) temperature. Any reduction in any of these factors reduces the amount of transmitted power needed for the system.

However, we must remember that the transmission rate is directly proportional to bandwith. The narrower the bandwidth of a system, the lower is the information con-

The Table gives noise figures obtainable with present techniques. The low noise figures around three kmc usually are obtained with traveling wave tube amplifiers. These tubes are highly desirable in ground equipment; because of their size, they may not be suitable for satellite installation.

In noise power formula, T poses problem

For the designer of satellite equipment, an interesting problem lies in assigning the proper value for T in the equation for noise power (P_N) in a given band:

 $P_N = KTB$.

where K is Boltzmann's constant; T, temperature (in degrees K); and B, bandwidth. At least part of the resistance of the system supplying signal power to the receiver is due to antenna radiation. The proper temperature that goes with this resistance depends mainly on the direction in which the antenna is pointed. An antenna looking into interstellar space will have a lower effective temperature than one

more on next page





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4.001-4.500	Longitudinal Long transverse Short transverse	68 68 62	/ 58 58 54	6 5 2
4.501-5.000	Longitudinal Long transverse Short transverse	68 68 61	58 58 53	5 5 2
5.001-5.500	Longitudinal Long transverse Short transverse	67 67 60	58 58 53	4 4 2
5.501-6.000	Longitudinal Long transverse Short transverse	67 67 59	58 58 52	4 4 2



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SATELLITE COMMUNICATION

pointed toward the earth.

The factors involved in the choice of a modulation system best be shown by a comparison of AM and FM. How power equipment, it is modulation circuits needed the either AM or FM are roughly equal in size and weight. However, as the power level go up, the modulators for A systems become quite large while those for FM system remain relatively constant.

Power tubes are better suited to FM

The power tubes used 10-30 kmc are electronical tunable types (klystrons, bac ward wave oscillators, etc. These are quite well suited FM but not to AM. The san holds true for voltage-tunab magnetrons at 1-10 kmc.

In wide band modulatic systems, equipment operatinat the higher frequencies againas a decided advantage. This because electronically turnable tubes are modulated owwide bandwidths without an loss in power output.

One of the clear drawback of AM transmission is the superptibility of AM signals in noise. From the relation: $[(S/N)_0FM^{\dagger}/[(S/N)_0AM] = SF$ it is obvious that, in exchang for bandwith the occupance FM gives much better transmission than AM.* However we should keep in mind that when the FM signal is we accompared with the noise, wid band FM is inferior to a equivalent AM system.

Phase irregularities in wide band FM

Wide band FM systems als suffer from phase irregularitie in the transmission path. Wit multi-path transmission, selec tive fading can cause phas variations even in the micro wave region. Such variation can also occur when the side

*H. S. Black, "Modulation Theory Van Nostrand, 53.

Circle No. 18 on Reader Service Card in Product Review Section

ands shift unequally because Doppler effects. In either use the distortions can be serias enough to make the signal aintelligible.

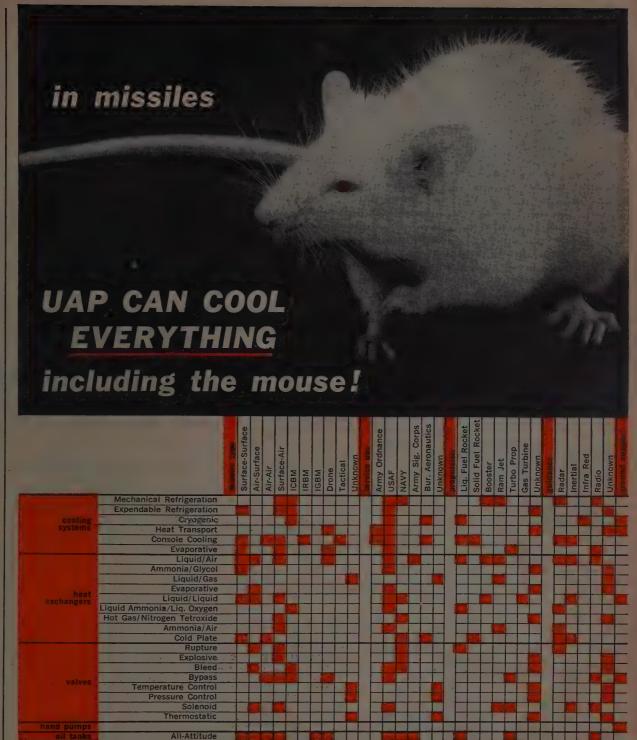
AM systems, on the other and, suffer little from selective fading and Doppler. All in the differences between AM and FM therefore are not as reat as is often believed.

Pulse modulation also should e considered by the designer f satellite communications. ienerally pulse modulation ystems are more efficient in uppressing noise than either M or FM systems. However, his advantage is gained only t the cost of more complex ransmitting gear.

Generally it's true that, as requency goes up, available ransmitter power goes down. But by itself this does not accurately account for propagation performance—it must be elated to such factors as ancenna gain and fading. There also certain major factors a equipment design that depend on frequency but not on propagation performance.

Probably the most imporant of these factors is effiiency of generation. Above 000 mc, this drops off very puckly. At 100-1000 mc it is cossible to get *overall* transniter efficiencies of 20-30 per ent, but at 20 kmc efficiencies of five per cent are seldom exeeded at the present state of the art.

This does not mean that he lower frequencies necesarily are the better ones. If andwidth, say, is a factor, hen high frequency equipnent certainly offers many dvantages. With voltageunable magnetrons and backard wave oscillators, bandvidths of around 5-10 kmc an be obtained at frequencies f 30 kmc. However, if either e power or the frequency tability of the generated sigal is a factor, then lower equency equipment offers better design basis, at least or the present.-End



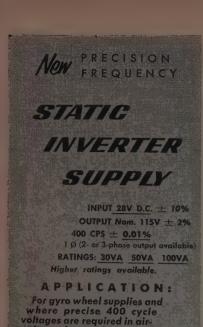
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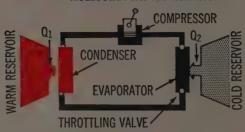
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Special Report

Space/Aero Electronics

MOLECULAR GAS REFRIGERATOR



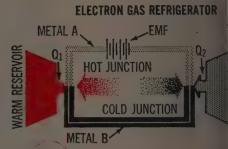


FIGURE 1: Comparison of refrigerators using molecular gas and electron gas. Q represents useful cooli

Wide use foreseen for

applied thermoelectricity

R&D engineers at Nortronics are turning up ways to relate the venerable Seebeck and Peltier effects to modern designs in such areas as environmental conditioning, solar power conversion, ion propulsion, and above all electronics cooling.

> by M. B. Grier, Project Engineer Nortronics Div., Northrop Aircraft, Inc.*

BY MEANS of applied thermoelectricity, it is possible to convert heat to electricity and electricity to heat or cold directly without moving parts. The scientific principle that accounts for this has been known for quite a long time. The conversion of heat to electricity, the so-called Seebeck effect, was discovered in 1822. The inverse Peltier effect, the conversion of electricity to heat or cold, was uncovered in 1834.

Jean Peltier, a French physicist, observed that, if a voltage

"Nortronics Div., Northrop Aircraft, Inc., Hawthorne, Calif.

is applied to a circuit consisting of two dissimilar metals, heat is absorbed at one junction of the circuit and emitted at the other. In 1911, Altenkirch and others applied this phenomenon to thermoelectric cooling. It soon became possible to draw an analogy between refrigerators using an electron gas (electrons have been considered a more or less ideal gas) and those using a molecular gas (Fig. 1).

In rocket vehicles, problems of aerodynamic heating and internal heat dissipation call for cooling systems of minimum weight and volume. In addition, greater reliability needed. Peltier refrigerato may well prove to be the a swer to these problems, since they have no moving parts ar can probably be designed to odd-shaped spaces as suppl mental coolers for hot spo and areas.

Peltier's phenomenon diffe from Joule heating, which produced by electric resistant in the circuit. Under the Peltis principle, the direction of the current flow determine whether heat is absorbed emitted at a given junction This can be shown in th following equation:

 $Q = IT_c(\alpha_1 - \alpha_2),$ where Q is amount of heatin or cooling; I, current flowin through the junction; Te, tem perature of the junction; an a_1 and a_2 , the thermoelec tric powers of the dissimila materials comprising the june tion. If we assume that hal of the I2R loss goes to the hot junction and the other hal

to the cold one, we can se

URE 2: Two-stage Peltier refrigtor used to increase efficiency infrared detectors (in this case PbS cell). Another version of s device (without envelope) is ly $\frac{1}{4}$ x $\frac{1}{4}$ x $1\frac{1}{2}$ in., says

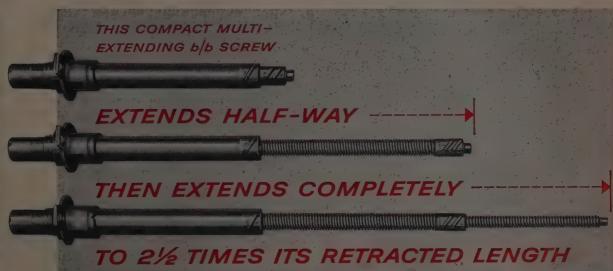
the lumped-parameter quation governing thermoectric cooling:

 $(I^2/2)(\rho_1L_1/A_1+\rho_2L_2/A_2)$ - $(K_1A_1/L_1+K_2A_2/L_2)(T_h-T_c)$ here subscripts 1 and 2 repesent negative and positive gs of the couple, respectively; resistivity; A, cross-sectional ea; L, length of a thermoouple element; K, thermal onductivity; T_h , temperature hot junction (in degrees elvin); and Q, useful cooling. In this equation, we find at the useful cooling at the old junction equals the Peltier poling less half the I²R loss nd the heat thermally conucted to it from the hot juncon. We may then calculate e maximum possible temper-

• cooling is proportional to more on page 49

ure drop as well as the curent required to produce it. lso, we can see that:

FAR-REACHING EXTENSION FROM LIMITED ACTUATOR SPACE



Now Saginaw supplies the answer to your most difficult actuator space problems with the Multi-Extending Saginaw Screw! Utilizing Saginaw's time-proved recirculating ball principle in multiple telescoping sections, the Multi-Extending b/b Screw conquers actuator space obstacles designers have been seeking to overcome for years! Here's why:

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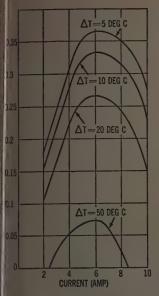


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ERMOELECTRICITY . . .



GURE 3: Useful cooling vs arrent for various temperature rops in a typical single-stage acro-refrigerator. Ingot is 0.89 in long and 0.09 cm² in area.

the difference in thermoelèccic power of the materials orming the junction, so that naterials with high thermolectric powers are needed for fficient cooling;

• high electric conductivity s required to minimize Joule neating; and low thermal conductivity is required to reduce the flow of heat from the hot to the cold junction.

Temperature drop lessens as load goes up

Most metals have low electric resistivity and also low thermoelectric power. In contrast, certain semiconductors, such as germanium and silition, have high thermoelectric power; but their low electric conductivity is a drawback. Furthermore, their crystal lattice structure contributes to high thermal conductivity.

The answer would seem to lie with intermetallic-compound semiconductors. In recent years, these have been

more on next page

SPECIAL REPORTS ON FINISHING NON-FERROUS METALS

NUMBER I—Decorative, Corrosion-Resistant Finishing with Iridite

Chromate conversion coatings are well known and accepted throughout industry as an economical means of providing corrosion protection, a decorative finish or a good paint base for non-ferrous metals. However, continued developments are so rapid and widespread that many manufacturers may not be completely aware of the breadth of application of this type of finish. Hence, this digest of current information; to bring you up to date on the many ways in which you can combine salable appearance with durability in one finish at a competitive price advantage. Report II on paint base, corrosion-resistant finishes and Report III on chemically polished, corrosion-resistant finishes are available on request.

First, as a basis for this discussion, a "decorative" finish is considered as any chromate film that is used as a final finish in itself. It may be truly decorative in that its sole purpose is to enhance the beauty of the product. For example, a bright chrome-like finish or a pleasing bronze appearance are among the many effects that can be obtained. It may be functionally decorative in that it reduces reflectivity for camouflage purposes or provides a means of color-coding parts. But, in all cases, the Iridite films protect the metal against corrosive attack.

'Iridite finishes are now available for all commercial forms of the more commonly used non-ferrous metals, including zinc, cadmium, aluminum, magnesium, silver, copper, brass and bronze. These films can produce a wide variety of pleasing appearances. The basic colors of the Iridite coatings are grouped below by metals.

ZINC and CADMIUM: Metallic bright, light iridescent, iridescent yellow, bronze, olive drab.

COPPER, BRASS, BRONZE: Metallic bright, yellow.

ALUMINUM ALLOYS: Clear, iridescent yellow, brown.

MAGNESIUM ALLOYS: Light brown, dark brown, black.

SILVER: Metallic bright.

. In addition, many films can be modified by bleaching or by dyeing. Among the dye colors available are various shades of red, yellow, green, blue or black.

Depending upon the metal and the Iridite used, corrosion resistance of clear and bright films ranges from mild passivity to as high as 500 hours in salt-spray; on heavier dark films, salt-spray resistance ranges from approximately 100 to 1000 hours.

.It is this combination of decorative and corrosion resistant properties that accounts for the widening use of Iridite finishes. For example, Iridites #4-73 and #4-75 (Cast-Zinc-Brite) make possible for the first time, a combination of lustrous chemical polishing of the as-cast surface of zinc die castings and good resistance to corrosion. Further, in many cases,

WHAT IS IRIDITE?

Briefly, Iridite is the tradename for a specialized line of chromate conversion finishes. They are generally applied by dip, some by brush or spray, at or near room temperature, with automatic equipment or manual finishing facilities. During application, a chemical reaction occurs that produces a thin (.00002" max.) gel-like, complex chromate film of a non-porous nature on the surface of the metal. This film is an integral part of the metal itself, thus cannot flake, chip or peel. No special equipment, exhaust systems or specially trained personnel are required.

sizeable savings in the cost of buffing and electroplating are realized

On many steel parts, a simple system of zinc or cadmium plate and bright Iridite is used instead of more costly electroplated finishes to provide a bright, decorative and protective finish with tremendous savings in material, equipment and labor.

In finishing aluminum, where corrosion resistance or paint adherence is the prime consideration, the aircraft industry has all but abandoned the anodizing process in favor of recently developed chromate conversion coatings, among them Iridite #14 and #14-2 (Al-Coat). These formulations and their method of application can be varied to retain the original metallic appearance while providing acceptable corrosion resistance, or to produce a fully colored brown finish that offers exceptional corrosion protection. Again, time and manpower savings are astounding—one company saved at least \$15,000 a year on maintenance of racks alone and another \$40,000 on materials and labor in only nine months. In addition, of course, hundreds of thousands of dollars are saved by eliminating the need for expenditures for generators, heating equipment and racks.

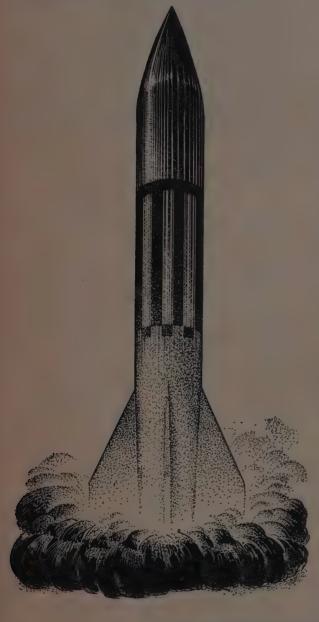
Iridites are widely approved under both Armed Services and industrial specifications because of performance, low cost and savings of materials and equipment.

In planning or designing, you should consider the many other characteristics of Iridite finishes which may enter into the specific problem. In addition to having decorative and protective functions, these chromate coatings form an excellent base for organic finishes and bonding compounds. They have low electrical resistance. Some can be soldered and welded. The Iridite film itself does not affect the dimensional stability of close tolerance parts.

You can see then, that with the many factors to be considered, selection of the Iridite best suited to your product requires the services of a specialist. That's why Allied maintains a staff of competent Field Engineers—to help you select the Iridite to make your installation most efficient in improving the quality of your product. You'll find your Allied Field Engineer listed under "Plating Supplies" in your classified telephone book. Or, write direct and tell us your problem. Complete literature and data, as well as sample part processing, is available. Allied Research Products, Inc., 4004-06 E. Monument Street, Baltimore 5, Maryland.

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Armco 17-7 PH and 17-4 PH—Grades widely used in missiles and aircraft. Provide high strength to 900 F and excellent fabricating characteristics, 17-7 PH produced in all forms, 17-4 PH in bar, wire and billets.

Armco 17-10 P—Non-magnetic precipitation-hardening grade with high mechanical properties. Available in bar, wire and forging billets, 17-10 P is useful for instrument, guidance and accessory parts requiring this unique combination of characteristics.

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THERMOELECTRICITY . . .

produced with high the emf and moderately low etrical resistivity. Good amples are Bi₂Te₃ and al of ZnSb and BiSb.

For a given set of mate parameters, the achieve temperature drop decrewith increasing load. In case sown in *Figure 3*, coing is 73 mw for a 50-ded drop at about six ampewhile for a 10-deg C drop the same current it is 330 materials.

To study Peltier cooling pabilities, Nortronics built single-stage electronic referator (Fig. 4). We found the amount of cooling value with the temperature range well as with the degree of timal isolation. Insulated cotton and cooled with a cotton and cooled a small tray water sufficiently to product block of ice within a few mutes.

Adequate cooling is part larly important for passive diation detectors (such as photosensors), whose sensity is far greater at very temperatures (about — deg C) than at ambient. forts in this area have yiel at least two miniature mechical cryostats — refrigera based on the Joule-Thomp effect. These are said achieve temperatures as low—195 deg C.

Moisture and impurities cause trouble

However, these relative expensive miniature deviare troublesome when it conto purging moisture and purities from the system. We is really needed to solve cooling problems for pass radiation detectors is a compact refrigerator that is in grally related to the detective of and can operate in pendently of external referant sources.

In experiments with a track stage Peltier refrigerator, have achieved a total temps

more on pag
Circle No. 24 on Reader-Service Ce



and now *Duralar** joins this famous family of fine pencils

pencil specifically designed for work on matteacing film of Mylar®, DURALAR is the newest in ete line of fine MARS drafting products. All are from West Germany and made to meet the highscional standards. Below • Bright-hued LUMOincils in 24 colors for color-coded drafting and reproduction • LUMOGRAPH graphite drawing 19 degrees; some degrees available with eras-

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FIGURE 4: Single-stage Peltie refrigerator devised at Nortronics. The legs of the couple are about three mm square.

ture drop of about 50 deg C starting from a hot junction temperature of 20 deg C. This is enough cooling power to in crease the useful sensitivity of such common photoconductive detectors as the lead sulfide (PbS) cell.

Figure 2 shows our photo sensor - refrigerator mounted within an evacuated aluminized envelope. The refrigerator-cel assembly fits on thermally mas sive copper support posts in serted in an insulating support disk. The supports have cooling fins and lugs to receivleads. The photoconductor (in this case a PbS cell) is fastened to the cold junction. The photoconductor leads are brough up and fed through the base

Driving emf for the device is about 250 mv, producing current of about 10 amp (2... w). Ample power for intermittent operation is supplied by zinc-silver oxide secondary celebrat furnishes about one volwith very high current level for long operating periods. For continuous operation, you'have to add a silicon rectifier

Direct conversion of nuclear energy

As we have noted, the ther moelectric refrigerator characteristically produces a sharpe temperature drop as the temperature of the hot junction of the cascade increases. Typically, a prototype unit that gave 50-deg C cooling with the hot junction at 20 deg C yielded only 30-deg C cooling.



Includes samples and descriptions of . . .

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clog pressure lines

You don't need high pressure

Electric power requirements are small

Few Peltier stages are needed for intermedite cooling

Cooling reduces cell heating at high Mach number

ith the hot junction at -20

Among the more promising plications for thermoelectric ower generation is the direct onversion of nuclear energy electricity without intermeate moving parts. As the engy source we might choose a eat-producing isotope (such plutonium 210) that emits sily shielded radiation in the rm of beta or alpha particles. ith the thermoelectric mateal shielded from harmful diation, the heat energy ould be allowed to pass.

An alternate method would e to use a radioactive source ombined with a thermoelecic material that retains useful nermoelectric characteristics hen transmuted. Obviously, e could use the more conentional reactor heat transfer onfigurations to minimize the ansmutation of the material y radioactivity. This would, f course, involve a pressurized as or fluid heat exchanger

Work at Nortronics has hown a thermoelectric solar onverter can be built that is omparable in efficiency—8-4 per cent—with the silicon olar cell under free space onditions. Using concentrators vith this device, you could get ven greater efficiency. Such a evice may be expected in the ear future.

For ion propulsion, thermolectric devices may be coupled with nuclear reactors to more on next page



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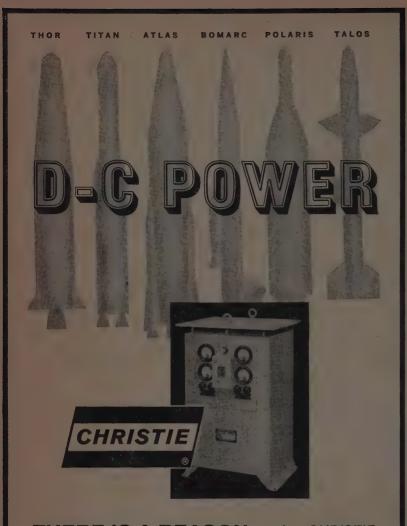
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produce an electric accelerating field. Also, solar energy converters will be a source of standby and auxiliary power for the larger space vehicles and may become the main power source for smaller ones.

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TYPICAL installation of spiral, scimitar, and exponential loop antenn

Spiral antennas meet needs of new vehicles

The first real breakthrough in antenna design for high performance aircraft and missiles came in 1956 with the application of the equiangular concept to the Archimedes spiral. Many advanced designs have since followed. The more important of them are reviewed in this article.

by Edwin M. Turner

Aerial Reconnaissance Laboratory, Wright Air Development Center*

N 1954, a bandwidth of one octave, or a frequency ratio of 2, was considered excellent. A frequency ratio of 3 was just about the ultimate. Today, we can produce efficient antennas with frequency ratios of 20 or more.

Notable among the new ar tennas are the "scimitar," th "valentine" (derived from the equiangular spiral ar tenna), and the many form of logarithmic periodic struc tures. Recently, both spira types and periodic structure have been made non-plana: This gives them a unidirec tional pattern with constar impedance over very larg bandwidths with or withou cavity backing.

The logarithmic periodi structures are especially inter esting, because they afford such great flexibility in con trolling polarization. Their ef ficiencies are also well up ove 95 per cent (provided good conducting materials and low loss dielectrics are used).

Until recently, the practice has been to build antenna with fixed patterns and polar izations. Since it is possible to null any fixed-polarization antenna, this type is no longe

PRINCIPAL CURRENT ANTENNA TECHNIQUES

9 MC

ed wire, trailing wire, tail & g caps, airframe, pods, horirtal & vertical stabilizer

59 MC

ips, tuned stubs, fixed wire, rite loops, pods, scimitars reraft structure), valentines reraft structure), spirals, logamic periodic structures, conl spirals

499 MC

ibs, biconical, partial sleeve, ploaded stub, helix (above of mc), plastic rods, pods, rels, scimitars (aircraft structes), valentines (aircraft structes), logarithmic periodic rectures, conical spirals

0 MC-19 KMC

ubs, helix, flat biconical, slotd waveguides, top-loaded stub, rns, traveling wave slot, cones, rabolas, Luneberg lens, printarrays, spirals, scimitars, valtine, logarithmic periodic strucres, conical spirals

39 KMC

ens, slotted helix, horns, slotted aveguides, printed arrays, spils, scimitars, logarithmic perilic structures, conical spirals

KMC & Up

inted arrays, lens, optical techques

dequate for many countereasure purposes. Designers by want to be able to shift the pattern and polarization most at will. Developments ong these lines show considable promise.

Basically, there are two pes of spiral antennas in the: (1) the Archimedes biral, for which R equals ae and (2) the equiangular spiral, r which R equals $e^{a\theta}$, where is the radius vector from the rigin to a point on the curve; the angle or rotation (in dians); a, a constant dening the conductor's rate of the spansion, and e the constant e^{aa}

.more on next page



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E. E.

FEEDING the two arms of a coplanar spiral antenna 180 deg out of phase produces a circularly polarized conical beam (left) about 80 deg wide in both planes. If the arms are fed in phase, a stub-type pattern is produced (right) with a null on the axis.

These antennas can be built as either single or double spirals, and can be fed either balanced or unbalanced with co-ax or waveguide. With the unbalanced feed the beam will tilt an average of about 10 deg from its axis.

Spiral antennas are unique in having a self-adjusting aperture that accounts for their constant beamwidth. The electric phenomenon responsible for this beam compensation is different for the two types of spirals; however, the far-field effect are similar.

I' per frequency limits for both antennas are determined the construction accuracy of the feed points. The lower limit is set by the circumference of the outer turns of the antenna.

If circular polarization is wanted, this lower limit has to be about one wavelength at the lowest frequency. Considerably smaller sizes can be used if circular polarization is not needed.

Coplanar antennas are particularly well suited to printed circuit techniques and can be

The OVER-RUNNING CLUTCH Wit



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The Formsprag clutch consists of a fucomplement of shaped sprags, wedges, located between concentrinner and outer races. Power is tranmitted from one race to the other kthe wedging action of the sprags. Eas sprag is so shaped that dimension Ais greater than BB. Rotation of or race in the "driving" direction cause the sprags to wedge, transmitting torquin full from one race to the other.

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Formsprag offers a size and model clutch for every over-running, indexing

or back-stopping application. Standard models are available in sleeve bearing, ball bearing, miniature and large bore backstop types. All are illustrated in the new Formsprag catalog, complete with specifications, performance data and design ideas. Formsprag engineers also offer a complete design service on special clutches, including prototype production facilities. Write for the complete Formsprag catalog, or send details of your application for a consultation with the Formsprag Engineering Department.

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World's Largest Exclusive Manufacturer of Over-running Clutches



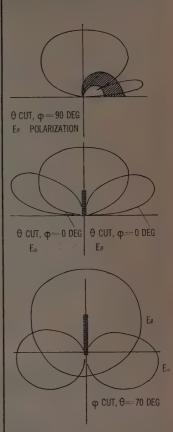
a expanding coil spring keeps the rags in light contact with both inner douter races. There is thus no lost point, the driving torque being interaction, the driving torque being interaction to the formsprag Clutch is so signed that it will transmit a greater rque in relation to its size and weight, an any other comparable type of atch... specify Formsprag on overnning, back-stopping and indexing plications.

Forcing a ball or roller into a curved, wedged space is an old over-running clutch principle. The sprag is, in effect, a "roller" of increased diameter with greater contact surface in a given annular space. Formsprag Clutches engage at constantly changing contact points. Clutch life is prolonged and backlash eliminated. Also, with the inclined sur-faces discarded, more sprags can be inserted to increase torque capacity.



Specify Rawson Automatic Centrifugal Clutches and Clutch Couplings. They protect the prime mover, eliminate need for costly reduced voltage starting equipment, permit use of less expensive smaller motors to start high inertia loads. Completely automatic and entirely mechanical in operation. Rawson clutches provide full overload protection, never need adjustment. Write for the new Rawson Clutch Catalog now.

SPIRAL ANTENNAS . . .



TYPICAL patterns of co-planar scimitar antennas.

made cheaply. They inherently lend themselves to flush mount-

Used with a cavity, spiral antennas become unidirectional, with a gain of three decibels over that of free-space units. However, the cavity reflections do restrict the pattern bandwidth. With conventional cavities, bandwidths of about 3:1 are practical.

The inherent impedance of the spiral antenna is relatively constant. It can be adjusted to almost any nominal value between 25 and 200 ohms by varying such parameters as the thickness of the conductors. the taper rate of the conductors, the spacing of the cavity backing, etc.

Non-planar spirals have patterns and impedance characteristics similar to those of the

more on next page



coplanar types, except that the patterns are always unidirectional. When the spiral is projected onto a con with a taper of 20 deg or less, the front to back ratio exceeds 25 db.

A non-planar spiral is stable over at least a 20:1 frequency ratio, with very good circularity. It can be built either with a conventional co-ax line or with an exponentially tapered flat conductor. (With the conductor, it can be made about 25 per cent smaller but is somewhat harder to build especially in small quantities.) The antenna feed must be balanced, however, to prevent beam tilt.

The non-planar conical spiral is relatively insensitive to a cavity environment provided that the apex of the cone is pointed in the direction of radiation.

The scimitar is a derivative



LOGARITHMIC periodic tures have essentially cor impedance and highly eff patterns over band-width great as 20:1.

of the spiral antenna. basically a folded monwhose conductor expand ponentially.

This antenna presen small narrow cross-sect area to the air stream. I great mechanical strength ing large at the stress pe and is not isolated from ground plane.

The curve for the ou dimensions of the antenna given by:

 $R = ke^{a\theta}$.

56 details of stainless steel jet engine component brazed in three steps with no distortion or oxidation by Wall Colmonoy . . .



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Design and Manufacture of Electro-Mechanical Timing Devices

there k is the scaling factor and a a constant denoting the steep of expansion of the outdeend inside curves of the similar. The inside curve is efined by the same equation, acept that the value of a is maller. The origin of the street is displaced from the sed point.

The efficiency of the scimar antennas is well above 5 per cent over their entire seful band. Bandwidth is sually limited to a 4:1 freuency ratio, because of patern shifts. However, if the attern requirements are not citical, it is easy to get a requency ratio of 10:1.

The lower cutoff frequency usually determined by the ength of the inside conductor. It must be one quarter wavenigth or greater. The upper nd of the band is limited by the co-ax or waveguide feed the cutoff.

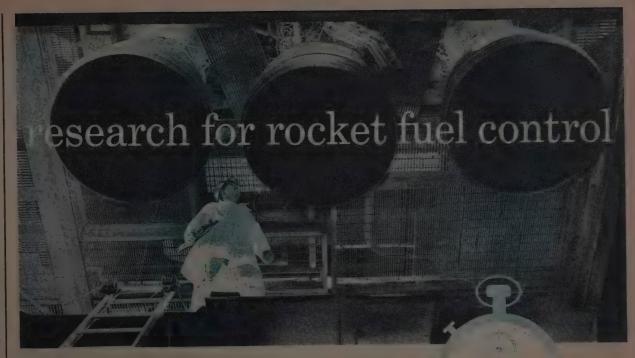
This type of antenna is asily fed by either co-ax or vaveguide. It is also relatively asensitive to the ground plane.

lost butterfly scimitars inearly polarized

The coplanar scimitar is inearly polarized, but the prientation is different for all points in space. The butterfly cimitar is mostly linearly polarized (with a conical beam perpendicular to the ground plane). The rosebud scimitar has a very broad hemispherical peam that is linearly polarized that is linearly polarized hear the ground plane and becomes circularly polarized as one approaches the axis perpendicular to the ground plane.

Similar to the scimitar are the exponential loop and valentine antennas, which are essentially two scimitars back to back. Their impedance characteristics are almost identical with those of the scimitar for comparable sizes, and they are just as admirably suited for integration into an airframe structure.

more on next page



Pumps, turbines, cryogenic hardware and fuel systems—everything short of firing the rocket—will soon be tested by a CEC process control system. Working with dangerous propellants and limited time, the system will provide complete, automatic programming, rapid control, and data in 30 seconds. Write for the complete story in Bulletin CEC 3015-X10.

Systems Division

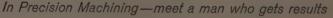
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Tom operates a precision jig borer in this picture, but he is an expert at every machine in the plant. Earlier in his career with us, he won the Minnesota Employer's Association award as the state's Outstanding Apprentice of the Year. He is active in community affairs, participates in many company activities. Tom Nelson—another of the many good reasons our customers say, "At General Mills, we get results."

MECHANICAL DIVISION

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Its." To wider worlds—through intensive research • creative engineering • precision manufacturing
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General Mills

If the antennas are sm (compared with a wa length), they act essentially transformers that excite entire structure around the If they are large, they ha a fat lobe whose center oriented in the direction the feed points.

These antennas are wide used as feed structures is secondary apertures. They come be fed unbalanced either from a co-ax line by a so-called finite balun or parasitical from a waveguide.

Logarithmic periodic strutures are fast becoming one the most useful antenna d signs. They have essential constant impedance and high stable patterns over ban widths as great as 20:1. T patterns are bidirectional whethe antennas are coplana with non-planar construction they have a front-to-back rat of about 12 db.

Relatively insensitive to cavity environment

The non-planar antennifire off the nose and are hor zontally polarized when twelements are stacked vertical one above the other in a conplementary manner. They can have any polarization—from linear to either right or leelliptical or circular—whe stacked over a rectangula cone. Four or more units musbe used, with suitable phase shifters or switches betwee them.

These antennas can hav many form factors so long a Babinet's principle on complementary structures is properly observed. They are bal anced but can be fed from co-ax line with an infinite balun.

Recent work at the University of Illinois indicates that these antennas are relatively insensitive to a cavity environment. The bandwidth that can be obtained from these antennas when they are mounted within a cavity is about two octaves.—End



Accepts varying input speeds from 3800-7000 RPM.

2 —GOVERNOR

Maintains $\pm 0.25\%$ steady state speed over 200° F temp. range, $\pm 0.5\%$ for 400° F temp. range and 0.5 second maximum full load transient recovery time.

3 -VARIABLE DISPLACEMENT PUMP/MOTOR

Either pump or motor, depending on whether it is adding to or subtracting from drive speed.

4 —OVERSPEED GOVERNOR

- Senses a predetermined excessive speed to prevent drive overspeeding.

5 -MAIN SYSTEM RELIEF VALVE

Overload protection for hydraulic system.

6 —DIFFERENTIAL GEAR SECTION

Ring, sun and planet gears. Also includes take-off gears for governors, scavenge and make-up pumps.

7 —FIXED DISPLACEMENT MOTOR/PUMP

Drives, or is driven by, sun gear to control output speed at 6000 rpm in response to fluid flow exchange with variable delivery unit.

Differential Type Constant Speed Drive For Aircraft Alternators

Tests now in progress on this 60 kva Vickers Differential Type Constant Speed Drive are proving its outstanding performance. (See curves) It combines Vickers dependable hydraulic pumps and motors with planetary gearing to provide a more efficient and lighter "package" than drives now available.

The differential type constant speed drive is a direct engine-driven transmission that can be either engine pad mounted or universal shaft driven. It employs the "differential drive" principle wherein the planetary gear train is the principal power vehicle and the hydraulic components differentially add or subtract speed and power to maintain constant output speed. For further information, write for Bulletin A-5221.

Performance of Vickers 60 kva Differential Type Constant Speed Drive at three different loads. The high efficiency requires minimum support equipment for cooling with minimum penalty for high temperature operation (400°F oil temp.)

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Special Report

Space/Aero Electronics



BLACK BOXES in the nose of this interceptor contain navcom fire control gear. Common design mistake is to overbeef structu members to take care of mechanical "unknowns".

Mathematical analysis for minimum weight and space envelope, top reliability

Airborne electronic equipment all too often is the victim of overdesign-because no careful computation was made of the static and dynamic loads to be expected in service. For efficient design and top reliability, a thorough analysis of shears. bending moments, tions, etc., is a must.

by David Ehrenpreis

Consulting Engineer

N THE mechanical design of airborne electronic and electromechanical gear, the margin of safety is a prime factor. It can make the difference between minimum and exce weight and geometric envelo and between reliability and to

The margin of safety

MS = F/f - 1,

where F is the allowable stre and f the actual stress. It's o vious that probability of fa ure is a function of MS. MS is negative, the mechanic or structural member w probably fail. If it is a lar positive number, the equi ment will most likely be ove weight and/or oversized.

Obviously, for efficient m chanical design, an optimu balance has to be struck b tween reliability and size as weight.

Damping Analysis

ing the procedure outlined in article, an analysis of a loal missile electronic equipintervealed these interesting

the use of a sandwich-type struction.

Increased damping is obtained permitting the structure to ate elastically and deform at builture injects.

Transmissibility at resonance, the case of most electronic uipment, decreases as the naal frequency of the structural

Transmissibility at resonance electronic equipment (expt on gimbaled structures) creases with increased amtude input at resonance ex-

Allowable stresses for calcuting MS can be found in pubhed data.* Actual stresses ust be found by a rigorous athematical analysis of the manic properties of the sysm and its environments.

Briefly, this analysis should etermine such parameters as e required wall thickness of ritical members, gimbals, astings, extrusions, and builtformed members with the esired margins of safety gainst fatigue, shear, or peranent deformation. It should so determine the best vibraon and shock isolation sysm, if a suspension system needed. As the analysis roceeds, it will become obvius to the designer just where e can "shave off" weight and ulk without compromising reability. Just such a procedure as been worked out and sucessfully used to find the dyamic properties and minimum eight and envelope of airorne electronic and electro-

lowable stresses for most structurierials (aluminum, steels, and majums) are found in ANC-5 and for dwich construction in airborne equipit racks and modules in ANC-23.

echanic equipment.

more on next page



Radome designed and built by Long Sault Woodcraft Limited, St. Andrews East, Quebec, for the United States Air Force RADC. Looking upward from the inside of the world's largest stressed skin sandwich radome built of translucent fiberglass panels, securely joined by hundreds of DUAL-LOCK fasteners.

Radar antennae along the upper perimeter of North America's defense system are enclosed by protective domes which stop ice, snow, and gales up to 150 mph.

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DUAL-LOCK is ideally adapted to panel fastening for military shelters, demountable shipping containers, aircraft cowlings and guided missiles.

Features:

- High load characteristics. The standard No. 1 DUAL-LOCK withstands 2500-lb. tension, and with modifications, tension loads of 7000 lbs. and over.
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- Positive-locking, Trigger action insures fully open and fully closed positions.
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The regime is defined as the t of all dynamic matrices, re sion equations and distribution of masses and inertias nee to find the dynamic propertie the electronic (or electromechical) equipment caused by excition in one axis.

The recursion equations de the deflection, slope, shear moment from substation to s station along one subsystem

A substation is a critical pealong a subsystem of the equipment.

A subsystem is a primi structural or mechanical mem of the electronic equipment.

In a typical case, it might to some 60 dynamic equations define a single regime. The equations define, within the bounds of the subsystem, ternal rotation (F), internal sering (H), internal shear (K), ternal moment (K'), external moment (M), external shear external deflection (y) and ternal rotation (a).

For example, the equation the externally applied deflect (y_a) at point "a", one of bounds of the subsystem i, $y_a = \alpha_b H_a{}^{ab} + Q_b H_a{}^{Qb} + y_b H$ Here H_a is the internal resist deflection at "a"; $H_a{}^{ab}$, the ternal deflection at "a" cauby a unit applied rotation

*For typical dynamic equations in tregimes, see D. Ehrenpreis, "Ar sis and Theoretical Investigation of Military Electronic Missile and Airc Borne Equipment;" IRE National vention, March '58.

The first step in the produce is to set up a mathen ical model of the system unstudy. The model consists three mathematical regindefining for each of the themutually perpendicular at the dynamic forcing funct

Each of the three regime made up of the defining resion dynamic equations duced to linear algebfirst-order, first degree for dynamic matrices, and the cusion tables defining inputs, responses, natural quencies, damping characisites, deflections, rotati



gime*

a second bound of subsys-I; a,H,ab, the internal deion at "a" caused by a unit tion at "b" times the true tion at "b", or the true innal deflection at "a" caused the true rotation at "b"; H_aqb, the true internal deflecat "a" caused by the true ar at "b"; and y_b H_ayb, the e internal deflection at "a" sed by the true shear at "b". The equation for the externalapplied bending moment (Ma) 'a" is similar:

 $-M_a = O = a_b K'_a a^b +$

 $Q_b K'_a^{ab} + y_b K'_a^{rb}$. The $e_b K'_a^{ab}$ is the true internal ment at "a" caused by the e rotation at "b"; $Q_b K'_a Q_b$, the internal moment at "a" sed by the true shear at "b"; y_b y_b , the true internalment at "a" caused by the deflection at "b".

hroughout the regime equas, the quantities y, a, Q and ppear as non-zero coefficients the dynamic matrices. H, F, and K' are polynomials in ns of ω , the natural frequen-Substations along the equipnt's subsystems are analyti-y linked to each other by rsion equations.

is a simple matter to plot r function vs assumed natufrequency on a digital comer. Where the error crosses x-axis, the electronic equipt has a fundamental resot frequency or asymptote.

rnal and external shears, rnal and external bending nents, torsional shears and ments, thrusts, and elasy deformations at all crisubstations along each

lach potential design change ears as an "open param-" and can be "plugged " the matrices, recursion ations, or recursion tables the mathematical model. solution then reveals effects of the parameter nges on reliability and dynic performance.

more on next page



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Typical open param now might be material th ness, supporting technic stiffness, and damping. L

it is expected, it will be

sible to plug in cost estin

schedule time, fabricab

Some Ideas for your file of practical information on drafting and reproduction



One of the ways to judge a skilled craftsman is by the tools he uses. They're invariably the best he can find - chosen to lighten his work, sharpen his skills. And, if the craftsman is a draftsman, they are, more often than not, products of K&E.

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The most remarkable feature of this compass is the speed and ease with which you can change settings-from diameters of 12 inches to 1/16 inch. With one hand, you can increase or decrease radii instantly and exactly. To go from small to larger radius, just press a spring release, and the legs will leg pencil compass, and the N1070 combination with interchangeable pen and pencil inserts. Both come with a box containing leads and spare needles. And with the N1070, a pen handle is provided for the pen insert which permits its use as a ruling pen. The compass can also be used as a divider by substituting one of the spare needle points for the lead in the pen-



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K&E Marathon Long Line and Wide Line Ruling Pens (1092) hold an extra large ink supply - draw lines up to eight times longer than ordinary ruling pens. And because they are pre-set, line widths are always uniform, easy to match with complete accuracy. Ink flow is regular and even, lines are always sharp and clean

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expand automatically. Stop approximately where you want, and make precise adjustments with a micrometer screw. To go from large to small, simply squeeze the legs of the compass together, then adjust

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serviceability, and availab as open parameters. The forcing functions military aircraft load fa determination include tak landing, catapult, flight neuver, gust, and buffe Specs for missile and speed aircraft attempt to fine peak probable load ditions as envelopes of a eration and plots of ampli

vs frequency of excitation. Mil-E-5270, Mil-E-5 and special airborne electro equipment specs give the quired test procedures for onance search, cycling, resonance dwell phases. ditional test procedures inc sudden impulse shock and tained acceleration tests.

The most difficult phase the test is the dwell at e resonant point frequency each axis. At resonance, high amplification of m structural frames of electro equipment cause perman deformations and fatigue 1 ures.

High Mach missiles exp ence vibration load inputs high frequencies. Transpo. tion and shipping loads electronic equipment are I shock loads and low freque sustained vibration loads. G fire loads are high intens low frequency steady-state bratory loads, with rag shock pulses added. Flight neuver loads are gener treated as sustained accel tions. Takeoff loads are sidered by many as rand vibrations and are often si lated in tests as a cycling a cedure from five to 1500 cm varying acceleration and plitude.

The actual stress is de mined from the analysis each equipment member.

When rotation and def

affect performance, marof safety must be found the members that might be osed to these effects. The lysis determines—for each ical substation of every subem-orders of magnitude deflection, rotation stress, ding moment, thrust, shear, ional moment, and allowbuckling. Also, it deteres natural frequencies, renses at inputs, and damping. The analysis is used at three es of the electronic or elecnechanical equipment's de-

1) Preliminary Design—A sory potential and kinetic rgy relationship analysis be made to find orders of enitude of critical deflecs, rotations, stress, natural uencies, and responses at cal substations in the nework. The study should ermine optimum geometric pes of every structural and chanical member for sevdifferent layouts of alter-

ve designs.

2) Final Design and De-Status-A rigorous, threeme analytic approach using igital computer is recomided. The precise actual sses in combined bending, ion, shear, thrust, and kling as well as the allowstresses in fatigue and kling must be analyzed and pared to determine marof safety. All critical les and their harmonics uld be investigated to make no members are overght and all members have proper MS against failure ermanent deformation.

3) Failure To Meet Specs f the equipment fails to t the dynamic and stress lysis requirements of Mil-272A, Mil-E-5400, or Mil-422, this analysis is opum. It may be applied to ermine quick fixes at critpoints to add damping at nance or revise the critical les of natural frequency reduce deflections, rota-



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When you send out bids for a bridge, a power station or a consumer product, you demand rigid adherence to your specifications. Why not with your drawing pencil, the most important working tool of your profession? Here is our set of specs — you may have your own ideas. See if we agree.

- 1. SMOOTHNESS-The pencil lead must be absolutely free of grit and hard spots. It must glide across the paper like an Olympic skater on ice.
- 2. STRENGTH-The pencil lead must be strong, without brittleness or splintering, and able to take needle-point sharpness in all drafting degrees, yet withstand firm drawing pressure.
- 3. UNIFORMITY—The pencil lead must have a steadfast consistency of hardness in each degree, pencil after pencil, lead after lead—dependable grading in a full range of 20 degrees, $\overline{88}$ to $\overline{10H}$.
- 4. BLACKNESS—The pencil lead must have pure graphite-saturated consistency, assuring positive opacity. It must not be treated with foreign oily

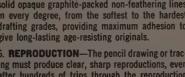
substances to give the illusion of black. Natural graphite blackness eliminates greasiness and reduces stains and smudges to the absolute mini-mum. Erasing must be quick and easy, and leave

- 5. DENSITY—The pencil lead must lay down solid opaque graphite-packed non-feathering lines in every degree, from the softest to the hardest drafting grades, providing maximum adhesion to give long-lasting age-resisting originals.
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design digest

Parami scores missiles in real time

PARAMI (Parsons Active Ring-Around Miss Indicator), produced by The Ralph M. Parsons Co., 151 S. De Lacey Ave., Pasadena, Calif., is an electronic miss distance indicator that gives you a real-time digital record of miss distances from 10 to 3000 ft range. It links target and missile transponders through wide band, pulse-modulated distanced data on two UHF carriers. The stronger target transponder telemeters data to the ground station. Fast ground-to-target calibration sharpens system accuracy before the test run. Among the missiles equipped with Parami are Nike-Ajax and -Hercules, Falcon, Genie, and Sidewinder.

by Bernard Kovit, Assistant Electronics Editor

RINGING FREQUENCY METHOD - Target transponder transmits a signal pulse that is received in missile on carrier frequency f1 (252 mc) and retransmitted on carrier f2 (291 mc). The signal is picked up on f2 by the target receiver and retransmitted on f1. Information frequency appears as amplitude modulation on each stable carrier. With enough gain in the system, the signal passes around the loop in a regenerative manner. Ringing frequency depends on the fixed electronic delay (t_o) in the equipment and the variable delay of the radio gap between the transponders. The latter is the information we want to get-as precisely as possible. The formula used to obtain this value is:

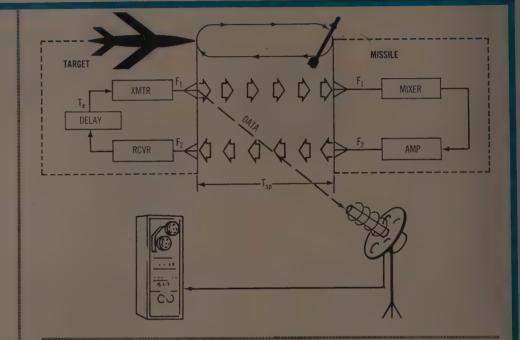
$$f_{data} = \frac{I}{t_o + 2R/c}$$

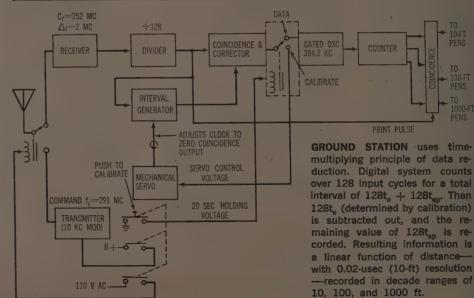
where 2R represents the round-trip distance between the transponders and c the speed of

The ground station is made up of a fivefoot-high standard electronics rack and a fourfoot-high helical antenna. A simple computer subtracts the two periods of electronic delay and counts in tens of feet exactly half of the round-trip radio delay $(t_{\rm sp})$ giving us the miss distance we're looking for. The recorder tape then shows the radial separation of missile and target to 4990 ft in 10-ft increments.

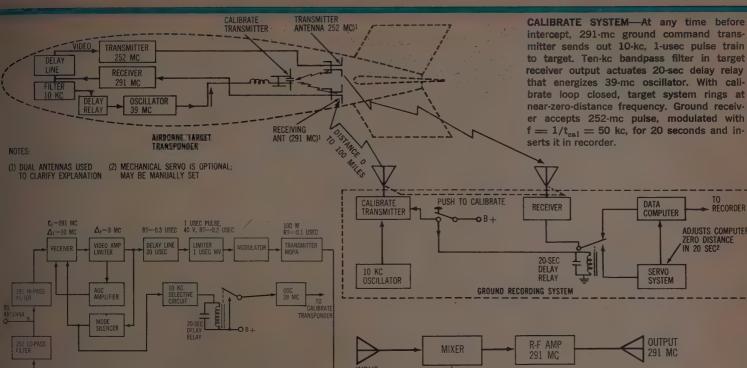
Parami Spec Summary

- Operating range: 10-3000 ft
- Operating frequencies: 252 and 291 mc
- System accuracy: 5-10 ft miss distance error
- Relative velocity: over 5000 fps
- Range of missile sizes: .50 cal bullet to large guided
- Power output: target transmitter, 100W; peak; missile transponder, 0.1 W (3000-ft version)
- Antenna beamwidth: about 40 deg
- Form of data: tens of feet, real time; time to get minimum miss disance and interpret record should not exceed one minute, says Parsons
- Data sampling rate: 195.3 samples per sec at 10 ft; 149.6 samples per sec at 3000 ft



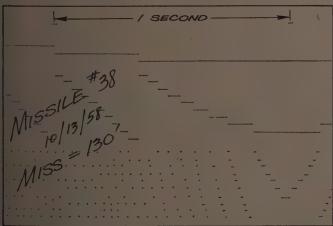


PRINT PULSE



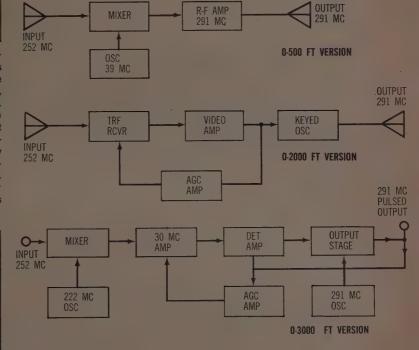
▲ TARGET TRANSPONDER ---Basic components are wide band receiver, delay line, and transmitter. Receiver is six-stage grounded-grid, TRF type. It is delaystable ±0.008 usec from 20 uv to 0.1 V over a 4-mc band centered at 291 mc. AGC handles intercept speeds up to 3000 fps. By including a simple calibrate transponder mounted within two feet of the main unit, Parsons makes it possible to calibrate out any delay variables.

MISSILE TRANSPONDER—To ▶ fit wide range of missile sizes (from smaller than Mighty Mouse to bigger than Nike-Hercules), Parsons uses three basic types. Simplest is a low gain (20 db) unit for short ranges (0-500-ft version). Most complex is 0-3000-ft unit with delay stability of ±0.01 usec. Its circuitry resembles that of a typical beacon. The third transponder is used for smaller, higher speed missiles than the long range unit.



INTERCEPT RECORD—Facsimile was made from an actual Parami flight record. Notations "Missile #38 . . ." were made by ground station operator. Using a scale

to identify traces, operator reads off distance of closest approach. Missile's speed can be found by converting distance traversed in one second to rate.



Frequency MDI vs Other Types

	CW Doppler	Phase Comparison	Frequency
Accuracy	5-10 per cent	about λ _m /100	5 ft
Ambiguity	none in the second	± λ _m /2	none
Missile equipment	several tubes	several tubes	several tubes
Telemetering needed	precise phase frequency	precise phase	none
Multipath rejection	none '	depends on bandwidth	depends on bandwidth
Data reduction	usually several hours .	real time	real time

TO RECORDER

ADJUSTS COMPUTER ZERO DISTANCE IN 20 SEC²

SERVO SYSTEM

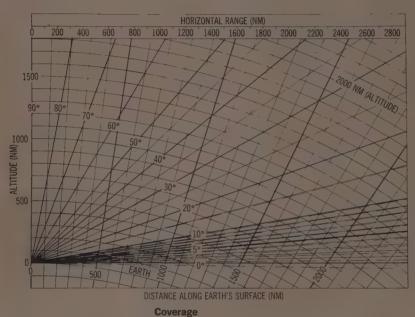


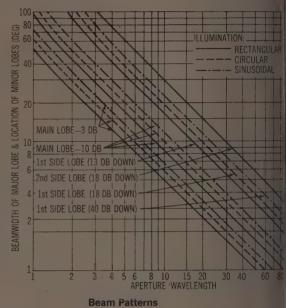
Special Report: Space/Aero Electronics

THE RESIDENCE OF THE PROPERTY OF THE PROPERTY

Pages from an Engineer's Notebook

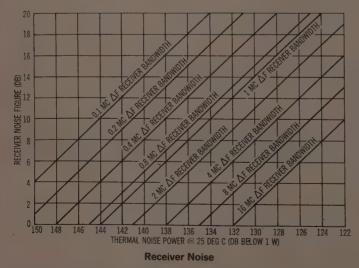
Radar parameters



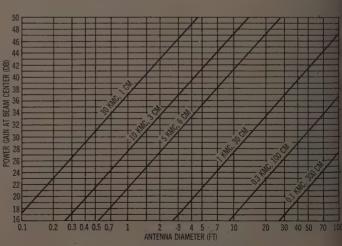


NOTE: Dashed lines show elevation angles corrected for atmospheric refraction according to observations by McCready, Pawsey, and Payne-Scott at 200 mc.

NOTE: Both aperture and wavelength are given in feet. For wavelength, this value is four dividing 983 by frequency (in megacycles).



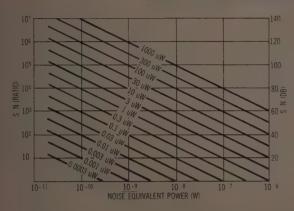
These graphs are from "1959 Data Book," Defense Electronics Div. General Electric Co., Electronics Park, Syracuse, N.Y.

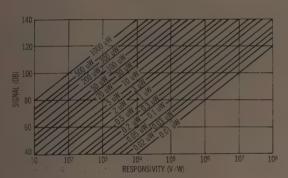


Gain and Beamwidth of Circular Parabolic Reflectors

NOTE: Based on 55 per cent power in main lobe.

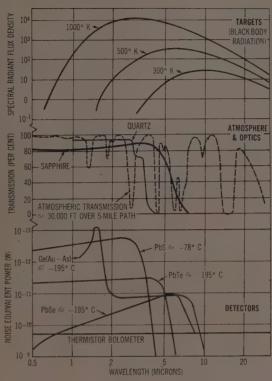
Infrared system parameters





NOTE: Signal reference level is one microvolt.

S/N vs NEP and Signal vs Responsivity for Various Incident Radiant Fluxes



Detector, Target and Atmosphere & Optics Curves

The graphs on S/N vs NEP and signal vs responsivity are from "Infratron Lead Sulfide Photoconductors" (Technical Bulletin 2), Infrared Industries, Inc., P.O. Box 42, Waltham 54, Mass. The third graph is from "1959 Data Book," Defense Electronics Dv., General Electric Co., Electronics Park, Syracuse, N.Y.

Conversion of decibels to degrees

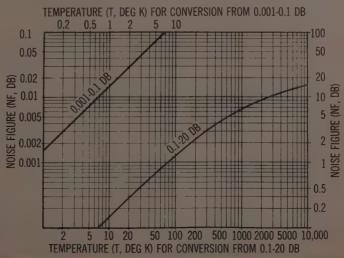
HIS Graph for the conversion of noise figure in decibels to effective noise temperatures in degrees is based on the formula:

$$NF = 1 + T/290$$
, or $T = 290(NF - 1)$.

Noise figures were taken as power ratios in the calculation of the *Graph*. For example, a

noise figure of three decibels represents a ratio of 2. NF — 1 therefore equals 1. Inserted in the second equation above, this gives a value of 290 for T.

Prepared for SPACE/AERO-NAUTICS by Albert T. Isaacs, Stanford Electronics Laboratories, Stanford U., Stanford, Calif.



Special Report: Space/Aero Electronics—Pages from an Engineer's Notebook

Atmospheric attenuation of space communication

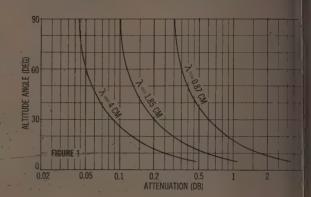
FOR the past 12 years, Collins Radio has been engaged in radio astronomy research. As part of this program, atmospheric attenuation has been measured with radio telescopes and sextants operating at wavelengths of 0.87 and 1.85 cm (34.5 and 16.2 kmc).

Much of the data was gathered by a combined 0.87-cm radio telescope-sextant that continuously records received

power while it automatically tracks the sun. Atmospheric attenuation can be found from the power reduction as the sun sets. For later, additional measurements, a 1.85 - cm radio telescope was slaved to the 0.87-cm sextant.

To explain the experimental attenuation data, a theory was developed on the basis of the quantum mechanics of molecu-

more on page 76



Gyro conversion tables

ANGULAR MOMENTUM

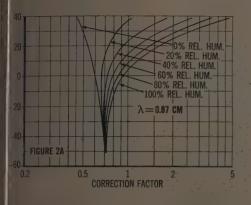
	Required Units*	lb-sq ft/sec	lb-insec.	oz-sq in./sec	oz-insec	slug-sq ft/sec	g-cm²/sec	kg-m²/sec	dyne-cm-sec	new-m-sec	oz-sq inrpm	g-cm²-rpr.
	lb-sq ft/sec	1,.	3.7297 × 10 ⁻¹	2304	5.9675	3.1081 × 10 ⁻²	4.2140 × 108	4.2140 × 10 ⁻²	4.2140 × 10 ⁸	4.2140 × 10 ⁻³	2.2002 × 101	4.0241 × 1
	lb-insec	2.6812	1	6.1775 × 108	16	8.3333 × 10 ⁻²	1.1299 × 108	1.1299 × 10 ⁻¹	1.1299 × 10 ⁸	1.1299 × 10⁻¹	5.8990 × 104	1.0790 × 1
	oz-sq in./sec	4.3403 × 10⁻⁴	1.6188 × 10 ⁻⁴	1	2.5901 × 10 ⁻⁸	1.3490 × 10 ⁻⁵	1.8290 × 10²	1.8290 × 10 ⁻⁵	1.8290 × 10 ²	1.8290 × 10 ⁻⁸	9.5493	1.7466 X 1
	oz-insec	1.6757 × 10 ⁻¹	6.250 × 10 ⁻²	3.8608 × 10 ²	1 1	5.2082 × 10~3	7.0615 × 104	7.0615 × 10 ⁻³	7.0615 × 104	7.0615 × 10⁻³	3.6869 × 108	6.7435 × 1
	slug-sq ft/sec	3.2174 × 10	12	7.4129 × 104	192	1	1.3558 × 10 ⁷	1.3558	1.3558 × 10 ⁷	1.3558	7.0786 × 10 ⁸	1.2947 X 1
Units	g-cm²/sec	2.3730 × 10 ⁻⁶	8.8506 × 10 ⁻⁷	5.4675 × 10 ⁻⁸	1.4161 × 10⁻⁵	7.3757 × 10 ⁻⁸	1 1	10-7	1	10-7	5.2211 × 10 ⁻²	9.5493
Given	kg-m²/sec	2.3730 × 10,	8.8506	-5.4675 × 10 ⁴	1.4161 × 10 ³	7.3757 × 10 ⁻¹	107	1 1	107	1 .	5.2211 × 10 ⁵	9.5493 X 1
	dyne-cm-sec	2.3730 × 10 ⁻⁶	8.8506 × 10 ⁻⁷	5.4675 × 10 ⁻³	1.4161 × 10⁻⁵	7.3757 × 10 ⁻⁸	1 1	10-7	1	10-7	5.2211 × 10 ⁻²	9.5493
	new-m-sec	2.3 7 30 × 10	8.8506	5.4675 × 104	1.4161 × 10 ²	7.3757 × 10 ⁻¹	107	1	107	.1 ,	5.2211 × 10 ⁸	9.5493 X 1
	oz-sq înrpm	4.5451 × 10⁻⁵	1.6952 × 10⁻⁵	1.0472 × 10 ⁻¹	2.7123 × 10 ⁻⁴	1.4127 × 10 ⁻⁸	1.9153 × 10	1.9153 × 10 ⁻⁶	1.9153 × 10	1.9153 × 10 ⁻⁶	1	1.8290 × 1
	g-cm ² -rpm	2.4850 × 10⁻⁻	9.2680 × 10 ⁻⁸	5.7254 × 10 ⁻⁴	1.4829 × 10 ⁻⁸	7.7236 × 10 ⁻⁹	1.0472 × 10 ⁻¹	1.0472 × 10 ⁻⁸	1.0472 × 10 ⁻¹	1.0472 × 10 ⁻⁸	5.4675 × 10 ⁻⁸	1

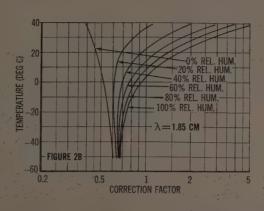
^{*}To find required units, multiply by the listed factors.

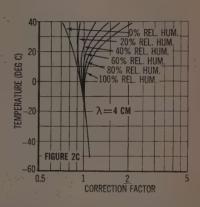
ANGULAR VELOCITY

Required Units*>	deg/min	deg/hr	rad/sec	mils/sec	min/sec	rpm	min/hr	mils/hr
deg/min	. 1	60 /	2.90888 × 10 ⁻⁴	2.90888 × 10 ⁻¹	. 1	/ 2.77778 × 10 ⁻³	3600	1.04720 × 10 ³
deg/hr	1.6 6667 × 10 ⁻²	1	4.84814 × 10 ⁻⁶	4.84814 × 10 ⁻⁸	1.66667 × 10 ⁻²	4.62963 × 10⁻⁵	60	1.74533 × 10 ¹
rad/sec .	3.43775 × 108	2.06265 × 108	1 1	108	3.43775 × 108	9.54930	1.23759 × 107	3,600,000
mils/sec	3.43775	2.06265 × 10 ²	10-3 .	, 1 ° - 1° 5° 1	3.43775	9.54930 × 10 ⁻⁸	1.23759 × 104	3600
min/sec	1	60	2.90888 × 10 ⁻⁴	2.90888 × 10 ⁻¹	1, 47	2.77778 × 10⁻³	3600	1.04720 × 10"
rpm ,	360	21600 ⋅ . `	1.047 <u>2</u> 0 × 10 ⁻¹	1.04720 × 103	360	1.	1,296,000	3.76991 × 10 ³
min/hr	2.77778 × 10 ⁻⁴	1.66667 × 10 ⁻³	8.08023 × 10 ⁻⁸	8.08023 × 10 ⁻⁸	2.77778 × 10-4	7.71605 × 10 ⁻⁷	1 .	. 2.90888 × 10 ⁻¹
mils/hr	9.54930 × 10 ⁻⁴	5.72958 × 10 ⁻³	2.77778 × 10 ⁻⁷	2.77778 × 10 ⁻⁴	9.54930 × 10 ⁻⁴	2.65258 × 10 ⁻⁶	3.43775	1

*To find required units, multiply by the listed factors.







MOMENT OF INERTIA

Required Units*	lb-sq ft	oz-sq ft	lb-sq in.	oz-sq in.	slug-sq ft	g-cm ²	kg-m²	new-m²	dyne-cm ²
lb-sq ft	1	16	144	2304	3.1081 × 10 ⁻²	4.21401 × 108	4.21401 × 10 ⁻²	4.13253 × 10 ⁻¹	4.13253 × 108
oz-sq ft	6.250 × 10 ⁻²	. 1.	9 ;	144	1.9426 × 10°	· 2.63376 × 10 ⁶	2.63376 × 10⁻³	2.58283 × 10 ⁻³	2. 58283 × 10 ⁷
lb-sq in.	6.94444 × 10⁻³	1.111111 ×₹10-1	. 1	`16	2.1584 × 10 ⁻⁴	2.92640 × 108	2.92640 × 10 ⁻⁴	2.86982 × 10⁻³	. 2.86982 × 10 ⁸
oz-sq in.	4.34028 × 10 ⁻⁴	6.94444 × 10 ⁻³	6.250 × 10 ⁻²	. 1	1.3490 × 10⁻³	1.82900 × 10³	1.82900 × 10⁻⁵	1.79364 × 10 ⁻⁴	1.79364 × 10 ⁸
slug-sq in.	3.2174 × 10	5.1478 × 10 ²	- 4.6331 .X 10 ⁸	7.4129 × 104	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1.3558 ·× 10 ⁷	1.3558	1.3296 × 10	1.3296 × 1018
g-cm ²	2:37303 × 10 ⁻⁸	3.79686 × 10 ⁻³	3.41717 × 10⊸	5.46747 ★ 110 ⁻⁸	7.3757 × 10⁻8	1. 4 - 1	10-7	9.80665 × 10 ⁻⁷	9.80665 × 10 ²
kgm-met ²	2.37303 X310	3.79686 × 10 ²	.3.41717 × 103	5.46747 × 104	7.3757 × 10 ⁻¹	107	. 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	9.80665	9.80665 × 10°
newton-met ²	2.41982	3.87172 × 10	3.48454 × 10²	5.57527 × 10 ⁸	7.5211 × 10 ⁻³	1.019 72 × 10 ⁸	1.01972 × 10⁻¹	* i *.**.	100
dyne-cm²	2.41982 × 10 ⁻⁹	3.87172 × 10 ⁻⁸ .	3.48454 × 10 ⁻⁷	5.57527 × 10 ⁻⁸	7.5211 × 10 ⁻¹¹	1.01972 × 10 ⁻¹	1.01972 × 10 ⁻¹⁰	10-10	5 5 7 1

"To find required units, multiply by the listed factors.

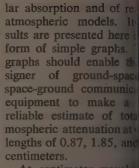
TORQUE

Required Units*>	dyne-cm	newton-meters	oz-in.	ft-16	In-lb (oz-ft	· g-cmi
dyne-cm	1	10-7	1.41612 × 10 ⁻⁸	7.37562 × 10 ⁻⁸	8.85074 × 10 ⁻⁷	1.18010 × 10 ⁻⁶	1.01972 × 10 ⁻³
newton-meters	107	1	1.41612 × 10 ²	7.37562 × 10⁻¹	8.85074	1.18010 × 10	1.01972 × 104
oz-in.	7.06156 × 104	7.06156 X 10~3	1/3/2	5.20833 × 1.0 ⁻³	6.250 × 10 ⁻²	8.33333 × 10 ⁻²	7.20075 × 10
ft-1b	1.35582 × 107	1.35582	192	1. 1.	12	16	1.38255 × 104
inlb	-1.12985 × 10 ⁸	1.12985 × 10 ⁻¹	1 16	% 8.33333 × 10 ⁻² 1	1 1 1/1	1.33333	1.15212 × 10 ⁸
oz-ft	8.47388 × 10 ⁸	8.47388 × 10 ⁻²	12	6.250 × 10 ⁻² ,	. 7.50 × 10 ⁻¹ ···	1 1 1 1 1	8.64090 × 10 ²
g-cm	9.80665 × 103	9.80665 × 10⁻⁵	1.38874 × 10 ⁻²	7.23300 × 10 ⁻⁵	8.67961 × 10 ⁻⁴	1.15728 × 10 ⁻³	. 1

To find required units, multiply by the listed factors.

Equivalents: 1 pound (1b) = 453.5924 grams (g), 1 g = 980.665 dynes, 1 newton (new = 100,000 dynes, 1 slug = 32.174 lb, 1 radian (rad) = 57.29578 degrees (deg).

These tables are from "A Handbook on Floated Integrating Gyros" ('58), Reeves Instrument Corp., Roosevelt Field, Garden City, N.Y. They use values from Industrial Research Service Conversion Factors and tables by Zimmerman and Lavine.



At centimeter waveler only the diatomic oxygen ecule and the water mol cause significant molecula sorption. There is also attation because of scattering absorption by dust parand water droplets, but has been neglected here "Satellite Communications 38).

The attenuation per length caused by mole absorption at any point is atmosphere can be expr as a function of tempera pressure, humidity, and vlength. If we adopt a sphely stratified model atmosp the total attenuation along transmission path can the found by integration.

Figure 1 shows the theoretical one-way attenuthrough the entire atmosp as a function of altitude a for three wavelengths based on the NACA Stan Atmosphere, which has a stant relative humidity of per cent throughout the traphere, a surface pressur 1013 millibars, and a suffer temperature of 15 deg C. model is the approximean of conditions over U. S. at a latitude of 40 december 15 deg C.

For wavelengths beta 1.85 and four centimeters attenuation values will lie tween the corresponding ct of Figure 1. The situated different for wavelengths tween 0.87 and 1.85 cm. this region contains the learning of water vapor abstion.

If surface temperature relative humidity depart





ABINED 8.7-mm radio sex-

ciably from the values of NACA Standard Atmostre, the attenuation value in Figure 1 must be multied by a correction factor in Figure 2. (The temperator correction assumes a noth transition from the CA Standard Atmosphere the USAF Standard Hot i Cold Atmospheres.)

The calculations on which se graphs are based allowed the curvature of the earth. wever, some deviations at be expected because of a standard lapse rates and rizontal gradients in the teorological parameters. It is been found, though, that the teorological parameters are not large. The theorical curves normally are equate for system design.

These graphs were prepared SPACE/AERONAUTICS by one R. Marner, Walter R. ff, and John M. Holt, Radio stronomy Group, Collins adio Co., Cedar Rapids, wa.

leferences—D. O. McCoy, "An All ather Radio Sextant:" Navigation of Dec. 55). G. R. Marner, "Radio ric Measurement of 8.7-MM Atmos ric Attenuation;" Collins Engre. Re 479 ("56). G. R. Marner, "New commissions of Atmospheric Micro e Absorption by Radio-Astronomica ans," URSI-IRE Joint Meeting, Apr For refraction and emission measure tas see G. R. Marner & R. M. Rin p. "Atmospheric Refraction of 8.7 I Radiation", IRE Convention Record 5, 36, and G. R. Marner, "Atmospheric Rediation Received by Directiona Message 1981.



AT CONVAIR. This Model CF 25-50 Bath Radial Draw Former has produced hundreds of titanium and other critical parts for the F 102A and other programs at Convair.



AT DOUGLAS. Anti-icers, retainer seals, canopy rails, and many other parts for RB 66C, B 47 are Radial Draw Formed. DOUGLAS-TULSA STAFF PHOTO.



AT MARTIN. This Model CF 12 1/2-25
Bath Radial Draw Former forms titanium,
stainless and aluminum parts for many of
the Martin airplane and missile programs.

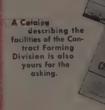


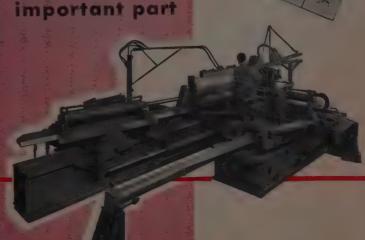
32370 AURORA ROAD • SOLON, OHIO (LOCATED IN THE GREATER CLEVELAND AREA) Manufacturers of Radial Draw Formers • Dies • Tools Press Brakes • Tangent Bending Sequence Presses • Press Type Brakes • Special Machines

Great credit is due the aircraft, jet and missile industries for their fantastic progress in the last few years. The Aircraft Industry has taken us through the sonic barrier into the thermal thicket . . . truly, progress that fires the imagination. The titaniums, the inconels and other tough metals needed for this air progress yield best to forming by the BATH combined stretch and compression methods of Radial Draw Forming. One-piece construction in parts involving compound contours or changes in cross section throughout the contour have become possible with consequently greater strength and reduced weight.

The Cyril Bath Company is proud of the part they are playing in America's most progressive industry.







AIR

PROGRESS

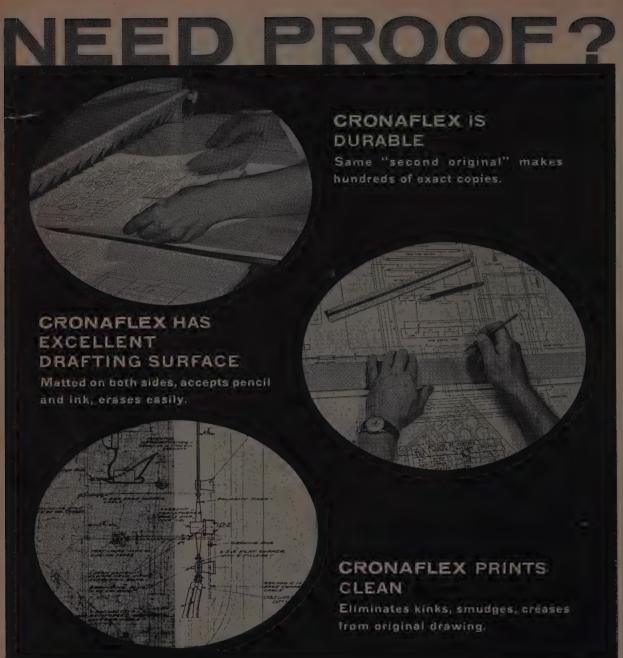
IMAGINATION ...

... and BATH

radial draw

formers are

playing an





Better Things for Better Living

. . . through Chemistry

All Du Pont Cronaflex* engineering reproduction films are dimensionally stable, resisting size changes caused by processing and temperature-humidity variations. Also, because of the clarity of its Cronar** base, Cronaflex intermediates provide faster print-through speeds on direct reproduction equipment. Working drawings are produced faster, saving time and expense. Cronaflex films are available in four types: 1. Direct Positive Film, 2. Contact Film, 3. Projection Film, 4. Unsensitized Cronaflex. Ask your Du Pont technical representative for more information on this line or write:

E. I. du Pont de Nemours & Co. (Inc.), Photo Products Department, Wilmington 98, Delaware. In Canada: Du Pont of Canada Limited, Toronto. ELECTRONICS OUTLOOK . .

Continued from page 23

These Developments . .

the space/aero electronic market

- New types of directional ar acceleration sensors (e.gyros and/or acceleromete depending on electron or niclear phenomena)
- Successful anti-missile tec niques
- Infrared vidicon
- Space communications fast than the speed of light
- Successful manned spanflight
- Detection device not confine to line of sight

der's technical staff and maragement.

Judging by the success we're having with anti-aircraft birds such as Bomarc and Nike, on last manned interceptor might very well prove to be Nort American's F-108 long rang Mach 3 fighter, now in earl development. Many industrial skeptics are even prediction that this plane may never react production.

The global Mach 3 B-70 also in early development a NAA, in turn might be one of the last bombers. Both F-10 and B-70 will contain more electronics than any previous production aircraft (other than special early warning an electronic recon types).

An educated guess is the each F-108, with a price tage \$2 million, will carry 25 pecent of its value in electronic (Besides the normal CNI at FCS complement of an aweather interceptor, the F-10 will carry HF communication a self-contained long ran navigation system, positic keeping and rendezvous equiment, and an infrared fire cotrol supplement.)

The value of the B-70 electronics system is estimate at 40 per cent of the total \$1 million unit cost. This include the electronics in the bomber Jefensive counter - measure votem, which in turn included ecoy missiles.

^{*}Du Pont's trademark

^{**}Du Pont's registered trademark for its polyester photographic film base

Electronics' Share of Missile Dollars

Share of Cost per Missile 40-88 ground equipment dance & control Ion-Inertially guided missile 10-70 nertially guided missile ctronic ground support equipment Air-to-air missile 10 Shipboard-to-air missile actical surface-to-surface missile 20 round-to-air missile

Strategic surface-to-surface

In a nutshell, the transition om aircraft to missiles means ore dollars for electronics. so it means a new distribun of the money, since many craft electronic suppliers ven't yet penetrated the mise electronics market.

An aircraft carries more ectronics than a missile built the same mission. But there l be more missiles, and there ll be more electronics on the ound in a missile system (for ound support and, in most n-inertial systems, for ound parts of the guidance.) The military should spend out 21/2 times more for mise electronics than for airaft electronics in the next cade. Based on estimated OD expenditures. This asmes electronics segments of per cent for aircraft and 33 per cent for missiles d does not cover R&D. 0.4 billion should be spent missile electronics alone in

cal 1958-68. One of the most drastic l come in guidance. Indusestimates based on known ssile projects predict that,

more on next page

Note the performance data on this new



CRYOGENIC CHECK VALVE

by LEONARD

Zero leakage at 0 to 35 G's from

Cracking pressures to a low of 0.25 PSI

Operating pressures above 4000 PSI

Flight certified for liquid or gaseous helium. oxygen, and nitrogen

Now available for line sizes \\\ " to 1.0"

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Circle No. 159 on Reader Service Card in Product Review Section

specialists in ELEMENTS

ELECTRO-FLEX

solves another difficult heating problem.

THIS TIME FOR ...

This quantity production element heats a resistoflex couple which is a flexible line 1¼" in dia., made of braided steel wire with hex fittings at each end. Line must be heated because it transmits liquid at very low temperature. Unit embodies snapaction thermostal sensing tubing temperature to hold 70 to 100°F.

If you have a special heating problem let us solve it for you. We are specialists in light-weight, tough, flexible elements of any shape or size. Units may be designed to operate under high pneumatic pressure. Insulation may be Silicone Rubber or Neone Flements may be impressed. prene. Elements may be immersable or resistant to corrosion. All units meet USAF specifications.



Rocketdyne Division of North American Aviation, Inc. Canoga Park, California Canoga Park, balliolina Material: Silicone rubber sponge, flexible enough to bend with the tubing. Rating: 200 watts at 115 V. (1250 V. breakdown)

Resistance Tolerance: ±5%

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FIRST WILL STANDARD ELEMENTS

Circle No. 160 on Reader-Service Card



which is taking 1/2500th second to snap shut upon a wafer. Talk about acceleration!—the upper jaw hits 60 MPH in three-eighths of an inch. The upper jaw and flying wafer fragments have been caught in many images along their paths of movement.

Photographed by means of a special process, it graphically illustrates the "snap" in Mueller's traditionally snappy springs. These springs provide a mean, corrosion-cutting bite for perfect test connections.

THE NEW "70 SERIES" ALLIGATORS FEATURE:

Simple, direct, streamlined design, and lower cost than the famous "60 Series".

Faster, easier connection, whether screw type or soldered. A new patented hinge, and cord strain relief ears.

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MINIATURE **AGASTAT** time delay relay

for ... missile, aircraft and electronic applications

INSTANTANEOUS RECYCLING . . . reset time—less than .020

UNAFFECTED BY VOLTAGE VARIATIONS time delay remains constant from 18 to 30 volts DC

ADJUSTABLE . . . time delays from .030 to 120 seconds

CHOICE OF OPERATION...for either energizing or de-energizing

SMALL . . . height—4%" . . . width—1-13/16" . . . depth—1½"

LIGHT ... maximum weight—15 ounces

MEETS ENVIRONMENTAL REQUIREMENTS OF MIL-E-5272A

This new AGASTAT time delay relay is an externally adjustable, double-pole, double-throw unit. It incorporates the basic able, double-pole, double-throw unit. It incorporates the basic AGASTAT timing principle, proved by a half-century of reliable operation on automatic aids to navigation, in a space-saving miniature unit, built to withstand the rugged environmental

conditions of missile and aircraft applications.

For specific information on the new AGASTAT relay for

your application, write to Dept. A29-119.



ELASTIC STOP NUT CORPORATION OF AMERICA

1027 Newark Avenue, Elizabeth, New Jersey Pioneers in pneumatic timing

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DOD Expenditures Largely Made for Electronics (billion

Fiscal Year	1957	1958	19591	19601	19611	19621	19631	19:
Major Procurement & Production ²								
Aircraft	7.9	8.4	7.2	5.8	5.2	4.5	4.5	4
Guided missiles	2.1	2.7	3.4	4.6	5.7	6.2	6.5	6.
Electronics & Communications ³	0.9	0.9	0.7	1.2	1,6	1.9	1.9	2.
RGD4	1.7	1.7	2.4	2.5	2.7	3.2	3.3	3.
Total	12.6	13.7	13.7	14.1	15.2	15.8	16.2	16.
Total for electronics1	3.506	4.0	4.5	4.5	. 4.5	4.5	4.5	64.

small electronic expenditures and not shown here are: Ships & Harbor, Ammunition fuzing systems and components), Military Public Works, and Reserve Components category covers mainly "Signal-Corps-type" ground-to-ground systems. (4) Actual on R&D by DOD come to 150-200 per cent of expenditures officially labeled as R& here. The extra funds are taken from the Major Procurement & Production allocation

by 1962, the total spent for guidance will rise from the present \$390 million per year to \$930 million. The amount spent on inertial guidance (IG) is predicted to increase by 380 per cent—from \$150 million to \$570 million in 1962. Non-inertial systems funds will also rise, but at a slower rate-from \$240 million to \$360 million in '62.

The reasons for the shift to IG are:

- IG's invulnerability to countermeasures;
- several inertially guided missiles can be launched at
- logistics problems are reduced, since IG is self-contained:
- Advances in the state of the IG art-Redstone is said to have a circular error probability (CEP) of 500 yd and Regulus II one of less than 600 yd; gyros, it is believed, are being produced with drift rates of less than 0.003 deg/hr and precision pendulous integrating accelerometers with ranges from 10⁻⁶ to 10 G; transistorization has provided drastic size and weight reductions—AC reportedly cut its basic IG system from over 1000 to well under 500 lb

mainly through transistoriza

The IG dollar is split three ways: 35 per cent for electro mechanical gear (gyros, celerometers, servos, etc.), per cent for electronics (a plifiers, computers, etc.), a the remainder for testing. (non-inertial systems, this sp works out to 10, 65, and per cent.)

AC Spark Plug has half of all IG business

The only IG systems know to have been produced those for Jupiter, Mace, Re stone, Regulus II, and Tho All except the Redstone's sy tems, three are made by A Spark Plug, which is now d ing about half of all IG but ness. However, once the systems for Atlas, Hound D Minuteman, Pershing, Polar Sergeant, Titan, and Wagt start to come off the line (1 portedly between now a 1962), the lead should go Arma and Autonetics (assu ing Titan is not phased of as persistent industry rume

Arma and Autonetics ha top developmental reputation in the IG field, but neither l

e a production system as Production capability will in especially strong factor letermining who is to get re IG business. So far, AC rk Plug and Ford Instruits are the only proven

riefly, here is how the IG xpected to be used:

Surface to surface—Some of the ballistic missiles wn to use IG should be t this year and almost three es as many in 1962. Long ge cruise types such as e, Regulus II, and Snark, use aided IG systems. duction will approach some of these in 1959, but their beyond 1960-1961 is in

market seen in Army's rt range birds

rmy short range missiles ht prove to be one of the t profitable markets for IG. out 10,000 of these birds being turned out annually. ong them, however, only geant has IG and this bird eing produced in limited

Air-to-surface — About of these birds (Corvus, en Quail, Hound Dog, tail) are using IG. The eased use of air-to-surface by long range bombersfor offense and decoy rvus and Quail) should n healthy production rates the next 3-6 years.

Air-to-air and surface-to--IG is not being used here studies are underway to k its feasibility. The idea use a Mach 6-9 bird; the nise, that the attacking ile would be too fast for target to maneuver out of lethal area in time. Inersystems probably would ide inexpensive guidance

rn back to beginning of xt article on page 24

HETHERINGTON

SWITCHES . INDICATOR LIGHTS . SPECIAL ASSEMBLIES

ENGINEERINGNEWS



A NON-SNAP SWITCH FOR "KEYBOARDS"

Developed specifically for the key-boards of electronic computers, cal-culators, and other business ma-chines, the Hetherington B5023 gives the smooth, yet precise action necessary for fast manipulation. Operating pressure is a scant ½ pound—just enough to retain the familiar keyboard "feel."

The versatile 2-circuit, momentary-contact circuitry of the B5023 can also be used as SPST or SPDT. Recent tests have shown a life of better than 1.7 million operations at 5 amps., 30 volts dc. This is far greater than the conservative rating reavill. Fed. in Sulletin S. 6. if the second you'll find in Bulletin S-6 if you write for it.

Circle No. 100 on Reader-Service Card

THUMB-OPERATED "TOGGLE" **CONTROLS 4 SEPARATE** SWITCHES -

A familiar sight to helicopter pilots is this versatile Hetherington Type F441 Four-Way Switch. Here it is housed, along with other Hetherington Switches, in a handgrip which fastens to the flight control stick. Moving the Switch's thumb-type knob up or down, right or left, operates control motors for hoists or lateral trim.

Housed in the 1-1/64" by 7/8" phenolic case of the F441 are 4 separate momentary-contact switches arranged 90 degrees apart around the base of a thumbcontrolled toggle lever. Normally, all switches are "open" and the toggle is centered.

Usually the switch is mounted so that the lever projects through eration to one switch at a time. Without the gate however, any two adjacent switches may be closed simultaneously. The toggle may also be held outward and moved in an arc to actuate all 4 switches in succession.

a cloverleaf gate that restricts op-

Admittedly, the F441 Switch is rather specialized. However, it offers industrial users interesting opportunities to simplify equipment design and operation - by saving space, by "foolproofing" critical circuits, or by making operations easier to understand.

No bulletin is available on this switch as yet, but ratings and a dimension drawing will be sent on request. Ask about the F441.

Circle No. 101 on Reader-Service Card



New SWITCHLITES

for "Human-Engineered" Panels

Wherever switches and indicator lights are used on the same panel, Hetherington Switchlites can play a major part in improving panel legibility and operating convenience. For the pushbuttons of these compact snap-action switches contain their own built-in indicator lights for easy, direct association of switch and light functions. Moreover, they save greatly on panel space, wiring, installation and removal time.

So many different Hetherington Switchlites are now available that only a few of their many features can be mentioned here:

Can be mentioned here:

Switch Circuits: SPST, SPDT, DPST, DPDT, or 2-circuit. Up to 15 amps @ 28 v. de.

Snop-Action: Push, momentary; Pull, momentary; Push-Pulh, maintained; Push-Push, maintained; Push-Pull, maintained with auxiliary momentary on "push."

Lomp Circuit: Case ground or separate gnd. terminal. "Hot" lead to separate terminal thru main sw. contacts, or thru separate built-in sw. circuit. 6, 14, 18, 28 volt AN3140-type lamps.

Pushbutton Lenses: 10 styles, each in 10 transparent or translucent colors. Lettering may be engraved on all.

tails on all Hetherington Switchlites.

New Bulletin S-7 gives complete de-

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Need Hetherington products for prototypes, breadboard models, special projects, or small pro-duction runs? Most popular Hetherington items are avail-able at local electronic parts distributors. distributors everywhere,

For the name of your nearest parts distributor and bulleest parts distributor and bulle-tins of the Hetherington Switches, Indicator lights, and Switchlites he carries, write directly to: DISTRIBU-TOR'S DIVISION, HETHERINGTON INC.. 26 Rittenhouse Place. Ardmore, Pa.

HETHERINGTON

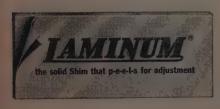
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better switch engineering for a pushbutton world



At Sikorsky, as at most other leading manufacturers of aircraft and missiles, laminated Shims of LAMINUM insure assembly of components with no margin for error. Shims of LAMINUM, which peel for a thousandth fit right at the job, always deliver a 3-fold advantage: they save time, improve quality and cut costs.



Laminum looks, acts like solid metal—yet quickly p-e-e-l-s to precision fit right at the job. In Brass, Mild Steel, Type 302 Stainless with .003" or .002" laminations. In Aluminum with .003" laminations or NEW with .002" laminations. Custommade, Laminum Shims eliminate extra operations: no machining, no grinding, no stacking, no miking. And no grit between layers—ever!



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Circle No. 164 on Reader Service Card in Product Review Section

INERTIAL GUIDANCE

Continued from page 24

three single-degree-of-freedom gyros for a complete directional reference system, whereas just a pair of two-degree-offreedom units will do the job.

Current gyro designs differ in a number of details, among them:

- degree of damping (in single degree of freedom types),
 - nature of spin bearings,
- suspension used to supplement flotation (e.g., jewel pivot, magnetic, flexural, hydrodynamic).
 - nature of angle pickoff,
- nature and accuracy of torque generator (both ac and dc types are used with accuracies varying from 0.01 to about one per cent).

Successful inertial gyros have been designed about all of these variations, but none of them enjoys a clear-cut lead. Most applications, though, use the highly damped, single-degree-of-freedom type. This has been found suitable for all but the most severe guidance applications under study.

Flotation and flexural suspension common

With few exceptions, IG accelerometers are basically pendulums kept centered by some type of torque feedback. Suspension methods include all those we have noted for gyros, with flotation and flexural types the most common ones. Devices using flexural suspension tend to be simpler and more compact, but are still subject to null shifts caused by imperfect coincidence between the zero torque angle of the suspension and the null angle of the pickoff.

Several conventional accelerometer designs use movingcoil, permanent magnet field devices for creating the force or torque feedback. The best examples of this approach give accuracies on the order of 0.01 per cent.

Still other pendulous accelerometers make use of the



Dorne & Margolin has designed more antennas f more different missiles ar aircraft than any other company in the nation.

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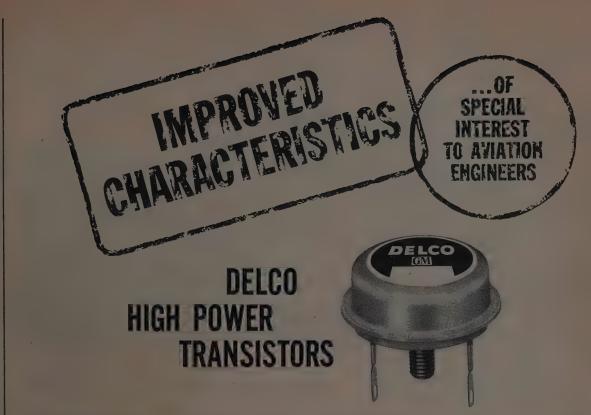
TWEIGHT airborne inertial orm.

ge the angular momentum wheel to balance the applied by acceleration so not the bob. Because the of change of a body's and momentum is exactly ortional to the applied set, this principle yields exely accurate acceleromate the two basic forms: the ulous gyro accelerometer the kinetic doubly intended.

the pendulous-gyro dethe gyro precession le is balanced against the fulum torque. The output its device is a shaft angle ortional to the first time tral of acceleration. In the tic, doubly-integrating acometer, the acceleration le of a flywheel is bald against the pendulum le. Its output is a shaft be proportional to the sectime integral of accelera-

here you're dealing with high velocities, the penus gyro type works best. low velocities, the kinetic, oly - integrating device is e satisfactory. Because of inherent accuracy of the de-balance principle, both uments can be made highcurate (substantially bethan 0.01 per cent). The error comes from uncer-

more on next page



UNEXCELLED FOR LIGHT WEIGHT, SMALL SIZE, EFFICIENCY, RELIABILITY, SWITCHING

TYPICAL CHARACTERISTICS AT 25°C.

	2N1100	2N1099	2N174A	ZN174	2N173	2N278	2N277	ZN443	2N442	2N441
Maximum Collector Current	15	15	15	15	15	15	15	15	15	15 amps
Maximum Collector Voltage (Emitter Open)	100	80	80	80	60	50	40	60	50	40 volts
Saturation Resistance	.02	.02	.02	.02	.03	.03	.03	.03	.03	.03
Thermal Gradient (Max.) (Junction to Mounting Base)	.8	.8	.8	.8	.8	1.0	1.0	1.0	1.0	1.0 °C/wat
Base Current 1B (VEC=2 volts, I _C =5 amps)	135	100	135	135	100	100	100	150	150	150
Collector to Emitter Voltage (Min.) Shorted Base (Ic=.3 amps)	80	70	70	70	50	45	40	50	45	40 volts
Collector to Emitter Voltage Open Base (I _C =.3 amps)	70	60	60	60	50	45	40	55	45	40 voits

*Designed to meet MIL-T-19500/13A (Jan) 8 January 1958 †Formerly DT100 ‡Formerly DT80

Check your requirements against the *new*, *improved* characteristics of Delco High Power transistors. You will find improved collector-to-emitter voltage . . . higher maximum current ratings—15 amperes, and extremely low saturation resistance. Also, note the new solid pin terminal design.

And of special importance to you is the fact that diode voltage ratings are at the maximum rated temperature (95°C.) and voltage.

Write today for engineering data on the new, improved characteristics of all Delco High Power transistors.

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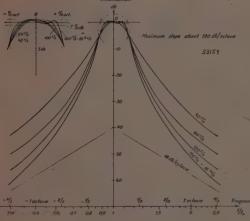
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B&K INSTRUMENTS, Inc. 1000 Inc. 1000 AUDIO FREQUENCY SPECTROMETER



Typical Applications include:
Recording Noise Spectrograms
Measurement of Reverberation Time
Measurement of Vibration Insulation
Distortion Measurements



Detailed filter characteristics.

Gentlemen:
Please send me information on the B & K
Model 2110.

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- * Reads Noise and Vibration Analysis in True RMS Values!
- * Gives Complex Analysis from 14 c/s to 36,000 c/s.
- * Combines with the B&K Level Recorder for Automatic Analysis and data plotting.

The new Audio Frequency Spectrometer Model 2110 features switch selection of the meter rectifier characteristic for either a true RMS, average, or peak reading of analyzed values.

The RMS indication is most frequently preferred for noise and vibration measurements because of its direct relation to the energy of the signal being measured. The true RMS characteristic in the meter circuit is obtained by approximating the square-law curve with straight line portions. An accuracy of 0.5 db is obtained for the measurement of signals with crest factors up to 5.

Thirty $\frac{1}{3}$ octave band filters are used in the Model 2110 to extend the frequency range for complex-signal analysis to be from 35 c/s to 36,000 c/s. Optional $\frac{1}{3}$ octave filters may be added to extend the low frequency range down to 14 c/s. These are available as a separate unit Model 1619 and may be plugged directly into a front panel jack on the Model 2110.

As a linear amplifier, the spectrometer Model 2110 covers the frequency range 2 c/s to 36,000 c/s, a feature especially valuable for vibration measurements.

The spectrometer can be mechanically switched by the level recorder Model 2304 for automatic measurement and recording of spectrograms on frequency-amplitude calibrated chart paper.

◀ Write now for complete information on B & K 2110!

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INERTIAL GUIDANCE

tainties in the degree of po

Vibration lowers the ac racy of all pendulous accel ometers. The error can reduced by increasing the namic restraint of the test m and lengthening the pendul arm. Ideally, the arm sho be of infinite length to eli nate this source of error. Pr tically you can suppress error under severe vibrati only by using heavy dynar restraint. This is why in so instruments closed-loop na ral frequencies have be pushed up to about 800 c

One interesting accelero eter design that doesn't within any of the categor we have listed uses a low disity mass suspended with friction in a body of viscoliquid. Spinning the body liquid at a fixed rate keeps test mass on the spin at Any acceleration along the axis exerts a force on the temass and so produces a mass a

It is still too early to sin out any accelerometer des as best for IG. The most co mon types are pendulums ing either moving-coil torque motors or the pend lous gyro principle.

Servo design poses so critical problems in IG s tems. You have to provide highest possible servo perfor ance to get the orders of

curacy needed.

Stable platform servos m
be able to keep the gyros
most perfectly nulled (wit
a very few seconds of arc)
all times. Good drift perfor
ance depends on keeping
disturbances that get to
gyros as small as possible.
eliminate gear backlash a
keep gear box friction and
flected inertia from produc
disturbing torques, most h
performance systems use
rect drive motors.

The relatively low efficier of the direct drive motor I always presented something a control amplifier problem



RTIAL integrating accelerom-(left) and advanced floated

arently licked now by the easing use of power tranrs. In fact, transistorized uitry is becoming fairly monplace in inertial guid-

le have already noted that y dynamic restraint is led in pendulous accelerters to keep vibration erdown. To keep delays t, many of these designs very high carrier frequen-In certain cases, the meical resonances become limiting factor for usable

arly IG systems used anacomputers almost excluly, partly because the availsignals were basically og and partly because the c sensors themselves limsystem accuracy. Also, ting digital machines were. oo bulky for most airborne

ecent refinements in sensor racy have focused attenon the accuracy limits of og machines. Also, with sistorization the digital puter has become a very pact, lightweight-device. these reasons, it is coming its own in IG.

ligital computer design for tial uses is following two n approaches: general-purtypes and the digital difntial analyzer (DDA). IG gners are still not in agreet as to their merits, and probably too early to tag

more on next page



USE NEW IRC MOLDED METAL FILM PRECISION RESISTORS

If you need the precision of a wire wound resistor, but in a space-saving, weight-saving size, IRC's new Type ME Metal Film Precision Resistors are for you.

Not only do they save weight and space but they are equal or superior to wire wound resistors in many respects as noted at right:

1. Higher environmental performance

2. Higher statistical performance on environmental tests

3. Higher ambients, extended lifes, extended moisture and temperature cycling

4. Better RF characteristics

5. Comparable or lower in cost

6. Resistance values are stable—just as stable for a 1% resistor as for a 0.10% resistor—just as stable for low T.C.'s as high T.C.'s.

TEMPERATURE COEFFICIENT—in order to meet the variety of T.C. requirements, precision metal films are offered in eight classifications. You may order T.C.'s as close as those for precision wire wounds...or, where it is not critical, T.C.'s up to ±100 PPM are available. Classifications are listed below:

Classification	PPM	Temperature span
T-1	±100	_55°C +165°C
T-2	±50	—55°C +165°C
T-3*	-0 +100	_55°C +165°C
T-4*	+0 -100	—55°C +165°C
T-5	± 25	+25°C +105°C
T-6*	+50 -0	_55°C +165°C
T-7*	-50 +0	_55°C +165°C
T-8*	±25	—55°C +165°C

*Special types.

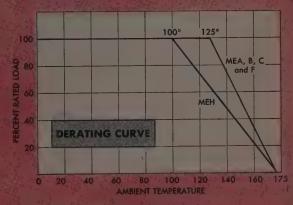
OTHER CONSTRUCTION ADVANTAGES—The new IRC Molded Metal Film Resistors eliminate two other bugaboos of wire wound resistors.

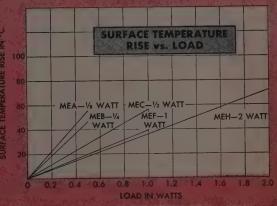
 No cold joint problem. There is no unsolderable esistance wire in the metal film resistor to be soldered to the terminals. 2. No failure due to excess winding stresses because there are no windings.

IRC Type	5 Wattage Ratings (125°C Ambient)	Maximum Continuous Voltage Rating	New Range Minimum** Ohms	New Range Maximum** Ohms		
MEA	1/8	250 V	30	500 K		
MEB	1/4	300 V	50	1 meg		
MEC	1/2	350 V	50	1.5 meg		
MEF	1	500 V	50	4 meg		
MEH	2*	750 V	100	10 meg		

*100°C Ambient Max, below 500 K ohms.

**NEW EXTENDED RANGES—These new ranges have lower minima and higher maxima by far than our former corresponding values.





Write for New Catalog Data Bulletin



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Circle No. 168 on Reader-Service Card opposite last page or send personal resume direct to address mentioned above. GIVE HOME ADDRESS

either approach as the more promising.

The trend to digital computers for IG has spurred an intensive program to produce basic sensors with direct applicability to digital systems. This would do away with the inefficient excessive use of analog-to-digital transducers.

The cost of IG equipment is tied closely to the very touchy manufacturing processes needed to get the desired accuracy. You can only get high tolerance manufacturing for a price. The complexity factor—involving such things as the quantities of diodes, transistors, etc., used in computers—is second to accuracy in pushing up the cost.

There's no absolute way to relate size and weight to accuracy. But experience shows there is a relationship—within limits, accuracy can be improved by going to bigger size. Advances over the past few years have brought radical cuts in size and weight. Whereas the earlier IG systems weighed in at thousands of pounds, current packages weigh in the hundreds and, where less accuracy is needed, break 100 lb.

In IG equipment, even a slight loss in performance brought on by wear or misadjustment makes the system worthless. In other words, reliability is closely tied to accuracy needs.

The trends in IG are not yet firmed up (save in a few cases such as the wide acceptance of digital computers and the swing to beryllium for gyro and accelerometer structures). The main reason for this is that the rapid advances in the art took place during the early years. The past few years have been more a period in which the early gains were consolidated.

Designers, faced with diminishing returns for their efforts, are avidly searching for new principles to work with. The most promising areas here include electrostatic suspension

MARMAN

Engineering Notes



Scientific sealing, ra than the brute-fo approach, is the ture of the Mar Conoseal that has cessfully exceeded sealing requirement tomorrow's aircraft missiles. As oppose

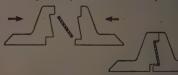
bolted flanges which invariably yield loosen during thermal cycling, the Man all-metal Conoseal closely approximates flexibility of organically sealed joints endures repeated extremes of thermal cyc while still maintaining a perfect seal.

In the low-temperature range (-320°F.) joint has been successfully tested with the oxygen and liquid nitrogen using mate transitions of aluminum flanges to stain flanges. Recent experiments conducted by leading airframe manufacturer have it trated why the safety features of the Coseal are a necessity. Impact sensitivity to conducted by this company reveal that tain materials, such as synthetic elastom cellulose base papers, thermo-plastics thermo-setting resins, including phenolics silicones would explode when saturated liquid oxygen and subjected to shock impact. The chemical inertness of the metal Conoseal precludes any possibility explosion during use of liquid oxygen, in effect, is an anti-explosion joint.

Rigorous laboratory tests at Marman Divisional Aeroquip Corporation, have proven the high temperature and pressure-sealing capabilities pipe or tubing. Prototype tests now be conducted by various customers have demonstrated successful applications in liquid metasuch as liquid sodium at 1600°F. In outstanding tests of other applications, a 14-in Conoseal used in a thrust vector confunctional applications, and the successfully withstood an nozzle temperature 4600°F. for 90 seconds duration.

The Engineering Department at Marman in not overlooked the requirements of picengineers and test equipment designers the development of a 6400-series, heavy-depipe Conoseal. This rugged industrial retains all the maintenance, assembly, a sealing abilities of the aircraft joint, but specifically designed for industrial usage.

Complete test reports are available up request, covering performance of the for ranges of joint types available for you aircraft, missile, and industrial requirement



Cross-sectional drawings of the CONOSEAL Joint show the Belleville-type metal gasket is purposely compre beyond its elastic limit to form a high pressure seal of coubalanced radial and axial forces

May boil.

ENGINEERING MANAGEMENT DIVISION, AEROQUIP CORPORATI



100% Metal! That's Why MARMAN CONOSEAL loints Assure Unlimited Shelf Life With Perfect Seal

Marman's all-metal CONOSEAL Joint provides two basic advantages over organically sealed tube joints:

1-shelf life is practically unlimited

2—seal is maintained from —300°F. to 2000°F.

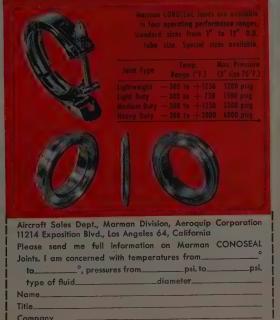
The CONOSEAL Joint utilizes a new concept of metal-to-metal sealing in which the conical metal gasket is completely encased by mating flanges. Compressed radially and axially, the gasket forms a superior seal that withstands extreme pressures and wide thermal cycling. Distortion, shock, even minor linear deflections are absorbed without loss of seal.

The compact design of the Marman CONOSEAL Joint minimizes envelope clearance needed. Single bolt fastening simplifies installation. Ideal for fluid transfer lines and structural joints, it is available in four weight/strength configurations for both air and ground installations. Mail coupon for complete new catalog.

CONOSEAL is an Aeroquip Trademark.



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for gyros and accelerometers, gas-lubricated spin bearings, nuclear spin gyros, and applied cryogenics.

For relatively long airborne missions, inertial systems still don't have enough long-term accuracy to go it alone, though advances in gyro mechanics have cut significantly into drift error. Until this problem is licked, hybrid systems using other sensors to supplement the intertial ones will be needed for some time to come. Among the more common combinations are the stellar-inertial and Doppler-inertial systems.

Stellar-inertial systems use an automatic star tracker to determine drift and provide correcting signals. In some instances, the tracker is used only for correcting heading direction, so that it needs a line of sight to only one celestial body. This type can take a fix on the sun during the day and on stars at night.

Doppler - inertial systems may use the radar data either for damping purposes only or as the prime source of velocity information. In the latter case, the inertial system is used basically to get the high frequency data on vehicle motion, allowing heavy filtering of the Doppler signal.

Space travel poses some special problems for IG, which are under close study. There's little doubt that inertial systems will be used for the launch phase, but once the spacecraft is underway, you have virtually negligible accelerations (actual acceleration will always be in balance with G-effects). Thus we'll have to rely on computations based on celestial mechanics to determine paths through space. Only when we want to change course will inertial methods again apply.—End

Turn back to beginning of next article on page 25

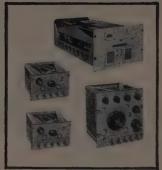
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PRODUCTS THAT SPEAK FOR **THEMSELVES**



IR engineers have reco nized for some time not on that many types of militar targets radiate profusely wavelengths above 3 micror but also that the backgroun problem is less severe her than in the 1-3-micron region The commercially available lead sulfide (PbS) detector sensitive in the latter regio and is used in many of th

STARFIGHTER'S IR gives nighttime capability in co

in fire control system.

junction with optical gunsig

INFRARED

One infrared scanning de vice with a PbS detector ha reached operational status a part of the AN/ASG-14 fin control system in the Lockhee F-104. In conjunction with th optical gunsight, this installa tion provides nighttime capa bility. Similar IR scanners ar slated for other line aircraft.

systems under development.

New search and surveillance IR systems are taking form steadily. The many differen operational uses for these classes of IR equipment has led to a good deal of work with various design parameters Where high sensitivity is im portant, we're studying field division by means of multipledetector systems. Small optica and mechanical devices tha scan the field of view with high efficiency are under development.

Rapid-scan detectors for better IR recon

For some time now we'v been applying discrimination techniques against unwanted backgrounds in scanning sys-



THE MAGNAVOX CO. . DEPT. 33 . Government and Industrial Division . FORT WAYNE, IND. Circle No. 170 on Reader Service Card in Product Review Section



ariety of special forms of Ifide detectors is avail-

Both spectral filtering by interference filters and image plane are being

ge-forming devices, such reconnaissance systems, enefiting from the rapnew 3-5-micron deteconide, for instance, give more detailed picture at scanning speeds than ower detectors. New opand mechanical techs for rapid scanning are eing pushed.

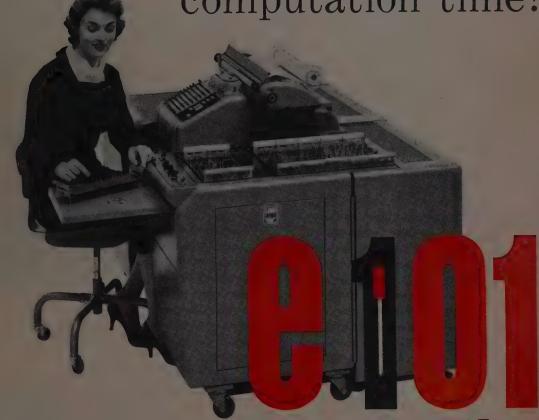
tracking systems have ed operational status in applications as the Sider and Falcon GAR-2A es, using PbS detectors. nt work on tracking sysis aimed chiefly at using 5-micron region for more ive operation against such emperature, gaseous radias targets seen otherwise from the tail aspect.

sign problems involving ivity, discrimination, look and multiple-target caty are being tackled with optical and mechanical ods. For the 3-5-micron n, you need optical matewith suitable transmission cteristics for tracking as as other types of systems. e detection systems must through air at supersonic s, you're faced with the

more on next page

IN AVIATION, E101 is widely used to reduce preliminary design and test data on aircraft, missiles and components . : . to provide accuracy checkpoints on complex problems for large-scale computing equipment. Pinboard programs reduce the data from test equipment readings; the results are directly printed.

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Those who have professional questions or desire additional information are invited to write Dr. William Karush, Head of the System Development Corporation Operations Research Group at 2424 Colorado Avenue, Santa Monica, California.

"A THEOREM IN CONVEX PROGRAMMING"

A paper by Dr. Karush is available upon request. Address inquiries to Dr. William Karush at System Development Corporation.



SYSTEM DEVELOPMENT CORPORATION

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problem of providing a able temperature-resistant dow over the optical system

Underlying the general plems of IR is the need for quate data on the radialevels of targets and bigrounds. Special radiominstruments have provided ful data on the IR outputvarious aircraft and missing Generally, the problems radiometric design resenthose of operational IR equent.

Some of the major IR of ponents are still being de oped. Perhaps the most crit of these is the detector.

Though not as commoused as the more convention phototubes, PbS detectors in a satisfactory product state. Such problems as bility with time and resistate to wide temperature rar and other ambient condition have been solved.

In some of the most rec designs, there's a trend to tectors of more elaborate of figuration for the 3-5 mice region. Several of these ty are in limited production under development. General production techniques for the detectors haven't progressed the level of the PbS units.

The most advanced det tors, in terms of sensitivi response speed, and spect response, are diffused-juncti indium antimonide, lead se nide, p-type germanium, a lead telluride. In their mosensitive forms, these element must be operated well beloambient temperatures.

For adequate sensitivity, d fused-junction indium antim nide, germanium, and le telluride must be at temper tures approaching that of li uid nitrogen. Some detecto such as lead selenide, PEI type indium antimonide, a photoconductive indium an monide, can be operated room or solid CO₂ temper tures. However, these units a available only in limited quatities and at high prices.

uirements for Design of Id Track Systems

Search System	rack System
20-180 deg	1-20 deg 4 to 100 about 0.05 deg 5 to 10 one only position (2 coord.); rate of change of position
presentation	holding one target in presence of reveral

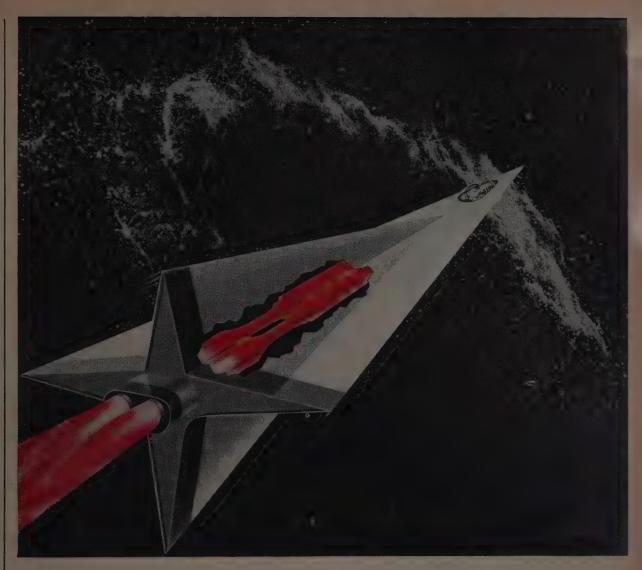
The low temperature opering requirements of the 3-5-icron detectors pose the oblem of a practical cooling ethod. Since detectors and otics are generally small, the oling system also must be nall. Several approaches to etting a cooling device proceeding—196 deg C and below the being worked on. Liquid strogen refrigeration using iniature Joule-Thomson examsion looks promising.

What is really needed is a ght, self-contained package at can operate on call. Probms of reliability for long eriods of operation must still a faced.

Optical materials usable as fractors in the 3-5-micron agion are being widely sought. The substances receiving the cost attention include germatum, silicon, arsenic trisulfide, alcium aluminate glass, and everal in the crystalline group ynthetic sapphire, magnesum oxide, and the alkali alides). As they have a high dex of refraction, low recetance surface coatings have the developed for optical ements.

With the use of large apertre optical systems, you face be problem of building sufciently large mirrors. Ways f making these large elements ith minimum weight and ade-

more on next page



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pneumatic controls • duct couplings & supports • heat exchangers Circle No. 173 on Reader Service Card in Product Review Section quate mechanical stability being looked into.

Aspheric optical syste have the big advantage of sign flexibility, which spurred efforts to deve these specialized compone Replicating techniques, us highly stable plastics, are be tried.

The window, an integ part of the optical systel must be of a material that exceptional optical quality: is mechanically stable in up supersonic wind-streams. gineers are working on problem with silicon, synthe sapphire, and magnesia oxide, among other materia

The multi-layer interf ence filter, so effective for visible spectrum, is being a plied to IR in a number of i portant designs. The adap bility of this technique to va ous filter requirements mal it most useful to the design

Electronic processing cuitry, an important part the IR device, is receiving much attention. Basically, accepts the signal generated the detector and processes to a form acceptable by t presentation system. Engineer are looking for new techniqu that will give optimum S characteristics in all applic tions. Electronic filtering is b ing tried to get maximum u of the signal while reducing the unwanted noise background.

Since it has to meet a var ety of operating evironment the processing circuitry is usi ally custom-designed. To me the need for small size an low power consumption, trar veloped.

At the output end of mo current IR systems is sor means of presenting the dat Closed-loop systems such trackers use the output da for reorienting their optic However, in search, survei lance, and recon systems, the output information is presente visually. Typical of the speci presentation devices that a

Firth Sterling ...

PIONEER IN POWDER AND MOLTEN METALLURGY

oming more popular are age-type cathode ray tubes two-dimensional indicat-tubes (e.g., National on's Videotron).

at the same time, we're imving techniques for reding the output of scanning-t devices on film. Magnetic es continues strongly as a ful medium for recording a from IR scanning systems. The error of the scanning description in the scanning techniques with the scanning that the scanning developed with the scanning that the target image have a developed. These gives a more direct and easily appreted presentation.

nsitivity limit is liation noise

Until we're stopped by conlons outside the IR system, ther development of sysis and components promis big benefits for the state of art. Detectors, for innec, can take improvements sensitivity up to the point which they are limited by incident radiation noise. In best operational detector produced, we're still short this limit by no less than actor of five.

Present optical materials are e for improvement in specl transmission and mechanl properties even under rent operational conditions. Sector cooling equipment, o, is still in a relatively early ge of development. We can ake it more reliable, and ore flexible.

Our information processing stems have a long way to go. on - linear devices perhaps il improve the overall S/N io. Further work on presention techniques can't help t make for more effective of output data.—End

Turn back to beginning of next article on page 28

1					
	ALLOY	TENSILE	DATA	RUPTURE DATA (100 Hrs.)	METHOD OF
	Useful Temperature Range	Yield (psi) Elongation (%) Temperature	Tensile (psi) Reduction of Area (%) Temperature	Stress (psi) Elongation (%) Temperature	METHOD OF MELTING
	HWD Up to 1000°F	140,000 14% 1050°F	170,000 52% 1050°F	170,000 12%+ 900°F	AIR ARC
	GREEK ASCOLOY † Up to 1000°F	135,000 17% 70°F	155,000 56% 70°F	42,000 35% 1050°F	AIR ARC
	16-25-6 Up to 1300°F	71,000 33% 70°F	129,000 54% 70°F	42,000 12%+- 1200°F	AIR ARC
	A-286 Up to 1350°F	98,000 22% 70°F	150,000 42% 70°F	62,500 7% 1200°F	AIR ARC STERCON
	STERVAC 3000 (M-308) Up to 1350°F	121,000 14% 1200°F	134,000 20%+ 1200°F	65,000 1300°F	STERCON STERVAC
	STERVAC 1000 (Waspaloy**) Up to 1500°F		9,440	36,000 14% 1500°F	STERCON STERVAC
	STERVAC 2000 (M-252) Up to 1500°F	98,000 20% 70°F	170,000 22% 70°F	34,000 1500°F	STERVAC
	STERVAC 5000 (Udimet 500***) Up to 1600°F	8%+ 1200°F	175,000+ 10%+ 1200°F	25,000 30%+ 1650°F	STERVAC (**)
WANTED.	STERVAC 4000 (Rene 41*) Up to 1650°F	110,000+ 8%+ 1400°F	120,000+ 12%+ 1400°F	25,000 25%+ 1650°F	STERVAC
WANTED=	UNKNOWN	TO ME	ET YOUR	REQUIRE	MENTS

new applications for high temperature alloys

Charted above are high temperature alloys produced by Firth Sterling metallurgy to meet today's specifications in the aircraft and missile industry. There are, no doubt, applications for these which do not yet exist—just as there are applications for which no satisfactory materials have been developed. We'd like to know your needs—and apply our experience to your problems.

For over 68 years, Firth Sterling metallurgists have pioneered the development of tougher, more heat-resistant metals. Their experience in all

three melting methods: air arc, consumable electrode (STERCON) and induction vacuum (STERVAC) is exceptional. And this important technological "know how" is being applied to newer metals such as Zirconium.

If your requirements involve the engineered performance of alloys at higher and higher temperatures, our background in this field could prove helpful. Please let us hear from you. Firth Sterling, Inc., Dept. 15A, 3113 Forbes St., Pittsburgh 30, Pa. Offices and warehouses in principal cities.

Trade Mark *General Electric Co. * **Pratt & Whitney Div., United Aircraft Corp. * ***Utica Drop Forge & Tool Div., Kelsey Hayes Co.

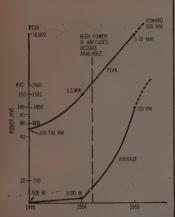
†Allegheny Ludlum Steel Corp.

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SINTERED TUNGSTEN CARBIDES . HEAVY METAL . CERMETS . CHROMIUM CARBIDES
ZIRCONIUM . STERVAC & STERCON SUPER ALLOYS

Continued from page 25



MICROWAVE POWER has been on the upswing over the years. Greatest boost came with development of axial beam (Otype) and crossed-field (M-type) tubes. Average power has increased more sharply than peak power. Reason is that high peak powers are limited by component breakdown. Design accent has been on getting more energy out by upping average powers.

gree of beamwidth. In turn this adds up to a problem of energy generation by the transmitter and one of focusing by the antenna.

Modern long range radars are putting out peak powers in the tens of megawatts and average powers in the hundreds of kilowatts. Soon, it is expected, these powers will approach 100 Mw peak and one megawatt average. Developments in high power klystrons, wave tubes of the O- and M-types, and other crossfield devices such as the platinotron have paved the way for these tremendous powers (see AvAge, "New Breed of Microwave Tubes," Apr. '58, p. 22).

The days of the magnetron—except in simple radars—are numbered. Because of their high power-weight ratios, magnetrons will have a longer life in airborne and portable ground systems, but even here they are being ousted rapidly.

Almost all new high power ground systems and many airborne ones are using oscillator-power amplifier combinations of either klystrons or wave

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dous versatility and flexibility. Exclusive feature of A-MP Systems is
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for clean contacts. A-MP Universal
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Panels are excellent for digital computers, data processing equipment and
automatic test equipment. A-MP
Shielded Patchcord Programming
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a complete series for all programming
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Bulletin Number 58

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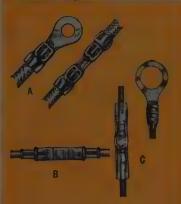
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(C) AMP COMPONENT TIPS . solder dipping cycle . . . permit bridging ind thru-plating on two-sided boards, by excellent solder wicking charac-

Bulletin Number 81



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Bulletin Number 37

A-MP products and engineering assistance are available through whollyowned subsidiaries in: Canada • England • France • Holland • Japan tubes. In non-microwave radars (200-900 mc), in which you'll find today's super-power sets, triode oscillator-power triode or tetrode combinations are used. Megawatt (peak) microwave amplifiers are available for the L- and S-banus and high kilowatt amplifiers of the same families for the Xand K-bands.

In addition to high power, oscillator-power amp combinations have excellent pulseto-pulse frequency stability and can change frequency rapidly. Pulse-to-pulse coherence is especially necessary in some of the sophisticated signal enhancement and countermeasures schemes in use and planned and also for Doppler radars and radars equipped with modern MTI (moving target indicator) system.

The tuning ranges of high power klystrons are usually one or two per cent electronically and up to around 15 per cent machanically. In high power backward and forward wave (TWT) amplifiers, electronic tuning of up to 15 per cent can be had. These transmission bandwidths are too small to suit the military, which would like to get at least 50 per cent tuning-for greater counter-countermeasures flexibility. This is one area in which much of the midnight oil is being burnt.

Klystrons in use have medium power

Some power klystrons are used today in aircraft and missiles, but they are medium power types. Here weight has been traded off for pulse stability (as for airborne MTI).

Hughes Aircraft is working on a master oscillatorpower amplifier (MOPA) transmitter for its newest interceptor fire control system that, it claims, will give some 20 times more power than the present average as well as the stability needed for pulse Dop-

more on next page



New FN-12 Series Featherweight Locknuts provide 125,000 psi minimum at temperatures up to 550°F, yet weigh 12% to 72% less than comparable sheet metal (NAS 679) and AN Series nuts. The FN-12 is installed with standard tools, incorporates new burr-free locking feature which permits highly accurate torquing.

New SPS Featherweight locknuts save you up to 72% in locknut weight

Smaller hex also requires less installation area —helps you design more compact bolted joints

Designed for fastening structural skin and panel assemblies, the new SPS FN-12 Series Featherweight Locknut offers you weight savings of 12% to 72% over widely used sheet metal and AN Series nuts. Yet despite its light weight, the cold forged FN-12 sacrifices none of the static or dynamic properties of its heavier counterparts. Meeting all requirements of MIL-N-25027, it has a tensile strength exceeding 125,000 psi. Its vibration resistance is 150% of specification minimum. And it gives greater bolt tension-tension fatigue strength than any other lightweight locknut tested.

Besides offering you valuable weight savings combined with high strength and reliability, the FN-12 locknut, because of its new configuration, can be installed closer to vertical bulkheads than any other aircraft nut now in use. This permits further weight reductions through miniaturization of joints to be fastened.

FN-12 Series Featherweight Locknuts are available in heat treated alloy steel in sizes #4 through ½ in. They are furnished cadmium plated, with optional molybdenum disulfide coating. For complete information request new Bulletin 2426. Aircraft/Missiles Division, STANDARD PRESSED STEEL Co., Jenkintown 54, Pa.

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WEIGHTS OF FN-12 AND COMPARABLE LOCKNUTS

(all weights expressed as pounds per 1000 pieces)

Size	FN-12	NAS 679*	% Weight Saved by FN-12	AN Series†	% Weight Saved by FN-12
# 4-40	0.4	0.8-1.0	50%-60%	1.3	69%
# 6-32	0.7	1.3-1.7	50%-60%	1.8-2.5	61%-72%
# 8-32	1.2	2.1-2.5	43%-52%	2.8-4.2	57%-71%
# 10-32	1,5	2.5-2.8	40%-47%	3.3-4.6	55%-67%
1/4-28	3.5	3.9-4.7	12%-26%	5.4-8.5	35%-59%
5/16-24	5.4	6.4-7.2	16%-25%	8.7-11.8	38%-54%
3/8-24	7.3	9.3	2 1%	11.5-19.5	37%-63%

*Range of four most commonly used sheet metal nuts of NAS 679 type \dagger Range for AN 363, 364, 365

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SPS research is continually developing fasteners with higher standards of predictable performance. By installing SPS high-reliability fasteners in your assemblies, you increase overall product reliability.

For more information on the full meaning of reliability, write for a copy of the new SPS booklet "High Reliability."

Table 1: The Best in Low Noise Crystal Mixen

Frequency Range (kmc)	Crystal	Microwa Mixer-1 Amp Noi Figure (db)
3.955.85	1N23EMR	5.8—6.
2.6 —3.95	1N21EMR	5.8-7.
1.7 -2.6	1N21EMR	6.0—7.:
1.12—1.7	1N21N	5,0-6.0

IF amplifier used Cascade GL-6299 frior
in front ends with average noise figure
one decibel at 30 mc. Until the advent
parametric amplifiers, these receivers we
considered to have the lowest noise figure
possible within state of the art. Figure
from Rome Air Development Center.

pler. (Present systems use 250 kw-peak mechanically tune-magnetrons.) This MOPA is believed to be a backwar wave oscillator working into string of TWTs.

The development of hig power switch tubes and diode has kept pace with power in creases in transmitters. Hydrogen thyratrons are still bein used, but their switching capabilities have increased several orders of magnitude in power the military is sponsoring several development program on high power hard tube an magnetic modulators.

Statistical concepts put into circuit form

The application of statistic concepts may prove to have the greatest influence on i ture radar designs (see S/ "Information Theory Pla Key Role in Radar Design, Dec. '58, p. 138). Since must consider random var ables-such as scintillatio glint, clutter, noise, and jar ming—the radar problem quires statistical analysis accurate solution. While t has long been realized, it has been only recently that design ers found ways of applying statistical concepts to succes

more on page

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BW BORG WARNER

No. 182 and Service Card roduct Review Section

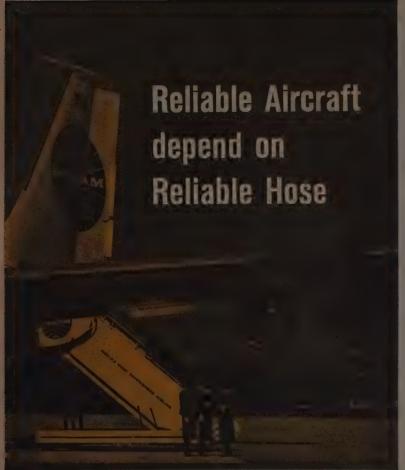


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*T.M., Pat. No. 2,853,319

Fluoroflex is a Resistoflex trademark, reg., U.S. pat. off.
 Tefion is DuPont's trademark for TFE fluorocarbon resins,

Resistoflex

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10 BEST CRISTAL TWTS & EWAS (C.S. DB)

8 S (C.S. DB)

PARAMETRIC AMPS (C.1-2 DB)

PARAMETRIC AMPS (C.1-2 DB)

MASSERS (1D K. EFFECTIVE NOIS TEMP)

TEMP)

NOISE FIGURES have been progressively lowered. Crystal mixers and the "medium-low" noise TWTs are available. Others are still in various stages of development. A limited number of PN junction parametic 'amplifiers is being used in special applications.

ful circuits and hardware.

One of the best-known techniques here is pulse compression. With PC, the designer can use relatively long transmitted pulses to get higher average powers without sacrificing the range resolution and high E/N_o of a short pulse, high peak power system. By getting away from high peak powers, he can put more energy on the target without running into arcing and voltage breakdown problems.

A common PC method is to vary the RF of the transmitted pulse. While the pulse appears to the generators as, say, a 10-usec pulse, because of the RF variations the receiver might see a train of one-usec pulses.

Matched filters, in various forms, are used in receivers to correlate the incoming wave's shape to that of the outgoing wave and so distinguish from false signals and noise.

Another statistical technique for signal enhancement is signal integration (loosely also a form of PC). The aim is to delay and integrate target returns over a period of time. Since the desired signals are regular in character, they will tend to add; the noise, being random, will tend to cancel.

There are two forms of integration: post detection (video) and predetection (at RF or IF). The former has been used for a number of years

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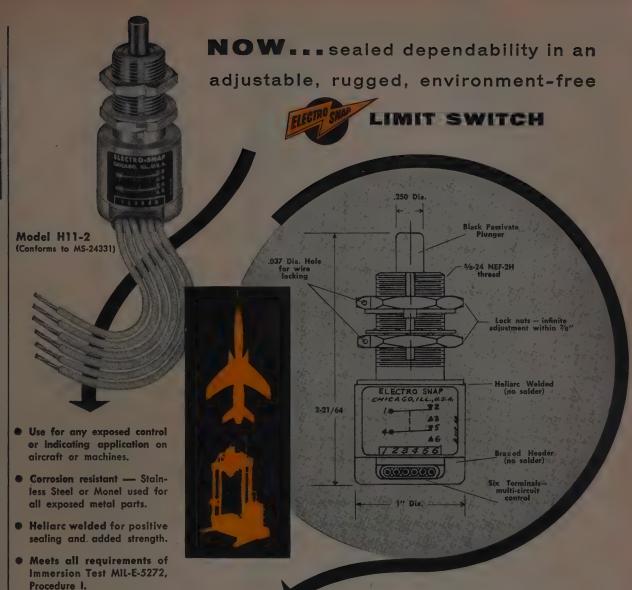
RE radar developed by shows trend to produce of almost photographic This radar is for ground ance and mapping.

vill be found in most new range radars. It works hen noise has a constant ge value and is almost s when average noise s are widely distributed. like video integration, tection integration res carrier phase continuity equency stability, and the t can't vary its range than 1/4 signal waveduring the integrating d. This third requirement n't be met until the dement of Doppler radar. a new technique—still in prototype stage—is posthat measures target and feeds it in as a corn factor. Effectively, tarnge no longer varies dure integrating period.

noise components ensitivities

e most advanced of these enhancement schemes rapped tightly in security. ast a dozen, in the protostage and/or developare competing for acnce by the military. The ns that have been built not a little in size and lexity to their radar sys-For the present, they bly will be applied only very long range radar E/N_o ratios are poor nay even be below unity. ually, if bulk and com-

more on next page



Header is brazed (not soldered)

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operating characteristics

Contact Arr.	angement.			2 -S.I	P.D.T.
Pretravel .				040	Max.
Movement 1					
Overtravel .				250	Min.
Operating (Cycles			25,000	Min.
Break Distar	nce			010	Min.
Difference of Between	of Operati Each Pole	ng & Re	set Pt.		Max.
Operating F					
Overtravel	Force			. 30 Lbs.	Max.
Release Fore	ce			4 Lbs.	Min.
Electrical Ra Sea Level		00 Ft		3 VDC, 4A	
			2 A. I	nd. – 4A <i>l</i>	Motor
Ambient Te	mperature	Range.	10	0° to +22	21° F.
247 2 2 4			01	2/ 0 0 .	



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plexity can be cut, signa hancement should find u shorted range systems ar airborne radars.

An important effort is underway to improve perfance by lowering the figure (NF). As with power, this is largely a tion of component devenent.

First came better of mixers and "quieter" tube preamps and IF strips, brought minimum figure less than 7½ db (see Table

Then came the Twhich gave lower NFs even the best crystal m and, at the same time, vided RF amplification an octave of bandwidth. Nof the newer receivers in eration and on the dra boards are designed an TWTs. Developmental T and backward wave ampl show promise of dropping to four decibels.

The highest hopes, how are pinned on the solid UHF and microwave as fiers of the parametric maser types and also on vacuum tube electron si parametric amplifier. maser, for instance, is tically noise-free. On the of programs underway USAF Rome (N. Y.) Air velopment Center (RAI NFs of 1-2 db have been dicted for PN parametric plifiers in the S-, L-, P-bands. For next year, N 0.1-0.5 db are predicted.

Table II also reveals main drawback of the state amplifiers: narrow be width. Their bandwidths have to be increased substially if designers are to be to take full advantage of low noise qualities. Judgin the large number of top engaged in parametric an fier and maser researchances for solving this plem are good.

Airborne Instruments L for example, reported get eight per cent bandwidth

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0 db gain at 400 mc with a N junction parametric amplier operated in the sum-freuency mode. Many of the dierse modes of maser and arametric amplifier operation re yet to be fully explored.

The electron stream parametric amplifier, a four-inch acuum tube jointly discovered y Stanford U. and Zenith, seems to have unlimited bandwidth potential. A prototype howed 30 db gain with a one ecibel NF. Chances look good or this device in future radar esigns.

The maser probably won't be used widely in active adars. One of the main drawacks is that its crystal element must be kept at liquid delium temperatures. Another is that it requires a sizeable magnetic field. Also, it is quite ifficult to align and tune. Most radar designers see the main use of the maser in spe-

cial-purpose equipment such as radio telescopes.

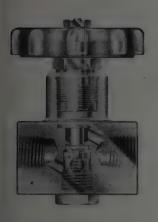
Low noise receivers also will pose a new set of problems. To put it briefly, by making a device sensitive to microscopic signals, the designer also makes it sensitive to microscopic noise sources ignored by present "noisy" receivers.

In conventional radars, the receiver contributes most of the receiving system noise. In receivers with effective noise temperatures of a few hundred degrees or less, most of the noise will be contributed by what the antenna sees and by resistive losses in the antenna, the waveguide, and connecting joints. These last should prove especially troublesome, since they both attenuate signals and generate noise.

Four sources of noise that is picked up by the antenna will have to be reckoned with: (1)

more on next page

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TUBE TESTERS

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MODEL 1230—Card programmed to automatically set-up predetermined voltages and conditions for each tube under test. Self-calibrating and accurate within 1 to 3% of best known standards for transconductance and plate current measurements. Reject levels set in accordance with military specs. Also preprint special purpose specs. Also permits special purpose tube tests to meet specific circuit re-quirements. Portable style with



MODEL 1575—Meets Western Electric specifications as a Mutual Conductance tester. 0-60,000 micromhos in 7 ranges. Two separate meters for line and grid bias voltages indications. Has 4 signal voltages. Features new, voltage regulator tube test. Built-in roll chart. Portable style with cover.

RESEARCH LABORATORY

MODEL (Not Illustrated)
MODEL 1700—For tube tests under
handbook conditions, All voltages and
currents are variable and metered
(9 meters). 0-60,000 micromhos in 8
ranges. Electronically regulated and
metered power supplies.

OSCILLOSCOPES

WIDE RANGE—VERSATILE

WIDE RANGE—VERSATILE MODEL 1810 — 5" flat face, post accelerated CRT and illuminated, calibrated screen. 4 mc band vertical amplifier. 10 millivolts peak-to-peak per cm sensitivity. Triggered or recurrent sweeps—variable from 2 cps to 30 kc. Built-in, direct reading, regulated square wave voltage calibrator. AC or DC vertical amplifier inputs.

INDUSTRIAL'S 5" PORTABLE

MODEL 685—Features identical amplifiers DC to 750 kc. 0.5 microseconds rise time. Sensitivity—20 millivolts RMS vertical (30 0.5 microsection is the control of the control of 10 to 1. External sync for triggering and sync for triggering and synchronizing the built-in sweep generator.

3" RACK MOUNT

MODEL 387R—Identical horizontal and vertical amplifiers. Triggered or recurrent sweeps from 1 cps to 100 kc. Slow sweeps (10 seconds or more) with external capacitors. 10 millivolts RMS per inch vertical sensitivity (14 horizontal).

3" PORTABLE

MODEL 385 CSM—Rugged lightweight designed for field service. Features untized circuit modules. DC amplifiers (horizontal and vertical) with frequency coverage to 4 mc at .075 RMS volts per inch. 2 MC bandwidth has roll-off per IRE standard, Telescopic light shield and shock mounted carrying case.

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MODEL 710-Both sine and square MODEL 710—Both sine and square wave outputs in one generator with wide frequency range (20 cps to 1 megacycle in 5 ranges). Sine wave total harmonic distortion below 1%. Hum level better than 90 db down. Square wave rise time less than 0.1 microsecond. Portable—easily adapted to rack mount.

SQUARE WAVE

MODEL 1715—Wide frequency range of 1 cps to 1 megacycle, continuous in 6 decade bands with constantly in 6 decade bands with constantly variable frequency available throughout the entire range. Rise time is 0.02 microseconds. Consistent output level is independent of frequency setting. Electronic regulation of power supply. Includes symmetry control and provision for external synchronizations.

VACUUM TUBE VOLTMETERS

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MODEL 1600—Accurately indicates AC voltages from .05 to 300 volts in frequencies from 20 cps to 700 mc (up to 3000 mc are possible). Includes special thermionic diode probe. Excellent plus or minus DC volts, readability from .05 to 1000 volts. 7-range ohmmeter with practical readability from 0.2 ohm to 1000 megohms. Power supply built-in. megohms. Pov Has 1% meter.

SMALL SIZE VOM



MODEL 456 — Has full-wave bridge type rectifier circuit. Features a patented automatic (button-reset) overload protection system to hold meter and associated circuits from damage due to accidental overload. 0-1200 AC or DC volts in 6 ranges. Sensitivity—20,000 ohms per volt DC (1000 AC). Resistance—0 to 100 megohms. 5 db ranges (-18 to +57).

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galactic and extragalactic. radio stars, (3) atmosph absorption, and (4) terrest absorption. In general the two decrease and the oth increase with frequency, though there are some quency windows where atm pheric absorption is virtue negligible.

Radar trend to 200 mc for long ranges

Other important trends and developments a

Frequency—For long rai search, the trend has be from S-band down toward 200-mc area, because of high power generators and b noise receivers available at lower frequencies. Now, w high power and low noise vices becoming available in: microwave region, the tree should be to go back up t scale. A good compromise tween high resolution and mospherics seems to lie tween 1000 and 6000 mc.

Polarization - The varie larizer, developed at RAD permits transmission and ception of any polarization elliptical, linear, or right left circular.

Displays—The trend is ward improved resolution, li earity and dynamic range a toward variable persisten and brighter indicators. Althere is marked trend to fe only processed data to displa A variety of special CRTs lows presentation of multip data, including numerals a symbols on black-and-wh and multi-colored displays. A tempts are being made to con up with CRT presentation that approach photo qualit

In future airborne systen displays may be presented the windshield combining gla with either optics or transpa ent, thin film CRTs. For larg area displays, hopes are pinne on electroluminescence. How ever, practical devices base on this principle are estimate to be 3-5 years away. In t

more on page

FULL VOLTAGE RESTORATION IN

I.5 MILLISECONDS

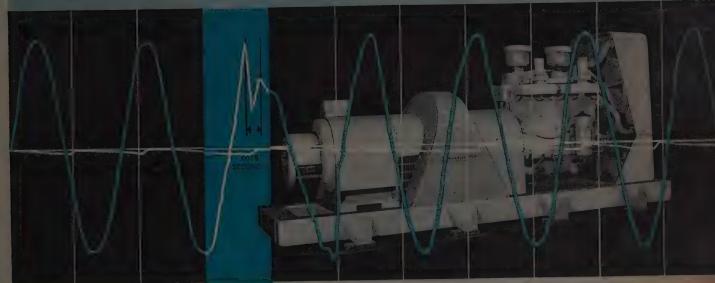
UNINTERRUPTED POWER FROM PRIME TO STANDBY

"Instantaneous" emergency power is just one of the many practical advanced engineering features of ConDiesel's Model UPS.*

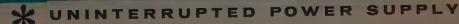
Model UPS* delivers this outstanding performance... Where even a few vital seconds count, you cannot afford to be without this unit.

Basically the unit consists of a synchronous alternator and flywheel connected to a full diesel, liquid cooled engine, through a dry-type magnetic clutch. When in stand-

by, the diesel does not run, thus greatly reducing operating and maintenance costs. The synchronous alternator and flywheel operate on prime power and when prime power fails or falls below established minimums Model UPS switchgear disconnects commercial power and energizes the magnetic clutch, causing the rotating flywheel to start the diesel. When commercial power returns to normal, the Model UPS will check its quality for a predetermined period, then automatically cut-out and return to its standby condition.



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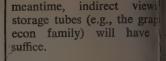
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POWER EQUIPMENT DIVISION STAMFORD, CONNECTICUT

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vibration. Electrical, hydraulic,

pneumatic, structural, Physical.



"Computer" storage median the offing?

Available direct view steage tubes suffer from limitedynamic range of inputs ar low target resolution. "Computer-type" storage media margind use in future system.

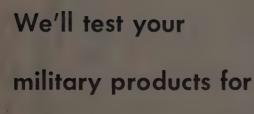
Counter - Countermeasur.

—Radar designers seem to lone jump ahead of the electronic counter-measures three. One of their best tools is programed transmitter frequency variations coupled with a wice band receiver. One technique changes carrier frequency from pulse to pulse. Unique modulation schemes and matche filters allow systems to sethrough many types of jamming and to discriminat against decoys.

Airborne Fire Control—Th C- and X-bands still look be here. Present systems use me chanically tuned magnetrons Future will use electronical tuned wave tube chains with less peak but some 20 time higher average power. Anten nas are getting larger, becaus more range is needed. Forty inch hydraulically scanned dishes will be seen on future systems. Designers would like to get rid of the dish altogether and replace it with an electronically scanned slot array that is part of the vehicle's struc-

Monopulse tracking will be replaced by pulse Doppler to get 100-200-mile ranges and minimize ground clutter. The advantages of Doppler are partly offset by problems of range ambiguity and design complexity.—End

Turn back to beginning of next article on page 26





tol miniature pressure switch res ultra-reliable precision pressure element. sive design provides outstanding resistance to , vibration, acceleration and overpressures.

Bristol miniatures, widely proved in modern aircraft, signed for switching electrical circuits in response to re changes in air, fuels, lubricants, hydraulic fluids,

tol's specially designed Ni-Span element is silver to the stainless steel base assuring greater reliability rdinary soft-soldered construction. Result: accurate, , repeatable performance in any position, at temres from -65° F to +250° F, and under Mil Spec enental requirements.

for Bulletin AV2010 on Bristol Miniature Gage and te, Adjustable and Differential Switches. The Bristol ny, Aircraft Components Division, 150 Bristol Road,



SPECIFICATIONS (Fixed pressure setting models)

Normal Working Range - 0 to 100 psi absolute, gage, or

Burst Pressure - exceeds 250% of normal working pres-

Electrical Ratings — 5 amp at 125 v, 00 eyes, manual or resistive
4 amp at 30 vdc resistive
2.5 amp at 30 vdc inductive
Dielectric Strength — 500 v rms between terminals and from terminals to case (MIL-S-8801)
Life at Rated Electrical Load — 40,000 cycles at 125 vac
25,000 cycles at 28 vdc
High Temperature Exposure & Operating —
(MIL-S-8801) 250°F

Low Temperature Exposure & Operating — (MIL-S-8801) -65° F Shock, 30 g, 3 axes — (MIL-S-8801) no change Vibration — (MIL-S-8801) no contact chatter, no switch

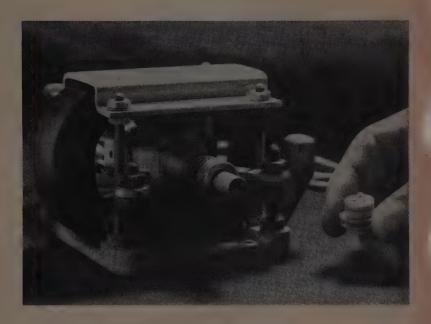
damage
300-600 cpm at 0.050" d.a.—set point change—none
operating differential change—none
600-4500 cpm at 0.036" d.a.—set point change—1/2 psi
operating differential change—1/2 psi
4500-30,000 cpm at 10 g—set point change—1/4 psi
operating differential change—1/2 psi

FINE PRECISION INSTRUMENTS

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space | aeronautics

product of the month



PACKAGED MAGNETRONS are pre-tuned and adjusted



JANUARY 1959

General Electric Co., Schenectady 5, N. Y., has developed packaged S-band voltage-tunable magnetrons consisting of tube, magnet, cavity, and integrated parts. They are pre-tuned and adjusted.

Tuning characteristics are linear over a 1.7:1 frequency range or greater, with changes in anode voltage. Mechanical tuning is not necessary. Wide-band range is 2200-3850 mc, with a

minimum CW power output of two watts across the entire frequency range. GE engineers say no additional circuitry is needed to adapt the package for oscillator service, since it is a completely pre-assembled unit. Approximate outside measurements are 4x4x31/4 ft. GE is developing voltage-tunable magnetrons to cover 50 to 800 mc and over, with corresponding levels of power output.

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This award is made in recognition of outstanding service performed through the development and manufacture of a product contributing to the advancement of aviation. tandolph Hawthome





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product review

PRESSURIZING WINDOW withstands 30 psi



A new pressurizing window, the BL 777, is announced by Bomac Laboratories, Inc., Dept. S/A, Salem Rd., Beverly, Mass. This Ku-band window has a maximum VSWR of 1.12 over a frequency range of 12.4 to 18.0 Kmc, and is for use in waveguide size RG

The window is designed for mounting between UG 541/U choke flanges, can be pressurized to a maximum pressure differential of 30 psi, and will operate up to 100 kilowatts max.

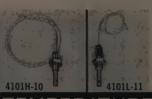
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MAGNETIC AMPLIFIER is plug-in type

Acromag Model 410 magnetic amplifier for signal mixing and summing is a completely self-contained, plug-in magnetic amplifier and power supply weighing less than nine ounces. It is designed for missile guidance, analog con trols, telemetering, and null-bal-ance detectors. It operates directly from standard 115 V 400 cycle power, says Acromag, Inc., Dept. S/A, 22519 Telegraph Road, Detroit 41, Mich.

A unique feature of the new unit is that regulated power sup-plies, bias supplies, and external required; the amplifier is inherently stable. Transimpedance, m is 25,000 ohms; 100 microamperes/ de control current give 2.5 V/d signal energy are required for full control; frequency response is do to 50 cps depending on circuits

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response:

EMPERATURE TRANSDUCERS



Other Specifications:

Calibration accuracy: 0.1-1.0%, depending on temperature range Repeatability and

within calibration accuracy

Resistance at 32 F: 100 ±5 ohms

Nominal temperature-resistance coefficient: 0.0018/°C

Output: 0-5 vdc, when Arnoux 100-ohm TME is used.

The newest line of Arnoux temperature transduct - 100-ohm resistance, 200-millisecond response permits accurate measurement of transient tempe tures such as those in missile and aircraft appli tions. The output signal is 0-5 vdc for as small a sp as 180 F, when Arnoux transistorized TME-1 TME-2 systems or similar equipment is used.

The fluid-immersion transducer (4101L-11), static or moving fluid, is LOX compatible and ave able in two calibration ranges: -302 F to -285

-320 F to +500 F.

The air transducer (4101H-10) is for static to hig velocity gases.

The surface transducer (2101H-15) is for materia of limited area and thickness, and has great mou

Both air and surface types are available in to calibration ranges: -100 F to +500 F. -100 F +1200 F.



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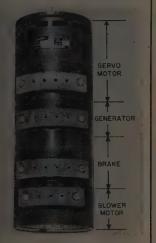
WIREWOUND RESISTORS for printed circuits

Two miniature precision restors are fixed non-inductively vire-wound types sealed in epoxy esin and suitable for operation in mbients up to 125 deg C. The P-2 esistor is ²⁷/₆₄ in long and ¹³/₆₄ in iameter. Termination is by means f two #20 axial leads at one nd of the resistor. Resistances up to 200,000 ohms may be supplied to tolerances as close as 0.1 per ent. The power rating for one er cent tolerance has ben tentively established at 0.3 W for 125 deg C ambient, says Shallross Mfg. Co., Dept. S/A, Colngdale, Pa.

For higher resistance values the 10S resistor is recommended. Restances up to one megohm may e supplied to tolerances as close s 0.1 per cent. The power rating or one per cent tolerance is 0.5 for a 125 deg C ambient.

Circle No. 111 on Reader-Service Card

' ANTENNA DRIVER is multi-purpose device



A servomotor-generator, 28-V de take and cooling blower have been combined into a single unit, the Type 28-MTG-6327-01, by the Dent Mfg. Co., Dept. S/A, Main St., Racine, Wisc. It is esigned for driving antennas and ther applications requiring self-boled drive motors.

The servomotor is a 6000-rpm nit rated at 1/15 hp, and the ch generator has an output of 5 V per 1000 rpm, nominal, with linearity of ±2 per cent up to 000 rpm. The eight-in. unit is railable in any desired combinator of its components.

Circle No. 112 on Reader-Service Card more on next page



for the toughest jobs in electrical circuit protection!

DELTA DART

DELTA DAGGER

BURRERS

Where reliability under extreme service conditions is imperative, airborne electrical systems are protected by Burndy LIMITERS. Here's why —

• Higher tripping temperature provides uniform, predictable performance in spite of ambient temperature variation...pinpoints the moment for

protection.Inherently fail-safe. No mechanical parts.

High interrupting capacity—in ratings from ¹/₄A to 125A.

Indicating LIMITER (above) for dead front panel mounting, ambients up to 300°F, and visible indicating LIMITER (not shown) conform to MIL-F-5372B and MIL-F-5373B.

LIMITERS now in development for 600°F operation and nuclear applications.

Voodoo

35. 2

OMATON DIVISION BURNDY NORWALK, CONNECT.; TORONTO, CANADA; IN EUROPE: ANTWERP, BELGIUM

Circle No. 47 on Reader Service Card in Product Review Section



GSE CONNECTORS

Connectors employed in Ground Support Equipment must be rugged and reliable—and easy to handle under any conditions. Fully meeting these particulars, AMPHENOL'S popular 89 series GSE connectors are being used in many top missile projects.

GSE connectors are completely waterproof and provide dependable service even when submerged in mud, ice or water. An internal rubber gasket in the cable clamp, a type "W" washer at the mating faces and another washer used with panel mounting receptacles provide assured protection.

To facilitate handling in rough weather, coupling rings are extra-long and heavily grooved. Flats are conveniently located for field-servicing with standard open-end wrenches. Caps & chains are provided for all connectors.

AMPHENOL GSE connectors are available in a large number of standard "MS" inserts. Complete catalog data is available upon request.

AMPHENOL ELECTRONICS CORPORATION

chicago 50, illinois

AMPHENOL

BLADE ANTENNA for L-band



This new L-Band Canoga Model 9926 Blade Antenna is designed to operate in the 950 to 1250 megacycle band for use with communication and navigation equipment. This blade antenna provides an all-metal leading edge for maximum strength and erosion resistance. The high aspect ratio with straight or swept-back leading edge insures minimum drag, says Canoga Corp., Dept. S/A, 5955 Sepulveda Blvd., Van Nuys, Calif. The polarization is perpendicu-

The polarization is perpendicular to the mounting surface and the resulting radiation pattern is circular. The Model 9926 has a maximum VSWR of 2.0 over its operating band, is 3.55 in. long, extends 1.70 in. from the skin of the airframe, and weighs 3½ oz.

Circle No 113 on Reader-Service Card

ABSORBER is permanent

This is a completely new Absorber, Type BL-48, for VHF, UHF and microwave ranges, for production and laboratory antenna testing, TV antenna test rooms or aircraft ramp testing of high-power-low-frequency radar. It is made of an inert plastic foam material and can be used for both indoor and outdoor applications. They maintain permanent attenuation characteristics with a frequency range from 40 mc to 35,000 mc, says McMillian Industrial Corp., Dept. S/A, Brownville Ave., Ispwich, Mass.

The absorber is supplied in wedge form, in blocks with a base

The absorber is supplied in wedge form, in blocks with a base of one by two feet and a height of four feet. The individual wedge block weighs five psf and has an average power reflection coefficient of 2½ per cent at normal inci-

Circle No. 114 on Reader-Service Card **←Circle No. 48 on Reader Service Card**

CAPACITORS come flat and round

These new miniature Mylar dielectric capacitors, type XPR and XPF, are designed for use requiring minimum size, high insulation resistance and exceptional capacitance stability. A Mylar polyeste outer wrap affords good protection against moisture, its ends being sealed with a plastic thermosetting resin. Voltage rating is 150 V at 85 deg C, 100 V at 125 deg C, says Astron Corp., Dept. S/A, 255 Grant Ave., E. Newark, N. J. Reliable performance is had over

Reliable performance is had over the entire operating range of 15t to +125 deg C. The new series is available in flat and round con struction.

Circle No. 115 on Reader-Service Card

RECORDING SYSTEMS for six and eight channels



A completely new series of sn and eight channel Direct Writing Oscillographic Recording Systems provide greatly reduced size, improved performance and greater reliability. The new 350 series are packaged in a single mobile vertical cabinet. Frequency response is flat to 100 cps at 10-division peakto-peak amplitude and three decibels down at 120 cps. Linearity is within 0.2 div. over the entire 50 divisions. Hysteresis level is less than 0.2 div., says Industrial Div., Sanborn Co., Dept. S/A 175 Wyman St., Waltham 34 Mass.

The rugged recorder features flush front recorder with paper take-up, transistorized plug-in amplifiers and power supply in 17½ in. of panel space. A limiter ahead of the current feedback power amplifier prevents saturation or cutoff, hence damping is never lost.

Circle No. 116 on Reader-Service Card

SPACE / AFRONAUTICS

POTENTIOMETER has high resolution

completely enclosed, high resion, two-inch-diameter potenneter consists of a small diamcylindrical Kohlrausch resiste element enclosed in a slim,
piece housing molded of glass
forced Alkyd, says DeJursco Corp., Dept. S/A, 45-01
thern Blvd., Long Island City

eries HP-200 can be supplied by or as multiple ganged units. Intings include pivoted servo or e tapped holes. Power dissipais three watts. Mechanical tion is 360 deg. Up to 16 taps be provided, depending on sing. Precious metal brush, color and tap contacts, together a hard gold plated terminals slip rings, are standard design ures.

Circle No. 117 on Reader-Service Card

RECORDER-TRANSCRIBER has 10 kc sampling rate

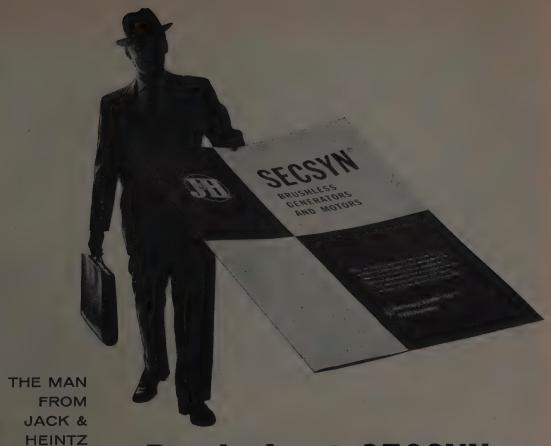


INVITES YOU TO .

An electronic recording and ascribing system that can active and process data "at the enomenal speeds needed in the toe age" was introduced by meapolis Honeywell Regulator. S. Dept. S/A, Davies Labs, itsville, Md. Using magnetic the new Digital Data Record-Transcriber can handle 100 ces of information at once on the of its 100 channels in measurather process variables.

I other process variables.
The system has two main units, recorder, located at or near the as source, absorbs information electrical signals and converts m into digits on magnetic tape, a transcriber, an electronic playak, reproduces the original tape, ects and re-arranges pre-deterned portions into block form for ther automatic processing in a mputer and records the blocks data on a final tape for direct mputer entry. As many as 0,000 bits can be processed in the minutes.

Circle No. 118 on Reader-Service Card more on page 112



Read about SECSYN

... Stationary-Exciter-Coil SYNchronous machines that run longer, faster, hotter and with less maintenance than any comparable-purpose machines now being used for ground support, missiles or aircraft.

This bulletin acquaints you with the design details of these unique machines whose unusual magnetic structure...proved in use... eliminates brushes, rotating windings, rotating rectifiers and other elements that impose limitations on conventional machines.

The SECSYN design can be used as a-c generator, d-c generator, synchronous motor, constant-speed motor, or as a synchro. In all applications, the design offers improvements in size, weight, operating speeds, operating temperatures, service life and maintenance.

SECSYN can be the answer to your most pressing design problem. Send today for this bulletin . . . the Man from Jack & Heintz is available to answer any questions concerning your specific application.



Circle No. 49 on Reader Service Card in Product Review Section

ADDRESS

uary 1959

SOLAR SAILING



EXPANDING THE FRONTIERS OF SPACE



SOLAR SAILING: Space travel with the aid of solar radiation pressure—an area of advanced research at Lockheed. Vehicle would employ a sail that would be raised and lowered in flight.

The artist has depicted Magellan's ship "Trinidad" to symbolize man's great voyages of discovery.

Lockheed Missile Systems Division is engaged in all fields of missile and space technology—from concept to operation.

Advanced research and development programs include—man in space; space communications; electronics; ionic propulsion; nuclear and solar propulsion; magnetohydrodynamics; computer development; oceanography; flight sciences; materials and processes; human engineering; electromagnetic wave propagation and radiation; and operations research and analysis.

The successful completion of programs such as these not only encompasses the sum of man's knowledge in many fields, but requires a bold and imaginative approach in areas where only theory now exists.

The Missile Systems Division programs reach far into the future.

It is a rewarding future which men of outstanding talent and inquiring mind are invited to share. Write: Research and Development Staff, Dept. A-16, 962 W. El Camino Real, Sunnyvale, California.

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CHNOLOGY

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Circle No. 81 on Reader Service Card in Product Review Section

THIN VERSATILE DUCTILE NETIC CO-NETIC FOIL

Aids Miniaturization & Reliability . . . Cuts to Any Shape with Scissors . . . Wraps like Tape . . . Shields Both High & Low Frequencies at Low Intensities

One or more layers are readily applied and their effects observed, permitting engineer to determine optimum location and number of layers needed for adequate shielding. Can be trimmed to any dimension or outline with ordinary scissors. Easily formed by hand. Dead soft; does not spring back into original position after forming. Aids miniaturization by reducing shielding bulk and by making possible crowding components closely together.

15" wide Co-Netic foil is non-

15" wide Co-Netic foil is nonshock sensitive, non-retentive, does not require periodic anneal-

does not require periodic annealing and attenuates low level fields. When used with 19%" wide Netic foil, higher ratios of attenuation for relatively intense fields are achieved. Both foils are immediately available from stock in .004" thickness in continuous rolls up to any length desired.

These versatile light weight time-saving foils open up a whole new shielding concept in laboratory, air borne, electronic, electrical and shielded room applications. Write for complete details today.

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Bellevue Hotel, San Francisco • Flamingo Hotel, Phoenix

Circle No. 83 on Reader-Service Card

LINEAR AMPLIFIER for low signal levels

Model 672A, AID Linear Amplifier is built to ORNL Spec. Q1326. It amplifies pulses of low signal level from radiation detectors such as ionization chambers proportional counters and scintillation counters to usable levels that can function in counting systems such as scalers, ratemeters and pulse height analyzers, says Victoreen Instrument Co., Dept. S/A. 5806 Hough Ave., Cleveland 3. Ohio.

Input sensitivity is from 200 microvolts to one mv. Gain, depending on bandwidth switch position, is 12,000 with switch set at 0.1 mc; 6000 at 0.5 mc; and 2000 at 2.0 mc (delay line). Output at high amplifier output is five V into 1000 ohms; at low amplifier output, five V into 150 ohms. Linearity is better than one per cent from nine to 90 V output; stability better than one per cent after 24-hour warm-up.

Circle No. 119 on Reader-Service Card

CAPACITORS have tantalum film



A series of solid electrolyte capacitors, Solitan, designed for transistor application in computer and military circuits contain absolutely no liquid electrolyte, residue or moisture of any kind. Tantalum oxide film on the anode is the means for producing this solid-state capacitor, says Cornell-Dubilier Electric Corp., Dept. S/A, So. Plainfield, N. J.

Advantages of these capacitors are: consilerably smaller sizes due to better space-saving factors; temperature range --80 to +85 deg C; better stability of capacitance with time and temperature; treedom from corrosion and effects of vibration; freedom from deteriorating effects of extended shelf-life and service aging.

Circle No. 120 on Reader-Service Card

more on page 115

advancedtechnique avionics



AVION

FOREMOST AVIONICS

Circle No. 84 on











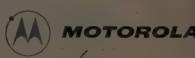


Working headgear for Motorola design engineers

From the arctic to the tropics...from jet altitudes to submarine depths...Motorola design engineers wear headgear matching the varied military applications of Motorola electronics. Working side-by-side with the Army, Navy and Air Force end user, these engineers provide technical assistance and, in return, gain first hand operational knowledge. The result—an understanding of user problems reflected in the design maturity of Motorola's military electronics systems.

And as for the many hats, those shown above have been worn on these representative programs:

AIntegrated Battlefield Communication B Supersonic Rendezvous and Station Keeping C High-speed
Tank Range Finder D Test Range Microwave Relay System E Ballistic Missile Radio Inertial Guidance
System F Shipboard Surface-to-air Missile Guidance G Member, Army Scientific Advisory Panel
H DEW Line Air Defense Radar I Strategic Bombing Radars J Amphibious Landing System
K Shipboard Air Defense Data System L Tactical Drone Guidance. For data on how Motorola's military
experience can be applied to your problem...or for detailed information on engineering career opportunities... write to: Motorola, Inc., Military Electronics Division, Dept. A, 8201 East McDowell Road,
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Working laboratories for Motorola engineers

Six Military Electronics Division plants in three locations...over 550,000 square feet of engineering and production space devoted exclusively to the design, development and manufacture of advanced military equipment.



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PRODUCT REVIEW

DATA CONVERTERS are flexible

Reversible analog-to-digital and digital-to-analog converters, and alarm limit monitors are included in a line of data converters that is based on assembly of a number of modular units. Over 50 standard models can be assembled from the basic plug-in building blocks, says Epsco, Inc., Dept. S/A., 588 Commonwealth Ave., Boston 15, Mass.

The transistorized Transicon Datracs, require no adjustments and use high-power outputs to drive external circuits directly. The units are designed for expansion, and eight-bit converters can be increased to ten, 12 or 14 bits. Codes may be changed from binary to binary coded decimal. The analog-to-digital Datrac can provide up to 25,000 conversions per sec, and the digital-to-analog unit handles up to 100,000 conversions per sec. Alarm limit monitors have a conversion rate of 200,000 conversions per sec.

Circle No. 121 on Reader-Service Card

COMPACT TESTER measures beta, ico, ibco



An accurate system that determines the basic parameters of a transistor has been designed into this new unit. Beta and Ico are examined as well as the Figure of Merit of a transistor, Ibco. The test set accurately measures Beta, Ico and Ibco of both PNP and NPN type transistors at any desired operating level. Component matching accuracy of about two, per cent is provided. Simplicity of the control panel and test procedure makes the tester compatible with the requirements of high rate quality control testing, says Armour Electronics, Inc., Dept. S/A, 10800 Ventura Blvd., Studio City, Calif.

Modern circuit techniques are used throughout the unit which weighs only 4½ lb and is enclosed in a cabinet five inches high, 11 in. wide and nine inches deep. All electrical power is derived from four mercury cells.

Circle No. 122 on Reader-Service Card more on next page Circle No. 86 on Reader Service



Sliding linear motions are nearly always troublesome. Thousands of progressive engineers and designers have solved this problem by application of BALL BUSHINGS on guide rods, reciprocating shafts, push-pull actions, or for support of any mechanism that is moved or shifted in a straight line.

Improve your product! Up-date your design and performance with Thomson BALL BUSHINGS!

LOW FRICTION · ZERO SHAKE OR PLA ELIMINATE BINDING AND CHATTER SOLVE SLIDING LUBRICATION PROBLEM LONG LIFE · LASTING ALIGNMENT

The various types cover a shaft diameter range of \%" to 4". Small sizes available in Stainless Steel. Write for literature and name of our representative in your city.

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Also Manufacturers of NYLINED Bearings Sleeve Bearing

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TRANSISTOR! are 60 and 100 v

Two new diffused-base silicon medium power transistors featur-four-watts at 25 deg C case temperature and one watt at 150 deg C case temperature dissipation ratings. Both new N-P-N- transistor-feature a typical saturation resistance of 20 ohms at 25 deg C plus an operating range of -65 deg to +200 deg C and are temperature stabilized at 215 deg C, says Texas Instruments, Inc. Dept. S/A, P. O. Box 312, Dallas Texas.

The 2N497 is a 60 V transistor for use with the 28 V power supplies employed in military aircraft. The 2N498 is a 100 V device for use in higher voltage applications such as servo amplifiers and regulated power supplies for missile applications.

Circle No. 123 on Reader-Service Card

TAPE HANDLER for digital computers



Increased tape speed of 75 ips and 40 start-stop cps in the new Model 101 digital magnetic tape handler comes in as few as six and as many as 20 tracks. Choice of tape speeds range from two to 75 ips. Complete remote control of start, stop, reverse and speed change functions is furnished as well as single or dual tape speeds as required, says Digitronics Corp., Dept. S/A, Albertson Ave., Albertson, N.Y.

Two solenoids start the tape in less than five milliseconds. Equally fast stops are obtained by dual braking solenoids that press the tape against stationary capstans as it is released from the driving capstans. On a 19 in RTMA standard relay rack it uses 24½ in and a 9 in depth.

Circle No. 124 on Reader-Service Card



FROM AN ORIGINAL PAINTING FOR CECO BY R. T. HANDVILLE

Contributing to superb performance

The Bell H-40, newest Army Utility Helicopter, is powered by a Lycoming T-53 gas turbine engine with *complete unitized fuel control system* engineered and built by Chandler-Evans.

Products, too, are "known by the company they keep", and CECO is proud to be airborne with many of the latest and finest military and commercial aircraft.



CECO's Model TA-1, a complete unitized fuel control system, is shown in position on cutaway model of Lycoming's T-53 gas turbine engine. It incorporates a dual-element main fuel supply pump, free power turbine governor, automatic altitude and temperature compensation, and integral emergency fuel metering system.



SYSTEMS CONTROLS

CHANDLER-EVANS

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Interesting, informative literature on many CECO products is yours for the asking. Please address your request to Department 18.

Circle No. 93 on Reader Service Card in Product Review Section

KLYSTRON SOCKET saves assembly time





Assembly time is said to be educed and rowork eliminated by ne use of a new moulded chassis apunted octal socket for the JAN-ZRP-2 K45 thermally tuned klytron tube. Self-aligning gold dated contacts are flexibly apunted in slightly over size orices of a glass filled dially abthalate socket body. The new ocket provides accurate alignment and probe penetration in the vave guide mount which avoids ny possibility of missmatch. says Slobe Electrical Mfg. Co., Dept. S/A, 1729 W. 134th St., Garlena, Calif.

Design of the socket eliminates he need for insulated bushings and any contact shorting to the vave guide tube mount. Contact abs are easily accessible for fast ecurate circuit assembly.

Circle No. 125 on Reader-Service Card

POWER SUPPLY is transistorized

This REL-203 Power Supply is a completely solid-stat unit employing the latest techniques of transistor switching to attain high efficiency and reliability, says Rheem Mfg. Co., Electronics Div., Dept. S/A, 7777 Industry Ave., Rivera, Calif. An r-f Filter is included in the 28 V input to prevent interference being conducted to radio equipment. The typical ref voltage level is down 50 db at 150 de 14 ke and down 80 db at 150 de 150 d

14 kc and down 80 db at 150 dc.

The unit weighs two lb. It has un input voltage of 24 to 30 V dc, un input current of 0.2 to 3.2 amps dc depending on load. Maximum power output is 80 W. (no heat ink -75 to +158 deg F.) Efficiency at full load is 88 per cent.

Circle No. 126 on Reader-Service Card

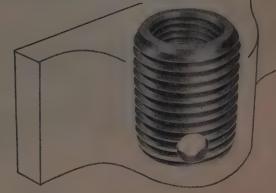
The TAP-LOK® threaded insert

taps its own thread

locks itself in...

in a

single operation!



TAP-Lok inserts provide strong wear-resistant threads in relatively soft machinable materials (wood, plastic, aluminum, etc.) . . . as well as in *harder* materials where repeated assembly makes excessive thread wear a problem.

Unlike ordinary threaded inserts, however, TAP-LOK inserts cut their own threads in the parent material. Thus, no additional assembly time is required with a TAP-LOK insert.

Its installed cost is the lowest of any threaded insert.

The locking action is achieved in this way: the tapered pilot section carrying the thread cutting edges is followed by the full diameter threads which force their way into the parent material. Once installed, a TAP-LOK insert is permanently locked in place.

Available in the types shown below, TAP-LOK inserts can be used wherever threaded inserts are required. Write today for complete information.



Slotted:—Full V-form external threads provide maximum locking.torque; permit wide choice of mating hole sizes. Recommended for soft aluminum, zinc die castings, sand castings, plastics. Class 2B internal thread — MIL-MS 35914.



H-Series:— A heavy walled insert with truncated root external thread and three-hole cutting edges for hard-to-tap higher-strength materials and to meet MIL and other spees calling for Class 38 thread fit for gag-ing after lebt-like.



W-Series: Coarse-pitch external-thread offers maximum strength; permits installation in small wooden sections without splitting. For furniture, cabinets and other wooden parts where strong, permanent threads are required.



P-Series:—This Tap-Lok insert was designed to eliminate thread wear and renew damaged threads in spark plug sockets in aluminum cylinder heads. It is available from stock for standard plug sizes to meet most needs.



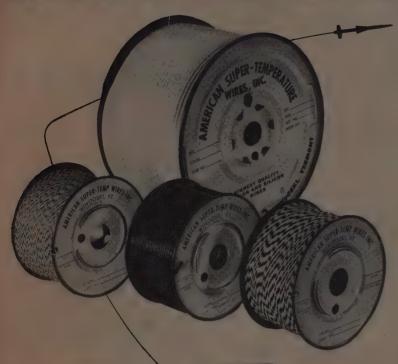
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Catalog

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Agents in principal electronic manufacturing areas

Circle No. 87 on Reader Service Card in Product Review Section

PRODUCT REVIEW

CONVERTER for analog to digital



This new type 525 Encoder (analog to digital converter) is an all-electronic converter providing precise and dependable conversion of analog input data to digital output voltages. It makes possible the use of versatile, noise-free digital transmission and computing techniques to a greater degree of accuracy than heretofore possible, says Avion Div., ACF Industries, Inc., Dept. S/A, 800 No. Pitt St., Alexandria, Va.

When used in combination with an Avion Model 1002 Decoder, it is possible to solve problems in telemetering analog computation, digital computation, PCM, and many other data processing applications.

Circle No. 127 on Reader-Service Card

DIGITAL SYSTEMS are all-transistorized



This all-transistorized Digital Systems, include a digital voltmeter, control unit, ratiometer, ohmeter, ac converter, master scanner and printer units. The voltmeter features polarity symbols, decimal point and ac symbol. Bridge linearity is greater than 0.01 per cent and bridge resistance is 50 K ohms, says Cubic Corp., Dept. S/A, San Diego, Calif. Automatic features include po-

Automatic features include polarity and voltage ranges. Accuracy is greater than 0.01 per cent. Input terminals are supplied at either the front panel or the rear connector panel. All units are of standard size; 17 in. wide, 3½ in. high and 11¾ in. deep.

Circle No. 128 on Reader-Service Card



FIGO... YOUR SPECIFYING and BUYING GUIDE of SHEET METAL PRODUCTS

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BUD RADIO CORPORATION

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Circle No. 88 on Reader Service Card in Product Review Section

SPACE/AFRONALITIO

INFRARED DOME made in one piece



Domes useful in transmitting trated radiation to detection uipment are being produced in the piece, according to Servo orp. of America, Dept. S/A, 0-20 Jericho Turnpike, New yde Park, N.Y. The Servofrax omes, which are made of arsenic sulphide glass, can withstand gh acceleration and subzero temeratures.

Their negative lens effect can metimes be used effectively as Maksutov corrector for conntric optical systems. The domes e available in diameters from ut to 12 in and in any spherical ction up to and including 180 are.

Circle No. 129 on Reader-Service Card

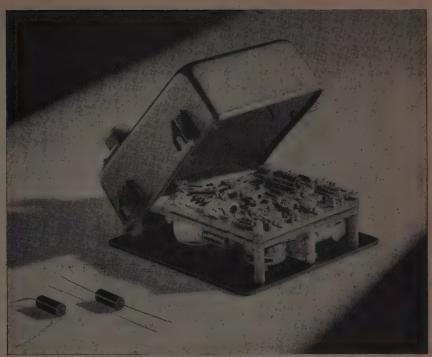
POTENTIOMETER has 2% resolution



A new 0.6 W wirewound minture precision potentiometer, fodel DM, is one-half inch in iameter, 0.1 ounce in weight, and available in stock values from m ohms to 50K ohms and feaures a resolution of 0.2 per cent, tys B-H Electronics, Dept. S/A, O. Box 25124, Los Angeles 25,

The unit requires no mounting ardware and mounts directly to hassis or printed circuit either lanually or by machine. Tinned adds (0.032 diameter) are E.I.A. addular spaced. Shock tested to 0C, three planes. No electrical pise under 25C, three planes, ther features include 320 deg rotton, range from -55 to +150 eg C and retention of setting uring vibration and shock.

Circle No. 130 on Reader-Service Card more on next page Airborne
Miniaturized
Mylar*
Capacitors
help reduce
weight and
bulk in
electronic
packages



Amplifier for oil temperature servo control system utilizes Airborne miniaturized "Mylar" capacitors to save weight and insure high reliability. Entire system, including rotary actuator, is Airborne produced and is designed to meet specifications MIL-E-5272A, MIL-E-5400A, etc. Environmental requirements include -65 to +200°F. and up to 60,000 ft. altitude.

Developed originally for motor start and run purposes, Airborne miniaturized "Mylar" capacitors are currently finding increasing application in electronic circuits where small size, light weight, and high reliability are of paramount importance.

Typical of such applications is one of our *own* servo control amplifiers, shown above. Used as a component of an Airborne-designed oil temperature control system for high performance aircraft, the amplifier consists of a .1% precision resistance bridge, stable feedback transistor amplifier, reference oscillator, phase demodulator, and relay output amplifier. Production units employ printed circuitry.

Two of Airborne's miniaturized

"Mylar" capacitors are utilized in this particular amplifier — a .1 mfd unit for tuning in the reference oscillator section of the amplifier and a .02 mfd unit for phase shift correction in the stable feedback transistor amplifier section. Capacitors are epoxy encased and are designed to meet or exceed Government specification MIL-C-25A.

Wound of thin metallized "Mylar" film, Airborne miniaturized capacitors are rated up to 600 v d-c, 330 v a-c and have an operating temperature range of -75 to +300°F. At 300°F they will withstand 120% rated voltage for 250 hr.

Write, phone or wire for more information on Airborne special design miniaturized "Mylar" capaci-

tors. Inquiries are also invited on complete electromechanical control systems.

*Du Pont's trademark for its polyester film

TYPICAL SPECIAL CAPACITORS



E-8109 Style

E-8104 Style

E-8107 Style

CATALOG 57B

Gives detailed information on Airborne special design miniaturized "Mylar" capacitors and Airborne R.F. filters. Write for copy.



LINEATOR® • ROTORAC® • TRIM TROL® • ROTORETTE® • ANGLGear® • ROTOLOK



AIRBORNE ACCESSORIES CORPORATION

HILLSIDE 5, NEW JERSEY

Represented in Canada by: WINNETT BOYD LIMITED • 745 Mt. Pleasant Rd., Toronto 12, Ont.

Circle No. 91 on Reader Service Card in Product Review Section

anuary 1959

POWER AMPLIFIER features high gain



The Model REL-10 power amplifier, which can be used with most FM transmitters, delivers from ten to 100 W of r-f with two W of drive, says Rheem Mfg. Co., Dept. S/A, 777 Industry Ave., Rivera, Calif. The 5½ x 3½ x 3-in. unit boosts signal power signifi-cantly in the 215 to 260-mc tele-

metering band.

It is very reliable in adverse environments of shock, vibration, and temperature, and it has a self-contained cooling system. It is suitable for missile-borne instrumentation systems.

Circle No. 131 on Reader-Service Card

COUNTERS for FM/FM telemetering



Simplified telemetering measurements that combine high resolution and short measurement time may be made with Model 2503 FM/FM Telemetering Counters. Channels are push-button selected and the visually displayed count is normalized to read deviation from midband directly in per cent of the midband period. A resolution by one part in 10⁵ is achieved with a one-second measurement period. The period can be reduced by front panel control to 0.1 sec, says Dynac, Inc., Dept. S/A, 395 Page Mill Rd., Palo Alto, Calif.

The unit has five decimal places and uses an internal countries.

and uses an internal crystal-controlled 100 kc time base or an external time base. It also measures frequency and ratio either directly or normalized. Two models, one for field and one for laboratory, are available.

Circle No. 132 on Reader-Service Card

MICRO SWITCH Precision

These 17 switches represent the widest for aircraft, missile, rocket, launcher,



4CE1—This weather-proof, corrosion-resistant switch is used as an indicator or limit switch on missile shelter, erecting and launching equipment. Has SPDT contact arrangement. Rating: 10a. 125vac; 5a. 250 vac. Operating force 6-10 lbs. Data short 151



1RS1 — High reliability telemetering switch for missile program sequencing, Hermetically sealed, Has rotary solenoid and 24 SPDT goldcontact "SM" switches, 74 terminals. Quick disconnect plug, Data sheet 151.



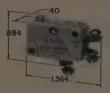
1EN42-R—A special design for high shock applications. Used on shipboard launching equipment. Meets requirements of U. S. Navy Bureau of Ordnance. Contains two SPDT switches. Quick disconnect connector. Ice and mud scraper on plunger. Operating force 6-12 bbs. Rating at 28vdc: inrush 24a., resistive 4a., inductive 3a., motor 4a. Data sheet 151.



54PB67-T2 — Space-saving lighted pushbutton assem-bly. Flexible indicably. Flexible indica-tion provided by indicator lamp and two-level position of button. Affinate action (on-off) of the action (on-on) of the switch contacts is provided. Four SPDT subminiature switches. All switch and lamp terminals are isolated. Data



1SE1-Smallest sealed environment-free switch. SPDT. Rating at 28vdc: ind. 3a., res. 4a., inrush 15a. Leaf and rollerleaf auxiliary actuators available. Catalog 77.



V3-1301—Operates reliably at 600°F. For jets, missiles, electronic gear. SPDT. 30vdc rating: ind. 10a., res. 10a., motor 6a. Cat. 74.



Hermetically sealed, high temsealed, high temperature DPDT
switch for reliable
use to 600°F. All
seals in accordance with Par.
5.2.2, MIL-E52.72 A. I ce
scraper ring on
actuator. Rating:
28 vdc. res. 52. 28vdc—res. 5a. ind. 2a. Operating force 6-12 lbs. Data sheet 122.



44EN1-6 (MS-24420-4)-

Rotary actuator adjustable through 360°. One 4-circuit basic switch. Rating: 28vdc-inrush 30a.; res. 15a.; ind. 10a. Operating torque—10 in. lbs. Catalog 77.



3EP2-E-Sealed enclosure. 3EP2-E—Sealed enclosure.
Rugged, two-hole side mounting. Two high capacity SPDT
basic switches. Sealed connector receptacle and roller
lever actuator. Rating: 28vdc
—inrush 36a.; res. 10a.; ind.
6a. Operating force 4-6 lbs.
Catalog 77. 201LS1-

Heavy duty, sealed "Plug-in Limit" switch. Can be replaced on the job in 20 seconds. Switch enclosure, with actuator posiactuator, positioned by dow-els and secured by two No. 10 screws. Available in six actuator designs.
Used in ground equipment as limit or indica-tor switch. Catalog 84.



ange of precision switches available narine and ordnance applications

Each of the switches illustrated on these pages represents a different series of MICRO SWITCH Precision Switches . . . Each switch has been designed to high reliability standards for a particular purpose . . . Each series includes many variations—sizes, weights, contact arrangements, electrical characteristics, housings, actuators, sealing, temperature, shock and mechanical characteristics . . . For example, the switches in which "EN" is a part of the catalog listing constitute a series in which nearly 100 different switches and assemblies are available ... No other source of precision switches offers as broad a line . . . Furthermore, no other can offer the services of as large a staff of Application Engineers—in branch offices at strategic points throughout the country. Consult the yellow pages.

MICRO SWITCH ... FREEPORT, ILLINOIS A division of Honeywell

In Canada: Honeywell Controls, Ltd., Toronto 17, Ontario



1EN75-R-High-impact shock 1EN75-R—High-impact shock switch for use on launching platforms, near jet after-burners, etc. Heavy duty push plunger with ice-mud scraper. Operating force 6-12 lbs., release force 4 lbs. Environment and immersion proof (MIL-S-901B, Navy and MIL-E-5272A, Proc. 1). Contains two SPDT switches. Available with straight (shown) or angle connector plug. Data sheet 152.



5ET1-6—Compact, magnetically-held toggle switch has a built-in 28vdc solenoid. Works as a momentary action toggle switch, but is converted to a maintained contact switch when solenoid is energized. Permits remote electrical release of the lever, thus eliminating auxiliary relays, etc. Toggle lever can be manually over-ridden. Environment and immersion-proof en-Environment and immersion-proof enclosure (MIL-E-5272A Proc.1). SPDT basic switch. 28vdc. rating: inrush 24a., resistive 4a., inductive 3a., motor 4a. Data sheet 121.





4TL1-1A—Same features as 2TL1-1 except toggle lever has safety feature which requires pull to un-lock and move the lever. (Available in a variety of locking arrangements.) Knob rangements.) Knob facilitates operation with gloves. 4-pole DT circuitry. Rating same as 2TL1-1. Data sheet 142.

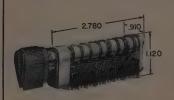


Catalog 77.

2TL1-1—Integral, stepped terminal de-sign. High impact strength, arc resist-ant case. Silicone seal on bushing and self-sealing sealant seir-searing searant between cover and case. 2-pole DT cir-cuitry. Rating at 28 vdc: res. 20a., lamp 7a., ind. 15a. Avail-able in variety of toggle positions.

Data sheet 139.





28vdc rating: ind. 10a., res. 25a.

17AS8-Compact rotary selector switch handles eight different circuits, requires only 1½ inch diameter panel area. Switching sequence is infinitely variable. Data sheet 86. PRODUCT REVIEW

VIDEO TERMINATION for 75-ohm coax lines

A new Video Termination, Type RF-300, for 75 ohm coaxial transmission lines is announced by Daven Co., Dept. S/A, Livingston, N. J.

Some of the unit's characteristics are: a turret type lug at the rear of the termination provides a connection point for an oscilloscope; negligible reactance (less than two deg phase shift) up to 250 MC; deposited carbon resistor rated at 1/2 W. Accuracy one per cent. RF resistance at 100 MC is within -1 per cent of the dc value. Terminations are housed in standard UHF type connectors.

Circle No. 133 on Reader-Service Card

INDICATOR TUBE in large size



The Jumbo Nixie, Type BD-307, is a three-in-dia indicator tube with a numerical display that can be seen at distance of over 150 ft, says Electronic Tube Div., Burroughs Corp., Dept. S/A, Plainfield, N. J. It is designed for

low power and ruggedness.

The all-electronic inline indicator contains the numerals zero through nine and utilizes a common anode for circuit simplicity.

The desired numeral lights when a voltage is applied between it and the anode.

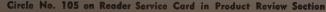
Circle No. 134 on Reader-Service Card

WAVEGUIDE FILTERS have sharp cutoffs

This new line of r-f waveguide filters features very sharp cutoffs and low insertion loss in the passbands, says Microphase Corp., Dept. S/A, Box 1166, Greenwich, Conn. One unit in the line, a lowpass type, cuts off at five kmc/s, has a maximum loss of 0.75 db, and a minimum attenuation of 60 db from 1.08f. out to 12 kmc/s.

The unit, which handles up to 100 W of power, has a UG-149A/U flange at one end and a UG-148B/U at the other. Units in the S through X bands can per-

> Circle No. 135 on Reader-Service Card more on next page



4TP1-1-Rocker-actuated key switch combines the advan-tages of both pushbutton and toggle manual control. Key is

toggle manual control. Key is flat, like a pushbutton and can be engraved, but has toggle feature of visible position. Also enhances panel design and provides edge-light indication. Construction of case is rugged and sealed, with integral, stepped terminal design. 4-pole DT circuitry (others available). 28vdc rating: res. 20a., lamp 7a., ind. 15a. Data sheet 141.



KEARFOTT COMPANY, INC., LITTLE FALLS, N. J.
A Subsidiary of General Precision Equipment Corporation
Sales and Engineering Offices: 1378 Main Ave., Clifton, N. J.
Midwest Office: 23 W. Calendar Ave., La Grange, Ill. South Central Office: 6211 Denton Drive, Dallas, Texas
West Coast Office: 253 N. Vinedo Avenue, Pasadena, Calif.

Circle No. 94 on Reader Service Card in Product Review Section

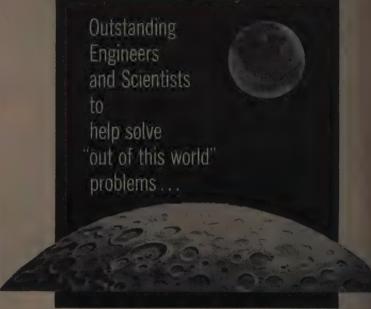
RESISTORS are micro-miniature

High resistance, very small size, And precision tolerances are featured in two micro-miniature, wirewound resistors made by Dale Products, Inc., Dept. S/A, Columbus, Neb. The smaller resistor, the RS-1A, is ¹³/₅₂ x ⁹/₅₂ in, and the other the RS-1B, 176 x and the other, the RS-1B, 17/32 x

Both resistors are coated with a silicone covering for protection from thermal shock, humidity, salt

air, and similar environmental conditions. The devices are available in tolerances of 0.05, 0.1, 0.25 0.5, 1, and 3 per cent. The RS-1 ohms, and the RS-IB, a range of 1-7500 ohms, and the RS-IB, a range of 1-10,000 ohms, depending on tolerances. The smaller unit is rated at one W at 25 deg C and the larger at one W at 50 deg C.

Circle No. 136 on Reader-Service Card



Today artificial satellites orbit in space. Missiles can span continents. Conservative scientists calmly talk of landing on the moon.

Just as they have contributed to other aeronautical sciences, the various divisions of United Aircraft Corporation have made significant contributions to these new fields of missiles and space

Recently the outstanding scientists and engineers who had specialized in missiles, missile guidance and space penetration problems in each division were brought together to focus their combined skills on advanced concepts and systems. A new Division was created . . . the Missiles & Space Systems Division.

This division is only weeks old. It is in an explosive growth period. Yet it has a built-in stability factor . . . the advantages of the brainpower, the "know-how", the financial resources and the unique facilities of a billion-dollar corporation that is already pre-

eminent in aeronautics

This combination of newness and stability should be significant to every alert engineer or scientist. It should suggest a unique opportunity to demonstrate ability and win the advancement and other rewards that ability deserves.

If you are looking for opportunity, we suggest that you contact us immediately.

Positions are available at all levels in . . .

ELECTRONICS: Guidance, Radar, Countermeasures, Computers, Telemetr SYSTEMS ANALYSIS - SYSTEMS INTEGRATION - MILITARY REQUIREMENT RELIABILITY - GROUND SUPPORT - SPACE TECHNOLOGY: Astrophysics, Ast nautics - AERONAUTICS: Preliminary Design, Performance, Aerodynami Structures, Propulsion.

Please send your complete resume, including salary requirements, to Mr. John C. North.

MISSILES & SPACE SYSTEMS

DIVISION OF UNITED AIRCRAFT CORPORATION . EAST HARTFORD 8, CONN.



ACTUAL SIZE

CLIFTON PRECISION PRODUCTS CO., INC. Clifton Heights, Pa.

IN SILICONES OR RUBBER WHAT'LL YOU HAVE?



SPONGE? Chemically blown, closed-cell silicone sponge is available in fine, medium, and firm density; skin thickness can be varied to suit. Excellent for gaskets, seals, shock mounts, light duty press pads, die cut parts.

SHEET? Solid sheet is produced from 1/32 to 1 in. thick, 20 to 80 durometer, in compounds to meet all AMS, ASTM, and military specifications. For gaskets, seals, bushings, die cut parts, diaphragms, heavy duty press pads. Laminated press pads and throw sheets are also available.

MOLDED PARTS? Diaphragms, pipe coupling gaskets, regulator parts, medical kit liners, bumpers, flexible coupling discs, spring inserts, aircraft test sleeves, and many other components are produced to rigid specifications.

Hewitt-Robins, prominent in development of aircraft refueling hose, makes many silicone and rubber products for the aircraft and missile industries. These components are fabricated using all elastomers, including silicones by themselves or with various fabric or metal reinforcements.

Specialists in our Aircraft Products Department can help you put today's new compounds to best use in aircraft, missiles, and rockets. For information, service, or your copy of comprehensive Product Bulletins, contact your local H-R representative, or Hewitt-Robins, Stamford, Connecticut.

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Amsterdam, Holland • Johannesburg, South Africa • London, England • Montreal, Canada • Paris, France

Circle No. 99 on Reader Service Card in Product Review Section



data review

ANALOG-DIGITAL CONVERTER— Its new shaft analog-to-digital converter is described in Brochure D1.03 by Data Instruments Div., Dept. S/A, Telecomputing Corp., 12838 Saticoy St., North Hollywood, Calif. Typical pulse output waveforms are shown.

Circle No. 137 on Reader-Service Card

TAPE PROCESSING—Special tape processing equipment is described in a six-page brochure by Digitronics Corp., Dept. S/A, Albertson, N. Y. Units are available singly or in combinations.

Circle No. 138 on Reader-Service Card

CAPACITOR—Its high-Q voltagevariable Semicap capacitor is described in Bulletin SR-205 by International Rectifier Corp., Dept. S/A, El Segundo, Calif. Design and performance specs are given.

Circle No. 139 on Reader-Service Card

MAGNETIC SHIELDING — Manual 101-122 summarizes design and fabrication techniques for Netic and Co-Netic magnetic shielding, made by Magnetic Shield Div., Perfection Mica Co., Dept. S/A, 1322 N, Elston Ave., Chicago 22, Ill. Test techniques for shielding evaluation are also covered.

Circle No. 140 on Reader-Service Card

PULSE TRANSFORMERS—A series of encapsulated miniature pulse transformers is described in a technical bulletin by Technitrol Engrg. Co., Dept. S/A, E. Allegheny Ave., Philadelphia 24, Pa. The units are wound on high-permeability ferromagnetic cores.

Circle No. 141 on Reader-Service Card

VOLTAGE REFERENCE SOURCES-

One binary and two decade models of its voltage reference sources are described by Epsco, Inc., Dept. S/A, 588 Commonwealth Ave., Boston, Mass., in four-page Bulletin 95806.

Circle No. 142 on Reader-Service Card

RESISTORS—Catalog Gr-20 tells of bobbinless precision wire-wound resistors and is available from General Transistor Corp., Dept. S/A, 91-27 138th Place, Jamaica 35, N. Y.

Circle No. 143 on Reader-Service Card

SEMICONDUCTORS — Important of impurities in semiconductor alloys is one of the things illustrated in the technical bulleti offered by Accurate Specialties Co. Dept. S/A, 37-11 57th St., Wood side 77, N. Y.

Circle No. 144 on Reader-Service Ca

POTENTIOMETERS — Data Shet 1362 tells about the Helipot Serie 5200 all metal, 1½6-in.-dia servo-mounting precision potent ometers. It is available from Helipot Div., Beckman Instruments Inc., Dept. S/A, Fullerton, Calif.

Circle No. 145 on Reader-Service Car

RADAR SYSTEMS—The radar at tenna system designed for ultra precision tracking of the Tale missile is described in a bulleti published by Special Product Div., I-T-E Circuit Breaker Co Dept. S/A, 601 E. Erie Ave., Phila 34, Pa. This 4-page bulletin describes the designing, fabricatio and testing of a prototype antenn system to specs within 13 month.

Circle No. 146 on Reader-Service Ca

TRANSISTOR DIGITAL CIRCUIT

An 8-page booklet is bein offered by Epsco Inc., Dept. S/A 558 Commonwealth Ave., Bosto 15, Mass. It concerns the transiste digital circuits series 100.

Circle No. 147 on Reader-Service Car-

WAVE TUBES—A bulletin concerning backward and forwar wave amplifiers and backward wave oscillators has been pullished by Huggins Laboratorie Inc., Dept. S/A, 711 Hamilto Ave., Menlo Park, Calif.

Circle No. 148 on Reader-Service Ca

TEST EQUIPMENT — An 84-pay catalog describing its expanded lin of microwave and UHF test equipment and components is being published by The Narda Microwave Corp., Dept. S/A, 118-16 Herricks Road, Mineola, N. Y.

Circle No. 149 on Reader-Service Ca

FILTERS—Burnell & Co., Dep S/A, 10 Pelham Parkway, Pelhar N. Y., is publishing Catalog 10 which stresses the importance toroids, filters and related network in military and industrial communications.

Circle No. 150 on Reader-Service Ca

ctronics

HANICAL STOPS—Data sheet 80-10 gives details on the menical devices made by Keafott Inc., Dept. S/A, 1378 Main, Clifton, N. J., for stopping tion at predetermined intervals. Circle No. 151 on Reader-Service Card

LURE PREDICTION — Bulletin I.1 describes a test procedure lab, production, and field mainnee developed by Associated earch, Inc., Dept, S/A, 3777 Belmont Ave., Chicago 18, Ill., a wide variety of electronic and tric equipment to forecast fail-

Circle No. 152 on Reader-Service Card

AR POLARIZATION — Report 8 describes the circular polarion feed designed by I-T-E Cir-Breaker Co., Dept. S/A, 601. Frie Ave., Philadelphia 34, Pa., airport radars.

Circle No. 153 on Reader-Service Card

erference meters — Five etins—NM-10A, -20B, -30A, A, and -50A—from Stoddart Airt Radio Co., Inc., Dept. S/A, 4 Santa Monica Blvd., Hollydd 38, Calif., gives specs and lication data on radio interference and field intensity measuring for 30 cps to 1000 mc.

Circle No. 154 on Reader-Service Card

ULATED WIRE—A technical etin is available from Hitempes, Inc., Dept. S/A, 1200 mes Dr., Westbury, N. Y., on company's Ceramatemp insudwire for continuous operaat 1000 deg. F.

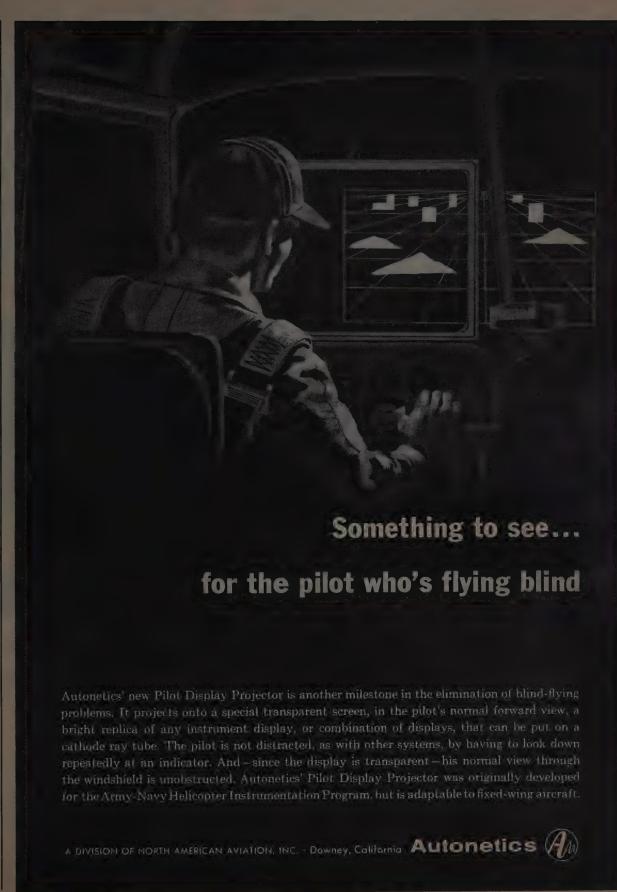
Circle No. 155 on Reader-Service Card

ritude Reference—Its twoo, all-attitude master reference, sified for several recent fighters, escribed by Lear, Grand Rapids , Dept. S/A, 110 Ionia Ave., V., Grand Rapids 2, Mich., in a catalog sheet.

Circle No. 156 on Reader-Service Card

TROWAVE PARTS — Catalog T8 has been issued by T.E.M., pept. S/A, 71 Okner Pkwy., ingston, N. J., describing signal splers, monitor tees, and detect and blocks for the microte region.

e region. Circle No. 157 on Reader-Service Card

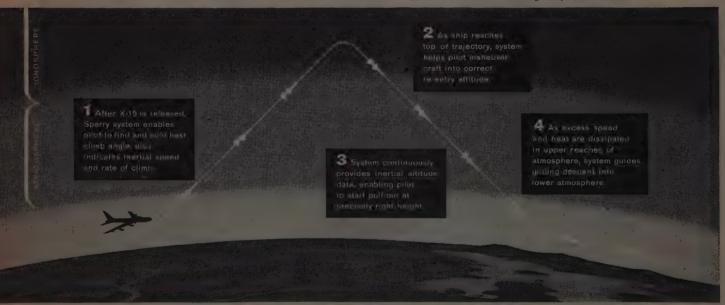




HYPERSONIC X-15, powered by a mighty rocket engine, will follow bullet-like trajectory during flight, similar to path of ballistic missile.



WEARING SPECIALLY-DESIGNED SPACE SUIT, X-15 pilot is protected against environmental extremes as he maneuvers the rocket to the edge of space.



X-15 WILL THRUST 100 MILES INTO SPACE UNDER CONTROL OF NEW INERTIAL SYSTEM

Strapped firmly into his cockpit, an Air Force pilot soon will ride the North American X-15 rocket research ship 100 miles and more into the sky at speeds above 3600 mph—over a mile a second.

A highly advanced Sperry inertial system, developed in conjunction with the Flight Control Laboratory of Wright Air Development Center, will supply control data for this historic venture. When the pilot and his revolutionary craft drop from a B-52 jet bomber, the inertial system will give him data for maneuvering and navigating the X-15 with extreme accuracy. As the stainless steel research

plane flashes outward through the thinning atmosphere, the Sperry system will "show" the pilot how to correct for even the smallest deviation from flight path. It will display flight information on specially developed instruments; at the same time feeding the data to ground and airborne recorders for a permanent record of the flight.

An important additional contribution of the Sperry system will be to guide the pilot in bringing the X-15 safely back into the earth's atmosphere. During this critical phase of the flight, attitude of the X-15 on re-entry must be precisely con-

trolled to avoid exceeding its structural limitations.

On this daring journey into space, the super-sensitive Sperry system will have to function perfectly under conditions ranging from extreme acceleration to complete weightlessness, through temperature swings that may heat the X-15's exterior to 1,000 degrees in a few seconds.

SPERRY GYROSCOPE COMPANY

Great Neck, New York

DIVISION OF SPERRY RAND CORPORATION

Circle No. 104
Reader Service
in Product Rev
Section

aircraft · missile engineering intelligence

Atlas tests to be speeded up after successful full range flight

FIRST FULL RANGE (6300-mile) firing of Convair Atlas ICBM from Cape Canaveral will result in a speed-up in Atlas tests. Present plans call for the firing of one Atlas a month.

AF's jubilation over the full ICBM flight will not result in any downgrading of the Martin Titan ICBM, however. USAF still considers the Titan (about one year behind the Atlas program) a more sophisticated weapon system.

IRBM orders may be smaller than expected

THE ATLAS FLIGHT may influence the IRBM program. Defense Department, under orders to choose either Douglas's Thor or ABMA-Chrysler's Jupiter for large scale production, will do just that. But orders for the IRBM that is selected may be smaller than expected. USAF will be less dependent on foreign IRBM launching bases now that Atlas has proved completely successful.

Valkyrie may carry IRBM armament, says USAF

NAA B-70 "CHEMICAL" BOMBER, now under development, will probably fire an air-launched IRBM, according to Lt. Gen. Roscoe C. Wilson, USAF's Deputy Chief of Staff for Development.

Components of this solid propellant WS-199B IRBM are already being tested by Lockheed and Martin. One version of it is identified as Bold Orion.

Two-stage IRBM will be larger than Bullpup, may also be carried by B-47 and B-52

IF IT CAN BE READY before the B-70 Valkyrie bomber is delivered, WS-199B may also be used on B-47s and B-52s. The two-stage missile is similar to, but larger than, the Bullpuo and White Lance air-to-surface missiles Martin is building for Navy and USAF.

Until the WS-199B is available, USAF will continue to hedge its bet with the NAA Hound Dog. First production contract for the GAM-77 (worth \$19 million) was placed by Air Materiel Command last month. Powerplant is the 7500-lb thrust P&WA J52. The bird will use an Autonetics self-contained inertial guidance system. Warhead is nuclear.

NAVY gave Martin-Orlando a \$20.4 million order for the short range Bullpup ASM. The Bullpup, which is just entering fleet service, is guided by the pilot by radio. An Aerojet-General rocket engine provides Mach 2 speeds. Range is 15,000 ft.

Latest A4D version can fly non-stop from coast to coast without refueling THIRD VERSION of the Douglas lightweight Skyhawk attack plane has been ordered by the Navy. A4D-2N will be produced at El Segundo under a contract totaling more than \$79 million.

Newest Skyhawk again uses C-W J65. It can reach top speeds of 700 mph and fly non-stop coast-to-coast without refueling. A night attack plane, it has basically the same airframe as earlier Skyhawk but is almost a foot longer. It can carry machine guns, missiles, rockets, and atomic weapons.

DC-8 jetliner test flights are in full swing at Douglas

WITH ITS LAST DC-7C delivered, Douglas is concentrating on speeding up testing and deliveries of the DC-8 jet transport.

First DC-8 powered by P&WA J75s has been flight-tested. Douglas now has four DC-8s flying, including three domestic versions fitted with J57s. It expects to use nine planes (including some powered by Rolls-Royce Conways) in its CAA test program.

Airline service with the DC-8 begins next fall.

GE comes up with closed-center hot gas servo in record time CLOSED-CENTER HOT GAS SERVO was developed by GE's Aircraft Accessory Turbine Dept., Lynn, Mass. Up to now, many application engineers felt a practical closed center system—if possible at all—was still a few years off (see S/A, "What's the Outlook for Hot Gas Servos?", Nov. '58, p. 20).

FEATURES OF GE'S servo design include: high frequency response through pressure-type control, electrically controlled damping, adjustable stiffness and

more on next page

accuracy characteristics independent of hardware changes, poppet valves to reduce leakage and sensitivity to gas flow contaminants, and symmetrical construction to reduce thermal shock problems.

SIMPLE method for re-entry studies has been worked out at Temple University's Research Institute, Elverson, Pa. Nose cone scale models made of refractory materials are inserted into extremely hot rocket exhaust flames, and the rate of ablation or sublimation is measured photographically.

Some correlation work is needed to draw conclusions about aerodynamic heating from these tests. Still the high heat flux of the rocket exhaust is sufficiently similar to that of re-entry to make the Temple researchers' relatively cheap setup-built around a rocket with a two-inch diameter exhaust nozzle-worth while.

\$29.3 million contracted for Titan underground launch

UNDERGROUND launching system for Titan ICBM will be developed by American Machine & Foundry. Design and development contract for \$29,-300,000 includes launcher system and related handling equipment.

PROPELLANT ignition problems on top of combustion instability are hampering development of large solid boosters in the 80,000 lb-and-over class for Polaris, Pershing, and Minuteman. Specs call for ignition reliability at temperatures below 10 deg F. The huge solid

grains can be ignited at these temperatures, but they resist uniform ignition. Once they do ignite, they start violently and in many instances quench the initial combustion process.

U. S. NUCLEAR turbojet engine is expected to be flying by '61. Reports of thrust rating vary from a low of 70,000 lb to high of over 80,000 lb.

Dual bypass planned for U.S. airborne reactor?

NUCLEAR engine design is said to have a dual bypass feature. (Russians pooh-pooh practicability of such an engine.) Idea is for turbojet to operate on chemical combustion heat during takeoff, on nuclear reactor heat during cruise.

INTAKE diameter reportedly is more than eight feet. During takeoff, air bypasses the reactor to enter annular conventional combustion chambers clustered around the reactor. Products of combustion are then re-routed to follow normal exit.

During cruise, air is directed through the reactor core, passes through the turbine wheel, and is ejected at subsonic speeds at around 1400 deg F.

LOBBER cargo ballistic missile developed for the Army by Convair was originally designed to deliver supplies (ammunition, rations, medicines, etc.) to advance troops isolated by enemy action or terrain. But Army feels Lobber may have other uses, tooamong them laving down radio transmitters as radio bench marks for rapid forward area surveys, sending cables across impassable waters for pontoon bridge construction, and stringing steel communications wire to forward observers.

W2F-1 EARLY_WARNING turboprop being developed by Grumman has been called "highly advanced and more heavily instrumented" replacement for the Navy's interim WF-2 Tracer. Power for the new plane will be supplied by two Allison engines.

ORENDA IROOUOIS turbojet scheduled to power the onagain, off-again CF-105 Avro Arrow has successfully logged over 100 hours in Lewis wind tunnel. Orenda-patented system for normal relights at critical Mach numbers following flameout worked well up to simulated altitudes of 60,000° ft (the maximum altitude capability of the tunnel).

WITH THE PROTOTYPE of the T-38 jet trainer due to make its first flight this month, AF gave Northrop \$17 million order to produce the supersonic T-38A as a production item.

The order is the first of a series of USAF contracts for the trainer, which will be used to transition pilots from the subsonic T-37 jet to/century series fighters.

ARDC tentatively agreed to a Lockheed Aircraft proposal to retrofit the F-104 with upward ejection seats. Lockheed's experience shows most F-104 problems unexpectedly have been at low altitudes (with power settings critical).

The F-104 was one of the first production aircraft to be equipped with downward ejection seats.

LOCKHEED'S XO-5 supersonic ramjet target drone was successfully intercepted by the Army's Hawk and Nike-Hercules. It was shot down by Hercules at altitudes over 60,-000 ft while traveling at 1500 mph. Hercules has also proved effective against the Pogo-Hi target at altitudes over 100,-000 ft. -

Valves most critical in missile servo hydraulics

SERVO valves are most critical components of hydraulic servo systems for missiles, J. W. Kastle, Jr., of Raytheon Missile Systems, told recent Vickers Hydraulic Conference in Detroit, Mich. Major effort is needed, he said, to perfect servo valve-actuator combination for pressures of 4000+ 5000 psi with much lower null

Null leakage, Kastle noted, will be increasingly important in large missiles of futurepumps will be held to 90 per cent minimum efficiency with only servo valves with flat P/Q curves specified. One point of improvement in these units is electric solenoids, many of which in today's systems are marginal at best, Kastle said.

CURRENT testing of hydraulic servo systems, Kastle said, indicated vibration is most likely to induce component failure and acceleration least

HOT GAS servo work at Vickers 1 o o k s promising. F. Klemach told the Confer-

ce. Present setup has hot generator feeding integral, sitive-displacement hot gas otor-hydraulic pump comnation. Positive-displacement achine was chosen because rotational speed closely atches that of either electric nerators or hydraulic pumps. is eliminates the gear box rmally needed with small, gh speed turbine drives.

ROPELLANTS are availle, Klemach said, that give mbustion chamber pressures 500-3000 psi, temperatures 1500-2500 deg F. Most ckers work thus far has been about 1000 psi and 2000 g F, but program is being

Solid propellants look best operating times of several nutes or less, said Klemach. r longer times, liquid prollants offer low sfc, which ually means lower installed ight and smaller space enelope.

ERIES of tests was run using ombustion of hydrogen and xygen. One such run resulted flame temperatures of 5000-200 deg. F. After 30 sec uner these conditions, Klemach aid, motor was found in exellent shape on disassembly.

LUIDS to meet all needs for 00-450-deg F hydraulic sysem operation will probably e available in next few years, R. D. Montgomery, Vickers ngineering section chief told he Conference. For temperaures over 400 deg F, he tated, metal or non-elastomerseals will probably be

Vickers is now studying vays of heating hydraulic luids from -65 to between -20 and zero deg F. Such neating would make it posmore on next page

How ESNA solves 7 typical aircraft fastening problems



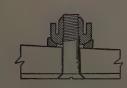
Speedier installation of access panels

ESNA gang channel nut strips eliminate the costly, time consuming instal-lation job of riveting individual nuts. Available in straight or curved sec-tions and even complete rings, custom designed for applications such as access doors or inspection covers.



Applications requiring guaranteed high reusability through more than 50 on-off cycles.

Where repeated tear-down and re-assembly or frequent readjustment is required, the exceptional elastic "memory" and non-galling characteristics of ESNA's standard nylon locking insert guarantees long lasting locking torque and fastening dependability. Available in all sizes and configurations of standard aircraft type nuts. Parts can also be designed to order and in any standard configuration, with guaranteed re-use factors as high as 300 on-off cycles. Where repeated tear-down and re-as-



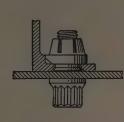
Bolting non-parallel surfaces

No more costly spot facing, step milling or hand selecting tapered shims! ESNA's counterbored, self-aligning types include one lug, two lug, gang channel, standard hex and high tensile types. Ball-and-socket relationship of nut and special base allow an 8° tilt in any direction from centerline to compensate for draft angle or taperate. to compensate for draft angle or tapered sections.



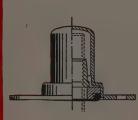
Attaching components in areas with limited wrench clearance.

Miniaturized insert-type hex nuts with across-the-flats dimensions as small as .109 in the 0-80 size . . or all metal (550°F.) nuts to AN365 or NAS 679 performance specifications with internal wrenching hexagon faces (which permit use of smaller wrench sizes) are available for use at locations where space and weight limitations are paramount. Complete lines of NAS miniature anchor nuts in carbon steel and A286 stanless steel are also steel and A286 stainless steel are also in production. Ask for your copy of the NAS/ESNA Conversion Book.



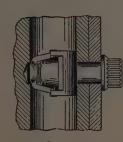
Fastening stressed joints subject to temperatures up to 1300°F.

For really "hot" applications such as jet engine flange assemblies or fire wall sections, where fastener dependability is critical, ESNA offers the "long-beam" locking device. The full cantilever of these sections assures protection against failures related to relaxation, creep and similar problems caused by the effects of extremely high temperatures upon metals. (Ask for ESNA Bulletin No. 5715 Design Manual for High Temperature Self-Locking Nuts.)



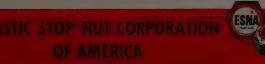
Sealing against fuel tank leakage

No danger of highly volatile fluids leaking past bolt threads with ESNA's self-sealing, floating anchor cap nut! The one piece cap unit is provided with "O" ring seal around its base which seals immediately the nut is riveted to the surface. The self-locking nut enclosed within the cap has .025" float to compensate for misalignment. Also available in gang channel nut strips.



Simplifying major substructure joining

An ESNA barrel nut doesn't have to be held for wrenching . . . doesn't need precisely mated bolt holes. The barrel-shaped fastener, is simply finger-pressed into a drilled or reamed hole until the special clip snaps into posi-tion at the bolt hole location. The tion at the bolt note location. The .030" float of the nut section of this fitting avoids misalignment problems and the bathtub recess for wrenching area is eliminated. New NAS 577 barrel nut (180,000 psi) now available. Also 160,000 and new 220,000 psi



WILLAT	ADE VOLLD	EACTENING	PROBLEMS?
- WHAI	ARE TOUR	PASIENING	PRUBLEMBIC

Dept. N49-150, Elastic Stop Nut Corporation of America 2330 Vauxhall Road, Union, N. J. Please send me details on the following: Installing access panels ☐ Simplifying substructure joining Assuring high re-usability Sealing against leakage ☐ NAS/ESNA Conversion Book Fitting in limited space ☐ Bolting non-parallel surfaces ☐ Bulletin No. 5715 High Temperature Design Manual

Raytheon Missile Projects



SPARROW III—the Navy's tenacious, lightningfast, cir-to-air missile—is intended for extensive use by Navy fighter aircraft in fleet air defense. Sparrow III is a Raytheon prime contract.



HAWK—the Army's defense against low-altitude attackers—carries out its destruction in the blind zone of conventional radars. Hawk development and production is under Raytheon prime contract.



TARTAR—A substantial contract for vital electronic controls for this Navy destroyer-launched missile is held by Raytheon. This equipment—a tracking radar and associated units—enables it to "lock on", cling to target's path, despite evasive tactics.



ADVANCED PROJECTS in aeronautical structures as well as missile quidance and control are now underway in Raytheon laboratories. New facilities are continually being added for this work.



PRELIMINARY NEW DESIGNS of tomorrow's missiles will result from the advanced work being done by today's missile engineers. Raytheon plays an important role in this area.

Raytheon diversification offers

JOB STABILITY FOR CREATIVE MISSILEMEN

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Diversified assignments—only possible in a company with Raytheon's wide range of missile activities—means security not found in one- or two-project companies. You apply your creative energies to the many projects you work on, and they in turn are your "insurance" against falling into a rut.

Individual recognition comes quickly from Raytheon's young, engineer-management—men who are keenly aware of the engineer's needs and contributions to missile progress.

Dynamic Raytheon growth—the fruit of this management's progressive policies—is best illustrated by the fact that Raytheon is already the only electronics company with two prime missile contracts—Navy Sparrow III and Army Hawk.

The next step is up to you. Why not get frank answers and helpful information on the type of job suited to your background and talents, its location, salary and other important details. Write, wire or telephone collect: The number is CRestview 4-7100 in Bedford, Massachusetts. Please ask for J. Clive Enos.

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RAYTHEON MANUFACTURING COMPANY
Missile Systems Division, Bedford, Mass.



ENGINEERING INTELLIGENCE

sible to use fluids with good high temperature properties whose pour points are too high for present specs.

CONTAMINANT - TOLER-ANT pumps are needed to meet increased reliability needs of future systems. Montgomery stated. Vickers pump of this type was subjected to contaminant three times higher than called out in military specs (filters were removed: contaminants included ferrous oxide, iron fillings, road dust) and survived 40-hr run. Leakage rate went up to 250 per cent, but volumetric efficiency only went down from 98 to 95 per cent. Standard lightweight units subjected to the same conditions, Montgomery noted, failed in less than one hour.

Thorough analysis, better materials gave tolerance

CONTAMINANT tolerance was achieved, Montgomery said, by careful analysis of all surfaces and replacement of conventional materials in high-wear locations with more wear-resistant material. For instance, Vickers went to modular ductile iron-type cylinder block and ductile iron bore instead of bronze units.

PROTOTYPE Huckbolt fasteners with automatic hole sizing feature were demonstrated by Huck of Detroit, Mich., a Los Angeles fastening symposium. One type has integral extruding land, other type four-tooth progressive broach integral with pin. Latter produces interference fit between hole that is pin-shank-controlled within 0.001 in. during manufacture, Huck engineers state. This close control, it is claimed, eliminates all secondary broaching or reaming when leakproof design is



Fantastic shapes for the space vehicles of the future already are on the boards. Even more radical designs are taking form in the minds of engineers. And their parts and components will just as radically differ from those produced today. New standards of precision and new methods of working new materials will be required.

One thing at least is certain: the same design, development and manufacturing experience which made the transition from aircraft piston engines to jets will be needed to produce these shapes of the future. Since the early 1920's, Ex-Cell-O has been among the major suppliers of machines, parts and assemblies to the aircraft industry. In that time it has built a reputation for extending the frontiers of precision.

as: rotors, blades, fuel nozzles, actuators, valves and fuel controls. Tomorrow? Well, perhaps you yourself have a problem which Ex-Cell-O's long experience in the production of precision controls and assemblies might help you solve. If so, why not contact Ex-Cell-O today?

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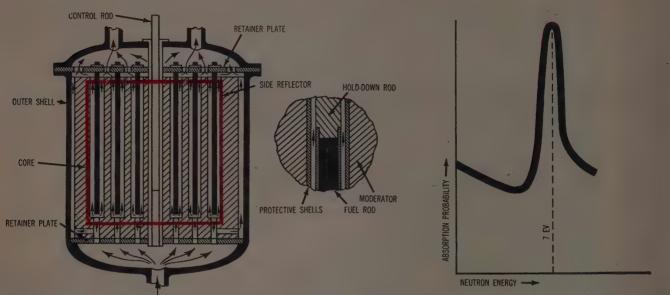




Today, Ex-Cell-O manufactures such components

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Propulsion



flow of liquid heat transfer agent for heat dissipation.

POSSIBLE basic layout for a nuclear reactor. Section at right shows FIGURE 1: Neutron absorption probability for uranium 238 increases sharply when resonance absorption comes into play.

Soviet development of

nuclear aircraft engines

It's generally acknowledged by now that the Reds are hard at work on atomic aircraft—and probably ahead of the U.S. in this field. Clues to the approach taken by Russia's nuclear designers are given in a detailed, authoritative Soviet report published in three parts by SPACE/AERONAUTICS. This first article deals with general design features; next month's article will take up control problems.

by G. N. Nesterenko, A. I. Sobolev and Yu. N. Sushkov*

ET us assume an atomic aircraft weighs 150 metric tons (165.375 tons), and has a very advanced aerodynamic form. To fly at Mach 1 at 11 km (36,091 ft), this aircraft must have a nuclear reactor putting out about 300,000 kw. (This is about twice the power of the reactor in the Soviet atomic ice-breaker.) For flight at Mach 1.5 at 11 km, the re-

actor would have to put out about 900,000 kw.

These high powers must be obtained within the smallest possible dimensions. This will keep down the weight of casing and shielding. The great weight of the shielding is one of the major obstacles in designing atomic aircraft.

The aircraft reactor must be a high temperature type. The higher the temperature in the reactors, the smaller and lighter and the more efficient will be the reactor. Calculations show that, to get the flight

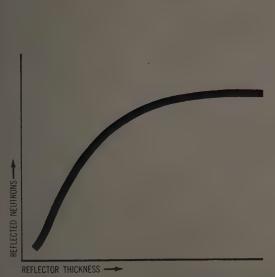


FIGURE 2: Effect of gradual increase in reflector thickness on number of reflected neutrons is sharpest at layers close to core.

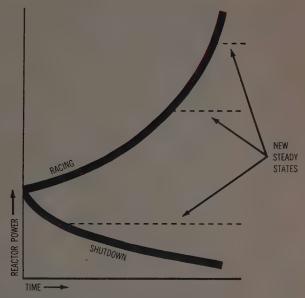


FIGURE 3: Effect of "racing" and shutdown at 0.0025 excess reactivity and new steady states that can be attained.

characteristics for chemically fueled aircraft now in series production in a nuclear plane, the reactor surfaces must be heated to not less than 1000 deg C.

An aircraft reactor finally must be highly reliable in operation. Unlike stationary reactors, it must function normally no matter what its attitude in space. Also, it must not be affected by inertia loads caused by changes in speed or direction.

The major processes in an atomic reactor are the controlled fission chain reaction and the dissipation of the heat generated by this reaction. The fission reaction occurs in the reactor's active section, or core.

Nuclear fuel in the form of cylindrical rods protected by metal casings is placed in grooves in a solid moderator. The heat of the reaction is dissipated by a liquid heat transfer agent. A rod made of a good neutron absorber controls the reaction.

The collision of neutrons with atomic nuclei of various

materials, including nuclei of fuel, isn't always accompanied by neutron capture. Often the neutrons bounce off the nuclei, transferring to them a portion of their kinetic energy. As a result, they slow down. In addition, a change in the direction of their motion, or neutron scattering, takes place. This moderation process is artificially induced whenever a rapid reduction in neutron energy is needed.

Scattering may slow down neutrons

Secondary neutrons escaping on nuclear fission have a very high kinetic energy at "birth" —1-2 mev for "fast" neutrons, which include more than half of all secondary neutrons. In the scattering process, the kinetic energy of neutrons may decline to the energy level of the thermal motion of the particles in the surrounding medium. The energy of such "slow," or thermal, neutrons depends on the temperature of the surrounding medium—at 20 deg C, it is about 0.025

ev; at 700 deg C, it is about 0.085 ev. Neutrons whose kinetic energy falls between the "fast" and "slow" ranges are "intermediate" neutrons.

Neutrons loss is caused by the capture of neutrons (1) by fuel nuclei without subsequent fission (2) by nuclei of other materials used in the reactor; (3) by nuclei of fission products. Like scattering, such neutron asbsorption is inevitable, since all known materials absorb neutrons to some degree.

At certain neutron energy levels, which vary with the absorbing material, the absorption probability increases sharply—because of the so-called "resonance" absorption (Fig. 1).

Finally, neutrons may be lost because they escape from the reactor without undergoing fission, or "leak." In a practical reactor, the fuel is spread more or less uniformly throughout the core. Neutron leakage depends not only on the amount of fuel but also on the dimensions and geometry of the core. As we have also seen, the various materials af-

fect neutron scattering and absorption (to degrees depending on their temperature). The properties and distribution of the core materials and their relative amounts therefore determine the mean neutron velocity at which the great majority of fissions of fuel nuclei takes place. This speed is the basis on which reactors are called thermal, intermediate, or fast.

For the reactor's critical state, at which the chain reaction first becomes self-sustaining, is characterized by a neutron flux constant in time and a constant rate of fissions. The latter determines the power. Calculations show that one kilowatt represents 31×10^{12} fissions per second. In practice, the critical state is reached by charging a specific amount of nuclear fuel into the core, since this usually is the last remaining variable.

Other conditions being equal, a reactor can function only so long as the amount of unused nuclear fuel remains above the critical level. When

more on next page

it is said an atomic aircraft will burn 500-600 g of nuclear fuel in a flight around the world, this doesn't mean such a flight can be made with merely 500-600 g of fuel aboard. The 500-600 g can be consumed only if they represent just a small portion of the total fuel in the reactor.

In the very best reactors now operating and being built, the "critical weight" of highly efficient nuclear fuel at the end of the operating period is not less than 80 per cent of the weight after initial charge. This means not more than 20 per cent of the initial charge can be consumed in operation.

The critical weight is reduced by any measures that tend to lessen neutron loss. It is lowest when the moderator and heat transfer agent are heavy water, and zirconium is the structural material.

However, these materials can't always be used. To get high operating temperatures, it sometimes is necessary to use fused metals as the heat transfer agent and heat-resistant nickel alloys as structural materials. Since these materials have high neutron-absorbing properties, less of them must be used within the core to lower the critical fuel weight.

This critical weight can also be reduced by reflection of escaping neutrons back into the core. For this, a scattering substance surrounding the core is used.

Heavy water is the best reflector material. Graphite, beryllium, ordinary water, and zirconium follow in order of diminishing effectiveness. As the thickness of the reflector increases, so does the number of reflected neutrons. The sharpest increase is produced by the layers closest to the core (Fig. 2).

With a reflector, there is an increase in the rate of fission near the core boundary. Efficient use of this peripheral zone enables the designer to make the reactor core smaller



AIR VALYES—A wide variety of electrically and pneumatically operated air valves for temperature and pressure-control applications. Butterfly, check, poppet, slide, and automatic drain types.



ACTUATORS — Rotary and linear types featuring a wide range of gear reswitching and mounting details. Available either as standard units or special to fit various applications. Designed to applicable military specifications.



designed for precision applications, new SYLC actuator meets the most exacting specifications on every count



Indicative of the Barber-Colman ability to build aircraft and missile controls that meet extreme requirements is this special, lightweight SYLC linear actuator. Designed for a high-speed jet fighter elevator trim application, it passes these specifications:

- Servo mode response rate
- Low backlash less than .005"
- Small differential ball detent clutch (90 lb carry 110 lb full slip)
- Less than 40 milliseconds' stopping time
 Position indication .002" resolution
- Ambient temperature range __65°F to +275°F
- Travel limit and centering switches
- Nonjamming positive stops
- Radio noise filter
- Thermal overload protection
- Weight 3.4 pounds

The Barber-Colman line of compact SYLC linear actuators is designed to handle loads in excess of 150 pounds and withstand 20 G vibration. They feature low backlash and high positioning accuracy... optional radio noise filter... adjustable limit switches... optional centering switches... permanent magnet d-c, split-series d-c, or 400 cycle a-c.

drive motors...dynamic braking and thermal overload protection if desired. For information on use of these actuators on your applications, call the Barber-Colman engineering sales office nearest you: Los Angeles, Seattle, New York, Boston, Baltimore, Fort Worth, Montreal, Rockford, or write direct to address below for literature.

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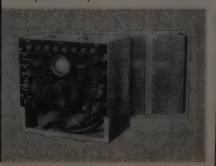
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THERMOSTATS — Units for sensing compartment, duct, windshield temperatures. CEDAC system senses, controls temperature and/or airflow in ducts,



TEST EQUIPMENT—Compact electrical test units for quickly checking all components of a Barber-Colman control system installed in an aircraft. Special units for checking many electrical systems.



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NUCLEAR ENGINES 2007

than it could be in a reactor without a reflector.

It is true the overall dimensions of a reactor are greater when a reflector is used. But we must not forget that the reflector is an integral part of the shielding. The thicker it is, the less special shielding will be needed.

For high temperature aircraft reactors, graphite is the only cheap reflector material worth considering. The smallest reactor dimensions are possible with reflectors made of beryllium, which melts at 1315 deg C and is inert to neutron irradiation. However, it costs a lot.

A steady state, in which neutron flux and reactor capacity remain constant in time, is observed when the reactor is in the critical state by the fact that the neutron multiplication constant (K) equals unity. K is the ratio of neutrons undergoing fission in a given generation to neutrons that have fissioned in the preceding generation.

If the reactor has more than the critical amount of fuel, K will exceed unity. A constant increase in power results. A reactor is placed in this "supercritical" state to "race" it. A special case of racing is the starting of the reactor.

Control possible if k_{ex} drops below 0.01

As an example, let us assume that, in a slow reactor in the critical state, we have suddenly increased K to 1.1 or introduced an excess reactivity k_{ex} of 0.1. Calculations show that, at this excess reactivity, the reactor power will increase 2.7 time in 0.002 seconds, 1500 times in 0.01 seconds, and about 20,000 times in 0.02 seconds. A breakdown is almost inevitable.

In a fast neutron reactor, power increases about 150,000 times in 10⁻⁶ seconds at 0.1 excess reactivity. This rate of evolution comes close to that

more on next page



MISSILE COMPONENTS Bulova safety and arming systems protect ground, air and sea crews from load to launch, then take over in flight. Safety factor of one in a million is specified and reliably delivered by Bulova's precision production facilities. Powder-driven gyros and fuzing systems are among other Bulova developments for 18 key missiles.



AIRCRAFT INSTRUMENTS Bulova's new Servo Altimeter assures maximum reliability through unprecedented sensitivity, accuracy and repeatability. At 40,000 feet, it detects 4-foot changes... is correct to 40 feet. Safety is improved in traffic control and flight over difficult terrain. As a control instrument, it is readily adaptable to guided missiles.

Bulova reliability helps to solve today's most challenging problems

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Milestones along the way are the electronic and electro-mechanical devices created by the Bulova capability—the uncommon blend of pioneering vision and precision production experience.

These Bulova developments, distinguished



by their advanced design and consistent high performance, help our nation's defense and industry stake solid claims on the frontiers of science and space.

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BULOVA RESEARCH AND DEVELOPMENT LABORATORIES, INC.



INFRA-RED COMPONENTS Bulova infra-red cells are designed to unerreingly guide Sidewinders to target. Bulova-improved production processes Increase yield and product reliability. Other infra-red developments include filters, reticles and thermistor bolometers, as well as advanced research in mosaic and lead selenide cells.



PHOTOGRAPHIC SYSTEMS Bulova's new high performance 70mm recon-camera features 8 frame/sec. and pulse operation...vibration-free exposures to 1/4000 sec. From the smallest 16mm gun camera ever built to units of 9x18" format size, Bulova developments include optical, data recording and instrumentation, and special sequence cameras.

of an atomic explosion.

Even at an excess reactivit of 0.01 it still isn't possible t control a nuclear reactor. soon as kew drops below 0.0! though, the picture change sharply. The reason is that during the very brief time re quired for fission, only 99 pe cent of the number of neutron escape. The remaining "delayed" neutrons appear groups with a delay of up t 80 seconds. They are emitte in "fragments" of the fissione nuclei as these undergo radio

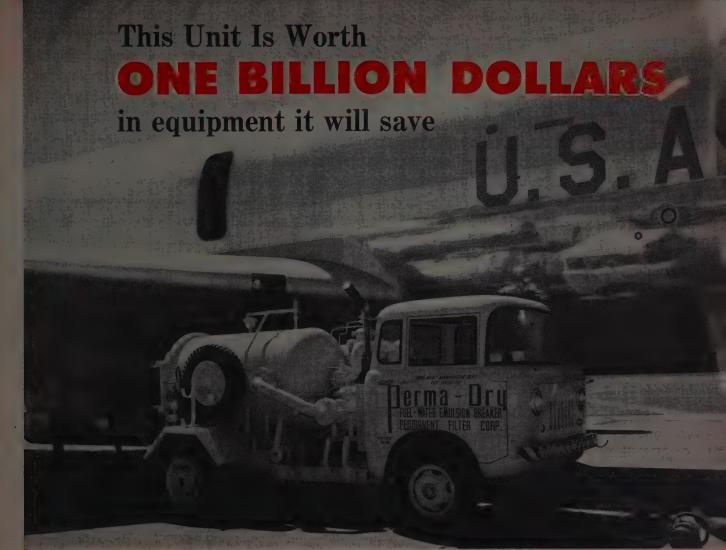
The exact percentage of de laved neutrons is 0,755. There fore, to guarantee safe opera tion during start, kex must b less than 0.00755. Abou 0.005-0.006 is recommended In this k_{ex} range a slow reacto will need about six seconds t increase its power 2.7 time and 30 seconds to increase 500 times.

Power increases in a fas reactor takes place at about the same rate, which insure reliable regulation of a chair reaction. Power reduction in practice also depends on de layed neutrons (Fig. 3).

Reactivity can be changed for control purposes (either manually or automatically) One way of doing this is to shift the control rod.

When the rod is pushed into the core, neutron absorption increases and reactivity de creases. If the reactor was in a critical state, it will now become subcritical. Withdrawa of the rod will increase the reactivity again.

Spontaneous changes in reactivity occur when the temperature is changed. An increase in temperature usually decreases the reactivity—all materials become less dense as they are overheated, and the greater distance between their nuclei means fewer of these will collide with the neutrons. This relationship be tween reactivity and temperature makes it easier to control the reactor.—End



600 GPM Jeep-mounted PERMA-DRY Fuel-Water Separator,

PERMA-DRY ... The only proven unit for JP or RP fuel-water separation

Developed for the missile, jet and rocket program and proven by rigid Air Force and Government tests, PERMADRY exceeds Government specifications (600 GPM).

The PERMA-DRY Fuel-Water Separator will break an emulsion from any fuel and deliver it 10 to 15 PPM below the saturation point and sub-micronically clean.

Designed for highest dependability, the PERMA-DRY has been tested successfully more times and in more places than any other separator. Utilizing inexpensive elements, PERMA-DRY is easy to use and maintain. It is available in sizes from 5 to 1500 GPM in either mobile or stationary mountings. Dirt-holding capacity more than double that called for in military specifications insures longer life between servicings of the PERMA-DRY Unit.

Now, the use of a PERMA-DRY unit will assure you of absolute jet fuel reliability . . . give safe, high-speed

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Dark or light, fair or foul... ABLE finds True North fast

Aline your missile-launchers and your mobile radars fast with celestial accuracy—any weather, any time of day. Do it with ABLE, Autonetics' Base Line Equipment. ABLE is a portable field instrument designed to find True North or any other base line—automatically. ABLE is reliable, easy to operate and maintain.

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Where Durability, Hi-Temp Resistance Are Needed

Coromo thermocouple and thermocouple extension wires are designed specifically for severe conditions-conditions where ordinary T/C wires are inadequate. First developed by Thermo



Electric, "Ceramo" wires incorporate ceramic insulation with overall metal sheathing. They'll solve your problems of high temperature, abrasion, chemical or corrosive action, pressure and difficult installation. You can bend them to almost any shape. As extensions, you can often use them without conduit. And under these conditions they'll outlast comparable standard types many times-with no significant difference in response. Choice of conductors in all standard calibrations and various sheath materials for temperatures up to 3000°F. Conductors-36 to 12 gage. Overall diameters-1/25" to 7/16" for thermocouples; 1/8" and 1/4" for extensions.

Write for Bulletin No. 31-300-A

Thermo Electric Co.. Inc. SADDLE BROOK, NEW JERSEY In Canada—THERMO ELECTRIC (Canada) Ltd., Brampton, Ont.



Aircraft Missile Electronics

Production Engineering



TYPICAL setup of electric discharge machine.

Microinch tolerances with electric discharge machining

Many of the newer temperature resistant, hard metals that practically defy conventional machining can be handled by electric discharge methods. This process holds surface finishes to 10-20 in. -without scratches, heat checks, waviness, or chatter ---and drill holes and cavities to 0.0005 in.

by W. F. Wagner

Supervisor, Design Producibility & Standards, Northrop Aircraft, Inc.*

LECTRIC discharge machining is similar to electrolytic and ultrasonic machiningall three methods erode the work piece. But there the similarity ends.

The electric discharge meth-

od uses an intermittent high frequency spark to remove the material by erosion. Both the electrode and the work piece must be submerged in a nonconductive or dielectric fluid.

Electric discharge machining is done by a series of discharges that occur at a rate of 20,-300,000 per second. With each buildup of voltage electrons pile up on the surface of the electrode until the stress becomes so high that they break through the surface barrier and speed toward the work piece.

The collision of these electrons with atoms on the surface of the work piece leave the atoms minus an electron, or with a net positive charge.

osion of Brass Electrodes

terial up	Relative E	rosion Rate
	Tool	Work
ngsten		1. 1
arbides of steels	3/4 1.4.	
inless steel	Ì.	
llites	11 13	
oper,		
pronze, &	1/2	
minum &	ĺ	
oft metals		15

s turns a tiny portion of the rk piece into superheated tallic vapor carrying a posi-

Since dissimilar charges are awn to one another, the charged particle om the work piece is atcted to the negative elecde, or tool. As the particles gin to move, they are cooled the dielectric fluid before ev reach the electrode and

Theoretically, none of the articles ever reaches the elecode, and the electrode does t erode. Unfortunately, this not true. Mainly because of nperfect flushing, a small perentage of the particles does ach and erode the electrode. A variety of precise electric scharge tools (electrodes) e available today for drilling, e sinking, forming of irregur internal shapes, and tool d surface grinding. A conderable amount of repetitive e sinking is being done ecoomically with Elo-Met as the

Elo-Met is a 50-50 tin-zinc loy and melts at 640 deg F. the die sinking application, is cast into a machined protype die. After cooling, the asting is squeezed back into ne die under pressure to over-

These castings may be acidtched to any degree of underze to produce roughing and nishing electrodes. In some ises, the useful life of the

more on next page



THE F-105-D FLIGHT SIMULATOR- BEING HUMAN ENGINEERED, DESIGNED AND BUILT BY ERCO PROVIDES COMPLETE, RELIABLE, EFFECTIVE TRAINING

FOR THE ENTIRE MISSION PROFILE OF THE LATEST AIR FORCE ADVERSE WEATHER AIRCRAFT.

FEATURES.

Automatic Self-Checking System (aural-optical) provides component failure indication by general area and exact location. The simulator is operative in a matter of minutes.

Reliability—in excess of 95% utilization.

Human Engineering-complete environmental realism.

Man-Machine Task Analysis Design—maximum learning transfer for both operator and instructor.

Erco's manpower—facilities—capabilities and mission training experience help the Air Force "to fill the suit .. to man the plane."

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ERCO TRAINING TO FILL THE SUIT .. TO MAN THE PLANE



NUCLEAR PRODUCTS - ERCO, DIVISION OF QCF INDUSTRIES, INC., RIVERDALE, MARYLAND

* CARTER CARBURETOR & SHIPPERS CAR LINE & W-K-M ADVANCED PRODUCTS AVION

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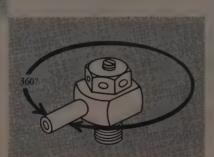
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- Cracking pressure available in 20 to 60 psi range



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- Internal leakage drops/hr (lapped steel seat)
- Low pressure drop both directions
- Ports --- AND-10050-16



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- System may be bled while under pressure
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Relief Valves - All Four Basic Types Restrictors Special Nonstandard Types

Combinations of Two or More Types in a Single Body.

Automatic, Hydraulic Pressure Dropping — **Time Delay Valve**

- Combines functions of six valves in single body
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VARIETY of tools or electrode used in elect discharge mach ing is limited on by the user's i genuity.

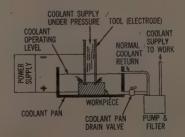
SURFACE FIN-ISHES on carbide cutting tools produced by electric discharge machining (left) and conventional machin-



tool reportedly was extended nearly fifty fold by etching undersize and then building back up to the desired size with hard chrome plating.

Some work has also been done on the contouring of large honeycomb sections whose cover skin is drawn into a contoured female electrode. In this case, the skin itself is used as the tool, or electrode, to shape the honeycomb core. Matching the skin and the core produces a good, close fit for bonding and brazing.

Electrodes, or tools, can be made from a variety of materials—yellow brass, copper tungsten, silver tungsten, tungsten carbide, carbon, Elo-Met, etc. It all depends on the ap-



BASIC arrangement of elements in electric discharge machining.

plication and material to be machined. The most common ly used materials are yellow brass and Elo-Met.

Electric discharge machining is particularly well suited to producing holes and cavities to tolerances of ± 0.0005 in Virtually no heat is generated in the work piece, so that it possible to drill precise holes spaced no more than 0.005 in apart without any distortion of mutilation of the remaining web. Holes as small as 0.002 in, in diameter have been drilled through 3/8-in.-thick steel plate.

Electric discharge grinding techniques are used for surface finishing and machining The "grinding" wheel is usually made of free machining brass but contains no abrasives whatever and never touches the work.

The wheel is rotated at 100-200 rpm and maintained at a fixed gap distance from the work piece by a servo unit The metal is removed by the action of the high frequency discharge.

This method has many advantages over conventiona abrasive grinding. It is possible get a 10-20-uin. mat surce finish without any heat lecks or feather edges. As the heel never touches the work, ere is no mechanical prestre or spindle vibration. This rtually eliminates all wavi-

Cuts of any width or depth imited only by wheel size) to possible in a single pass, here is no need to reciprocate ther wheel or work piece. The cost of electric discharge grinding of otherwise expensive carbide form tools is in many cases only 35 per cent that of conventional grinding. On top of that, the finished tools can be operated at appreciably higher feeds and speeds than conventionally ground carbide tools, produce smoother surface finishes, and have 4½ times longer service lives between grinds.—End

The further we progress, the more important experience becomes in solving today's most urgent problems. One company actively and continuously engaged in advanced ECM research and development since 1952 is . SERVICES research and development · electronic equipment production · reliability evaluation EQUIPMENTS—communications · radar · countermeasures · reconnaissance · infra red devices · heat exchangers · pulse generators · antennas

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4 Ounce Contact Force Gives Relay Reliability

Contact force of 4 ounces per contact on 50 "G" models and 2 ounces per contact on 30 "G" models of "Diamond H" Series R and Series S miniature, hermetically sealed, aircraft type relays is one of the most important factors in their proven high reliability.

Though absolute reliability of any similar device is impossible to guarantee—a bitter fact of life recognized by all electronic engineers—close approach to this goal by the relays manufactured by The Hart Manufacturing Company is the basic reason they are found today on many of this country's headline-making missiles.

In addition to contact force far beyond that found on other relays, "Diamond H" relays have greater contact cleanliness. Selfcontamination is virtually eliminated by a completely inorganic switch mechanism, as well as use of coil materials which will not dust, flake or out-gas.

Finally, the high degree of reliability that is designed into these relays is maintained in their manufacture by high quality workmanship and a stringent inspection policy at every stage.

In addition to missiles, and their ground control systems, Series R and S relays are designed for use in jet engine controls, computers, fire control, radar and similar critical applications.

4PDT units, they offer an extremely broad range of performance characteristics, including temperature ranges from —65° C. to 125° and 200° C.; ratings to 10 A., 120 V., A. C., and 26½ V., D. C., with special ratings to 400 ma. at 350 V., D. C., or down to millivolts and milliamperes. Dry and wet circuits may be safely intermixed.

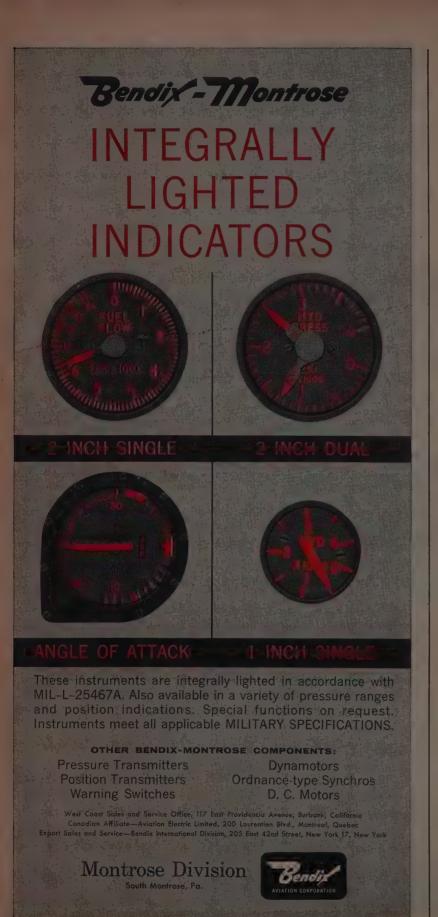
For more information, write today for Bulletins R250 and S260. For quick facts about "Diamond H" switches, thermostats and other devices, ask also for a copy of the "Diamond H" Check List of Reliable Controls.



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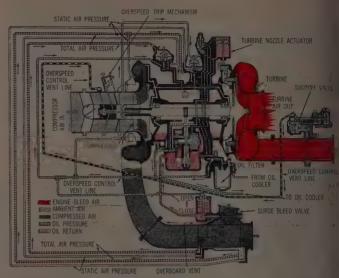


FIGURE 1: Operational flow diagram of turbo compressor. Besides speed controls, safety features include: overpressure shutdown on shutoff valve to preclude effects of pressure regulator malfunction, variable-area nozzles aerodynamically loaded to fail closed, hydraulic nozzle actuator spring-loaded to fail closed.

Cabin pressurization now needs built-in reliability

As production runs become shorter and planes more expensive, systems reliability no longer can be built around failure feedback from testing—it has to be built in right from the beginning. Here is an outline of what this means for cabin pressurization design.

by Carl W. Jonasson

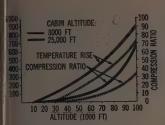
Group Engineer, Mechanical Engineering Staff, Transport Div., Boeing Airplane Co.*

RELIABLE cabin pressurization is becoming ever more important as the industry makes the transition to jet transports in the commercial

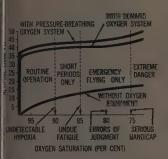
field and to spacecraft in the military. USAF's Lt. Gen. C. S Irvine recently stated the case for reliability: "In the past, we have used such techniques as failure-feedback reportings in reliability and durability. These procedures worked fairly well with long production runs.

"In the future, however, when short production runs will be the norm, backtrack corrections will be totally unacceptable. This means that full time, maximum performance capabilities must be de-

*Transport Div., Boeing Airplane Co., Renton, Wash. This is a condensation of a paper, "Standard for and Reliability of Cabin Pressurization Equipment," presented at the USAF-Industry Symposium on Aircraft Pressurization & Oxygen Systems, last Sept. 23-24, in Los An-



IGURE 2: Cabin pressurization enalty as variation of temperaare rise and compression ratio ith altitude.



IUMAN reaction to altitude with supplemental oxygen. (Hypoxia s loss of useful consciousness because of crease in partial pressure of oxygen with exposure to increasing altitude.)

signed, developed, and produced into first and subsequent items... Functional reliability and structural durability should be virtually 100 per cent."

Cabin pressurization systems can be generally divided as in Figure 3. Up to 50-60,000 ft altitude, the external air supply system with outflow pressure control is the most logical choice. In this range, the cost of pressurizing ambient air isn't excessive. A system providing a fresh air ventilating rate high enough so no air must be recirculated will normally provide air in excess of that required to make up for structural leakage.

As Figure 2 shows, the airplane penalty for pressurize ambient air increases very rapidly at 40,-100,000 ft. It's log-

more on next page



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4340 stands alone among medium carbon steels in its ability to provide maximum strength, ductility, toughness and resistance to fatigue in parts of medium to heavy cross section.

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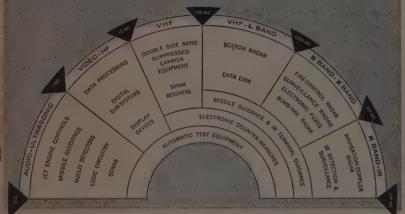
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Reliability Comparison of Component Arrangements

Number of	Sei	ries	Parallel Parallel		
Identical Components	System Reliability	Flights Without Failure	System Reliability	Flights Without Failure	
1728	0.9	10	0.9	· . 10	
2 / - :	0.81	5.26	0.99' .	. 100	
3	0.729	3.69 1 11	0.999	- 1000	
41	0.6561	2.91	0.9999	10,000	
5	\ 0.59049	2,50	0.99999	100,000	

ambient air to maintain the desired pressure in the cabin. This system will have inflow pressure control. Recirculation of cabin air will then be needed to provide proper air velocity, temperature, and humidity

As we approach 100,000 ft, a careful study of all the penalties imposed on the vehicle by the cabin pressurization system

ical to supply only enough may show that the best compromise would be a combined system. Here the cabin air would be enriched with oxygen, so less pressure would be needed for adequate physiological protection. This extends the altitude for which external air can be used. Ram pressurization would also become more feasible.

Above the altitudes where it's no longer practical to pres-

PRE-FLIGHT RADIO TESTER

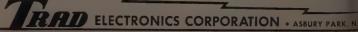


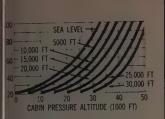
The RT-500 tests — IN PLACE — the performance of ALL aircraft and vehicular communications equipment in the 190 kc — 400 mc frequency range. The functions of Transmitter Monitor, AC-DC Voltmeter, Milliwatt Meter and two wide-range Signal Generators are combined in one compact, easily transportable, precision instrument.

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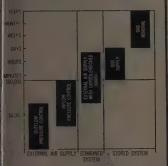
RECEIVER TEST SE

Write for details and specification





YGEN concentration vs cabin ssure altitude.



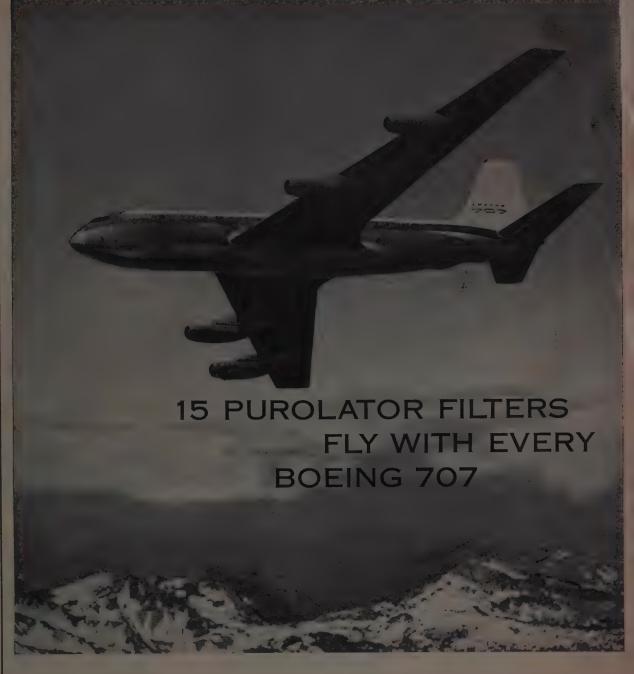
IGURE 3: Basic categories of bin pressurization systems.

urize ambient air, a stored stem, using liquid gas, or the ke, is required. For flights mited to several hours' cruise these altitudes, only a semiealed cabin is needed. After ne gas has been used to cool e cabin, remove CO, water apor, and odors, it will be disharged overboard. The rate at hich the gas will be used is function of cabin pressure, nd temperature, and mininum ventilation. The best sysm would be one in which the akage rate at full pressure bout equals that required for ooling during cruise condi-

For the longer orbital and pace flights, the vehicle must e self-sufficient. This will require a hermetically sealed cabin with provisions for controlling pressure, oxygen content, temperature, humidity, CO₂, and odor. Rejuvenation of the atmosphere and re-use of human waste products will be essential. The overall reliability must be extremely high.

The reliability of a system s a combination of the reliabilities of the system compo-

more on next page



The Boeing 707 is America's first jet transport. Performance is so outstanding that 11 airlines already have ordered 707's.

Effective filtration is essential on the 707 just as it has been on all aircraft for the past 25 years. Fifteen Purolator Micronic filters ensure that fluids move freely, and give optimum performance, by filtering out sub-micronic particles of foreign matter...particles that could cause serious trouble in a jet transport going 600 miles an hour.

With filtration such a must in any aircraft built today ... or planned for tomorrow ... there's no reason to settle for anything less than Purolator.

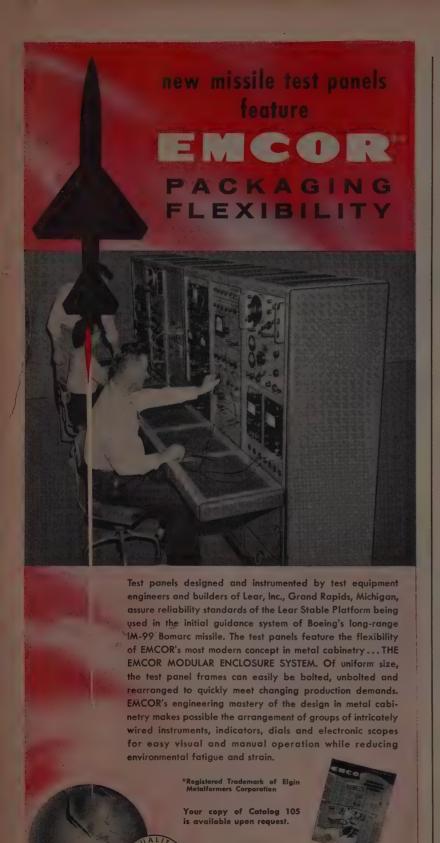
For every vital fluid used in aircraft ... fuels, lubricants, hydraulic fluids and air . . . nothing performs better than a standard Purolator filter. The same will hold true when new needs arise. All fluids need filtration. And none need get along without the extra margin of reliability of Purolator filtration.

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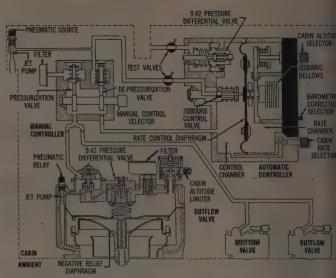
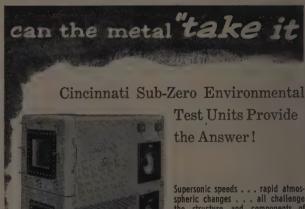


FIGURE 4: Cabin pressurization system for current jet transports.

nents. These component reliabilities can combine in series -when any component fails, the system fails. They can also combine in parallel, so that,

if one component fails, another can perform its function. A the Table shows, the overall systems reliability is greater with components in parallel.



Supersonic speeds . . . rapid atmos-spheric changes . . . all challenge the structure and components of aircraft, rockets and missiles. "Can the metal take it?" must be answered in advance.

MODEL ASU-120-6-HC Altitude Test Unit del has a temperadel has a tem

New 6 cu. ff. model has a temperature range of plus 300 to minus 100° F. Equipped with temperature and altitude recording controllers. Altitude range at sea level to a copy of the new Catalog.

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6- & 8-CHANNEL DIRECT WRITING SYSTEM

The chief factors to be condered for overall reliability e: human reliability, penalty the vehicle, effect of other stems, mechanical reliability, d emergency provisions. If human limitations are ex-

eded because of conditions eated by the cabin pressuration system, the decrease in liability is chargeable to that stem. These limitations aren't ways exceeded because of echanical failures, but somenes occur because of imoper design objectives or proper operating procedures. Many human errors that ay crop up in servicing, aintenance, and operation ould be compensated for the iginal design through good man engineering. For inance, built-in connections in e system can eliminate many rvicing mistakes. Designing uipment so it can't be missed can prevent operating er-

Excessive penalty to the vecle in the form of extra eight, volume, drag, or power eds can reduce reliability. op up in efforts to gain durpility, a compromise must be ade for optimum reliability. Cabin pressurization, or the ck of it, can reduce the over-I reliability of the vehicle by eating a condition which may use some other essential uipment to fail. For exame, increased arcing and deeased cooling of certain elecic or electronic equipment ay be the direct result of inifficient pressure.

An emergency provision can a piece of equipment or an perating procedure. A typical tample of the former is the tample of the former is the tample of the former is the tample of the new jet transports. If cabin ressure is lost, the system atomatically presents oxygen lasks to each passenger and rovides him with enough oxyen to prevent marked hypoxia.

A typical emergency procemore on next page If you want a practical direct writing system for straightforward recording in the range from DC to 100 cps — such as computer readout, telemetry recording — look what the new Sanborn "850" offers in compactness, reliability and operating convenience. A complete 8-preamplifier module with power supply, plus an 8-channel flush-front recorder package containing power amplifiers and power supply at rear, occupy only $24\frac{1}{2}$ " of "850" panel space.

PERFORMANCE characteristics of an "850" include flat frequency response 0-70 cps, down 3 db at 100 cps (10 div. peak-to-peak amplitude)... thermal drift eliminated by current feedback power amplifiers... limiting at input to prevent amplifier saturation or cut off, so that damping is never lost... drift less than 0.2 div. for 20° to 40° C. changes, line voltage changes from 103 to 127 volts... gain stability better than 1% with 20° C. and 20 volt changes... linearity 0.2 div. over 50 divisions... clear, permanent, inkless recordings in true rectangular coordinates.

IN RELIABILITY, "850" features include fully transistorized power amplifiers and power supply...rugged galvanometers with low impedance, high current, enclosed coil assemblies and velocity feedback damping... JAN components wherever practical, such as MIL-T-27 hermetically sealed power transformers, MIL-approved electrolytics in power supplies, etc.... forced filtered air cooling for stable operation.

And in operating **CONVENIENCE**, an "850" system provides such advantages as nine electrically controlled chart speeds, selected by pushbuttons... a choice of interchangeable Preamplifiers (DC Coupling and Phase Sensitive Demodulator presently available, with others in development)... remote control of chart drive, speeds, timer and marker... monitoring connection points... a Recorder that loads from front and has built-in paper take-up and paper footage indicator.



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stainless steel

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ATLAS ICBM

on first

full range flight!

U.S. missile program one step nearer to complete operational capability.

On the evening of November 28, 1958, a 100-ton ATLAS lifted from its pad at Cape Canaveral and arched majestically into the heavens, 30 minutes later its nose cone shot into the Atlantic, marking the first successful completion of its fully-programmed distance of 6300 statute miles.

The main part of the ATLAS structure is literally a huge fuel tank, the hell of which is thin gauge MicroRold stainless steel. Important factors in selection of stainless steel for the outer skin of the ATLAS are-great strength at both high and low temperatures, resistance to corrosive exotic fuels and good workability.

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dure is that of descending after loss of cabin pressure. The new jets have high descent rate capabilities to make this maneuver practical at altitudes of 40,000 ft.

Although human beings can generally survive several minutes without oxygen, recovery would be very doubtful following exposure to altitudes over 65,000 ft for more than about 15 seconds. The effects of gas expansion, boiling of body fluids, and lung damage would hinder recovery from hypoxia. Emergency pressure is thus needed. The military probably will use pressure suits. For transports, automatic opening of a ram air scoop as soon as cabin pressure drops to a predetermined value looks prom-

At altitudes at which emergency provisions can no longer sustain life, the aim will be to gain extremely high overall reliability and take the risk of having a catastrophic failure

Detail design, of course can't be ignored. Reliabilit must be designed into eac unit. Examination of currer jet transport units shows son of the changes in design think ing that have led to improve reliability.

One example is the turb compressor. This is basicall an air-turbine-driven, single stage compressor with a the bine wheel and a compresso wheel on a common shaft. Th air for driving the turbin comes from the aircraft engine Its quantity is governed by th outflow of the compresso

The rotational speeds of the unit are limited to safe value by any of three controls:

- The normal L-control regulates the speed to me cabin cooling, heating, or presurization needs and keeps below 46,500 rpm.
 - If the normal control fai



David R. Grossman Co., 155 East 44th Street, New York 17, N. Y.

Engis Equipment Company, 431 South Dearborn Street, Chicago, Illinois,

Manufacturing Associates, Los Angeles 24, California,

BRITISH MANUFACTURED BEARINGS CO., LTD.,

B.M.B. Sales Ltd., High Street, Crawley, Sussex, England. Circle No. 273 on Reader Service Card in Product Review Section

mit the speed, a hydraulic sing speed control limits unit to 47,000 rpm by closdown on the turbine noz-

If the speed continues to ease, a trip speed control shut down the unit at 1000 rpm. Both the turbine off valve and the pressure dating valve go to the ed position. The unit can't eset in flight.

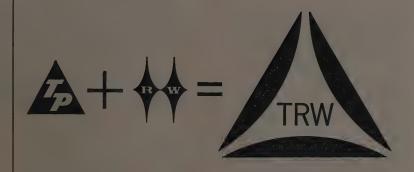
flow valve design has nimproved

Figure 4 shows—as part of complete pressurization system—an outflow valve detected to today's reliability dards. Its important reliatory features are:

The ambient - pressuresing line is an integral part the valve. This reduces the bability of failure in the as well as the probability malfunction due to failure connect the line or to contion to the wrong port.

- The pneumatic relays isovalves that malfunction ause of improper control d pressure. In other words, component reliability searrangment has been shortd to increase system repolity.
- Jet pumps hold the outw valve open whenever the ected cabin altitude is the me as or above the ambient tude. This prevents excese cabin bump when the bin air flow is turned on. It allows faster equalization cabin and ambient pressure, eventing a buildup of residpressure.
- A larger cabin vent filter duces chances of plugging, ich could cause valve malaction.
- The cabin altitude limit atrol is designed to make the no single component failth in the pressure control system will open the valve and pressurize the cabin to an arrow 15,000 ft.—End

second in a series



THE MERGER

The legal act of merging two companies into one does not of itself change the sum total of their capabilities. Thus, today the competence of the Ramo-Wooldridge Division for the development of electronic systems for military and commercial applications is indistinguishable from that of its predecessor organization, The Ramo-Wooldridge Corporation, while the skills of the Thompson Products group of divisions in the design and large-scale production of precision devices also remain unchanged. Soon, however, effects of the merger will begin to appear. One early effect will be an important addition of manufacturing strength to Ramo-Wooldridge programs, several of which have passed out of development and are in the prototype or manufacturing phases. Conversely, the special skills of Ramo-Wooldridge scientists and engineers in certain fields can usefully supplement the services that the Thompson Products divisions offer to their customers.

The formation of Thompson Ramo Wooldridge Inc. is intended to provide an unusual capability for the development and production of the complex electronic and mechanical devices and systems required by today's expanding technology.



Thompson Ramo Wooldridge Inc.

Main Offices • Cleveland 17, Ohio Los Angeles 45, California

DIVISIONS AND PRODUCTS

TAPCO: Missile and aircraft auxiliary power systems, ground support systems, fuel systems, pumps, accessories, hydraulic systems, pnematic systems; electronic control systems, microwave switches; frame structures, pressure vessels. Jet engine compressor blades, rotors, stators, and impellers; turbine buckets, rotors, and stators; structural and fabricated components. Rocket engine cases, nozzles and pumps. Nuclear reactor control rods, pumps, accessories, and core structures. Precision investment and continuous vacuum cast parts for aircraft, missiles, jet and rocket engines. Vacuum cast super-alloy ingot, billet and mill shapes.

RAMO-WOOLDRIDGE: Electronic reconnications ance and countermeasures systems, infrare systems, analog and digital computers, and navigation and traffic control, anti-submarine warfare, electronic language translation, information processing systems, nuclear energy applications, missile electronics systems, advanced radio and wire line communications.

AUTOMOTIVE and INDUSTRIAL PRODUCTS: Valves and associated parts for all types of internal combustion engines. Steering linkages, front wheel suspension ball joints, hydraulic cylinders and pumpš, cylinder sleeves, piston rings. Truck retarders. Diesel engine turbochargers. Rock drill bits and drill rods. Alloy pistons for automotive and aircraft; impact extrusions, permanent mold and die castings. A wide variety of automotive replacement parts distributed nationally and overseas through 7.000 distributors.

CONSUMER PRODUCTS: High fidelity amplifiers; FM-AM radio tuners; magnetic tape recorders; stereophonic sound systems, public address and intercommunication systems. Television cameras for industrial and broadcast purposes; complete low-power television broadcasting stations.

THE THOMPSON-RAMO-WOOLDRIDGE PRODUCTS CO.: Digital control computers and associated equipment for automatic control of industrial processes, data logging and computation, pilot plant operation and process research, test facility operation, and general computational use.

PACIFIC SEMICONDUCTORS, INC.: Germanium and silicon diodes and transistors, high voltage rectifiers, subminiature rectifiers, voltage variable capacitors.

Number of employees: 22,000
Estimated 1958 Sales: \$335,000,000
Plants in Los Angeles, Bell, Culver City and Long Beach, California. Denver, Colorado.
Michigan City, Indiana. Cambridge,
Massachusetts. Warren and Portland, Michigan.
St. Louis, Manchester and Sullivan, Missouri.
Cleveland, Euclid, Willoughby, Minerva, and Columbus, Ohio. Danville and Harrisburg,
Pennsylvania. Roanoke, Virginia. St. Catherines,
Ontario.

If the job calls for TEFLON* use R/M and be sure!





Minimize the risk of failure of a "Teflon" part or component at a critical point by getting it from Raybestos-Manhattan.

R/M has been in the forefront of "Teflon" development from the beginning. Our engineers know "Teflon" its unique characteristics its vast

R/M has been in the forefront of "Teflon" development from the beginning. Our engineers know "Teflon"... its unique characteristics, its vast potentialities, how it can best serve the needs of a particular industry. Our production men have the know-how and complete plant facilities to follow through. You specify it; they fabricate it.

We understand the complexities of

We understand the complexities of your problems—friction, extremes of temperature, the corrosive action of exotic fuels, etc.—and will give you all possible aid in designing, molding or machining the R/M parts or components you need. Make R/M your headquarters for all the "Teflon" products you need. Call upon your nearest R/M district office for the cooperation you require—or contact us direct.



R/M "Teflon" products for aircraft, missiles and space vehicles include thin-wall tubing (with or without identifying stripe); flexible wire braid covered "Teflon" hose; sheets, tape, rods and tubes; rods, sheets and tubes in certified and stress-relieved grades (X-ray films for sheets and rods are available); centerless ground rods held to very close tolerances.

*A Du Pont trademarl



RAYBESTOS-MANHATTAN, INC.

PLASTIC PRODUCTS DIVISION FACTORIES: MANHEIM, PA.; PARAMOUNT, CALIF.

Contact your nearest R/M district office listed below for more information or write to Plastic Products Division, Raybestos-Manhattan, Inc., Manheim, Pa.

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RAYBESTOS-MANHATTAN, INC., Engineered Plastics • Asbestos Textiles • Mechanical Packings • Industrial Rubber Sintered Metal Products • Rubber Covered Equipment • Abrasive and Diamond Wheels • Brake Linings Brake Blocks • Clutch Facings • Laundry Pads and Covers • Industrial Adhesives • Bowling Balls

Circle No. 275 on Reader Service Card in Product Review Section

Aircraft Missile Engineering

Materials



ANODIZED strip or wire can be made by this pilot machine Permaluster. In continuous process, material is anodized in minute, depending on thickness.

Aluminum oxide insulation looks good for high temperature wire

Multi-layer coatings with an aluminum oxide base may lead to strip and wire that can take as much as 1900 deg F, recent experimental developments indicate.

THERE is a prime need today for good insulation for wire and strip for solenoids, transformers, etc., that will take temperatures as high as 1000-2000 deg F. One possible answer to this requirement is the use of aluminum oxide coating, which has a melting point of 3660 deg F.

A recent advance in this area was the development of flexible anodic coatings by such firms as Permaluster, Inc., 2012 W. Burbank Blvd., Burbank, Calif. (see AvAge, "Anodized Aluminum Wire Takes 1000 Deg F," Sept. '57,

p. 64). Wire with such an ano dized coating—of 0.00015-in thickness—could be ben around a radius three time that of the wire without crazing or fracturing the film.

The same firm has since extended its process to provide flexible aluminum oxide coatings for aluminum strip. Work on using such insulation for other metals, such as silver promises to provide wire anstrip that will take temperatures as high as 1900 deg F

Company officials to lead to SPACE/AERONAUTICS that pilot line is in operation to produce anodized aluminum strip in widths up to 24 in Coating thickness on wire of strip can be controlled from allow as 0.00008 up to 0.001 in., they report. A number of transformer manufacturers are testing units using this material, since weight reductions of up to 50 per cent plus space.

ngs can reportedly be led over conventional ma-

. H. Hadley, general manof Robert M. Hadley Co., os Angeles, told SPACE/-CONAUTICS his firm is workon a three-phase, 400-cycle er transformer to supply ifiers for operation up to deg C. So far, he reports, s good. The unit has a d space factor, he reports, has stood up under humidand vibration tests (Mil-E-2). The prototype has been \pm 55 to \pm 125 deg C. It stood a temperature rise 100 deg C with little or no t on properties.

Pilot runs are also being de on copper wire and strip the aluminum oxide insulaa. The copper must first be ted to eliminate galvanic aca and diffusion in high temature operation. Permalusis making material with her silver or nickel plate, rious research studies sponed by the services are also king into other platings,

h as gold.

Work on actual uses of minum - oxide - insulated, kel-plated copper is still in early stages and little data presently available. In one et, however, Cannon Electroports successful continus operation for several days temperatures of 1200 deg F a prototype dc solenoid de of this wire.

One of the major problem cas is that aluminum oxide ulation tends to be hygroppic in the presence of moise, which can adversely after the dielectric strength. The dielectric strength is particularly true at the voltages. One solution ggested by Permaluster is to pregnate the oxide with such terials as silicone, zircoum, or magnesium oxide.

A major competitor of anozed insulation is ceramic-ornic insulation. WADC is

more on next page



...and now for everyone

The wraps are off RADAN! The civilian counterpart of GPL's famous military Doppler auto-navigators, is ready and available *now* — for anyone and everyone!

RADAN Systems have behind them millions of miles of experience in transcontinental, oceanic and polar flights... save precious time and fuel... provide a priceless margin of safety.

- RADAN accurate: within 1% for ground speed, within ½° for drift angle
- RADAN small: 4.4 cu. ft.-light: 89 lbs.
- RADAN operates without ground or celestial aids
- RADAN virtually maintenance-free

Now in quantity production, RADAN Systems are manufactured by GPL, who developed and is the nation's leading producer of Doppler air navigation systems. Address inquiries to: General Precision Laboratory Incorporated, Pleasantville, New York.



ENGINEERS - GPL achievements have opened up some unusual research and development opportunities. Send resumé to Personnel Manager.

Circle No. 187 on Reader Service Card in Product Review Section



When the finished product comes off the board and takes to the air, the individual skills of all who contribute are tested to the fullest.

Even the prosaic fastener is under stress scrutiny . . . and the men at Chandler are aware of the responsibility placed upon them.

Chandler cold forged fasteners are airborne all over the world in aircraft of all types. They are serving well because they have been produced from the finest steel stock available. They have been inspected and rechecked by skilled technicians.

Chandler rides herd on precision . . . to produce a fastener of the highest standards of precision, uniformity and economy . . . so that precision can ride herd with America's pilots everywhere.

> Specialists in Thread-Rolling After Heat-Treating



1493 Chardon Road • Cleveland 17, Ohio

Circle No. 3 on Reader Service Card in Product Review Section

High Temperature Test Results of Anodized Aluminum Wire*

10 Mil Wire	Maximum	Minimum	Avera
Mercury Dielectric Test	130 V	90 V	104
Twisted Pairs Dielectric Test 1.5-lb tension	560 V	460 V	1516N
After 24 hr @ 90° & 90% RH After thermal shock	430 V 510 V	410 V 430 V	420 V 473 V
Mandrel Dielectric Test 0.5-lb. tension	70011	20014	20.11
½ In. ¼ In. ¼ ₁₆ In.	300 V 340 V 320 V	260 V 300 V 290 V	284 V 324 V 806 V
Insulation Resistance Room temperature	300x103 r		141x103 megir
After 40 hr @ 600° C Dielectric Breakdown	87 meg-of		58.6 meg-ohr s
After 40 hr @ 600° C	325 V.	300 V.	308 V

12.6 Mil. Wire

Dielectric Breakdown On light coating at: 23 Deg C 200 Deg C	258 V	212 V	226 V
	255 V	220 V	241 V
350 Deg C	238 V	188 V	221 V
600 Deg C	244 V	225 V	237 V
Insulation Resistance 200 Deg C 350 Deg C 600 Deg C	9 5x109 ohms 7.6x108 ohms 11x107 ohms		4.6×109 ç 4.0×105 a 7×107 a

*Data from Genèral Electric test program.

ALINA DIAL TEST INDICATOR

A fully reversible indicator incorporating a clear easy-to-read, adjustable dial, graduated in half thousandths of an inch, a switch lever on the body to reverse action and TWO hardened chrome plated contact points of .030" and .100" diameter that are equipped with a non-slip ratchet. Furnished complete with a universal shank, body clamp, and plush lined steel case.



WRITE FOR LITERATURE AND COMPLETE CATALOG

CORPORATION ALINA

122 East Second St., Mineola, L.I., N.Y.



NNON ELECTRIC dc solenoid ng aluminum-oxide-insulated, kel-plated copper wire was rated for several days in nco furnace at 1200 deg F. non engineers emphasize, vever, the new insulation must 1 undergo rigorous test prom.



ANSFORMERS using alumimodian aluminum are being studied by Robert Hadley Co. Test units shown re reportedly have space factore to unity, are 50 per nt lighter than units using oper foil.

onsoring two major developent programs in high temrature wire insulation. In the, researchers at Georgia Intute of Technology are adying the use of frits for otecting copper wire. At high imperatures (1300-1500 deg b), researchers report, the ramic-organic materials they adied presented problems of etting the base metal and adresely affected electric propties.*

This indicated need for a

D. Walton & J. N. Harris, "Highoperature Insulation for Wire," DC Technical Report 58-13 (Part I)

more on next page

THE REQUIREMENT:

To design and build a quick-check, highly accurate, dual system, portable pneumatic test bench for sensing and locating air leakage in the canopy seal and cabin/cockpit of pressurized aircraft.

THE EQUIPMENT:

The SUN Model SX-106, Air Leakage Cabin/Cockpit Pressure Tester. Comprising two separately controlled pneumatic systems, operating from a common 100 psig shop air source, this unit provides a dynamic air flow of 200 SCFM with a pressure load up to 15 psig. Static pressure may be regulated up to 30 psig. Simple controls and instrumentation relate air temperature, pressure and true flow reading to pounds per minute air loss.

THE BUILDER: The Aeronautical Division of Sun Electric Corporation — one of the foremost experienced designers and producers of aeronautical and missile testing and support equipment. For any problem involving electrical, electronic, hydraulic, pneumatic, or related equipment, you'll save time AND money by consulting Sun's Engineers FIRST!

GENERAL SPECIFICATIONS

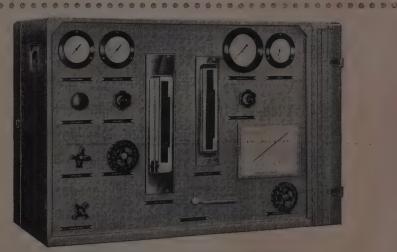
Length 38½ inches Height 28½ inches Depth 22½ inches Weight 290 pounds

Operating Range: 40°F to 120°F at 80 psig

(a)

High and Low Range
Flowmeters — 20 to 200 SCFM
and 2.9 to 29 SCFM
±2% full scale

Internal pressure relieving components protects aircraft against over-pressurization.



Model SX-106 Aircraft Cabin/Cockpit Air Leakage Tester. All connections between tester and aircraft made at rear of unit.



AERONAUTICAL DIVISION • HARLEM AND AVONDALE • CHICAGO 31, ILLINOIS

Circle No. 12 on Reader Service Card in Product Review Section

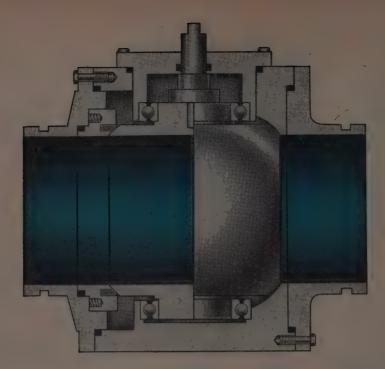


ALUMINUM OXIDE insulation for aluminum wire looked promising at temperatures approaching 1200 deg F in General Electric tests. Test coil is shown here after soak at 1200 deg F for 384 hours.

base coating of the wire. So far, indications are that aluminum oxide is best for this coating. The ceramic-organic layer applied to the anodized coating would then solve the hygroscopic problem.

Current programs indicate anodized aluminum wire is good at temperatures as high as 1200 deg F, Permaluster reports. General Electric tests of heavily anodized 10 mil wire showed an insulation resistance of 300x103 megohms maximum and 141x10³ megohms average for samples of Permaluster wire at room temperature. After 40 hours at 600 deg C, the respective values were 87 and 58.6 megohms. Dielectric breakdown after 40 hours at 600 deg C was 325 V maximum, 300 V minimum, and 308 V average.

Dielectric breakdown for lightly anodized 12.6 mil aluminum wire at 23 deg C was 258 V maximum, 212 V minimum, and 226 V average. At 600 deg C, the respective values were 244, 225, and 237 V. Insulation resistance at 200 deg C was 9.5x109 ohms maximum, 3x109 ohms minimum, and 4.6x109 ohms average. At 600 deg, the values were 11x10⁷, 2.2x10⁷, and 7x10⁷ ohms (see Table). Circle No. 60 on Reader-Service Card for more information.—IS



HYDROMATICS FLO-BALL VALVES

The accepted standard of maximum performance and reliability

100% FLOW EFFICIENCY

Hydromatics' exclusive FLO • BALL design provides a straight-thru unrestricted fluid path, exactly equal to the pipe line diameter.

PERFECT SEALING

Zero leakage is assured through the use of a precision ball—the ideal geometric form for perfect sealing contact.

LOW OPERATING TORQUE

Ball rotates in precision bearings which absorb all pressure loads. Pressure balanced valve seat further minimizes forces on ball, reducing frictional drag.

LONG LIFE

Seat is always in sealing contact with the ball surface, resulting in a self-wiping, self-lapping action that insures long, trouble-free life.

HIGH SPEED ACTION

Only 90-degree rotation is required to fully open or close valve. Full travel as fast as 5 milliseconds.

RELIABILITY

Simple construction, with only one rotating part, provides built-in reliability and rugged, dependable operation.

CRYOGENIC AND CORROSIVE APPLICATIONS

Hydromatics' FLO*BALL valves, with new diaphragm sealing and unrestricted fluid path, have been proved the best valves for operation with LOX, Liquid Nitrogen, Helium, Hydrogen Peroxide, Red Fuming Nitric Acid and Hydrazine.

MODULAR ARRANGEMENT

Only FLO®BALL design, with its rotating valve action, permits side-by-side grouping of several valves, all driven simultaneously by a single actuator.

VERSATILITY

Only FLO®BALL design makes possible the interchanging of manual, motor or pressure actuators without changing the valve body.

HYDROMATICS, the world's leading designer and manufacturer of high performance ball valves for military and industrial applications, offers the most extensive selection of designs to meet all your requirements; Manual, motor or pressure operated. For cryogenic, corrosive or general service media. Pressures from vacuum to 10,000 psi. Sizes from ¼ inch to 12 inches.

Hydromatics, Inc.

70 Okner Parkway, Livingston, New Jersey

Hydromatics, Inc.



HYDROMATICS FIELD ENGINEERING OFFICES:

Pasadena, 35 N. Arroyo Pkwy., RYan 1-7448 / Denver, 829 15th St., AMherst 6-2714 / Washington, 1413 K St. N. W., STerling 3-3612

CABLE JACKET ithstands extreme heat

hermazip is a cable jacket that protect components and cables a high temperature exposure missile blasts, and engine or the temperature aluminized, retive, asbestos fiber jacket without 2000 deg F of continuous 2000 deg F of continuous F for brief periods, says The pertubing Co., Dept. S/A, 752 can Pedro St., Los Angeles 14,

hermazip, which is closed by astic of metal zipper track, can applied very rapidly. It is ilable in id's from % in. up, acrements of 1/2 in.

Circle No. 204 on Reader-Service Card

THERMOCOUPLE GLAND seals eight wires



Up to eight bare wires can be minated and sealed at pressures m. 005 mu to 20,000 psi by Type C bare wire thermosple gland, says Conax Corp., pt. S/A, 2300 Walden Ave., ffalo 25, N.Y. An open-end tube available for mechanical protion of the wires in high velocor viscous fluids.

The gland is furnished for asnbly or completely assembled. is suitable for 14, 20 or 24ge wire. Immersion depth of the t junction is adjustable to any gth. The glands are made of 3 stainless steel and are equipd with neoprene, Teflon or lavadants, and ceramic insulators.

Circle No. 205 on Reader-Service Card

GATE VALVES for cryogenic use



Gate valves that can be used as pre-valves or fill valves under line pressures up to 60 psi and over a —320 to +250 deg F temperature range are available from Koehler, Aircraft Products Co., Dept. \$/A, 409 Leo St., Dayton, O. The valves are suitable for systems involving LOX, other cryogenic liquids, and standard aircraft and missile fuels and oxidizers.

Valve actuation can be accomplished manually, electrically, hydraulically, or by pneumatic pressure.

Circle No. 206 on Reader-Service Card

SERVO SYSTEM has feedback network



This completely-integrated, electro-hydraulic servo system is being used for control of hot gas turbine auxiliary power supply, according to Hydraulic Research and Mfg. Co., Dept. S/A, 2835 N. Naomi St., Burbank, Calif. The device has an output velocity proportional to the input and time derivative of the input electrical command.

A hydro-mechanical rate feedback network is used to generate the derivative computation. The network measures and feeds actuator velocity back to the servo valve flapper to achieve velocity anticipation.

Circle No. 207 on Reader-Service Card

VOLTMETER aligns missile controls

This phase angle voltmeter is being used for control equipment alignment in the Atlas Missile, says North Atlantic Industries, Inc., Dept. S/A, 603 Main St., Westbury, L.I., N.Y. The device may also be used for phase-sensitive null indication.

Another function includes use as a ratiometer for zeroing and testing precision synchros, servomechanisms, and transducers.

Circle No. 208 on Reader-Service Card

SENSOR is triple-shielded



Air temperature measurements substantially independent of solar and other radiation are possible with a temperature sensor housed in a triple-shielded enclosure, says Beckman & Whitley, Inc., Dept. S/A, San Carlos, Calif. The device is equipped with an aspirating blower and a support arm.

The thermal-sensitive element is surrounded by a stainless steel inner shield which, in turn, is housed in a cylinder of aluminum. The outermost shielding element is hat-shaped, and it reduces further the maximum solar radiation error. The shields provide radiation shielding effective to a maximum temperature deviation of 0.2 deg F.

Circle No. 209 on Reader-Service Card

TRANSFORMER has doubled range

The null position of the Series ES-L has been placed at the end of core travel for increased range, says Schaevitz Engineering Dept. S/A, Route 130 & Schaevitz Blvd., Pennsauken, N.J. The single-ended linear variable differential transformer has a range about twice that of a conventional transformer of similar size.

It has a nominal frequency range of 60 to 10,000 cps. Ambient temperature range is -65 to +200 deg F.

Circle No. 210 on Reader-Service Card more on next page



1/2" MOTORIZED FLO.BALL VALVE 100% flow efficiency — for general service media up to 3000 psi

Extremely light weight and compact, suitable for a wide range of aircraft and missile applications, both airborne and ground support. Provides perfect sealing and 100% flow efficiency for general service media (jet fuels, hydraulic fluids, water, alcohol, helium, hydrogen, oxygen, etc.) at pressures up to 3000 psi and temperatures from -65° to 200°F. The DC motor actuator includes automatic current shut-off and a positive Geneva-Lock mechanism. Valve is of bi-stable design — in the event of electrical power failure, the valve will remain in its last position, either open or closed. Included is a built-in microswitch for remote observation of valve position. Flanges, which are removable and interchangeable, may be specified per AND-10050, AND10056, NPT or ASA.



FLO-BALL PROPELLANT VALVE Dual line flow control of cryogenic media at 1000 psi

For simultaneous flow control of Liquid Oxygen and Fuel, this prop valve provides zero leakage control of a 5/8" LOX line and 1/2" Fuel line at pressures up to 1000 psi and temperatures as low as -350°F. A single-acting, spring-return pressure actuator drives a precision mechanical linkage which controls both valves, thus insuring constant timing. The valves are extremely fast acting, with response times as short as 10 milliseconds. All dynamic sealing is double with a vent between seals for maximum safety. A sealed, rotary action snap switch permits remote observation of valve position. Valve flanges, available in standard or special styles, are removable and interchangeable.

Hydromatics, Inc.

70 Okner Parkway, Livingston, New Jersey
Circle No. 189 on Reader-Service Card



SWITCH is very light



A tiny mercury switch that weighs 1.8 gm, including leads, offers sensitive, trouble-free operation in applications such as computers, scales and other devices, says Micro Switch Div., Minneapolis-Honeywell Regulator Co., Dept. S/A, Freeport, Ill. The AS419A1 may be mounted in any position through 360 deg around its longitudinal axis, and it may be actuated by slow, snap or fast-tilting action.

The low shift of mass involved in actuation facilitates gang-mounted assemblies. The spdt switch is rated for a resistive load of one amp, 30 V ac or dc, and an inductive load of .75 amps, 30 V ac or dc.

Circle No. 211 on Reader-Service Card

PRESSURE SWITCH is adjustable



A switch designed for all aircraft hydraulic systems requiring a pressure sensor to actuate the electrical apparatus has been developed by Consolidated Control Corp., Dept. S/A, Bethel, Common The unit, Type 6512, features an enclosed snap action and an operating point that is adjustable within the pressure ranges of 506 to 2000 psi and 2000 to 3500 psi

The switch is made with lapped piston or an O-ring sealer piston hydraulic actuator. In addition to standard aircraft fluids, operating media include MIL-15606 hydraulic fluid and MIL-1. 7808 oil. The spdt unit has a contact rating of five amps resistion 2.5 amps inductive at 28 Media.

Circle No. 212 on Reader-Service Card

Split-Second Starters with Super Finishes

Advanced retainer design and rigid manufacturing control for MPB bearings eliminate torque pulses due to hang-up; micro-finished balls and raceways assure low torque with minimum variations and maximum reliability. Write MPB Inc., 1401 Precision Park, Keene, N. H., for catalog



MPB helps you perform miracles in miniaturization

MPB

MINIATURE PRECISION BEARINGS INC. Circle No. 191 on Reader Service Card in Product Review Section

SPACE/AERONAUTIC

COUNTER has built-in readout

h-speed electronic counting have been combined in a frecy-period counter with a e of zero to 220 kc. The Model B, whose applications include and piston engine test cells, amometer test stands and misountdown tests, has a memory it that holds readings while counter cycles to accumulate data, says Computer Measurets Corp., Dept. S/A, 5528 eland Ave., North Hollywood,

ne period range is ten usec to 000 sec. Numerals are 21/4 in. each digit consists of 40 small bulbs that make readings ble up to 150 ft. The device is gned with a temperatureulated, crystal-controlled time surements. For period measment, six internal standard fre-

Circle No. 213 on Reader-Service Card

FOIL in 15 in. width

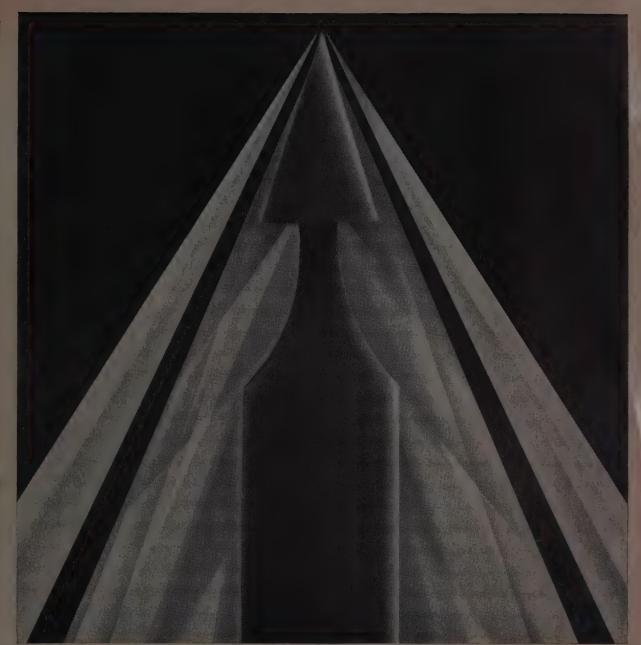


Co-Netic magnetic elding foil is now offered in in. width. The new width simfies the enclosure of a much ger volume by minimizing the mber of junctions needed when e foil was available only in ir inch widths, says Magnetic ield Div., Perfection Mica Co., ppt. S/A, 1322 N. Elston Ave.,

nicago, 22, Ill.

The foil can be trimmed to any nension or outlined with a pair ordinary scissors and is easily med by hand. One or more vers can readily be applied and ir effects observed. Co-netic foil non-shock sensitive, non reten-e, does not require periodic anling and attenuates low level

Circle No. 214 on Reader-Service Card more on next page



Shock wave around a simulated missile

SUPERALLOYS for supersonic performance

We develop them. And through modern vacuum melting produce them to the closest possible limits of chemical control and uniformity of mechanical properties for use in such critical components as aircraft turbine blades, buckets and skin material for missiles and rockets. UDIMET 500, one of our well established vacuum melted alloys, combines unsurpassed stress-rupture life with excellent ductility and fatigue strength in the 1200°F to 1800°F range. At 1200°F, for example, its tensile strength is 180,000 psi -at 1600°F, 100,000 psi. For further information concerning applications of these superalloys to your products, write to address below.

VACUUM MELTING provides these properties

- High temperature
- Increased ductility
- Extreme cleanliness
- Precise chemical control
- Longer stress-rupture life
- Increased tensile strength • Better fatigue resistance
- Greater yield strength
- Greater impact resistance

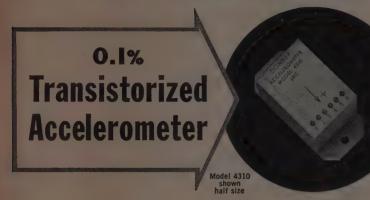
• Greater creep properties

KELSEY-HAYES CO. OT. M. REGISTERED

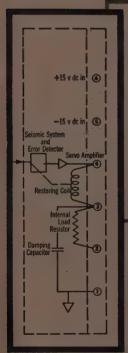


SOME ALLOYS COVERED BY U.S. PATENT #2009110

HIGH OUTPUT -HIGH RESOLUTION



Donner's new transistorized Model 4310 accelerometer is designed for demanding measurement and control functions under severe environmental conditions. With applications in telemetering, navigation, control, gyro-erection, and short range inertial guidance, the 4310 is especially appropriate where low weight, small size, and high output are important considerations. When an especially small light weight sensing unit is needed, the acceleration sensing portion of the instrument can be separated from the servo-amplifier. Typical applications in this category include gyro and platform measurements of acceleration, velocity, and displacement.



OPERATIONAL HOOKUPS OF 4310

A variety of specific output arrangements are possible with the standard Model 4310 Accelerometer. For a full scale voltage output of approximately $\pm 7\frac{1}{2}$ volts, jumper terminals 1 and 2, read output across 1 and 3. For a full scale current output of approximately ± 1.5 ma, connect a series load between terminals 1 and 2. For a standardized voltage output, connect a resistive voltage divider network across 1 and 3 with a total resistance of approximately 5000 ohms to yield any prescribed output below $\pm 7\frac{1}{2}$ volts full scale. Standard modifications of the 4310 include provision for biasing of output; fluid-filling for mechanical rejection of high frequency vibration; and operation from 0-28 volts dc or ±28 volts dc with higher output,

KEY SPECIFICATIONS

STANDARD RANGES

Within 0.05% deviation from best fitted straight line Between ± 0.1 g full range and ± 30 g full range. Lower and higher ranges available on special order. Better than 0.0002% full scale

RESOLUTION WEIGHT OUTPUT EXCITATION DOMESTIC PRICE

±71/2 v dc and/or ±1.5 ma full scale Plus 15 v dc, 5 ma max.; minus 15 v dc, 5 ma max. Standard instrument \$450.00 F. O. B. Concord, California, Modifications extra

Donner engineering representatives are located in principal areas. For the name of your nearest representative and complete technical information, please address Dept. 091.

ONNER SCIENTIFIC

CONCORD, CALIFORNIA

Phone Mulberry 2-6161 . Cable "DONNER"

Circle No. 192 on Reader Service Card in Product Review Section

RELAY for dry circuitry



SWITCH SERIE

is explosion-pro

This HG-4SL series relay in size and mounting dimensions, meets proposed MS drawings and MIL-R-25018 relay specifications. Rated for dry circuit operation through ten ampere contacts, it is available in one, two, three, and four pole Form A, B, or C contact arrangements, says HI-G, Inc., Dept. S/A, Bradley Field, Wind-

sor Locks, Conn.

It can be operated from voltage sources from four volts through 250 V. It meets MIL-S-901B shock specifications.

Circle No. 215 on Reader-Service Card

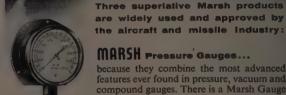
A new series of explosion-pro switches feature double condi openings. Designed the Exseries, they carry full Underwrite Laboratories listings for hazardor atmospheres of Class 1, Group (ethyl ether vapor) and Group (gasoline, petroleum naphtha, cohol, acetone, lacquer solvent pors, and natural gas); Class I Group E (metal dust), Group (carbon black, coal and coke dust and Group G (grain dust); switches have adjustable roller-a actuators with non-sparking lers, says Micro Switch, Den S/A, Freeport, Ill.

The aluminum alloy housing the EXL series has ample room for wiring through both condu

openings.
Circle No. 216 on Reader-Service Can
more on page 16

First choice of the rocket

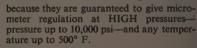
and missile industry ...



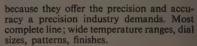
the aircraft and missile industry:

MARSH Pressure Gauges... because they combine the most advanced features ever found in pressure, vacuum and compound gauges. There is a Marsh Gauge for every conceivable application.

MARSH Needle Throttling Valves ...



MARSH Dial Thermometers...



All Marsh products available with AND threads





Marsh Instrument & Valve Co., (Canada) Ltd., 8407 103rd St., Edmonton, Alberta, Canada, Houston Branch Plant, 1121 Rothwell St., Sect. 15, Houston, Texas

Circle No. 193 on Reader Service Card in Product Review Section SPACE/AERONAUTION



rugged reliability . . . at a production price.

* 2° to 6°F differentials available



Circle No. 194 on

STEVENS manufacturing company, inc. Mansfield, Ohio





Compact! Easy to Read! **Union Data Display Indicators**

Union Switch & Signal makes two types of electro-mechanical, DCoperated data display indicators: digital types, displaying 10, 12, or 16 characters on a wheel; and alpha-numerical types, displaying up to 64 characters on a Mylar* belt. Character assignments can be furnished as required.

TRANSLATION Both Digital and Alpha-Numerical Indicators operate directly on binary codes on a null-seeking basis. This eliminates the need for external equipment for translation from binary to decimal code, as required with other display devices.

VISUAL READ-OUT Indicator packages are designed for quick, easy readability, even when indicators are mounted in rows.

INFINITE RETENTIVITY The indicators require power only during the response time, because they are of the null-seeking type. Once positioned, the indicators retain the data visually and electrically until a new code is transmitted.

ELECTRICAL READ-OUT The design of the decoding and control portions of the indicators provides electrical read-out of data in the same form as the input. The data can be read continuously or periodically without erasing the stored information.

USES These indicators can be used in the output of digital computers, in teletype receiving equipment, in telemetering systems, or wherever data needs to be displayed. *Dupont's synthetic fiber.

Bulletin No. 1015 gives you complete information.

"Pioneers in Push-Button Science"

UNION SWITCH & SIGNAL DIVISION OF WESTINGHOUSE AIR BRAKE COMPANY

PITTSBURGH 18, PENNSYLVANIA

Circle No. 195 on Reader Service Card in Product Review Section

TORCH IGNITER for turbo-iets

This torch igniter, designed for high altitude turbojet requirements, features the use of a combination of oxygen and IP-4 fuel to provide a high velocity ignition flame eight to ten inches in length at a calculated temperature of over 3000 deg F. Each igniter unit produces 950 btu/min, or 7500 joules, says Carleton Aviation Co. Inc., Dept. S/A, East Aurora, N. Y.

a wide range of external pressure conditions without variation in fuel or oxygen pressures. The igniter is impervious to high or low operating temperatures.

Circle No. 217 on Reader-Service Card

COUPLING passes severe burst test

The Type 5-5002-8 quick-disconnect coupling for missile and aircraft hydraulic and fuel lines has withstood 22,000 psi in a burst test, according to On Mark Couplings, Inc., Dept. S/A, 4440 York Blvd., Los Angeles 41, Calif. Because the device contains neither steel balls or dogs, it can manually connected or disconnec ed under high pressure.

The design is also available remotely actuated couplings. I locking mechanism of the dehas a 360-deg gripping action, a it will compensate itself as wer

Circle No. 219 on Reader-Service Ca

SHIELDING in large size

A neutron shielding materia for atomic power plant use is n available commercially in its larg est plate and sheet sizes to dat says Aluminum Co. of Americ Dept. S/A, 1501 Alcoa Bldg Pittsburgh 19, Pa. It may be ob tained in sizes of 48x120 in., an 36x96 in. thicknesses for both size are .250 and .125 in.

Boral plate .250 in. thick is ported to have an effective shie ing power equal to that of a 2 in. concrete slab The material available in two standard tempers, annealed and as fabricated

Circle No. 220 on Reader-Service Car

more on page 16:

\$1064 to Overhaul Aircraft Generator at Factory \$60 to Rhodium Plate Commutator and Slip Rings

WHICH ONE WOULD YOU PREFER?

Dalic plating makes the difference. Using a specially developing solution and Dalic selective plating equipment, useful operational time of the main generator of a bomber has been



increased to some 350 hours. Formerly, time between mandatory overhauls was only about 70 hours.

SAVINGS: Two or three overhauls, at \$1,064 each, plus cost of time and labor to remove and reinstall generator.

If you're concerned with military or civil aircraft, missile controls, servos, high-quality electrical equipment, investigate rhodium plating with Dalic equipment. It offers the following

Longer commutator life • Reduced contact resistance • Higher starting and running torques • Elimination of static and noise in radio and other electronic equipment • Reduces temperature by as much as 80° C • Increased power/current ratio

For further information on Dalic plating, write to:

ARLANE DEVELOPMENT CO. 153 East 26th Street New York 10, N. Y. PIDDINGTON & ASSOCIATES LTD. 1219 East Foothill Blvd. asadena, Calif. OHIO METACHEMICAL, INC. 2742 Second Street Cuyahoga Falls, Ohio D & S AVIATION CO., LTD. 671 Lauretides Blvd. Pont Viau, Quebec

DALIC METACHEMICAL LTD.

121 Judge Road, Toronto, Ontario

expanding the frontiers of technology... over the full spectrum of advanced electronics



BRUBAKER ELECTRONICS, INC. subsidiary of TELECOMPUTING CORPORATION

are dynamically attacking, and overcoming the highly specialized electronic barriers associated with space-age technology. A skillful blending of technical ability, competitive production capabilities, and extensive testing facilities has established Brubaker Electronics as top-flight experts, in the research, design, and development of complex electronic systems and components for both military and industrial applications. Brubaker's experience, personnel, and capabilities, together with a well-integrated research program, are the reasons why Brubaker equipment is operational on so many of the nation's vital weapons systems.

Past achievements show why Brubaker is superior in the area in which it operates: coding and decoding systems, radar, radar beacons, IFF, telemetering, communications and custom test equipment, highly classified military electronic systems – and such components as networks, delay lines, pulse transformers, switches and relays. If you have a problem in advanced electronics, Brubaker engineers have a solution. Wire, write or phone:

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Telephone: TExas 0-6441
TWX Culver City, Calif. 7239

cie No. 197 on er Service Card Product Review Section





Up a stairway on a hurry-up job,

nothing to it

with the big, 10-inch diameter, semi-pneumatic balloon tires.

Emergency repair work across the plant,



compact Tool Truck carries full tool selection to the job.

Job in another building,



the Tool Truck

eases heavy loads over rough pavements or tracks. Drawers

are held shut during travel by a padlocked bar.



The new Snap-on Tool Truck is invaluable wherever tools

or parts transport is needed. Plenty of tool space with four

husky drawers.



Upper panel section is par-

titioned for nuts, bolts, parts.



Ample storage space

throughout.



A demonstration will prove the Tool Truck's money-saving ability. Call your nearest Snap-on branch or write us direct.

SNAP-ON TOOLS

8080-A 28th Avenue • Kenosha, Wisconsin
Circle No. 198 on Reader Service Card in Product Review Section

INERTIA SWITCH actuated by vibration

Once actuated, the 1SA1 inertia switch will remain open and cannot be jarred back into an unactuated position. To reset the switch, a manual reset button is provided. Electrical reset may also be provided, says Micro Switch Div., Minneapolis-Honeywell Regulator Co., Dept. S/A, Freeport, Ill.

The switch operates when subjected to a thrust in any horizontal direction. A vibration of thrust under the minimum duration of 1/30 of a second will not actuate the switch. Force-of-actuation settings are factory adjusted and may be supplied from 1.5 to 10 or more G, depending on requirements.

Circle No. 221 on Reader-Service Card

CLAD-COPPER WIRE for high temperatures

Oxalloy 28 is a copper wire that has been clad with a chrome-iron alloy to enable it to withstand high temperature (1300 deg F), and corrosive and oxidizing conditions, says Sylvania Electric Products, Inc., Dept. S/A, Warren, Pa. The conductor is said to be ideal

for use as leads to electrical appl ances and vacuum tubes, and a hook-up wire.

The wire is available in roun wire sizes from 0.005 to 0.250 in dia. and in tempers rangin from soft annealed to full hard, ribbon form comes in widths u to 0.125 in.

Circle No. 222 on Reader-Service Card

SWIVEL FITTING has low torque rating

Torques as low as one lb-in under all pressure conditions are featured by the Series 131J10 dual swivel fitting, which provide 360-deg swiveling at each end says Altair, Inc., Dept. S/A, 5 MacQuesten Parkway, South Mount Vernon, N.Y. Other characteristics include low pressure drop no leakage, and long life.

The right-angle swivel, which is not adversely affected by impulse conditions, is said to be particularly useful where positioning excess flexing, vibration, or twisting are problems. It has an operating pressure of 3000 psi and a temperature range of -65 to +300 deg F. It is available in 1/4 in. tube sizes.

Circle No. 223 on Reader-Service Care

BARCO

Flexible Joints for Handling Liquid Oxygen and other missile fuels

BARCO has the specially designed flexible joint which has passed the official "Qualification Test":

- Used in piping on fueling lines handling liquid oxygen, JP4 and JP5 fuel, white and red fuming nitric acid, and other chemicals.
- Provides swivel motion to allow for thermal expansion and contraction.
- High corrosion resistance. Other special designs available. For temperatures from −320°F. to +1,000°F., and higher. Also joints for gas and hydraulic service.



BARCO Serving Industry Since 1908
MANUFACTURING CO., §75B Hough St., Barrington, Illinois

Circle No. 199 on Reader Service Card in Product Review Section

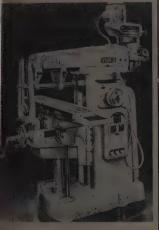
BATTERY CHARGER has constant current

In this automatic constant curate Battery. Charger each gnetic amplifier-silicon diode dule supplies an adjustable out-t of one to five amperes ±1 per t dec. The current is constant to a silver cell battery system aging from one to 50 cells. This smits automatic charging of any types of cell configurations the same charging unit, says to Electric & Mfg. Co., Dept. 4, 2806 Clearwater St., Los ageles 39, Calif.

Automatic shutoff is provided the four meter relays shown. her units available.

Circle No. 224 on Reader-Service Card

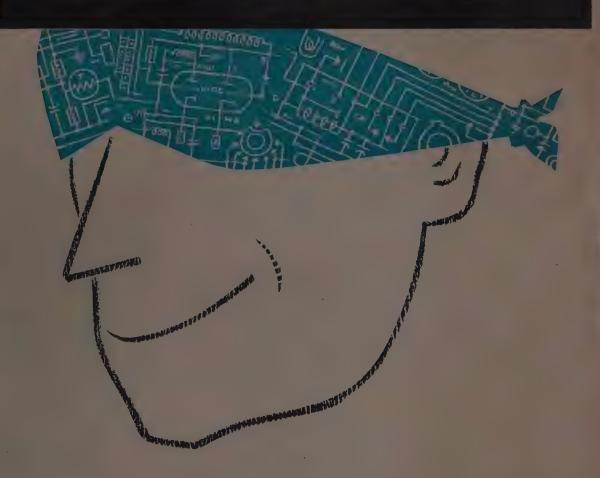
has triple purpose



The Morey-Ram VSR2 is a combination milling machine that flectively includes three different illers in one machine tool, according to Morey Machinery Co., nc., Dept. S/A, 383 Lafayette t., New York 3, N. Y. The evice's vertical and horizontal pindles, powered by a 3½ and we hp motor, respectively, can be sed simultaneously to mill the

The vertical spindle has 16 peeds, from 65 to 4750 rpm, and he horizontal geared head spindle has 12 speeds, from 30 to 1400 pm. The miller provides up to 8-in automatic longitudinal feed with preselective-type table feeds f ½ to 22 ipm, or rapid traverse f 100 ipm. Its 48x10-in table will scillate and can be used in surace grinding when table stops are a place. The lead screw, which uns in oil at all times, is three in diameter.

Circle No. 225 on Reader-Service Card more on next page Simmonds engineers are accustomed to seeing their way clearly through challenging problems, because precision and high performance are a must in all our products . . . sensitive electronic fuel management systems, fuel injection systems, electronic instruments and precision mechanical equipment. The size of our organization permits flexibility and speed in the application of new ideas. We are well known also for the exceptional service we render our customers.



We remove Blueprint Blindfolds...



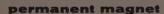
GENERAL OFFICES: TARRYTOWN, NEW YORK - BRANCH OFFICES: GLENDALE, CALIFORNIA - SAN DIEGO, CALIFORNIA - WASHINGTON, D. C. - DAYTON, OHIO ST. LOUIS, MISSOURI - DALLAS, TEXAS - DETROIT, MICHIGAN - SOLE CANADIAN LICENSEE: SIMMONDS AEROCESSORIES OF CANADA LIMITED, HAMILTON, ONTARIO

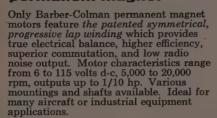




d-c small motors

compact, powerful-up to 1/10 hp







split-series

Barber-Colman split-series motors are available in two frame sizes with available in two frame sizes with continuous duty outputs up to nine millihorsepower. Outstanding efficiency due to excellent magnetic design and symmetrical lap-type armature winding. Electromechanical brakes can be supplied for these motors.



gearhead

Barber-Colman gearhead motors can be used as small actuators to drive used as small actuators to drive switches, programing devices, camera mechanisms, autopilots, and for remote positioning in industrial automation. Standard EYLM motor with gearhead, illustrated, provides up to 10 lb-in. torque output. Gear ratios from 9.5 to 55,446.1



with blowers, filters

Barber-Colman d-c motor-blower units Barber-Colman d-c motor-blower units quickly dissipate heat from hot tubes, circuit components, and other confined equipment. Air volume for a typical 1½-in. centrifugal unit is 20 cfm at 0 static pressure and 70°F. Voltages range from 6 to 115 volts d-c.

Barber-Colman permanent magnet motors also available with lightweight, compact, integrally mounted radio noise filters.

TECHNICAL BULLETINS ON COMPLETE LINE OF BARBER-COLMAN ELECTRICAL COMPONENTS

Detailed specifications, performance data, product and circuit drawings on polarized relays, resonant relays, d-c motors, tach generators, choppers. Write for your copy on any or all products.



BARBER-COLMAN COMPANY Dept. M, 1823 Rock Street, Rockford, Illinois

Small Motors

Automatic Controls

Industrial Instruments

Aircraft Controls Electrical Components • Air Distribution Products • Overdoors and Operators Molded Products • Metal Cutting Tools • Machine Tools • Textile Machinery

Circle No. 201 on Reader Service Card in Product Review Section

MICROSYN is miniature

The T-813 microsyn is a miniature unit useful for position-indicator applications in gyros, process control, and motion and torque amplification, among others, says Sterling Precision Corp., Dept. S/A, 17 Matinecock Ave., Dept. Washington, L. I., N. Y. Size is % in. od by %2 in. overall, and frequency range is 400 to 160

The T 813 S 1 signal generator has a 0.3 mv/mrad/ma sensitivity at an excitation of 400 cps. The T 813-T 1 torquer has a maximum torque of 3000 dynecms and a rated continuous torque

of 50 dyne cms.

Circle No. 226 on Reader-Service Card

SUBMINIATURE SWITCH is explosion-proof

The Type SS subminiature, snap-acting switch is a unit that may be safely used in explosive atmosphere, says Unimax Switch.
Div., The W. L. Masson Corp.,
Dept. S/A, Ives Rd., Wallingford, Conn. It is sealed against moisture, including splashing liq-uids, and dust.

The switch, which can be plied with integral and auxilia type actuators, features a flex bushing compressed around the actuator inside the phenolic of Case halves are bonded togeth Ratings are: five amps at 125, 1 V ac, or 30 V dc, resistive; 2½ amps at 30 V dc, induct Circle No. 227 on Reader-Service C

TRANSFORMER for transistor us

The units in this line compri the most complete U.S.-manufa tured group of transistor tran formers available from stock, a cording to Crest Transform Corp., Dept. S/A, 1834 North Ave., Chicago, Ill. The unit are all cored with 48 per ce nickel-silicon steel, are wound nylon bobbins, and are boxed translucent plastic housings.

They are made in two core size and are available with leads with T-type straps and stiff, tinne leads for printed circuits. Dime sions are 3/x3/4x1 in. and 5/x5/x8/4 i There are more than 45 transform ers in the line.

Circle No. 228 on Reader-Service Ca more on page It

Circle No. 202 on Reader Service Card in Product Review Section



is your answer!

The state and town to Adjust the series of t

REFASIL IS IN LARGE SCALE PRODUCTION, IN A VARIETY OF PHYSICAL FORMS, TO MEET THE DEMANDS OF THE MISSILE INDUSTRY.

itSFROCITE is not a unit of 15,000° F for short the above the constraint

Write for your copy of the new HITCO CAPABILITIES Brochure.



STROLITE

Product Bulletin No. PB7-24 describes its astonishing resistance to ultra-hightemperatures. Write



If you are a Missile or Spacecraft Designer or Manufacturer—with a need for ultra-high-temperature materials, you know the vital importance of low ABLATION rate.

ASTROLITE is a remarkable **Refrasil**-reinforced plastic with impressive resistance to Ablation **and** extremely high temperatures—**up to 15,000°F.** for short-duration applications!

Fabrication techniques for ASTROLITE have been perfected to give optimum fiber orientation to resist Ablation—while retaining maximum insulation properties.

Many major airframe and missile manufacturers are using ASTROLITE today in their Space Technology programs.

Perhaps you can use ASTROLITE for insulation of Rocket Nozzles, Nose Cones, Deflector Vanes, Blast Tubes or Combustion Chamber Liners. Engineering counsel is yours for the asking. Please write to Director of Research.

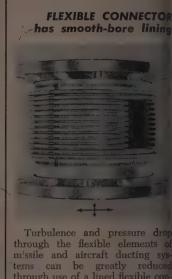


H. I. THOMPSON FIBER GLASS CO.

1733 Cordova Street, Los Angeles 7, Calif. • REpublic 3-9161

WRITE OH CALL YOUR NEAREST HITCO REPRESENTATIVE: EASTERN - Tom Kimberly, 38 Crescent Circle, Chesnire, Connecticut, BRowning 2-6544; Fred W. Muhlenfeld, 6659 Loch Hill Rd., Baltimore 12, Md., VALIEY 5-3135 - MIDWEST Survice L. Weddle, 1547 Penoryly, 16-51, iodianapolis J. Ind., ME 5-5681 - SOUTHWEST - Marshall Merits, 28506 W. Serry, Rm. 12, Fait Worth, Talay, Walhold Jearys - Morthwest J. 1. Laisen, 5157 Hallawer Place, Seaths, Walhold, Parkway 5-1517 - CAMADIGH FLANT, THE H. I. THOMPSON CO. OF CAMADIGH FLANT, ST., Guelon, Ontario, Tolerators, 24004, 24004.

Circle No. 231 on eader Service Card in Product Review Section



Turbulence and pressure drop through the flexible elements of missile and aircraft ducting systems can be greatly reduced through use of a lined flexible connector, according to Flexonics Corp., Dept. S/A, Maywood, Ill. A smooth-bore, stainless steel flexible tube inside the corrugated connector element eliminates turbulence caused by gas or liquid flow across corrugations.

The interlocking convolutions of the liner permit the full offset or axial motion for which the connector is designed, and the liner is relieved so that the corrugated element is the sole pressure carrier. The connector is reported to be especially useful in liquid oxygen systems and pneumatic power pining.

Circle No. 229 on Reader-Service Card

uses new principle

This relay, the Mark II, designed on a "revolutionary" new principle, employs a wedge action to achieve a positive contact in both the energized and deenergized conditions. It involves constantly increasing contact pressure during the over-travel period after initial engagement, says Electro Tec, Dept. S/A, South Hackensack, N. J.

sack, N. J.

The relay is a six-pole, double-throw, miniature, hermetically sealed unit meeting Mil-R-5757C and Mil-R-25018. Important features are its operating ambient temperature range from -65 to +200 deg C, its operating vibration resistance to frequencies from five to 2000 cps at 30 G, its operating shock resistance to 100 G, and its ability to switch currents reliable from dry circuit levels to

two amperes.

Circle No. 230 on Reader-Service Card

10 A WHOM B CONTE CANA BEARINGS

When Airspeed Makes the Weather Horizontal...

When the weather comes in at hundreds of miles per hour, no seal can keep rain out of exposed anti-friction bearings.

For such applications, Torrington Aircraft Needle Bearings and Cam Followers are made available in stainless steel. This material resists corrosion pitting and prevents the formation of areas of increased stress in bearing contact surfaces. Life expectancy is greatly increased for exposed airframe applications.

These advantages of stainless steel, coupled with the high inherent capacity provided by the full complement of rollers, make Torrington Aircraft Type Needle Bearings ideal where space and weight are at a premium. The Torrington Company, Torrington, Conn.—and South Bend 21, Ind.

TORRINGTON BEARINGS

District Offices and Distributors in Principal Cities of United States and Canada

NEEDLE • SPHERICAL ROLLER • TAPERED ROLLER • CYLINDRICAL ROLLER • BALL • NEEDLE ROLLERS • THRUST

Circle No. 203 on Reader Service Card in Product Review Section

TYPE RT-high strength stud

with rollers and outer race designed to roll on a hard-

TYPE NBC—inner race, outer

race, rollers and washers

securely fastened to inner

TYPE NBK-inner race, rol-

lers and self-aligning spheri-

cal OD outer race mounted in spherical ID ring.

TYPE NBF-extra heavy

outer race for heavy rolling

loads. Also available in

double row NBL series.

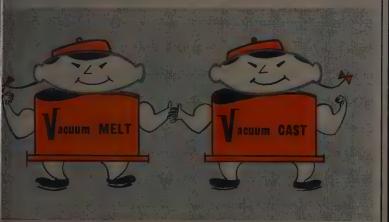
ened steel track.

nall, fixed-displacement cartype gear pump for aircraft issile use contains only two parts and can reach effiex piston types, says Adel on Products, Dept. S/A, Vanowen St., Burbank, Calif. amp is available for capacito 25 gpm, speeds up to rpm, and pressures to 4000

HYDRAULIC PUMP designed for efficiency

are furnished in four basic sizes. Cartridge construction provides hydraulically and thermally-control-led positive clearances: thus, mechanical drag for all pressures and operating conditions is eliminated. The pump will handle all ordinary oils, fluids and fuels, features low starting torque, and operates over a wide temperature range.

Circle No. 291 on Reader-Service Card more on next page





INTRODUCES "DOUBLE V"

A technique to enhance the properties of metals demanded by the aircraft, missile and space age vehicles.

"DOUBLE V" signifies the manufacture of cast components using vacuum melted primary ingots subsequently remelted, poured in vacuum, using investment or MONO-SHELL molds.

Misco engineers have designed vacuum equipment which permits continuous operation for High Volume Production. Charging of primary ingots, introduction of MONO-SHELL molds, melting and pouring, are carried out in vacuum through the ingenious use of charging locks.

For maximum benefits, melting temperatures are controlled optically and a vacuum pressure of 10 microns or less is maintained on mold and metal during pouring.

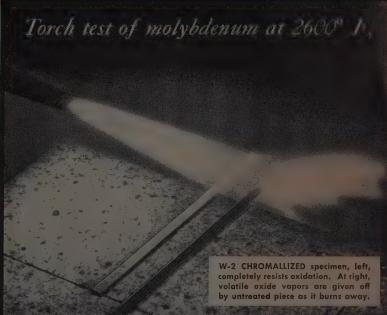
ORDER BY REQUESTING "DOUBLE-V"



Misco now provides a complete range of casting facilities to meet your requirements — AIR — INERT ATMOSPHERE — VACUUM, for your airfoil shapes now and in your future. Send us your inquiries and details of your requirements.

Misco Precision Casting Comp

PRODUCERS OF AIRCRAFT AND INDUSTRIAL INVESTMENT CASTINGS



CHROMALLIZING FOR METAL PROTECTION AT JET-HOT TEMPERATURES

Improve oxidation and thermal shock resistance of metals used in jet engines and similar superheat applications with CHROMALLIZING. This patented method of diffusing chromium with other elements into the surface provides an alloy case which is integral with the base metal. It can't peel or flake; the chromium and other elements diffuse uniformly into recesses, pores, cracks and even blind holes.

Alloy	Usual Operating Temperature	Operating Temperature of CHROMALLIZED Alloy
Iron Base (including stainless steels)	1500° F	SA CHROMALLIZED 310 and 321 stain- less steels show no failure after 18 hours at 1950° F in an atmosphere containing lead bromide and lead sulfide,
Nickel Base	1800° F	U CHROMALLIZED nickel base alloys are unattacked after 200 hours at 2000° F.
Cobalt Base	1800° F	SAC CHROMALLIZED cobalt base alloys are unattacked after 150 hours at 2200° F.
Molybdenum	Over 2000° F	W-2 CHROMALLIZED molybdenum shows no failure after 400 hours at 2350° F, after 48 minutes at 2800° F, and after one minute at 3400° F.



Chromalloy Corporation Divisions

- · PROPELLEX CHEMICAL DIVISION, EDWARDSVILLE, ILLINOIS Propellants, contridge actuated devices, explosives and special chemicals
 - . ELYRIA FOUNDRY DIVISION, ELYRIA, OHIO Quality aray fron costings.



Full complement needle bearing capacity, precision accuracy to NAS and AFBMA standards, non-separable construction, lightweight and high material selection standards qualify McGILL RT, NBF, NBL and NBC bearings for aircraft cam, track, and guide roller applications. Surfaces plated as required.

RT and HRT SERIES—for use when a cantilever type stud mounting is required. Plating or surface treatment and lubrication holes supplied as application dictates.

NBF AND NBL SERIES — for use in applications where the bearing O. D. is unsupported and will support rolling loads in cam or track support.

NBC SERIES — for applications where bearing O. D. can be supported in a housing. Suitable for slow rotation and oscillation.

Consult our Engineering Department for Special Aircraft bearings built to specification —

Write for free Workbook Guide to Series and sizes



engineered electrical products

precision needle roller bearings

McGILL MFG. CO., INC., Bearing Division 621 N. Lafayette St., Valparaiso, Indiana

Circle No. 234 on Reader-Service Card

CONNECTORS have removable contacts

Removable contacts, silicone inserts, and crimp-type terminations that replace solder pots are featured in these miniature, snap-in connectors, according to The Deutsch Co., Dept. S/A, 7000 Avalon Blvd., Los Angeles 3, Calif. The DS Series of quick-disconnect connectors have precision-machined pins and sockets that are silver and gold plated for maximum corrosion resistance.

The connectors permit no air leakage at 30-psi differential, offer continuous dielectric separation with no voids.

Circle No. 292 on Reader-Service Card

SWIVEL JOINT is self-lubricating

The design of this cryogenic swivel joint permits liquid oxygen and other liquified gases in transfer to circulate in the thrust bearing races. The technique takes full advantage of the fluids' lubricity and contributes to improved temperature stabilization and the absence of contaminating lubricants, says Chiksan Co., Dept. S/A, 330 N. Pomona Ave., Brea, Calif.

The joint provides low-torque freedom of movement on a many planes as desired. Joints under two in. in dia handle pressures to 2500 psig over a —320 to +160-deg F range. Units over two in. handle pressures to 150 psig. Available sizes range from ½ through 16 in. and feature flanged, threaded, or beveled-for-welding end connections.

Circle No. 293 on Reader-Service Card

AC GENERATORS for ground power

The ac generators in this line are well suited to the power carts being developed for use with jet airliners, says Jack & Heintz, Inc., Dept. S/A, 17600 Broadway, Cleveland 1, O. The Model G180, one of the units, is rated at 120/160 kva, continuous duty, at 180/240 kva for five min intermittent duty, and at 240 kva for five see

to the one the time the time the time the duty, and at 240 kva for five sec. The G180 weighs 260 lbs, said to be one-third the weight of comparable commercial generators. It has the same power and electrical design characteristics as generators aboard the new superliners. Other units in the line are rated at 30, 40, 60, 90 and 120 kva.

Circle No. 294 on Reader-Service Card



McGILL

High capacity in limited space plus precision accuracy mak these bearings ideal for use a track, guide and support roller in missile and rocket ground support equipment.

cf series cameot bearing feature an integral stud and hear duty outer race. They eliminate the cost of building up improvised bol and roller units and withstand the shock of heavy loading. Also available with seals and pre-lubricated

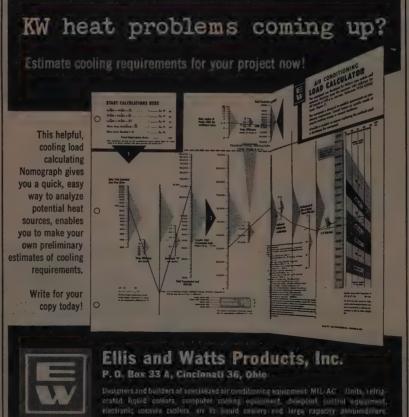
GR SERIES GUIDEROL bearing provide full complement roller of pacity with self-guided rollers for true-running trouble-free performance. SG SERIES sealed GUIDEROL bearings are also available for a plications where contamination must be sealed out and lubrication sealed in.

For data on the complete line of McGill Bearings, send for Catalog No. 52A



McGILL MFG. CO., INC., Bearing Division 621 N. Lafayette St., Valparaiso, Indiana

Circle No. 236 on Reader-Service Care SPACE/AERONAUTICS







Boosted into space by the fiery thrust of three huge rocket engines, the seven-story Atlas intercontinental ballistic missile roars upward from its Cape Canaveral launching pad. Quickly it sheds the frost encrusting the liquid oxygen tank and races to its predetermined destination in the far reaches of the globe. In its size and range and capability, the Air Force Atlas is a

commentary, for all the world to heed, of the necessity to maintain the peace. RCA's Missile and Surface Radar Department has been privileged to design and develop ground check-out, launch control and cabling equipment as a major subcontractor to Convair (Astronautics) Division of General Dynamics Corporation, the Atlas prime weapons systems contractor.



Bendix - Montrose SYNCHROS



TINCH SQUARES

SIZES 11, 15, 18, 23, 31, 37 ALSO TYPES 1, 3, 5 AND 6 TO MIL-S-2335

ON ALL COUNTS BENDIX IS BEST

- 1. Quality—MIL-O-5923C is the only quality standard in the entire plant. There is no lower or commercial quality standard. Constant analysis results in unequaled quality.
- 2. Engineering—Bendix research and engineering represent the pioneering effort on synchro design and development for the industry. Bendix engineers' extensive training and experience are combined to formulate the most forward thinking in synchro science.
- 3. Production—Large quantities of precision production tools all operated by skilled workers who are proud of their craftsmanship.
- 4. Inspection—The most modern gauges, equipment, magnaflux, X-ray, etc., are used throughout the production cycle to assure the precision required to meet the quality demanded by our quality control.
- 5. Environmental—Continuous type testing of production samples to the various environmental conditions required on all types of units is another Bendix extra
- 6. Service—Experienced, well trained sales engineers are always on call to help with your synchro application and procurement of standard and special synchro
- 7. Delivery—Production planning and scheduling plus releases of quantities of various synchros in advance of requirements mean delivery on schedule at the lowest cost.

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AUTOSYN INDICATORS AUTOSYN PRESSURE TRANSMITTERS AUTOSYN POSITION INDICATORS AUTOSYN POSITION TRANSMITTERS

West Coast Sales and Service Office, 117 East Providencia Avenue, Burbank, California Canadian Affiliate—Aviation Electric Limited, 200 Laurentien Blvd., Montreal, Quebec

Export Sales and Service—Bendix International Division, 205 East 42nd Street, New York 17, New York

> Montrose Division SOUTH MONTROSE, PA.



Circle No. 237 on Reader Service Card in Product Review Section

data review

ROTARY JOINT - A waveguide rotary joint, three times larger than the previous largest joint, is described in a 4-page bulletin published by Special Products Div., I.T.E Circuit Breaker Co., Dept. S/A, 601 East Erie Ave., Phila.

Circle No. 295 on Reader-Service Card

DYNAMIC ANALYSIS - An 18page booklet entitled "Precision Dynamic Analysis Instrumentation" has been published by The Solartron Electronic Group Ltd., Dept. S/A, Thames Ditton, Surrey, Eng-

Circle No. 296 on Reader-Service Card

ELECTROMECHANICAL DEVICES

-Over 120 electric motors, linear actuators, solenoids, power units and other components are covered in the Electro-Mechanical Catalog. prepared by Hoover Electric Co., Dept. S/A, 2100 S. Stoner Ave., Los Angeles 25, Calif. Design data pertinent to the aircraft and missile fields are also included.

Circle No. 297 on Reader-Service Card

MISSILE DESIGN-An eight-na brochure outlining its capabil in the design and manufacture propulsion systems, rocket engir missile controls, and propellan among others, has been prepar by Astrodyne, Inc., Dept. S. P.O. Box 548, McGregor, Tex Research and development manufacturing facilities are

Circle No. 298 on Reader-Service Card

HEAT TREATING-An eight-pag catalog, "Heat Treatments in A mosphere Furnaces," which cover protective atmosphere furnace at nealing, brazing and hardening a wide variety of metals, is ava able from Ferrotherm Co., De S/A, 1861 E. 65th St., Clevela 3, O. The catalog also deals wit protective atmospheres, such dissociated ammonia, exothermic, argon and helium, continuous batch or pit-ty furnaces.

Circle No. 299 on Reader-Service Card



TWIST OUT

AIRCRAFT **OXYGEN OUTLET**

This unique, flush-mounted oxygen outlet features a retractable, self-purging dust cap that never needs to be removed or pushed aside. It's easy to operate, too - just a soft push to insert and a quick twist to remove. Lightweight - (under 1 oz.) - this outlet is functionally designed - only one moving part. Here is the same principle that has been proved in thousands of hospital pipeline installations - now adapted for aircraft use!

For complete data and free engineering aid from specialists in oxygen piping systems, please write to Aviation Dept.

Circle No. 246 on Reader Service Card in Product Review Section

INTERNATIONAL RECTIFIER CORPORATION



Highly Specialized Rectifiers for Military and Space-Age Projects Available Through "Prescribed Reliability" Program

Prescribed Reliability Specification Service

To simplify your procurement, proposed specifications in the latest military style, containing as a part of the requirement the design and life-test pattern found to be correlated with known field failure rates, will be supplied to qualified customers without charge. These specifications will be written to include any degree of reliability in a manner enforceable as a procurement document.

Write on your letterhead for Rectifier News RN458. This 8 page issue describes the steps taken to achieve "prescribed reliability" and the tools provided by International Rectifier to simplify your design and procurement program. There is no obligation, of course.

Prescribed Reliability . . . essential to space-age technology . . . supplies you with enforceable procurement documents that prescribe the reliability needed coupled with the engineering and manufacturing skills required to deliver rectifiers heretofore unavailable . . . designed, manufactured and tested to perform within the prescribed contract specifications!

tract specifications!

This concept is the basis for the development of highly reliable, ruggedized rectifiers now available through the Military Products Department of International Rectifier Corporation. Advanced design and manufacturing techniques enable this group to supply you with standard or special units all manufactured to meet the most rigid of existing military requirements plus physical and electrical reliability to any degree specified by you or our capable advisory group.

advisory group.

Information on existing military specifications governing rectifiers, as well as forthcoming revisions to specifications and similar data is available from our Military Products Group. If you are engaged in the design or production of military equipment, you are cordially invited to avail yourself of these services, and to learn the great



Typical silicon rectifier of advanced design produced by International Rectifier Corporation for military use. Rating: 12,500 volts PIV at 5 amperes.

extent to which they can prove valuable to you. Address your inquiries to: Military Products Department, International Rectifier Corporation, El Segundo, Calif.

All International Rectifiers

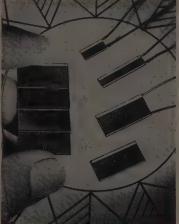
Designed and Manufactured to

Meet Rigid Military Specs.

As each new specification governing the design, manufacture or testing of rectifiers for military use was released during the past years, it became standard procedure in the manufacture of all rectifier types produced by International Rectifier Corporation for industrial as well as military exception.

New Techniques Yield Silicon Solar Cells With Conversion Efficiencies Up to 10%!

From the Astro-Power Division come a new series of rectangular silicon solar cells raising the conversion efficiency to a new high for military and industrial applications. New alloying techniques permanently bond the contact to the silicon wafer making the contact an integral part of the cell itself. This processing improvement results in a substantial gain in operating efficiency . . . minimizes series resistance. Batteries constructed of shingle-type modules will provide an output of approximately 9 watts per square foot of active cell area in bright sunlight. You are invited to investigate these cells for possible application to your industrial control system or remote power source. For engineering specifications, write for Bulletin SR-275, or . . .



Those for which rigid military specifications exist and are now available from stock are listed here:

JAN 1N253 per MIL-E-1/1024A

JAN 1N254 per MIL-E-1/989B

JAN 1N255 per MIL-E-1/990B

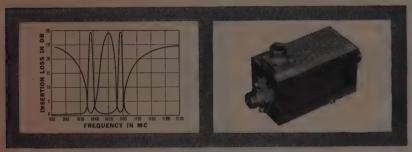
For high current applications: JAN 1N412B per MIL-T-12679/26

We would welcome the opportunity to assist you in the application of rectifiers to your military project. Contact our nearest field representative or our factory applications advisory group.

CIRCLE READER SERVICE CARD NO. 103

FOR SAME DAY SERVICE ON PRODUCT INFORMATION DESCRIBED ABOVE, SEND REQUEST ON YOUR COMPANY'S LETTERHEAD

Typical of Microphase Techniques: Bandpass / Band-Suppression Filter — for simultaneous operation of ATC equipments from a single antenna. Notch-valley response adaptable to similar and more complex requirements in other frequencies. Write for literature.



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DIPLEXERS, MULTIPLEXERS, PRE-SELECTORS

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Circle No. 238 on Reader Service Card in Product Review Section

ELECTRONIC ENGINEFR

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Group Leaders and Senior Electronic Engineers needed for assignments in the design and development of:

Electronic Packaging Airborne Digital Computers Antennas

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Analog Computers Switching Circuitry
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or Missile Engineering Divisions. Write today, sending complete resume of education and experience to:

Raymond F. Kaletta Engineering Employment Supervisor P. O. Box 516 St. Louis 66, Mo.



(LOCATED IN SUBURBAN ST. LOUIS)

CONTROL DEVICES-A four-page brochure briefly describing a variety of aircraft and missile hydraulic and pneumatic control devices is available from Hydra-Power Corp., Dept. S/A, Pine Court, New Rochelle, N. Y. Illustrated are powered flight controls, and solenoid and brake valves, among other devices.

Circle No. 300 on Reader-Service Card

GYROS — A 64-page, \(\text{illustrated}\) "Gyro Primer" has been prepared by Ketay Dept., Norden Div., Dept. S/A, United Aircraft Corp., Commack, Long Island, N. Y. It explains how gyros work, gyro terms, and gyro operating principles.

Circle No. 301 on Reader-Service Card

RESISTANCE TESTER—an illustrated data sheet describes the Arc Resistance Tester for quality control and material development. Other features and specs are included in this literature by Delsen Corp., Dept. S/A, Glendale, Calif.

Circle No. 302 on Reader-Service Card

FACILITIES-An extensive description of its activities in design en-gineering, environmental testing specifications, flight testing and other technical fields is included in a brochure by Nelson Technical Enterprises, Inc., Dept. S/A, 116 W. Main St., St. Charles, Ill. The publication also covers organization and facilities of the company.

Circle No. 303 on Reader-Service Card

GEARS - A profusely illustrated brochure on the techniques and facilities required for the design and fabrication of a large variety of gears has been prepared by Advance Gear & Machine Corp., Dept. S/A, 5851 Holmes Ave., Los Angeles 1, Calif. Typical steps in the creation of a gear are included.

Circle No. 304 on Reader-Service Card

SWIVEL JOINTS-Specific details on a variety of swivel joints for aircraft and missile use are contained in Bulletin 5-58, available from Chiksan Co., Dept. S/A, Brea, Calif. The joints, which feature 360-deg rotation in one, two and three planes, have been de-signed for missile fueling, ground support equipment, and aircraft hydraulic, pneumatic and fuel sys-

Circle No. 305 on Reader-Service Card

PUMPS-A four-page, illust catalog describing the operation its variable displacement inte external pump has been issue Pesco Products Div., Borg-W Corp., Dept. S/A, 24700 N. Rd., Bedford, O. Schematic d ings for zero and full displacer positions of the VIEP are inclualong with suggested applicat for aircraft.

Circle No. 306 on Reader-Service (

PRECISION SYNCHROS-A new lustrated catalog covering sync sizes and types for a wide re of applications has been publis by Induction Motors of Calif. D Induction Motors Corp., De S/A, 6058 Walker Ave., Maywo Calif. An important feature of new catalog is a section cover points to consider in synchro s tion such as types (control torque), military designations engineering data. Included are quiry sheets, and dimensional a electrical data and materials.

Circle No. 307 on Reader-Service

more on next po



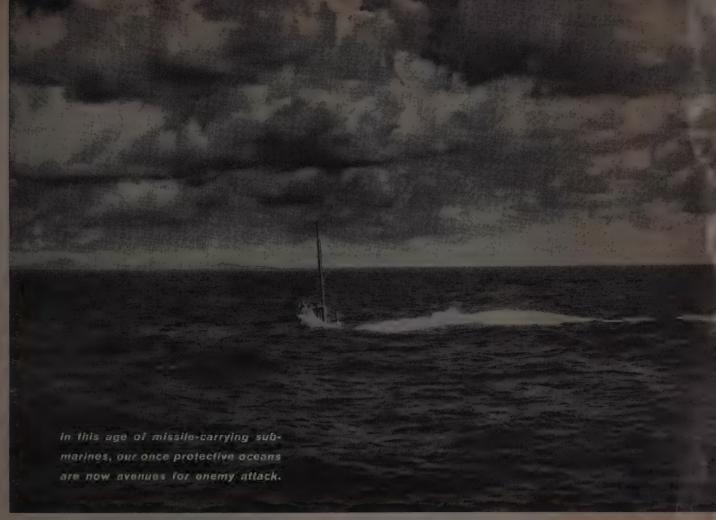
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Circle No. 240 on Reader-Service Card SPACE/AERONAUTIC



circuit development — system planning engineers...

help nab hit-and-run divers

Missile-packing subs able to "torpedo" Indianapolis from either ocean pose a sinister threat...call for even higher levels of detection sensitivity as well as totally new sub-hunting methods. Apparatus division of Texas Instruments has thousands of airborne submarine detector systems in the fleets of the world, but now is expanding its efforts to outstrip the new capabilities of the new generation of submersibles. Other active projects in this division include missile and space electronics, early warning, reconnaissance, airways control, and attack control. Any program you choose will be amply supported by the latest facilities and implemented by TI's ability to create basic components or complete systems.

TI's brisk, steady growth (20-fold over the past decade to a current \$90-million volume) creates corresponding needs for all levels of talent. Engineers, physicists... choose your design or production future—

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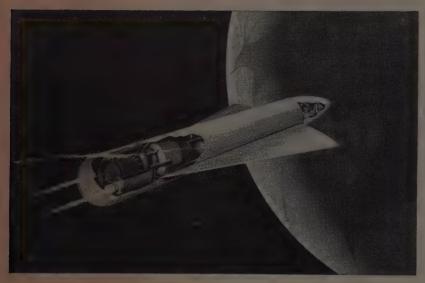


Mr. John Pinkston

apparatus division

rcle No. 241 on der Service Card Product Review Section

COUNT DOWN for the conquest of space



ROCKETDYNE ENGINEERS HAVE MADE MORE THAN 50 TRIPS TO THE NEIGHBORING PLANETS

Through the ship's viewing port looms a breathtaking sight—a gigantic red crescent spanning some 30° of deep black sky. A television camera, passenger on this strange new chariot, stares intently at a sight never before seen by man and beams home to Earth his first crude view of the planet Mars.

From dream to drafting board

Less than a decade will pass before this age-old dream of man is realized. Bold steps toward such an exploration of Space are underway now. An experimental ion rocket engine will soon be placed in operation at Rocketdyne's Propulsion Field Laboratory in the Santa Susana mountains. From this research tool will come design data for the efficient, low-thrust freight engines for Outer Space. These engines will be capable of operating for months at a time, and will make possible extended reconnaissance of the Solar System and detailed studies of the phenomena of Space.

But what of the journey itself?

Rocketdyne engineers have made more than 50 trips to the neighboring planets on huge computer machines. In these paper trips, they have studied the gravitational effects of as many as seven planets at a time. By watching closely the effects of such forces on their low-thrust ion vehicle they determined thrust programs to reach various planetary objectives. They showed the trip to Mars could be made with thrust to vehicle-weight ratios as low as 1 to 10,000.

Testing in Space conditions

Rocketdyne has been at work on ion rocket engines since 1955. While many be solved, extensive new facilities and three years of exhaustive studies are being applied to the job. Rocketdyne scientists will operate their experimental ion engine in simulated space conditions to unlock important answers to thrust chamber design, power conversion systems, nuclear heat sources, and propellants.

Hardware for defense and science

Today the operating hardware in the field of high-thrust rocket engines is designed and built by Rocketdyne: propulsion systems for the Air Force's Atlas and Thor, and the Army's Redstone and Jupiter...and for scientific missions such as the Explorer satellites and the NASA space probes conducted by the Air Force and Army. Based on this unequalled experience, Rocketdyne is already probing far into the future. Engineers are already at



PROBING TOWARD THE PLANETS Heaved bodily into Space by the Rocketdyne-powered Thor first stage, the Pioneer starts on its 80,000 mile sortie toward the moon.

work on the next and succeeding generations of high-thrust rockets, and high-specific-impulse engines to supdifficult design problems yet need to plement chemical rocket performance.

FIRST WITH POWER FOR OUTER SPACE

A DIVISION OF NORTH AMERICAN AVIATION, INC.

DATA REVIEW

REMOTE POSITIONING COM TROL-Two bulletins, F 8537 F 8342-1, from Barber-Coler Co., Dept. S/A, 1400 Rock Rockford, Ill., describe magne amplifier control systems, for in on-off, proportioning, or pul modulated floating temperature control applications as well as n mote positioning and synchronizing systems.

Circle No. 308 on Reader-Service Co.

SERVOMOTOR - Design and p formance data on the smallest 8 servomotor available are tained in the four-page Data S 1366, available from Helipot C Dept. S/A, Newport Beach, Co The Model 8 SW 420, a V, 400-cycle unit, is 0.840 long, weighs 1.1 oz, and has stall torque of 0.25 oz-in.

Circle No. 309 on Reader-Service Ca

DROP TEST MACHINE-A tech nical bulletin, on its Drop Tes Machine, Model 30K, which provides shock forces in excess of 7 g's on specimens weighing up to 400 lbs, is offered by Aerole Corp., Dept. S/A, 34-06 Still man Ave., Long Island City 1

Circle No. 310 on Reader-Service Care

MINIATURE BEARINGS - Details on its line of precision miniaturball bearings, including dimen sions, tolerances, torque value static and dynamic capacities an limiting speeds, are covered in the new six-page catalog supplement M1, available from Barden Corp Dept. S/A, Danbury, Conn.

Circle No. 311 on Reader-Service Car

NITROUS OXIDE-A new cight page brochure describing the type of nitrous oxide installations specifically designed for use in leak designed. tection of pressurized systems available from Ohio Chemical Surgical Equipment Co., Dep S/A, 1400 E. Washington Ave Madison 10, Wis.

Circle No. 312 on Reader-Service Card

GAS ANALYZER-Beckman/Proc ess Instrument Div., Dept. S/A 2500 Fullerton Rd., Fullerton Calif., announces the availability of a new, four-page bulletin, TC-4012, giving detailed information on the applications, features, principles. ciple of operation, and specifications of the Model 7C Therma Conductivity Gas Analyzer.

Circle No. 313 on Reader-Service Ca

astronautics intelligence

Biggest satellite yet put into orbit

Nose cone re-entry
periment may come at end
of Atlas' flight

Project Score comm system weighs 150 lb

Extensive, very clear radiation data gathered by Pioneer III on both legs of its trajectory

Most of ARPA's Discoverer satellite launchings to be made from Vandenbergh AFB

Discoverer vehicle itself will be tested first

Ventilated suit may enable spacecrews to take higher temperatures

STANDARD ATLAS with extra fuel aboard managed to get into orbit with an apogee of 625 miles and perigee of 118 miles. It's by far the biggest satellite put up by man so far.

ORBITING ATLAS' weight of 8500-8700 lb is well over minimum weight needed for an "austere" manned capsule. Unofficial reports are that some sort of animal life aboard Atlas has survived, will be kept alive for at least two weeks.

Nose Cone engineers speculate that, toward the end of Atlas' orbiting life, the nose cone will be separated for a controlled re-entry attempt.

COMMUNICATIONS EQUIPMENT in the nose cone weighs about 150 lb. "Score" system contains receiver and transmitter for re-broadcasting messages received from the earth.

Project Score (Signal Communications by Orbiting Relay Equipment) is being run by ARPA. Atlas is the first of several planned launchings.

MINITRACK TRANSMITTER was assigned two frequencies—107.97 and 107.94 mc. Communications system was tied into four stations near Los Angeles, at Ft. Huachuca, Ariz., Ft. Sam Houston, Texas, and Ft. Stewart, Ga. Messages from any station could be relayed to the others.

PIONEER III MISSION to the moon and possibly the sun was not completed. But the Army space probe sent back exceptionally clear data on the radiation belt surrounding the earth.

Radiation was measured on both upward and downward legs of the trajectory, so that intensity can be compared for regions over different parts of the earth. Eventually scientists would like to launch straight probes up and down for radiation studies. This may even be done seasonally—to determine the variation (if any) in intensity over the year.

ARMY'S SPACE COMMUNICATIONS system still has not been checked out as a result of the incomplete Pioneer flight. The HF system is expected to have a range of some 500,000 miles.

PACIFIC MISSILE RANGE at Vandenbergh AFB, Calif., will be the site of most of Project Discoverer satellite launchings planned by ARPA. It is expected a considerable number of launchings will be tried.

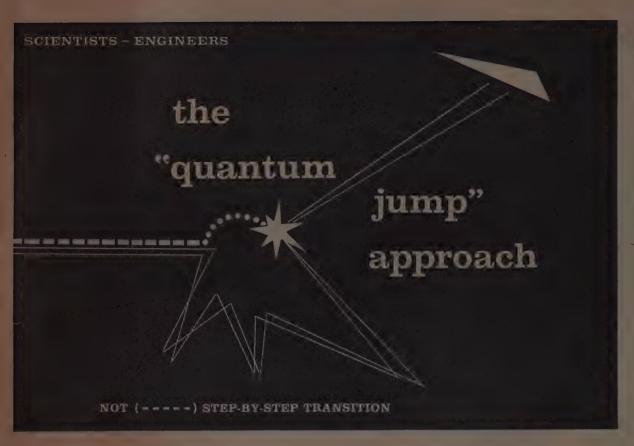
First series of Discoverer satellites will weigh about 1300 lb each. This includes weight of the second stage, which will orbit as an integral part of the satellite after burnout. Initial versions are designed to orbit for short periods of time at low altitude.

MAIN STAGE of Discoverer vehicle is a modified Thor IRBM. The second stage (from Lockheed) is new. It is powered by a Bell Hustler engine.

EARLY DISCOVERER LAUNCHINGS scheduled this year are designed mainly to test the vehicle itself, particularly its propulsion and guidance systems. Once the design has proved itself, a series of heavier satellites will be launched. These will contain biomedical specimens and live animals.

EVAPORATIVE COOLING SUIT is being developed to further man's tolerance of high temperatures. If results are successful, it may be possible for pilots to withstand much higher capsule re-entry temperatures than formerly thought possible.

Ventilated plastic film suit is worn under conventional pressure suit. Air circulated between the inner and outer suits reaches the pilot's skin through tiny pinholes in the plastic. It collects perspiration in the form of water vapor and seeps back out of the suit through larger holes in the plastic. Result: the body cools itself naturally by evaporation.



How Republic's Engineers & Scientists Attack Today's Missile, Aircraft and Space Problems

There's something in the environment at Republic Aviation that fires the imagination and sharpens the logical faculties.

You find it, at its quintessence, within the research groups, whose members have recently come up with a feasible design for a "Magnetic Pinch Plasma Engine" to operate in space. A design utilizing a novel application of basic knowledge in the field of magneto-hydrodynamics.

The same spirit of bold inquiry, of looking beyond the

immediate, conventional solution of a problem for new and better answers, permeates the entire professional staff. You'll find it in development engineers working on new missiles for new missions ...in experimental engineers studying the behavior of a prototype aircraft. You'll find it in the metallurgical and electronics labs.

THE RESULT—"quantum-jump" advances in many technologies vital to the conquest of upper atmosphere and space flight.

Engineers and Scientists who share Republic's views on Creative Problem-Solving are invited to discuss opportunities now open in the following areas:

THERMO, AERODYNAMICS

Theoretical Gasdynamics • Hyper-Velocity Studies • Astronautics Precision Trajectories • Airplane/Missile Performance • Air Load and Aeroelasticity • Stability and Controls • Flutter & Vibration • Vehicle Dynamics & System Designs • High Altitude Atmosphere Physics • Advanced Propulsion • Engineer Performance • Re-entry Heat Transfer • Supersonic/Hypersonic Wind Tunnels • Shock Tunnel • Plasma Tunnel • Magneto-Hydrodynamics • Ground Control Equipment

ELECTRONICS

Inertial Guidance • Inertial Navigation • Digital Computer Development • Radar Design • Systems Engineering • Telemetry-SSB Technique • Information Theory • Doppler Radar • Countermeasures • Radio Altimeters • Radome & Antenna Design • Microwave Circuitry & Components • Receiver & Transmitter Design • Airborne Navigational Systems • Missile Checkout Systems • Missile Arming & Fusing • Jamming & Anti-Jamming • Miniaturization-Transistorization • Ranging Systems • Propagation Studies • Ground Support Equipment

REPUBLIC AVIATION is investing \$35,000,000 in bold new R&D Program designed to arrive at Major Breakthroughs inTransition from Aeronautics to Astronautics.



Send resume in complete confidence to: Mr. George R. Hickman, Engineering Employment Manager

REPUBLIC AVIATION

Farmingdale, Long Island, New York

ASTRONAUTICS INTELLIGENCE

VOLUNTEERS wearing the suit have been able to take 160 deg F continuously for three hours without any loss of work efficiency. Limit, or point of appreciably diminishing effectiveness, for men without "ventilated underwear" is about one hour.

Next step is to see just how high the temperature can be raised without impairing efficiency. Extremely high temperature tolerance may even be possible for a few minutes.

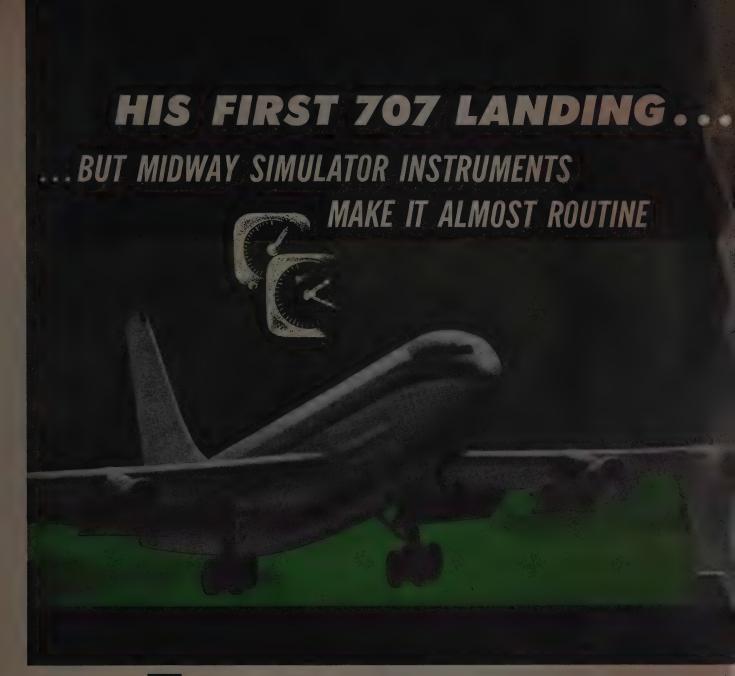
ELECTRIC impulse exercises may be the answer to keeping space travelers fit during lengthy confinement in cramped quarters. According to medics, it is possible to substitute electrically induced muscle contractions for normal exercise.

INITIAL trajectory of Pioneer III was picked up and followed by Collins tracking station in Puerto Rico. From there the probe's position was relayed to a larger station at Goldstone Lake, Calif., which then locked on to the signals from the low-power transmitter in the probe when it came over the horizon.

Collins teamed up with JPL to equipping the Goldstone Lake station. JPL provided the 85-ft dish and the pedestal and Collins the receivers, display units, telemetering, recorders, etc.

PLANET shots will be scheduled by the calendar as much as by technological advances. Assuming present rate of development is continued, best time for a Venus vehicle launching will be in June of this year. Next time chances of success will be as good will be September 1962.

OVER next 10 years, we will have only six good opportunities to send a space probe to Venus and only four for Mars.



Midway Simulator Instruments are now helping to train commercial jet pilots for the Boeing 707, Convair 880, the DC-8, and Lockheed Electra. Landing a Boeing 707 jet for the first time takes skill and thorough training. Today a new jet pilot faces this task with confidence for he has many simulated landings behind him . . . thanks to MIDWAY instruments.

MIDWAY Simulator Instruments realistically reproduce actual flight conditions — including emergency and bad weather situations. To meet stringent training standards, these instruments must be capable of many repeat performances at uniform high accuracy.

At MIDWAY no compromise is made with quality . . . for quality means reliability.

For information about MIDWAY instruments write to Publications Dept.

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der Service Card Product Review Section

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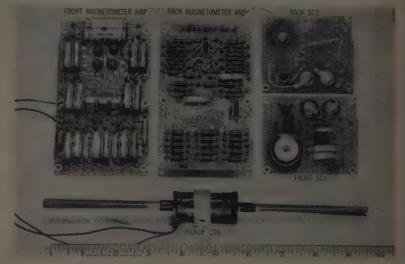
Honeycomb housing for pioneer instruments

USAF'S PIONEER lunar probe moon, which streaked about 71,000 miles into space last Oct. 12, was fourth stage of a launch vehicle weighing about 104,500 lb. First stage was a Douglas Thor; second, an Aerojet-General liquid rocket; third, an Allegany Ballistics Lab solid rocket. Pioneer itself weighed about 85 lb. Its honeycomb fiberglass package and supporting structure had a weight of less than 15 lb. Instrument package was developed by Space Technology Laboratories, P.O. Box 95001, Los Angeles 45, Calif., weighed about 30 lb.

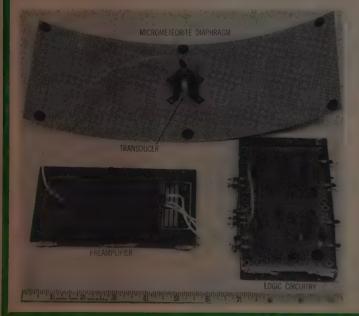
by Irwin Stambler, Engineering Editor

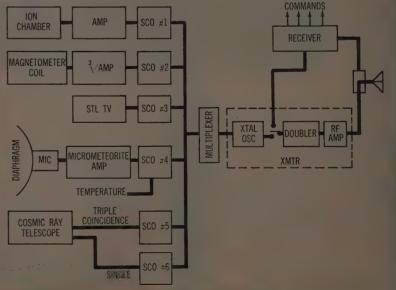


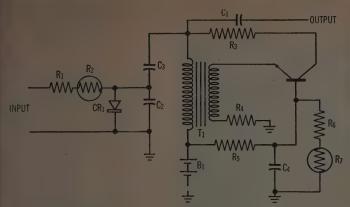
PIONEER INSTRUMENTATION—Package (above left, without cover) has Thiokol retro-rocket weighing about 35 lb. Above right; magnetometer circultry. Below left: Elements of micrometeorite assembly. Its data involve only counting, so internal ambient



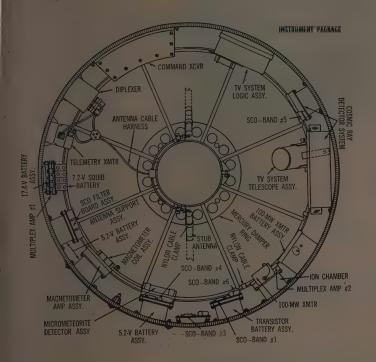
temperature can be multiplexed on top of it as shown in block diagram (below right). STL Microlock receiver is smaller than shoe box, can be acquired from ground to supply either Doppler readout (for which transmitter becomes transponder) or commands for verniers, retro-rockets, etc.

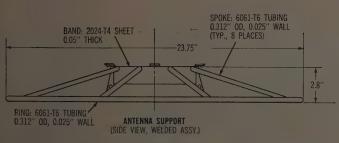






SUBCARRIER oscillator carried by Pioneer.

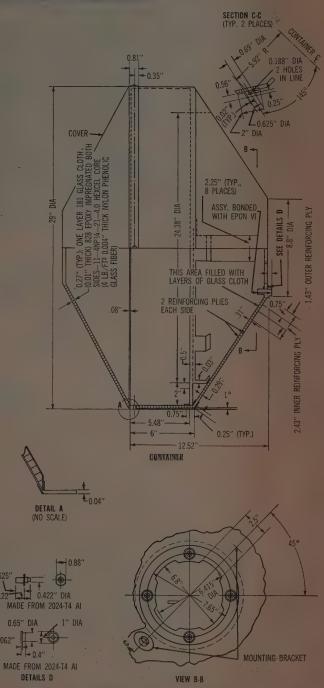




INSTRUMENT layout and structural design of Pioneer. Folded dipole antenna was installed on inner surface of cover. Honeycomb container was picked by STL engineers for minimum weight.



PAINTED, pressure-sensitive aluminum foil applied to container's outer surface formed blocking pattern for control of Pioneer's internal temperature. Pattern had to be changed daily while the vehicle waited for launch.



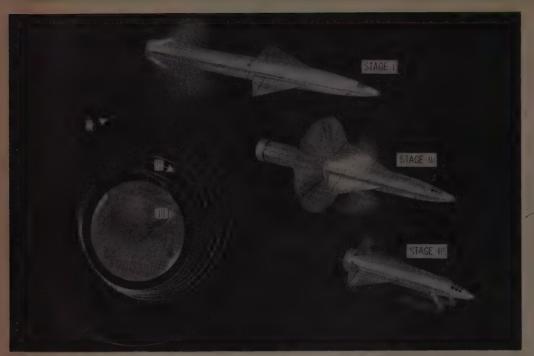


FIGURE 1: Combination of retrorockets and drag devices might prove workable for satellite recovery. During Phase 1, small retro-rockets near tail are used at high altitudes to change orbit ellipticity. When the vehicle has dipped into denser air, large, flaring drag device of Phase II causes the orbit to decay until it is nearly circular at under 100 miles altitude. During Phase III, larger retro-rockets are used for lift augmentation and deceleration to prevent spiral-in and excessive heating.

Retro-rockets and drag devices teamed for

satellite recovery

Pretty soon, satellite recovery will loom as large as satellite launching does right now. It's generally recognized that it will have to be done by means of retro-rockets. But just how are these going to work? Here is an article that answers this question—partly by pointing out that rockets can't do the whole job all by themselves.

by F. W. Ross, Chief of Aerodynamics, Solar Aircraft Co.*

THE problem of satellite recovery applies to just about all recoverable space vehicles that we can imagine as "reasonably" economical in operation: Even those among these vehicles that escape into space will have to go into an orbit before re-entering.

If retro-rockets are used for it, satellite recovery essentially turns into the reverse of the launching process. In launching, most of the energy is needed to accelerate the satellite. Comparatively little must be expended to raise it to altitude. In recovery, most of the energy must be expended on deceleration. Only a little extra must be used to lower the satellite from 300 miles to the surface of the earth.

The kinetic energy for a satellite orbiting about a homogeneous spherical body of

^{*}Solar Aircraft Co., 2200 Pacific Hwy., San Diego 12, Calif.

ss M at a distance h above

 $KE = kM/2(r_o + h)$, here ro is the radius of the anet and k the universal avitational constant from ewton's theory. The total eny for this case is:

 $TE = kM[1+h_r/(1+h_r)]/2r_o$

here h_r equals h/r_o .

Kinetic, potential and total ergies here aren't influenced such factors as air drag, opulsion efficiency, ellipticetc. They are determined tirely by the altitude of the biting mass. This is true of y motion in a central force

As Figure 2 shows, the magtudes of the energies make tellite recovery a formidable oblem. The amount of engy involved is over 19,000,-00 ft-lb per pound of orbitg mass, or 24,400 btu/lb. ne heat energy of a pound of hydrocarbon, such as gasoe, is only about 21,000 u/lb. And an oxidizer with veral times the weight of the el must be used to get this

If the entire orbiting mass ere hydrocarbon with the ost favorable ratio of oxidizer d all of it were used with 00 per cent efficiency, there only enough energy available slow down about 25-30 per nt of the mass. When we ke account of the fact that ere must be weight for strucre and payload, we see that

me form of staging is needed.

nergy below 300 miles early all kinetic

Figure 2 also shows that a tellite orbiting at high altide — say, 2000 miles — reires only about 1.15 times ore energy than one orbiting 200 miles. On the other and, a satellite weighing 3000 , as does Sputnik III, needs ore than 100 times as much ergy as a 30-lb satellite. - 1

Note also that the total engy for altitudes below 300

more on next page

New Mallory-Sharon 1100° Titanium Alloy





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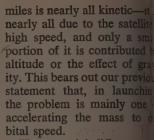
What this means in terms of jet engine construction, for example, is illustrated above. The weight-saving advantages of titanium can now be obtained in additional stages of hot Mach 3 engines through use of MST 881.

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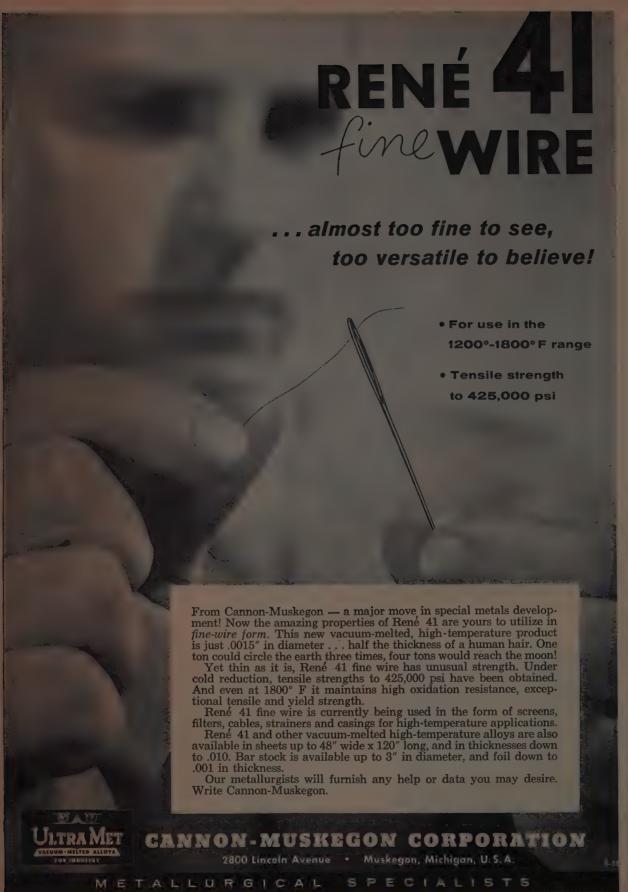
The essential difference between launching and recover is due to aerodynamics. Duing launch, aerodynamic draacts against the thrust, making it necessary to use morthrust to overcome this dra For recovery, the drag work in the same direction as the decelerating thrust and reduce the required thrust.

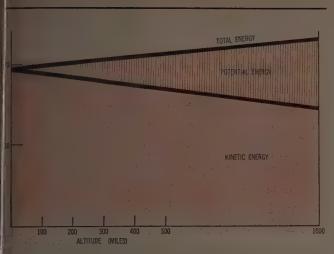
When staging is used, the launching losses caused by drag are larger because the are, in part, determined by the size of the larger initial stage. For a re-entry vehicle, the "larger initial" stages operated at or near the altitudes a which the air drag is lowes. The direction of these two effects tends to increase the propulsion needs in both these cases.

High altitude deceleration should be faster

Launching and recovery at closely similar in another re spect: Less fuel is needed for launching if high acceleration are used to permit ejection the initial stages at as low altitude as possible. Convers ly, for re-entry it's more ef cient to decelerate faster first before the initial stage gain speed through losing a titude. If this isn't done, the added energy (jet thrust) wi have to be expended to decel erate this extra speed of the initial stages.

From this we can see that if retro-rockets are to be used as the sole means for recovery of a satellite from orbit, the re-entry vehicle would operate much as the launch vehicle did —but in reverse. The necessary propulsion units will have





IURE 2: Kinetic, potential, and total energies of typical satellite iting above the earth's surface.

out the same relative magtudes in both cases.

For Vanguard, a gross eight of 22,600 lb is needed place about 220 lb in orbit, ving a launch-orbit weight atio of about 100. For a rern trip, therefore, the ratio forbiting weight to recovered eight is also 100. Conseuently, the overall launchcovery weight ratio becomes 0,000. To recover a 1000-lb tellite solely by means of tro-rockets we would need initial launch weight of 10,-0,000 lb with present chemal fuels and staging methods. bviously, recovery by retrockets alone is not what we e looking for.

We can, however, make use air drag to obtain a workde recovery system (Fig. 1). The problem here is that, at gher altitudes, the air drag is altitude takes days or months. At lower altitudes (50-100 miles and below), on the other hand, the drag offers too much resistance—the satellite spirals in rapidly and finally dives, building up high temperature as well as high acceleration. By using rockets in conjunction with drag devices, this problem can be solved.

At high altitudes, jets can be applied opposite to the orbit speed at apogee to increase the ellipticity and reduce the perigee to lower altitudes. There, with the aid of large drag 'chutes or balloons, the higher drag at perigee can be used to dissipate the orbital energy faster in the initial phases of recovery. The jet sizes and fuel energies needed to change the ellipticity are

more on next page

Comparative Recovery Energy Needs

		Mass & Total Energy Ratio to Earth	Radial Ratio to Earth	Equivalent Altitudes (miles)	
rth	1	1 1 7.5	21	. 300	
oon		. 0.0123	0.2738	82	
Brs		0.107	/ 0.5340	160.2	
nús		0.82	0.9720	292	
oiter		317.7	10.92	2760	

What makes the



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comparatively small—air drag accounts for most of the dissipation.

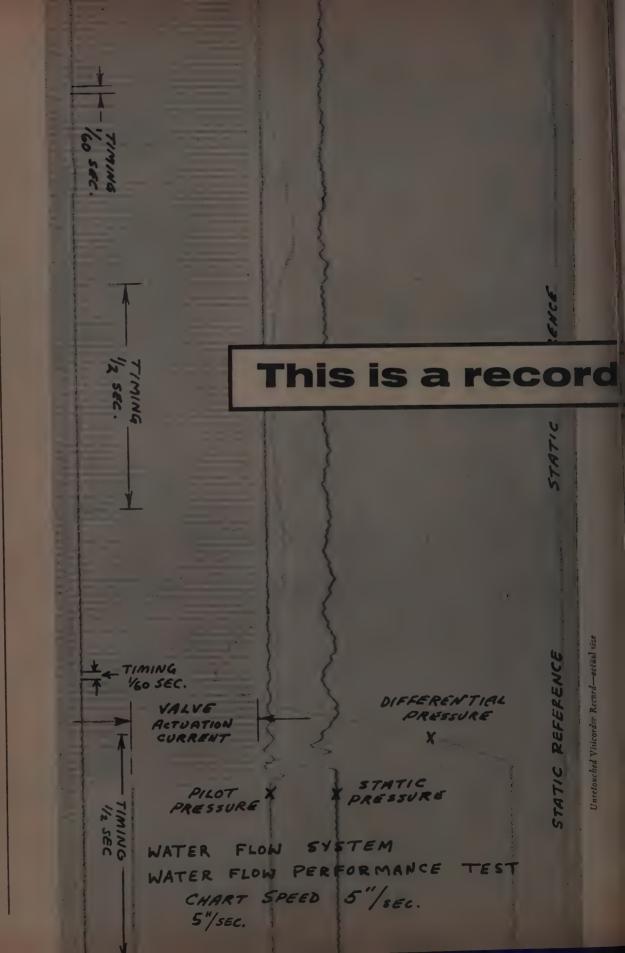
After the orbit has been decayed to nearly circular shape at about 35-50 miles altitude, the re-entry vehicle will tend to spiral in as drag increases. One way of preventing this spiraling might be to design the vehicle to provide lift. However, at the air densities and speeds involved here, aerodynamic lift is low and may be less than the drag.

Jets must again be used to keep the rate of descent low enough to eliminate spiral-in. By tilting the jets forward, the lift component can be held nearly constant, for the cosine of the angle of tilt will vary only very slightly from 1. This added deceleration reduces the heating factor and makes possible a steeper descent.

As a final note, observe that the energy requirements as stated in Figure 2 and Equation 2 are determined directly by the weight of the earth. If a satellite is orbiting around any other body, such as the moon, Mars, or Venus, at a proportionate height, the total energy—and hence the retro-rocket needs—are proportionately more or less, depending on the mass of the body.

As the table shows, a launch - and - recovery vehicle based on the moon would require about 1.23 per cent of the retro-thrust needed for comparable earth vehicles. For Mars, the retro-thrust needs are only 10.7 per cent of those for the earth. For Jupiter, though, they are enormous—317 times as much as for the earth.

The much lower retrothrusts needed for the moon and Mars cases are important, since these bodies are thought to have little or no atmosphere. Hence re-entry ill have to be done largely by means of retro-rockets without the aid of energy dissipation by air drag.—End





Wyle Laboratories in El Segundo, California, have used a battery of four Visicorder consoles like the one shown below to run a series of tests on a vital missile component. In the Wyle test project the unique Visicorder consoles are easy to operate. Most parameters are low frequency, requiring response on the order of 5 to 60 cycles.

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of a missile component



Tom Jackson, Wyle engineer, examines Visicorder record

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Reference Data: Write for Visicorder Bulletin Minneapolis-Honeywell Regulator Co., Industrial Products Group, Heiland Division 5200 E. Evans Ave., Denver 22, Colo.

books

Metals for Supersonic Aircraft and Missiles. This is the proceedings of the conference on Heat Tolerant Metals, held at the Univ. of New Mexico, Albuuerque, N.M., Jan. 28-29, 1957, jointly sponsored by the Univ. and American Society for Metals. American Society for Metals, 7301 Euclid Ave., Cleveland 3, Ohio. \$7.50.

Causes and Prevention of Corrosion in Aircraft, by T. C. E. Tringham. Included in the areas covered in this book are the causes of corrosion, testing methods and methods of applying corrosion resistant coatings. Pitman Publishing Corp., 2 W. 45th St., New York, N. Y. \$5.75.

Proceedings of the National Electronics Conference, Hotel Sherman, Chicago, Oct. 7-9, '57. This contains some 100 papers presented at the meeting covering most major areas in electronic design. National Electronic Conference, 84 E. Randolph St., Chicago 1, Ill. \$7.50.

Standard Aircraft Handbook, by Stuart Leavell & Stanley Bungay. Intended for production workers, this includes tables on riveting, bolting standards, materials, etc. Aero Publishers, 2162 Sunset Blvd., Los Angeles 26, Calif. \$2.75.

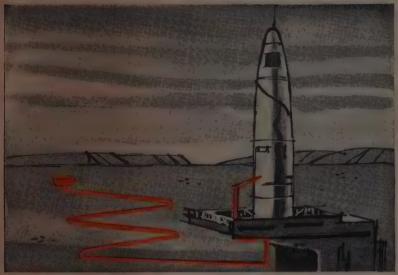
Graphic Science, by T. E. French & C. J. Vierck. This comprehensive book provides the designer with both fundamental theory and a wide range of applications of engineering drawing, descriptive geometry and graphics. Such subjects as functional scales, nomography, graphic calculus, vector geometry and applied geometry are discussed. A glossary of technical terms and a series of standards tables are also included. McGraw-Hill Book Co., Inc., 330 W. 42nd St., New York 36, N.Y. \$8.50.

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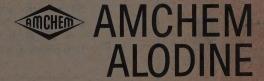
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Please write: Mr. F. A. Jamieson, Engineering Personnel Dept., 6633 Canoga Avenue, Canoga Park, Calif.

ROCKETDYNE IR

A DIVISION OF NORTH AMERICAN AVIATION, INC.



readers' round table

We must cut down development time!

THE West today is challenged as was the great Roman civilization just before its decline and fall.

Our response to that challenge is feeble. It is confused and contradictory.

The trouble is, we have not really analyzed the challenge. Look at Soviet achievements:

• Once a primitive, largely agricultural, illiterate nation, Russia has become the second industrial state of the world in 40 years. In this period she has been ravaged by war and revolution. Yet her rate of industrial expansion exceeded our own.

• Militarily, Russia is equal, perhaps superior, to the West.

• Russia has produced a very large intellectual elite in all fields except the humanities. She seemingly has an inexhaustible reserve of scientists, engineers, and technicians. Quite as important, there are great numbers of linguists who, among them, command the languages of the world.

• Russia's people, as a whole, are imbued with a fierce patriotism and sense of mission. Whatever their sufferings, they are convinced that Russia represents the wave of the future.

• Africa and Asia have been penetrated by Soviet aid and investment faster than by ours. The conditions for Soviet help appear to these peoples more favorable.

• The Reds and their satellites proceed with a higher de-

Contributions to this department may be on any subject, technical or nontechnical, about which readers would like to air their views. Names and professional affiliations will be withheld on request.

REAR ADM. JOHN
T. HAYWARD, Assistant Chief of
Naval Operations
for Research &
Development



gree of integration than does the West.

What are we, as free men, prepared to do and sacrifice to meet this challenge?

To get a feel for the problem, we must define just what national strategy is. It is the use by a country of economic, political, psychological, and military forces to reach national objectives.

Note that our strategy must be more than just military. Mere reorganization of the Defense Dept. isn't going to bring everything into order.

We must have a clearly spelled out national policy. This must identify the global interests of vital national concern. We must be prepared to fight any type of war against the Sino-Soviet coalition—with or without allies, and without waiting for the U.N. to determine that the Reds are guilty of "aggression."

From this starting point we can go on.

We need an R&D program aimed at definite, long range objectives. Now, we can't have everything we want. So we must have a selective process.

Mistakes made in this selection process are largely responsible for development time problems. Anyone can set up an operational requirement. But if it doesn't make sense in terms of the present and

future state of the art, you pay a penalty in development time

To get a reasonable progra we must ask ourselves:

- Are the essential weapons on hand or in development?
- Are we thinking of o pursuing more ways to mee a given situation than an needed?
- Are the developmentimes allowed for new weaponsystems reasonably well matched to the state of the scientificand technical arts?
- Are the performance gains great enough to warran a new system?
- Can the existing system be modernized to produce at equal gain in performance a less cost?
- Are the requirements unnecessarily complex, so that we will end up with an expensive system?
- What is the threat for the task you are trying to meet in the time you have to develop your system?

There are many more questions. While the best technical answer may not always be the best policy, those who make policy must know the best technical answer.

A look at the strategic bombardment missiles quickly shows that all the questions were not answered. In some cases they weren't even asked.

Circle No. 252 on Reader-Service Card or

development process nly can be shortened by l planning and work in lective process before the on is made to go ahead. 's look at the situation Navy faces. The biggest today is the submarinee combination. Our sucith Polaris shows clearly this weapon system rs the most urgent renents for a deterrent It is flexible. It is secure, dependent on foreign We in the Navy, howwould be first to say it's part of the overall deterposition.

e Red missile launchings a big threat to the U.S. its allies. So the anti-sub lem is high on the list.

call your attention to the that the Russian bloc is pped with modern weap-If our aircraft are to conthe air over the sea or they must be modern. As as the manned airplane th us, the aircraft carrier be here. The decision on sea as on the land rests the decision in the air.

review of events since and the situations that faced our armed forces some idea of the threat problems that we face in future. Our weapon sysmust be such that they the various problems the whole spectrum of as it exists and as it will in the future.

his calls for great selecy in choosing our weapon ems. It calls for short depment times. We cannot d seven years as in the

we answer the questions ve posed, we should make I decisions and cut down clopment time. I don't k this will happen overtt. But let me assure you we in the Navy R&D busiare going to do our best take it happen as soon as lible.—End

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